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## **NORMAL SPECTRAL REFLECTANCE OF ANODIZED COATINGS ON ALUMINUM, MAGNESIUM, TITANIUM AND BERYLLIUM**

*103-565-*

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SEPTEMBER 1961

AERONAUTICAL SYSTEMS DIVISION

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*SEPTEMBER 1961*

DIRECTORATE OF MATERIALS AND PROCESSES  
CONTRACT No. AF 33(616)-6191  
PROJECT No. 7312

AERONAUTICAL SYSTEMS DIVISION  
AIR FORCE SYSTEMS COMMAND  
UNITED STATES AIR FORCE  
WRIGHT-PATTERSON AIR FORCE BASE, OHIO

## FOREWORD

The report was prepared by the Honeywell Research Center under USAF Contract No. AF 33(616)-6191. The contract was initiated under Project No. 7312, "Finishes and Materials Preservation", Task No. 73120, "Surface Treatments and Coating". The work was administered under the direction of the Directorate of Materials and Processes, Deputy for Technology, Aeronautical Systems Division. The work was initiated with Mr. Bennie Cohen acting as project engineer, and was continued under supplemental agreement No. S1 (60-1001) to the completion of the project with Lt. N. M. Geyer as project engineer.

This report covers work conducted from January 1959 to May 1961. Eleven bi-monthly progress reports (1) were written during the course of the investigation under the title, "Measurement of Spectral Reflectance of Anodized Metals". The title of this final report has been changed to make it more descriptive.

The authors wish to acknowledge the contributions of K. E. Nelson of the Solar Radiation Laboratory of the University of California, Los Angeles Branch, and C. C. Shaw of Lockheed Missiles and Space Division, Lockheed Aircraft Co., Sunnyvale, California. Their reflectance measurements on several specimens prepared for this investigation provided a valuable comparison. The contributions of C. H. Jackson and R. L. Sampson of the Honeywell Research Center during the early phase of this study were also important.

## ABSTRACT

Selection of materials having proper thermal radiation properties permits the engineer to achieve passive temperature control in spacecraft. The object of this investigation was to study the effect of various variables in the anodizing process on the reflectance of anodized metals. The integrating hemisphere was modified by adding a diffuser and then locating the sensor external to the hemisphere so that measurements could be made over the range of 0.4 to 22 microns in vacuum with specimen temperatures from 100 to 1300 F. Methods for extending both the spectral and temperature ranges are discussed.

Normal spectral reflectance data on 158 specimens of anodized aluminum, magnesium, titanium, and beryllium are presented. Aluminum anodized in sulfuric acid gave a ratio of solar absorptance to infrared emittance of about 0.3 while A110AT titanium anodized in sodium hydroxide gave a value of  $\alpha/\epsilon = 7$ . In general, the anodizing process had a greater effect on the reflectance than did alloying elements in the metal. Measurements at elevated temperatures indicated a loss of water from coatings that contained water. In general, it was the presence of this water which produced high emittance in the infrared region and gave low values for  $\alpha/\epsilon$ , however.

## Publication Review

This report has been reviewed and is approved.

FOR THE COMMANDER:

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## TABLE OF CONTENTS

### PART I

	PAGE
INTRODUCTION	1
DISCUSSION	3
Black-Body Reflectometer	4
Integrating Hemisphere Reflectometer	6
Specimen Preparation	52
Anodized Coating Thickness	61
Results	63
CONCLUSIONS	67
RECOMMENDATIONS FOR FUTURE WORK	68
REFERENCES	70
APPENDIX TO PART I	71

### PART II

PRESENTATION OF REFLECTANCE DATA	75
----------------------------------	----

## LIST OF ILLUSTRATIONS

FIGURE	PAGE
1. Black Body Reflectometer.....	5
2. Energy Reflection in Integrating Hemisphere.....	8
3. Spectrometer.....	12
4. Hemisphere and Turret .....	14
5. Vacuum Chamber and Hemisphere.....	15
6. Integrating Hemisphere Reflectometer.....	17
7. Reflectometer Optical System.....	18
8. Hemisphere Reflectance Determination.....	20
9. Hemisphere Reflectance Determination.....	21
10. Hemisphere Reflectance Measurement Apparatus.....	22
11. Normal Spectral Reflectance of Evaporated Aluminum on Glass - 0.38 to 2.5 microns.....	27
12. Normal Spectral Reflectance of Evaporated Aluminum on Glass - 2.5 to 25 microns.....	28
13. Diffuser.....	29
14. Diffusivity of Diffusers.....	31
15. Diffuser Calibration Equation.....	37
16. Diffuser Reflectance.....	40
17. Reflectance of Magnesium Carbonate.....	44
18. Chromic Acid Anodized 1075 Aluminum.....	47
19. Chormic Acid Anodized 1075 Aluminum.....	48
20. Sulfuric Acid Anodized 7075 Aluminum.....	49

## LIST OF ILLUSTRATIONS (Continued)

FIGURE	PAGE
21. Sulfuric Acid Anodized 7075 Aluminum,.....	50
22. Martin Hard Coat Anodized 7075 Aluminum.....	51
23. Normal Spectral Reflectance of Specimen No. 50 Temperature, RT.....	106
24. Normal Spectral Reflectance of Specimen No. 50 Temperature, 600 F.....	107
25. Normal Spectral Reflectance of Specimen No. 50 Temperature, RT <sub>f</sub> .....	108
26. Normal Spectral Reflectance of Specimen No. 51 Temperature, RT.....	109
27. Normal Spectral Reflectance of Specimen No. 75 Temperature, RT.....	110
28. Normal Spectral Reflectance of Specimen No. 110 Temperature, RT.....	111
29. Normal Spectral Reflectance of Specimen No. 55 Temperature, RT.....	112
30. Normal Spectral Reflectance of Specimen No. 55 Temperature, 300 F.....	113
31. Normal Spectral Reflectance of Specimen No. 55 Temperature, 600 F.....	114
32. Normal Spectral Reflectance of Specimen No. 55 Temperature, 825 F.....	115
33. Normal Spectral Reflectance of Specimen No. 55 Temperature, RT <sub>f</sub> .....	116

## LIST OF ILLUSTRATIONS (Continued)

FIGURE	PAGE
34. Normal Spectral Reflectance of Specimen No. 56 Temperature, RT, HT-800 F.....	117
35. Normal Spectral Reflectance of Specimen No. 57 Temperature, RT.....	118
36. Normal Spectral Reflectance of Specimen No. 83 Temperature, RT.....	119
37. Normal Spectral Reflectance of Specimen No. 84 Temperature, RT.....	120
38. Normal Spectral Reflectance of Specimen No. 92 Temperature, RT.....	121
39. Normal Spectral Reflectance of Specimen No. 63 Temperature, RT.....	122
40. Normal Spectral Reflectance of Specimen No. 63 Temperature, 300 F.....	123
41. Normal Spectral Reflectance of Specimen No. 63 Temperature, 600 F.....	124
42. Normal Spectral Reflectance of Specimen No. 63 Temperature, 825 F.....	125
43. Normal Spectral Reflectance of Specimen No. 63 Temperature, RT <sub>f</sub> .....	126
44. Normal Spectral Reflectance of Specimen No. 64 Temperature, RT.....	127
45. Normal Spectral Reflectance of Specimen No. 65 Temperature, RT.....	128
46. Normal Spectral Reflectance of Specimen No. 66 Temperature, RT.....	129

## LIST OF ILLUSTRATIONS (Continued)

FIGURE	PAGE
47. Normal Spectral Reflectance of Specimen No. 73 Temperature, RT.....	130
48. Normal Spectral Reflectance of Specimen No. 74 Temperature, RT.....	131
49. Normal Spectral Reflectance of Specimen No. 77 Temperature, RT.....	132
50. Normal Spectral Reflectance of Specimen No. 78 Temperature, RT.....	133
51. Normal Spectral Reflectance of Specimen No. 120 Temperature, RT.....	134
52. Normal Spectral Reflectance of Specimen No. 67 Temperature, RT.....	135
53. Normal Spectral Reflectance of Specimen No. 68 Temperature, RT, HT-800 F.....	136
54. Normal Spectral Reflectance of Specimen No. 69 Temperature, RT.....	137
55. Normal Spectral Reflectance of Specimen No. 69 Temperature, 600 F.....	138
56. Normal Spectral Reflectance of Specimen No. 69 Temperature, RT <sub>f</sub> .....	139
57. Normal Spectral Reflectance of Specimen No. 71 Temperature, RT.....	140
58. Normal Spectral Reflectance of Specimen No. 71 Temperature, 300 F.....	141
59. Normal Spectral Reflectance of Specimen No. 72 Temperature, RT.....	142

## LIST OF ILLUSTRATIONS (Continued)

FIGURE	PAGE
60. Normal Spectral Reflectance of Specimen No. 80 Temperature, RT.....	143
61. Normal Spectral Reflectance of Specimen No. 81 Temperature, RT.....	144
62. Normal Spectral Reflectance of Specimen No. 11 Temperature, RT.....	145
63. Normal Spectral Reflectance of Specimen No. 85 Temperature, RT .....	146
64. Normal Spectral Reflectance of Specimen No. 86 Temperature, RT.....	147
65. Normal Spectral Reflectance of Specimen No. 87 Temperature, RT.....	148
66. Normal Spectral Reflectance of Specimen No. 87 Temperature, 300 F.....	149
67. Normal Spectral Reflectance of Specimen No. 87 Temperature, 600 F.....	150
68. Normal Spectral Reflectance of Specimen No. 87 Temperature, 825 F.....	151
69. Normal Spectral Reflectance of Specimen No. 87 Temperature, RT <sub>f</sub> .....	152
70. Normal Spectral Reflectance of Specimen No. 88 Temperature, RT, HT-800 F.....	153
71. Normal Spectral Reflectance of Specimen No. 89 Temperature, RT.....	154
72. Normal Spectral Reflectance of Specimen No. 129 Temperature, RT .....	155

LIST OF ILLUSTRATIONS (Continued)

FIGURE		PAGE
73.	Normal Spectral Reflectance of Specimen No. 13 Temperature, RT.....	156
74.	Normal Spectral Reflectance of Specimen No. 15 Temperature, RT.....	157
75.	Normal Spectral Reflectance of Specimen No. 15 Temperature, 600 F.....	158
76.	Normal Spectral Reflectance of Specimen No. 15 Temperature, RT.....	159
77.	Normal Spectral Reflectance of Specimen No. 20 Temperature, RT.....	160
78.	Normal Spectral Reflectance of Specimen No. 23 Temperature, RT.....	161
79.	Normal Spectral Reflectance of Specimen No. 82 Temperature, RT.....	162
80.	Normal Spectral Reflectance of Specimen No. 500 Temperature, RT.....	163
81.	Normal Spectral Reflectance of Specimen No. 52 Temperature, RT.....	164
82.	Normal Spectral Reflectance of Specimen No. 52 Temperature, 300 F.....	165
83.	Normal Spectral Reflectance of Specimen No. 52 Temperature, 600 F.....	166
84.	Normal Spectral Reflectance of Specimen No. 52 Temperature, 825 F.....	167
85.	Normal Spectral Reflectance of Specimen No. 52 Temperature, RT <sub>f</sub> .....	168

## LIST OF ILLUSTRATIONS (Continued)

FIGURE	PAGE
86. Normal Spectral Reflectance of Specimen No. 54 Temperature, RT.....	169
87. Normal Spectral Reflectance of Specimen No. 62 Temperature, RT.....	170
88. Normal Spectral Reflectance of Specimen No. 115 Temperature, RT.....	171
89. Normal Spectral Reflectance of Specimen No. 58 Temperature, RT, HT-800 F.....	172
90. Normal Spectral Reflectance of Specimen No. 59 Temperature, RT.....	173
91. Normal Spectral Reflectance of Specimen No. 59 Temperature, 300 F.....	174
92. Normal Spectral Reflectance of Specimen No. 59 Temperature, 600 F.....	175
93. Normal Spectral Reflectance of Specimen No. 59 Temperature, 825 F.....	176
94. Normal Spectral Reflectance of Specimen No. 59 Temperature, RT <sub>f</sub> .....	177
95. Normal Spectral Reflectance of Specimen No. 60 Temperature, RT.....	178
96. Normal Spectral Reflectance of Specimen No. 79 Temperature, RT.....	179
97. Normal Spectral Reflectance of Specimen No. 100 Temperature, RT.....	180
98. Normal Spectral Reflectance of Specimen No. 25 Temperature, RT.....	181

## LIST OF ILLUSTRATIONS (Continued)

FIGURE		PAGE
99.	Normal Spectral Reflectance of Specimen No. 25 Temperature, 350 F.....	182
100.	Normal Spectral Reflectance of Specimen No. 25 Temperature, RT, HT-800 F.....	183
101.	Normal Spectral Reflectance of Specimen No. 25 Temperature, RT <sub>f</sub> .....	184
102.	Normal Spectral Reflectance of Specimen No. 36 Temperature, RT.....	185
103.	Normal Spectral Reflectance of Specimen No. 138 Temperature, RT.....	186
104.	Normal Spectral Reflectance of Specimen No. 138 Temperature, RT, MF.....	187
105.	Normal Spectral Reflectance of Specimen No. 140 Temperature, RT.....	188
106.	Normal Spectral Reflectance of Specimen No. 140 Temperature, RT, HT-800 F.....	189
107.	Normal Spectral Reflectance of Specimen No. 140 Temperature, RT, MF.....	190
108.	Normal Spectral Reflectance of Specimen No. 155 Temperature, RT.....	191
109.	Normal Spectral Reflectance of Specimen No. 210 Temperature, RT.....	192
110.	Normal Spectral Reflectance of Specimen No. 211 Temperature, RT.....	193
111.	Normal Spectral Reflectance of Specimen No. 211 Temperature, 350 F.....	194

## LIST OF ILLUSTRATIONS (Continued)

FIGURE	PAGE
112. Normal Spectral Reflectance of Specimen No. 211 Temperature, RT <sub>f</sub> .....	195
113. Normal Spectral Reflectance of Specimen No. 211 Temperature, RT, HT-800 F.....	196
114. Normal Spectral Reflectance of Specimen No. 215 Temperature, RT.....	197
115. Normal Spectral Reflectance of Specimen No. 31 Temperature, RT.....	198
116. Normal Spectral Reflectance of Specimen No. 31 Temperature, 350 F.....	199
117. Normal Spectral Reflectance of Specimen No. 31 Temperature, RT <sub>f</sub> .....	200
118. Normal Spectral Reflectance of Specimen No. 35 Temperature, RT, HT-800 F.....	201
119. Normal Spectral Reflectance of Specimen No. 35 Temperature, RT.....	202
120. Normal Spectral Reflectance of Specimen No. 35 Temperature, 350 F.....	203
121. Normal Spectral Reflectance of Specimen No. 35 Temperature, RT <sub>f</sub> .....	204
122. Normal Spectral Reflectance of Specimen No. 152 Temperature, RT.....	205
123. Normal Spectral Reflectance of Specimen No. 152 Temperature, RT, MF.....	206
124. Normal Spectral Reflectance of Specimen No. 158 Temperature, RT.....	207

## LIST OF ILLUSTRATIONS (Continued)

FIGURE	PAGE
125. Normal Spectral Reflectance of Specimen No. 212 Temperature, RT.....	208
126. Normal Spectral Reflectance of Specimen No. 213 Temperature, RT.....	209
127. Normal Spectral Reflectance of Specimen No. 214 Temperature, RT.....	210
128. Normal Spectral Reflectance of Specimen No. 214 Temperature, 350 F.....	211
129. Normal Spectral Reflectance of Specimen No. 214 Temperature, RT <sub>f</sub> .....	212
130. Normal Spectral Reflectance of Specimen No. 214 Temperature, RT, HT-800 F.....	213
131. Normal Spectral Reflectance of Specimen No. 38 Temperature, RT.....	214
132. Normal Spectral Reflectance of Specimen No. 43 Temperature, RT.....	215
133. Normal Spectral Reflectance of Specimen No. 203 Temperature, RT.....	216
134. Normal Spectral Reflectance of Specimen No. 207 Temperature, RT.....	217
135. Normal Spectral Reflectance of Specimen No. 45 Temperature, RT.....	218
136. Normal Spectral Reflectance of Specimen No. 45 Temperature, 300 F.....	219
137. Normal Spectral Reflectance of Specimen No. 45 Temperature, 600 F.....	220

LIST OF ILLUSTRATIONS (Continued)

FIGURE		PAGE
138.	Normal Spectral Reflectance of Specimen No. 45 Temperature, 900 F. ....	221
139.	Normal Spectral Reflectance of Specimen No. 45 Temperature, $RT_f$ .....	222
140.	Normal Spectral Reflectance of Specimen No. 48 Temperature, $RT$ .....	223
141.	Normal Spectral Reflectance of Specimen No. 48 Temperature, 500 F.....	224
142.	Normal Spectral Reflectance of Specimen No. 48 Temperature, 1000 F.....	225
143.	Normal Spectral Reflectance of Specimen No. 48 Temperature, $RT_f$ (1000 F).....	226
144.	Normal Spectral Reflectance of Specimen No. 48 Temperature, 1300 F.....	227
145.	Normal Spectral Reflectance of Specimen No. 48 Temperature, $RT_f$ (1300 F).....	228
146.	Normal Spectral Reflectance of Specimen No. 178 Temperature, $RT$ .....	229
147.	Normal Spectral Reflectance of Specimen No. 37 Temperature, $RT$ .....	230
148.	Normal Spectral Reflectance of Specimen No. 41 Temperature, $RT$ .....	231
149.	Normal Spectral Reflectance of Specimen No. 47 Temperature, $RT$ .....	232
150.	Normal Spectral Reflectance of Specimen No. 47 Temperature, 500 F. ....	233

## LIST OF ILLUSTRATIONS (Continued)

FIGURE	PAGE
151. Normal Spectral Reflectance of Specimen No. 47 Temperature, 1000 F.....	234
152. Normal Spectral Reflectance of Specimen No. 47 Temperature, 1300 F.....	235
153. Normal Spectral Reflectance of Specimen No. 47 Temperature, $RT_f$ .....	236
154. Normal Spectral Reflectance of Specimen No. 197 Temperature, $RT$ .....	237
155. Normal Spectral Reflectance of Specimen No. 199 Temperature, $RT$ .....	238
156. Normal Spectral Reflectance of Specimen No. 201 Temperature, $RT$ .....	239
157. Normal Spectral Reflectance of Specimen No. 201 Temperature, 500 F.....	240
158. Normal Spectral Reflectance of Specimen No. 201 Temperature, 1000 F.....	241
159. Normal Spectral Reflectance of Specimen No. 201 Temperature, $RT$ , HT-1500 F.....	242
160. Normal Spectral Reflectance of Specimen No. 201 Temperature, $RT_f$ .....	243
161. Normal Spectral Reflectance of Specimen No. 162 Temperature, $RT$ .....	244
162. Normal Spectral Reflectance of Specimen No. 164 Temperature, $RT$ .....	245
163. Normal Spectral Reflectance of Specimen No. 166 Temperature, $RT$ .....	246

**LIST OF ILLUSTRATIONS (Continued)**

FIGURE	PAGE
164. Normal Spectral Reflectance of Specimen No. 166 Temperature, 400 F.....	247
165. Normal Spectral Reflectance of Specimen No. 166 Temperature, 800 F.....	248
166. Normal Spectral Reflectance of Specimen No. 166 Temperature, 1200 F.....	249
167. Normal Spectral Reflectance of Specimen No. 166 Temperature, $RT_f$ .....	250
168. Normal Spectral Reflectance of Specimen No. 173 Temperature, $RT$ .....	251
169. Normal Spectral Reflectance of Specimen No. 173 Temperature, 400 F.....	252
170. Normal Spectral Reflectance of Specimen No. 173 Temperature, 800 F.....	253
171. Normal Spectral Reflectance of Specimen No. 173 Temperature, 1200 F.....	254
172. Normal Spectral Reflectance of Specimen No. 173 Temperature, $RT_f$ .....	255
173. Normal Spectral Reflectance of Specimen No. 177 Temperature, $RT$ .....	256
174. Normal Spectral Reflectance of Specimen No. 177 Temperature, $RT$ , HT-1500 F.....	257
175. Normal Spectral Reflectance of Specimen No. 167 Temperature, $RT$ .....	258
176. Normal Spectral Reflectance of Specimen No. 174 Temperature, $RT$ .....	259

## LIST OF ILLUSTRATIONS (Continued)

FIGURE	PAGE
177. Normal Spectral Reflectance of Specimen No. 176 Temperature, RT.....	260
178. Normal Spectral Reflectance of Specimen No. 168 Temperature, RT.....	261
179. Normal Spectral Reflectance of Specimen No. 170 Temperature, RT.....	262
180. Normal Spectral Reflectance of Specimen No. 172 Temperature, RT.....	263
181. Normal Spectral Reflectance of Specimen No. 18 Temperature, RT.....	264
182. Normal Spectral Reflectance of Specimen No. 156 Temperature, RT.....	265
183. Normal Spectral Reflectance of Specimen No. 171 Temperature, RT.....	266
184. Normal Spectral Reflectance of Specimen No. 46 Temperature, RT.....	267
185. Normal Spectral Reflectance of Specimen No. 209 Temperature, RT.....	268
186. Normal Spectral Reflectance of Specimen No. 202 Temperature, RT.....	269

## LIST OF TABLES

TABLE		PAGE
1.	Reflectance of Aluminized Hemisphere.....	26
2.	Comparison of Aluminum Diffuser Reflectance Measurements.....	36
3.	Reflectometer Calibration Data.....	39
4.	Reflectance of Anodized Aluminum Measured at UCLA.....	45
5.	Anodized Coating Thickness.....	62
6.	Index of Specimen Variables.....	76
7.	Spectral Reflectance of 7075 Aluminum Anodized in Chromic Acid.....	79
8.	Spectral Reflectance of 1075 Aluminum in Chromic Acid.....	80
9.	Spectral Reflectance of 7075 Aluminum Anodized in Sulfuric Acid.....	81
10.	Spectral Reflectance of 1075 Aluminum Anodized in Sulfuric Acid.....	83
11.	Spectral Reflectance of 7075 Aluminum Hard Coat Anodized.....	84
12.	Spectral Reflectance of 1075 Aluminum Hard Coat Anodized.....	85
13.	Spectral Reflectance of 7075 Aluminum Anodized in Boric Acid...	86
14.	Spectral Reflectance of 1075 Aluminum Anodized in Boric Acid...	87
15.	Spectral Reflectance of AZ31B Magnesium Anodized in HAE.....	88
16.	Spectral Reflectance of HK31A Magnesium Anodized in HAE.....	89
17.	Spectral Reflectance of AZ31B Magnesium Anodized in Dow 17....	90

**LIST OF TABLES (Continued)**

TABLE	PAGE
18 Spectral Reflectance of HK31A Magnesium Anodized in Dow 17...	91
19 Spectral Reflectance of A110 AT Titanium Anodized in Sulfuric Acid.....	92
20 Spectral Reflectance of B120 VAC Titanium Anodized in Sulfuric Acid.....	93
21 Spectral Reflectance of A110AT Titanium Anodized in Sodium Hydroxide.....	94
22 Spectral Reflectance of A110AT Titanium Anodized by a Proprietary Process.....	96
23 Spectral Reflectance of B120 VAC Titanium Anodized by a Proprietary Process.....	97
24 Spectral Reflectance of Beryllium, 1% Copper Anodized in Sulfuric Acid.....	98
25 Spectral Reflectance of Beryllium, 1% Copper Anodized in Sodium Hydroxide.....	99
26 Spectral Reflectance of QMV Beryllium Anodized in Sodium Hydroxide.....	100
27 Spectral Reflectance of Beryllium, 1% Copper Anodized in Nitric Acid.....	101
28 Spectral Reflectance of QMV Beryllium Anodized in Nitric Acid..	102
29 Spectral Reflectance of Beryllium, 1% Copper Anodized in Chromic Acid.....	103
30 Spectral Reflectance of QMV Beryllium Anodized in Chromic Acid.....	104
31 Spectral Reflectance of Anodized Metals.....	105

## PART I

### INTRODUCTION

Means for passive temperature control are of primary importance in the successful operation of satellites and space vehicles. Since there is no atmosphere in space, heat transfer to and from an object occurs only by radiation. Solar energy incident on the surface is either absorbed or reflected, and internal heat from the vehicle is dissipated as infrared radiation. Solar energy is concentrated in the short-wavelength region of the spectrum -- 90% occurs at wavelengths shorter than 1.6 microns, whereas low-temperature radiation from a satellite or space vehicle is confined to the long-wave infrared region. Variations in the spectral radiation properties of the surface of such vehicles can, therefore, be used to achieve passive temperature control.

Bare metals are good reflectors for both short- and long-wavelength radiation. They are relatively poorer reflectors for solar energy, however, and therefore attain undesirably high equilibrium temperatures in space. Materials such as white paint have more desirable thermal radiation properties but are frequently unstable in the high vacuum of interplanetary space. Paint pigments also tend to be unstable in the intense ultraviolet radiation from the sun.

Anodized coatings on metals have many desirable physical properties, and the object of this investigation was to study the effect of the various variables in the anodizing processes and a vacuum environment on the thermal radiation properties of several anodized metals. The spectral reflectance of aluminum, magnesium, titanium, and beryllium anodized by a number of different processes was measured over the range of 0.4 to about 22 microns. Measurements were made on two different alloys of each metal in a vacuum of about  $10^{-6}$  mm of Hg with specimen temperatures up to 825F for aluminum, 350F for magnesium, 1300F for titanium, and 1200F for beryllium.

Reflectance measurements were chosen because there would be insufficient energy for emittance measurements in the case of most of the specimen temperatures to cover the desired spectral range. The requirements of variable specimen temperature and high-vacuum eliminated many of the measurement techniques commonly used. A Gier-Dunkle, black-body reflectometer especially designed for vacuum operation was originally proposed and tried. Serious evaporation and outgassing problems made it impossible to use this device, however. A modified Coblentz integrating hemisphere was then conceived and was successfully used.

## DISCUSSION

Four properties of materials are important in radiation calculations. These are emissivity ( $\epsilon$ ), absorptivity ( $\alpha$ ), reflectivity ( $\rho$ ) and transmissivity ( $\tau$ ). An energy balance for the case of radiation incident upon a surface shows that the radiation is either absorbed, reflected, or transmitted. Therefore

$$\alpha + \rho + \tau = 1 \quad (1)$$

Most solid materials--and metals especially--are opaque to thermal radiation ( $\tau = 0$ ) and

$$\alpha + \rho = 1 \quad (2)$$

Equations (1) and (2) are true for either the total radiation integrated over the entire spectrum or the spectral radiation at any particular wavelength.

$$\alpha_\lambda + \rho_\lambda = 1 \quad (3)$$

Kirchhoff's law states that for a surface at a given temperature

$$\alpha_\lambda = \epsilon_\lambda. \quad (4)$$

Equations (3) and (4) show that if any one property can be measured, the others can be calculated.

The temperature range of interest extends down to 100 F and the wavelength range extends down to 0.4 microns. The amount of energy radiated by a surface at 100 F in the 0.4 micron region is infinitesimal and it would be difficult if not impossible to measure the emissivity directly under these conditions.

Energy of any desired intensity (provided it does not heat the surface unduly) can be projected onto a surface for reflectance measurements, however. For these reasons spectral reflectance was chosen as the property to measure

## BLACK-BODY REFLECTOMETER

The original proposal was to use a black-body reflectometer of the type developed by Gier, Dunkle, and Bevans (2) for reflectance measurements. This device consists of a heated hohlraum (black-body enclosure) in which the specimen is mounted on a cooled mount as shown in Figure 1. A viewing aperture is located in the wall opposite the specimen. The hot walls of the hohlraum radiate to the specimen and some of this energy is reflected out the aperture to a sensor. One measurement is made of the energy reflected by the specimen and a reference measurement is made of the energy radiated from another part of the hohlraum. The reflectance is then equal to the ratio of the signals. If the specimen temperature approaches the hohlraum temperature, a correction for the energy emitted by the specimen is also required. A derivation of the equations and detailed discussion of this device is given in the appendix.

A ceramic hohlraum and vacuum chamber were built and measurements were attempted. In order to have sufficient energy for measurements at the short- and long-wavelength ends of the spectrum, it was estimated that the hohlraum would have to operate at about 2000 F. Specimen temperatures were to be varied from 100 to 1000 F. A serious evaporation problem was encountered at a hohlraum temperature of about 1200 F and a vacuum of  $10^{-4}$  mm of Hg, however. Under these conditions, material that evaporated from the hot parts condensed on the relatively cold (100-200 F) specimen and specimen holder. A black residue was scraped from the specimen and holder and was analyzed by x-ray and infrared spectrographic methods. It was found to be predominantly lead molybdate ( $Pb \cdot MoO_4$ ). The molybdenum obviously came from the molybdenum radiation shield which showed evidence of oxidation. The only source of lead in the system was soft solder used to seal the heater leads into the tubular hermetic seals. It was expected that some oxygen and nitrogen would be liberated from the ceramic parts during the warmup. Since these gases are transparent in the wavelengths region of interest, it was not expected that they would interfere with the measurements. Materials with vapor pressures well below the chamber

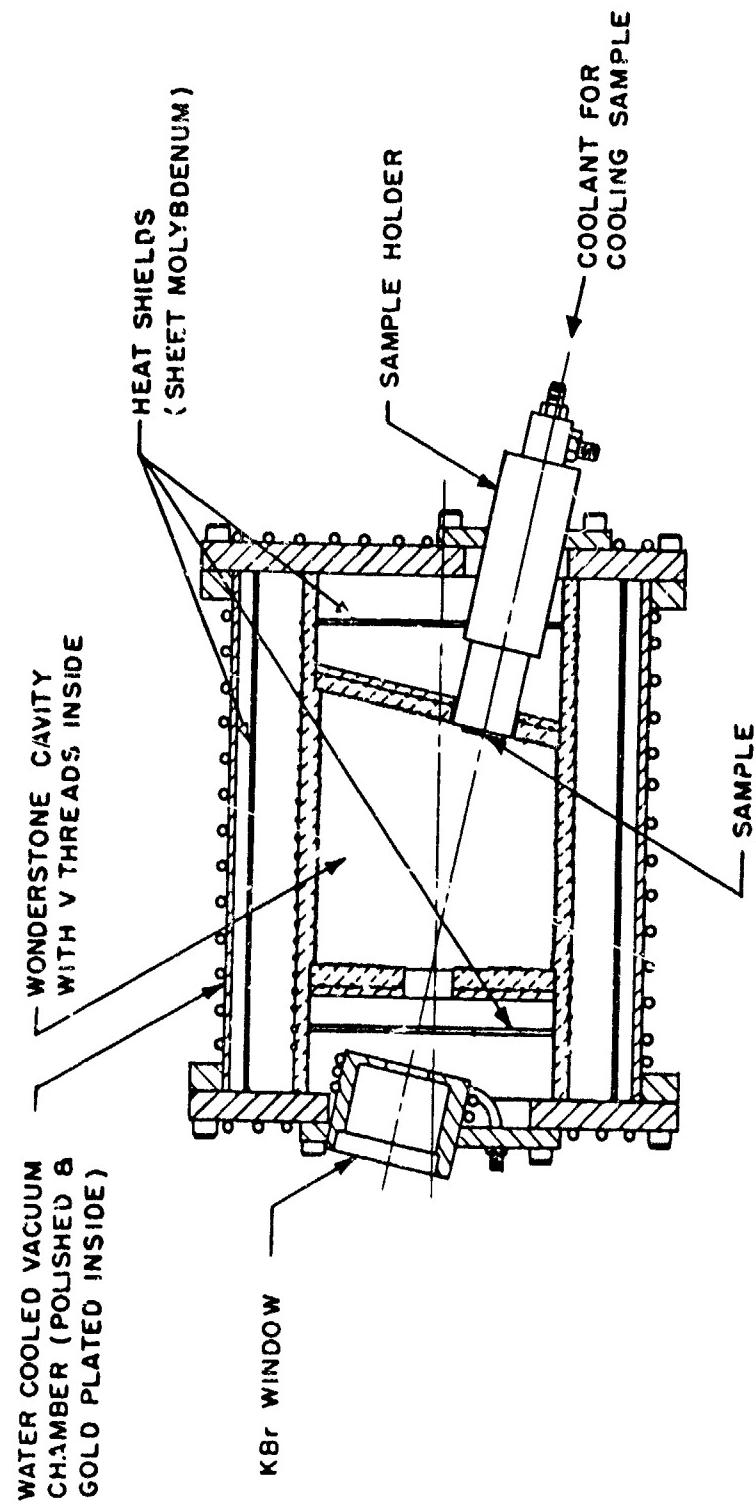


Figure 1 - BLACK BODY REFLECTOMETER

pressures, were selected for the hot parts. Apparently, however, the oxygen liberated from the hot ceramic attacked the molybdenum radiation shield and formed intermediate oxides with a relatively high vapor pressure which then evaporated at higher temperatures. Also it appeared that sufficient heat was conducted down the power leads to evaporate lead from the soft solder at the seals. This lead and molybdenum oxide vapor then must have combined and condensed on the specimen which was the coolest surface in the hohlraum.

This condensation on the specimen aggravated a second problem. The specimen in the form of a 1-1/8" diam. disk was clamped to the end of a water cooled mount by a ring around its edge. As the hohlraum temperature increased, the increasing heat flow through the specimen to the mount raised the specimen temperature slightly which produced sufficient thermal expansion to cause the specimen to "oil can" and separate from the specimen holder. Once this happened, the specimen cooling mechanism was virtually destroyed and the specimen temperature rapidly rose to the melting point.

Because of the very fundamental nature of these problems, it appeared advisable to conclude that the black-body reflectometer was not suitable for vacuum measurements and to abandon that technique. A modified version of the integrating hemisphere used by Coblentz was then conceived and used for the study.

#### INTEGRATING HEMISPHERE REFLECTOMETER

The earliest use of a highly specular reflecting hemisphere to collect the scattered energy reflected by a diffuse surface and concentrate the energy on a sensor is attributed to Paschen in 1899 (3). Coblentz published considerable data obtained with a similar instrument about 10 years later. In these devices a hollow, highly polished, glass hemisphere with a highly reflective evaporated metal coating was used to approximate an ellipsoidal mirror. A specimen was located a short distance to one side of the center of the hemisphere and a thermopile was located at a conjugate position. Radiant energy from a suitable

source was projected through an aperture in the hemisphere onto the specimen. The reflected energy was collected by the hemisphere and focused on the thermopile. A reference measurement was made by projecting the same energy directly onto the thermopile. The reflectance of the specimen was then equal to the ratio of the two measurements divided by the reflectance of the hemisphere.

In order to adapt this technique to spectral measurements and heated specimens, it was necessary to introduce a diffuser at one of the conjugate foci and use an external optical system to focus the energy into a spectrometer. The wide spectral range of interest and the low energy levels, required the use of a photo multiplier sensor for the visible region and a vacuum thermocouple for the infrared.

### Theory

Figure 2 shows a schematic of the integrating hemisphere. A diffuse reflector (diffuser) was located on a diameter of the hemisphere a short distance to one side of the center and the specimen was located at a conjugate position. The hemisphere was large enough compared to the distance between the diffuser and specimen so that it approximated an ellipsoid.

The energy from a tungsten lamp or globar was projected through an aperture and focused on the diffuser. The energy reflected from the diffuser was re-collected by the hemisphere and focused onto the specimen. The diffuser reflected with a good approximation to Lambert's cosine law, thus irradiating the specimen with a hemispherical distribution of incident energy. Some of the energy was reflected out the exit aperture and some was re-reflected back to the diffuse reflector. An infinite number of internal reflections between the diffuser and specimen occurred with some energy being reflected out the exit aperture to the spectrometer with each internal reflection.

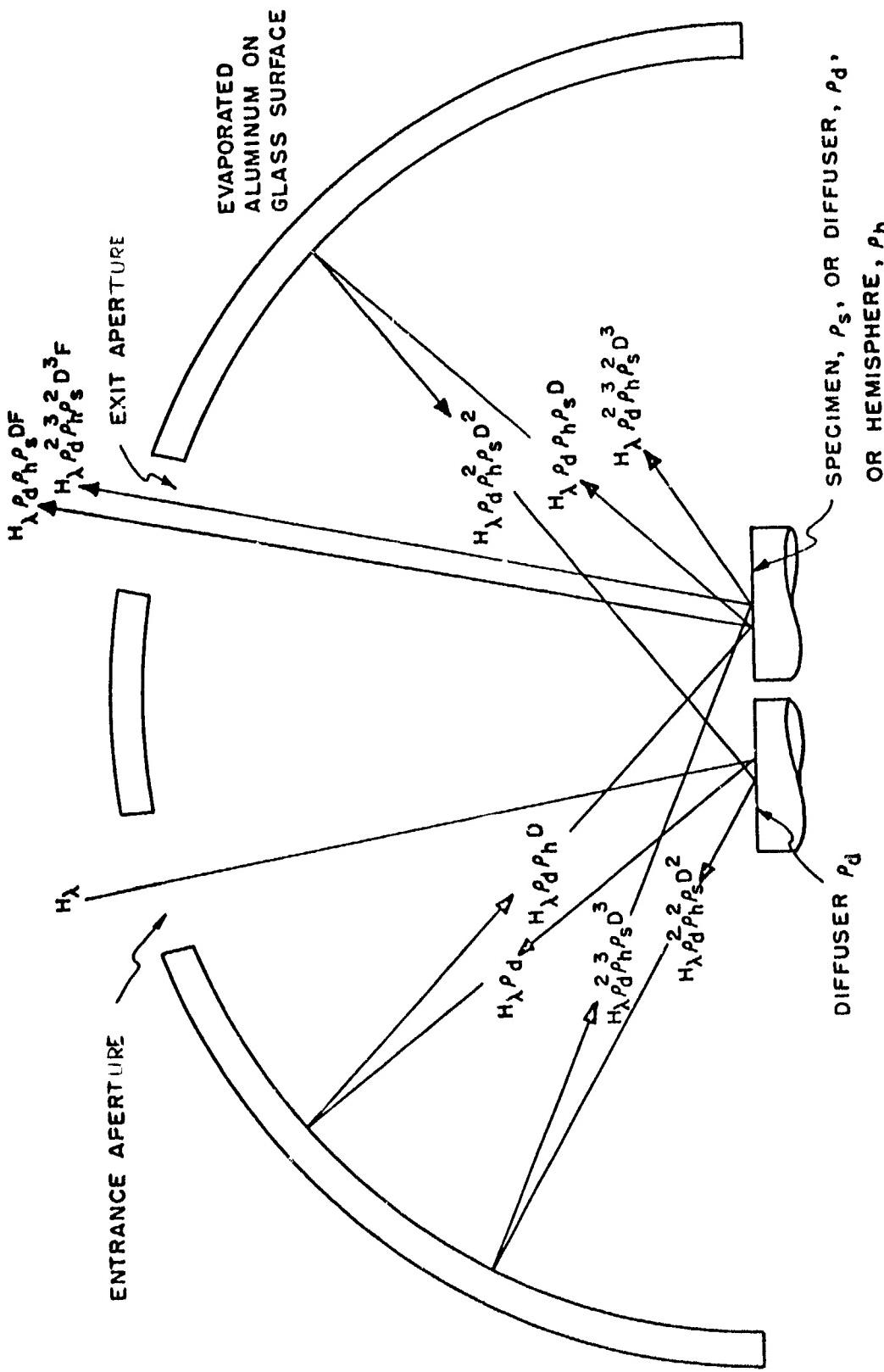


Figure 2 - ENERGY REFLECTION IN INTEGRATING HEMISPHERE

The equation for the energy reaching the sensor was derived by following a ray of energy through the system as shown in Figure 2. The summation of the energy reaching the sensor was a geometric series as follows:

$$G_{ds} = H_\lambda \rho_d \rho_h \rho_s D F \left[ 1 + \rho_d \rho_h^2 \rho_s D^2 + \rho_d^2 \rho_h^4 \rho_s^2 D^4 + \dots \right] \quad (5)$$

$G_{ds}$  = energy at sensor for diffuser and specimen at wavelength  $\lambda$  \*

$H_\lambda$  = incident energy at wavelength  $\lambda$

$\rho_d$  = diffuser reflectance at wavelength  $\lambda$

$\rho_h$  = hemisphere reflectance at wavelength  $\lambda$

$\rho_s$  = specimen reflectance at wavelength  $\lambda$

$D$  = factor to account for energy lost out apertures =  $1 - F_i - F_e$   
(see calculation page 24)

$F_i$  = solid angle between diffuser or specimen and inlet aperture

$F_e$  = solid angle between diffuser or specimen and exit aperture

$F$  = solid angle factor between specimen and sensor

The sum of a finite number of terms in a geometric series is given by

$$S = a \frac{1 - r^{n+1}}{1 - r} \quad (6)$$

The sum of the series for a finite number of internal reflections was therefore:

$$G_{ds} = \rho_s \rho_d \rho_h H_\lambda DF \frac{\left[ 1 - (\rho_s \rho_d \rho_h^2 D^2)^{n+1} \right]}{\left[ 1 - \rho_s \rho_d \rho_h^2 D^2 \right]} \quad (7)$$

---

\* Throughout this report energies and reflectances are spectral values at wavelength  $\lambda$ . Subscript,  $\lambda$ , has been omitted to simplify notation.

A similar equation was written for the case of two diffusers:

$$G_{dd} = \rho_d^2 \rho_h^2 H_\lambda DF \frac{\left[ 1 - (\rho_d \rho_h D)^{n+1} \right]}{\left[ 1 - \rho_d^2 \rho_h^2 D^2 \right]} \quad (8)$$

Also, if one of the diffusers was replaced by a surface identical to the hemisphere surface (h), the energy reaching the sensor was given by:

$$G_{dh} = \rho_d \rho_h^2 H_\lambda F \frac{\left[ 1 - (\rho_d \rho_h^3 D^2)^{n+1} \right]}{\left[ 1 - \rho_d \rho_h^3 D^2 \right]} \quad (9)$$

The loss factor, D, does not appear in the first term of the series from which equation (9) is derived when the specimen or second surface is highly specular. The reason for this is that the energy reflected by the specular surface to the sensor comes from a particular, small area on the hemisphere and the energy lost out the apertures after reflecting from the diffuser does not effect the energy incident on this spot. When the energy is reflected back to the diffuser for the second pass, this loss is integrated by the diffuser and must be included, however.

From the ratio of the signals  $G_{ds}$  and  $G_{dd}$ :

$$\frac{G_{ds}}{G_{dd}} = R_{sd} \frac{\left[ 1 - Z_d \right] \left[ 1 - (R_{sd} Z_d)^{n+1} \right]}{\left[ 1 - R_{sd} Z_d \right] \left[ 1 - Z_d^{n+1} \right]} \quad (10)$$

$$\text{where } R_{sd} = \frac{\rho_s}{\rho_d} \quad (11)$$

$$Z_d = \rho_d^2 \rho_h^2 D^2 \quad (12)$$

The specimen reflectance  $\rho_s$  could then be determined if  $\rho_d$ ,  $\rho_h$ , and  $n$  were known. The loss factor,  $D$ , is calculated later. Also from the signals  $G_{ds}$  and  $G_{dh}$  a second equation for specimen reflectance was obtained.

$$\frac{G_{ds}}{G_{dh}} = R_{sh} D \frac{\left[1 - (R_{sh} Z_h)^{n+1}\right] \left[1 - Z_h\right]}{\left[1 - R_{sh} Z_h\right] \left[1 - Z_h^{n+1}\right]} \quad (13)$$

$$\text{where } R_{sh} = \frac{\rho_s}{\rho_h} \quad (14)$$

$$Z_h = \rho_d \rho_h^3 D^2 \quad (15)$$

The procedures used in determining  $\rho_d$ ,  $\rho_h$ , and  $n$  are covered in the section on calibration.

#### Description of Apparatus

The integrating hemisphere reflectometer along with its optical system and vacuum chamber were designed for use with the Perkin-Elmer Model 112 infrared spectrometer. The spectrometer and special image rotating mirror on the entrance are shown in Figure 3.

A war surplus mbsight bubble was used as the hemisphere. It was actually slightly more than a hemisphere and had a steel mounting flange attached. The hemisphere was glass, precision ground and polished to approximately 9-3/8" inside diameter. Two 1" diameter holes were drilled in the hemisphere for the entrance and exit apertures and the inside was aluminized by vacuum depositing an opaque coating of high-purity aluminum.

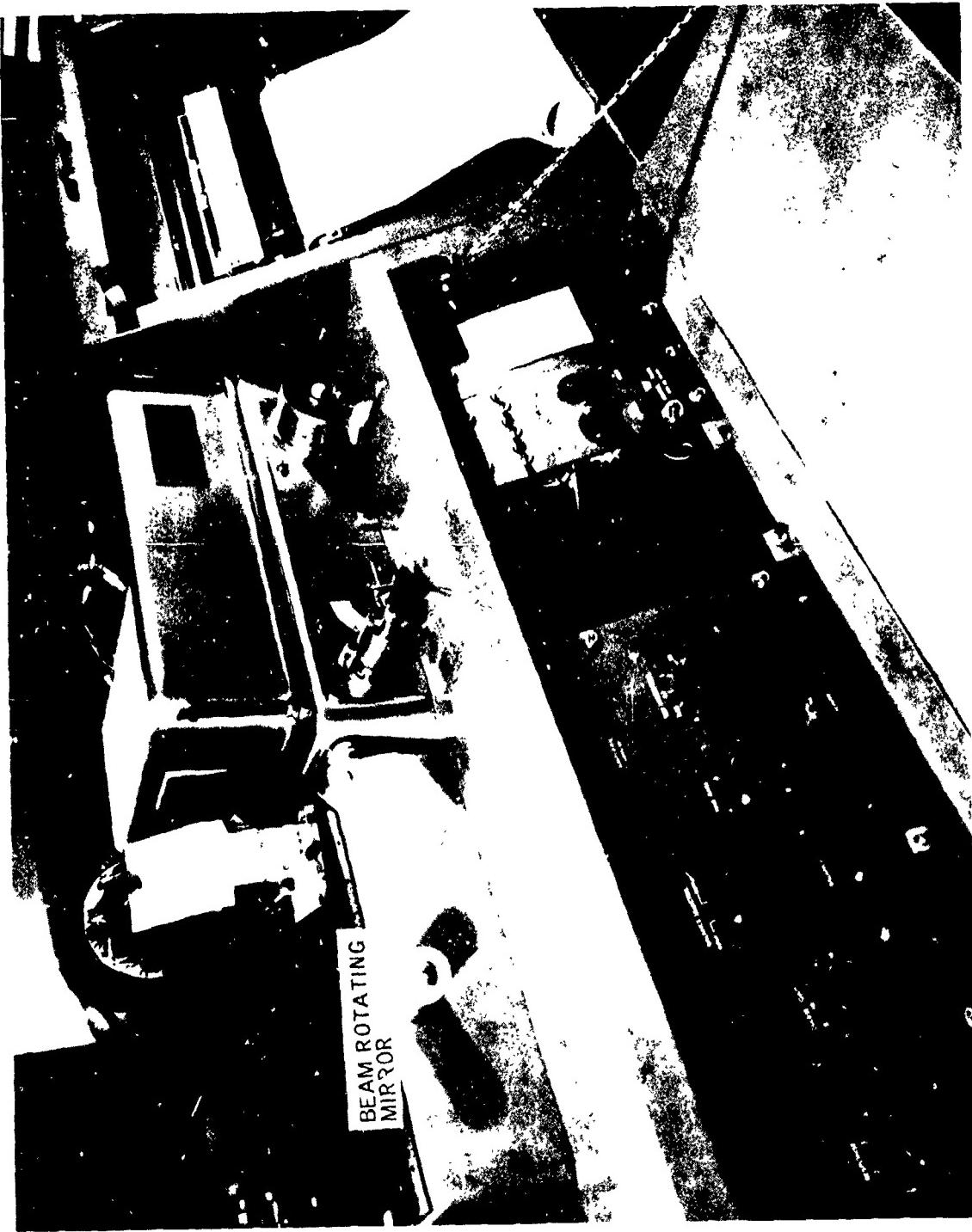


Figure 3 - SPECTROMETER

A turret with eight positions was provided for mounting specimens and the various diffusers and reference surfaces at the proper location with respect to the center of the hemisphere. Ceramic specimen holders with tungsten heating elements were provided for mounting specimens at two positions and a chromel-alumel thermocouple welded to a small flat nickel strip was held to each specimen surface with a spring-loaded clip made of tungsten wire. Specimen temperatures were recorded with a Brown recorder. Figure 4 shows the hemisphere, and turret with the various diffusers, specimen heaters and a specimen. The heaters were powered by storage batteries and a rheostat was used for control purposes. Water cooling was provided to keep the diffusers at approximately room temperature.

Optical alignment of the diffuser and specimen with respect to the integrating hemisphere was quite critical. An external nut on the turret tube and means for positive indexing were, therefore, provided to adjust the location of the specimen and diffuser surfaces with respect to the center of the hemisphere. The height of the individual surfaces on the turret could also be independently adjusted to bring any two or more to the same height and thus make allowance for variation of specimen thickness.

The hemisphere was mounted on the rear end plate of the vacuum chamber and the turret was mounted on a second plate bolted to the rear end plate as shown in Figure 5. This arrangement made it possible to remove the turret to change specimens without disturbing the hemisphere. A rotary seal consisting of two O-rings lubricated with diffusion pump oil was provided for turret rotation. Two 3-inch diam. potassium bromide windows were provided in the front end plate and a heater was provided to keep these windows at a temperature of about 100 F to prevent fogging. The vacuum gage was also mounted in the front end plate. Neoprene O-ring gaskets were used at all joints.

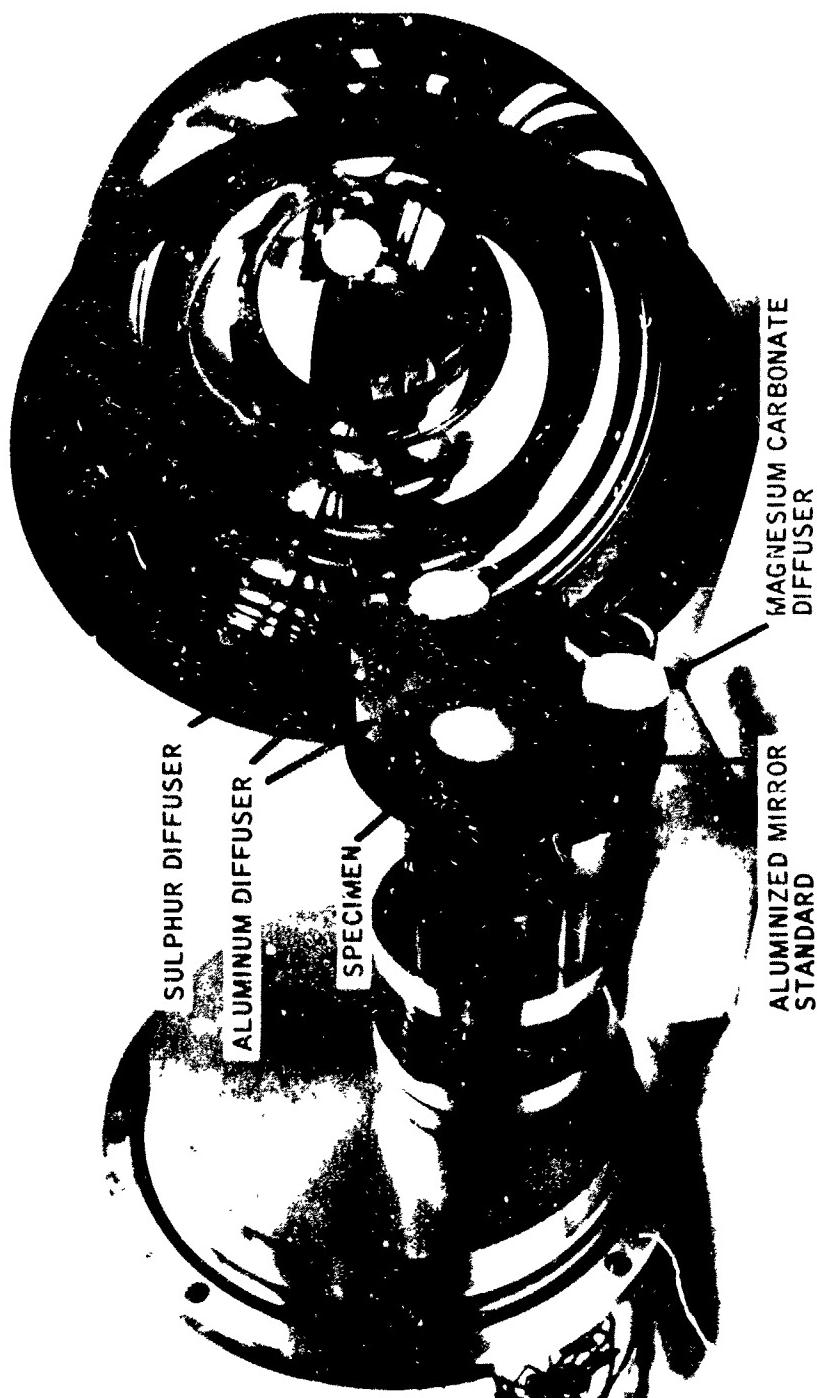


Figure 4 - HEMISPHERE AND TURRET

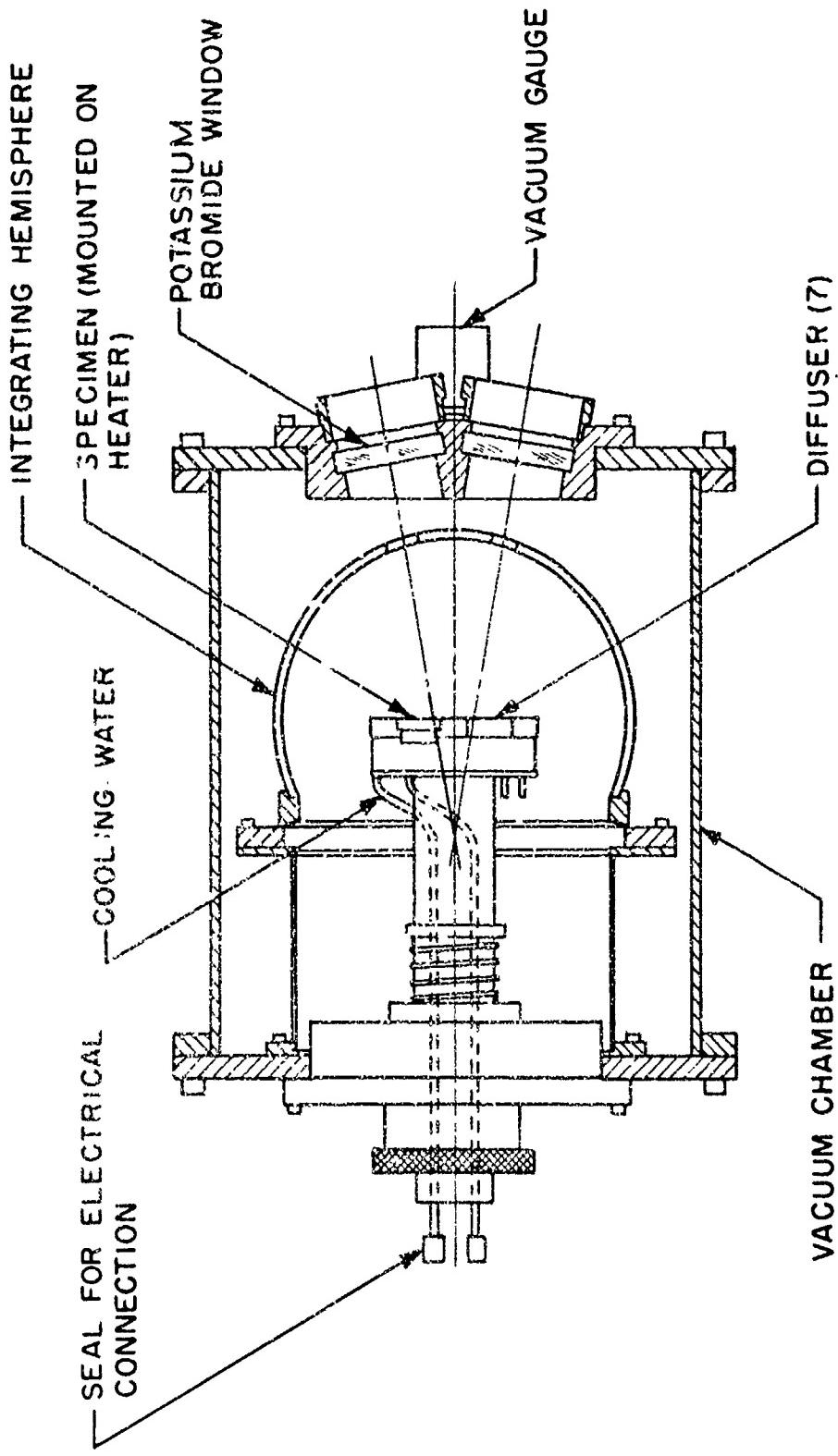


Figure 5 - VACUUM CHAMBER AND HEMISPHERE

The vacuum chamber and pump assembly constructed for the black-body reflectometer were also used with the integrating hemisphere. The chamber was a brass cylinder 12" diameter by 15" long. A 4" diam. tee connection fastened directly to the vacuum valve. The liquid nitrogen cold trap, water-cooled oil vapor baffel, and 4 in. oil diffusion pump were connected directly below the vacuum valve as shown in Figure 6. A large capacity fore pump was provided as a back-up for the diffusion pump. This arrangement of large-diameter, short connections provided maximum pumping speed. From a cold start, the system could be pumped down to a pressure of  $10^{-5}$  mm Hg in approximately one hour. If the diffusion pump was pre-heated, this time was reduced to 10 minutes.

The vacuum chamber, valve, pump, optical system, and associated equipment were mounted on a steel framework (Fig. 6) which was in turn bolted to the console of the spectrometer. The whole assembly was raised off the floor on screw jacks to achieve and maintain the desired optical alignment.

The optical system is shown in Figure 7. Since a globar did not provide enough energy at the short wavelengths, it was necessary to provide a tungsten ribbon filament lamp for the visible region. A two position, mirror was provided for switching from one source to the other (Section A-A, Figure 7). The slit on the entrance system was mounted in a 5/8" diam. brass tube which was cemented in a hole in the middle of the double sided 45° mirror. This mirror was aluminized on both sides. Two 6 in. diam. f/1 spherical mirrors were required to produce an image of the source slit on the diffuser. The image was about 1/8" wide by 5/8" long, and in line with a horizontal line connecting the center of the diffuser and specimen.

Energy reflected from the specimen out the exit aperture was deflected by a plane mirror to a 6 in. f/2.5 spherical mirror; it then passed through a beam rotating mirror arrangement (Section B-B Fig. 7) and finally was focused by the spectrometer source parabola onto the spectrometer entrance slits. The

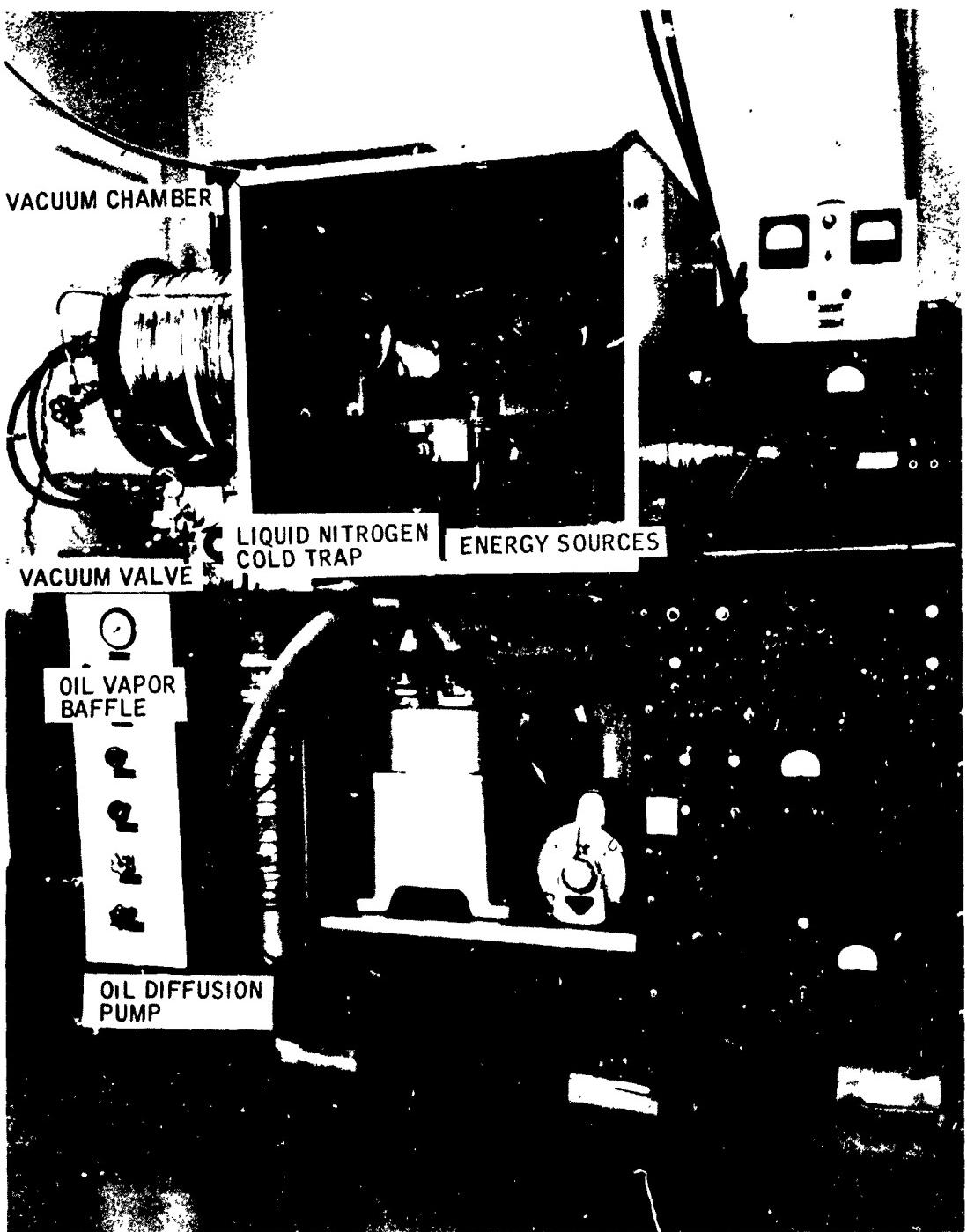


Figure 6 - INTEGRATING HEMISPHERE REFLECTOMETER

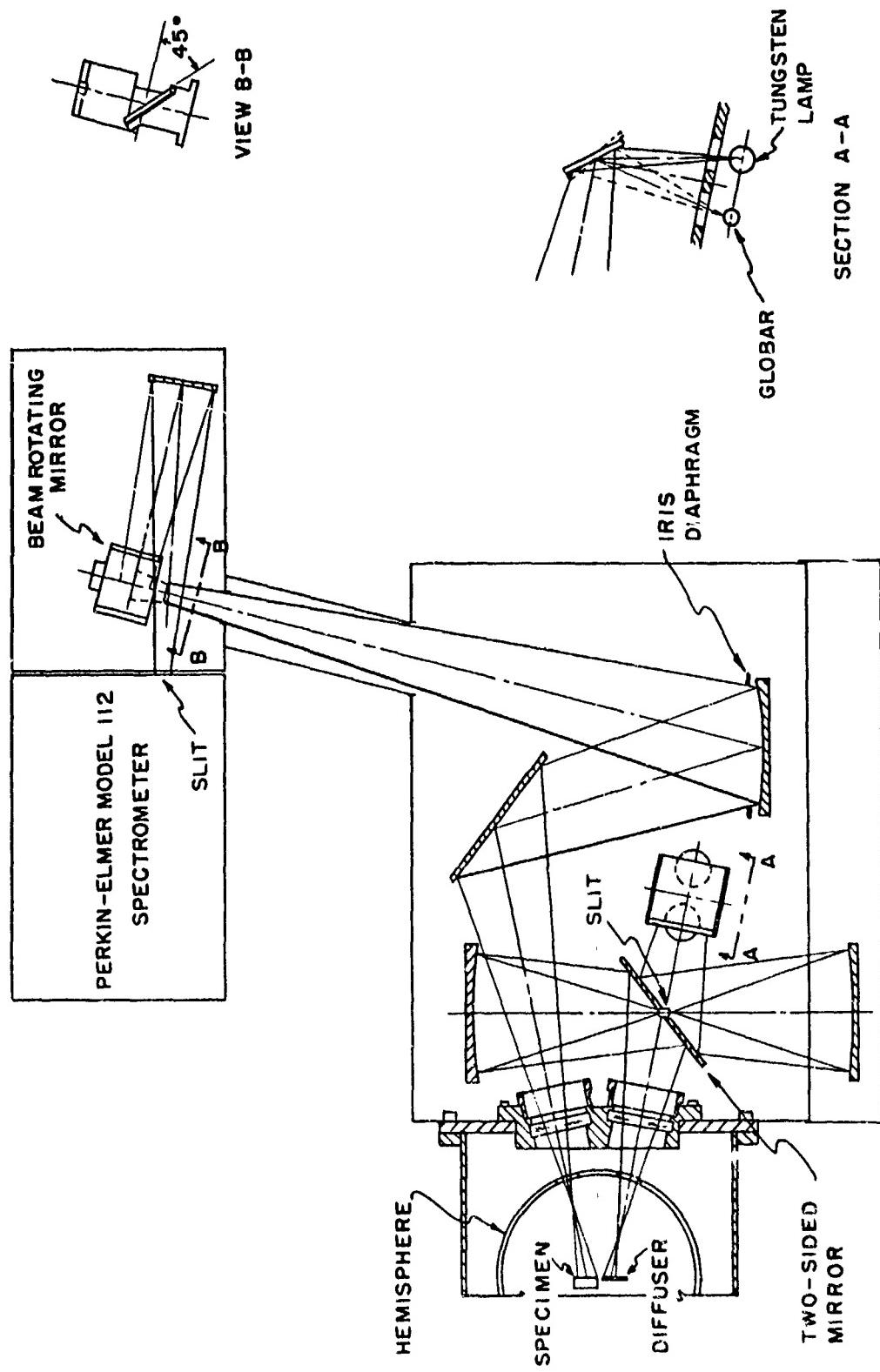


Figure 7 - REFLECTOMETER OPTICAL SYSTEM

images on the diffuser and specimen were horizontal and the spectrometer slits were vertical. Therefore, it was necessary to incorporate a beam rotating mirror in the optical system. Horizontal in-line images were used to reduce the effect of spherical aberration.

Provisions were made for adjusting all of the mirrors to achieve proper optical alignment. The sources and mirrors, with the exception of the beam rotating assembly and spectrometer source parabola, were all mounted on a single 1/2" thick aluminum plate to insure alignment. The entire optical system was enclosed in a light-tight housing which was kept dry with a pan of silica gel.

The sodium chloride prism was used in the spectrometer for the region from 0.4 to 10 microns, and the potassium bromide prism was used from 10 to 24 microns. A 1P21 photo multiplier was used as a sensor for the 0.4 to 0.8 micron region and a Reeder vacuum thermocouple was used for the region from 0.8 to 24 microns.

The tungsten lamp was used as the energy source for the 0.4 to 1.0 micron region and the globar for the remainder of the spectrum.

#### Hemisphere Calibration

It will be recalled from equations (10) and (13) that the spectral reflectance of the specimen,  $\rho_s$ , could be determined by the ratio of signals  $\frac{G_{ds}}{G_{dd}}$  and also from  $\frac{G_{ds}}{G_{dh}}$ , at any particular wavelength provided  $\rho_h$ ,  $\rho_d$  and  $n$  were known.

The hemisphere reflectance,  $\rho_h$ , was measured by an absolute method, and  $\rho_d$  was then measured relative to  $\rho_h$  utilizing the integrating hemisphere reflectometer.

The reflectance of the aluminized hemisphere was measured using a slight modification of Strong's method (5). With this method the specimen (No. 2) and two identical mirrors (No. 1 and 3) were located as shown in Figure 8.

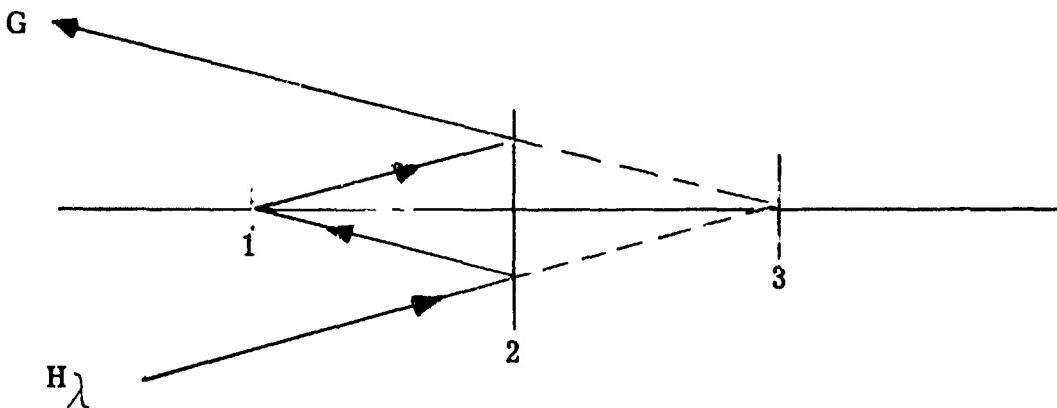


Figure 8 - HEMISPHERE REFLECTANCE DETERMINATION

The specimen (No. 2) was located midway between 1 and 3 as recommended by Strong, so that two reflections from No. 2 occurred. Care was taken to align all three surfaces in parallel planes. When the specimen surface was in place as shown, the energy reflected,  $G_2$ , was given by:

$$G_2 = \rho_1 \rho_h^2 H_\lambda \quad (16)$$

When the specimen was removed a reference measurement was obtained.

$$G_r = \rho_3 H_\lambda \quad (17)$$

Since mirrors No. 1 and 3 were identical,  $\rho_1 = \rho_3$  and the reflectance of the specimen of hemisphere material (aluminized glass) was given by:

$$\rho_h = \sqrt{G_2/G_r} \quad (18)$$

Measurements were made by this technique and by a second method with twice the sensitivity. In the second case, the specimen was located  $1/3$  of the distance from 1 to 3 as shown in Figure 9.

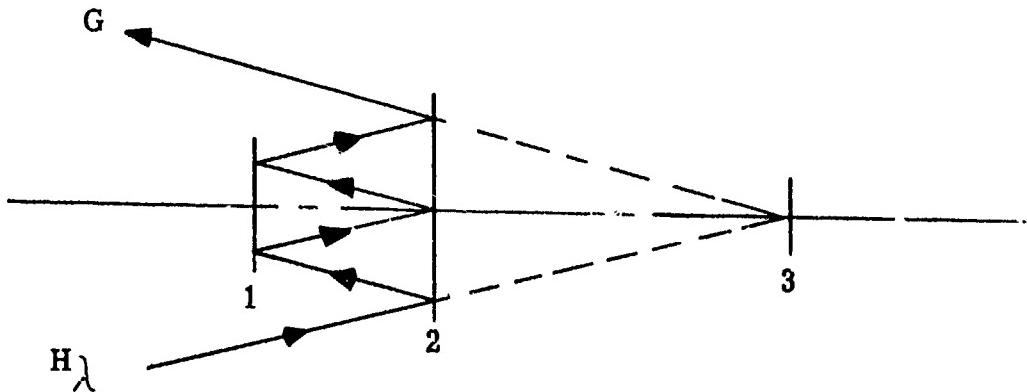


Figure 9 - HEMISPHERE REFLECTANCE DETERMINATION

Three reflections from No. 2 occurred with this arrangement and the energy reflected from the specimen was:

$$G_4 = \rho_1^2 \rho_h^3 H_\lambda \quad (19)$$

In this case mirrors 1 and 3 were the same material as the specimen.

When the specimen was removed the reference measurement was the same as before and the reflectance then was given by:

$$\rho_h = \sqrt[4]{G_4/G_r} \quad (20)$$

The apparatus used for these measurements is shown in Figure 10. This assembly was mounted from the rear end plate of the vacuum chamber after first removing the hemisphere.

Certain problems were encountered in this measurement which were not present in the other measurements. Since the surfaces were all specular and the optical lever arm was quite long, optical alignment of the surfaces was critical. The mirror mount was designed so that the assembly could be removed from the vacuum chamber and the mirrors were aligned with the aid of a transit and target located about 30 feet from the mirror. In addition the sensor collecting

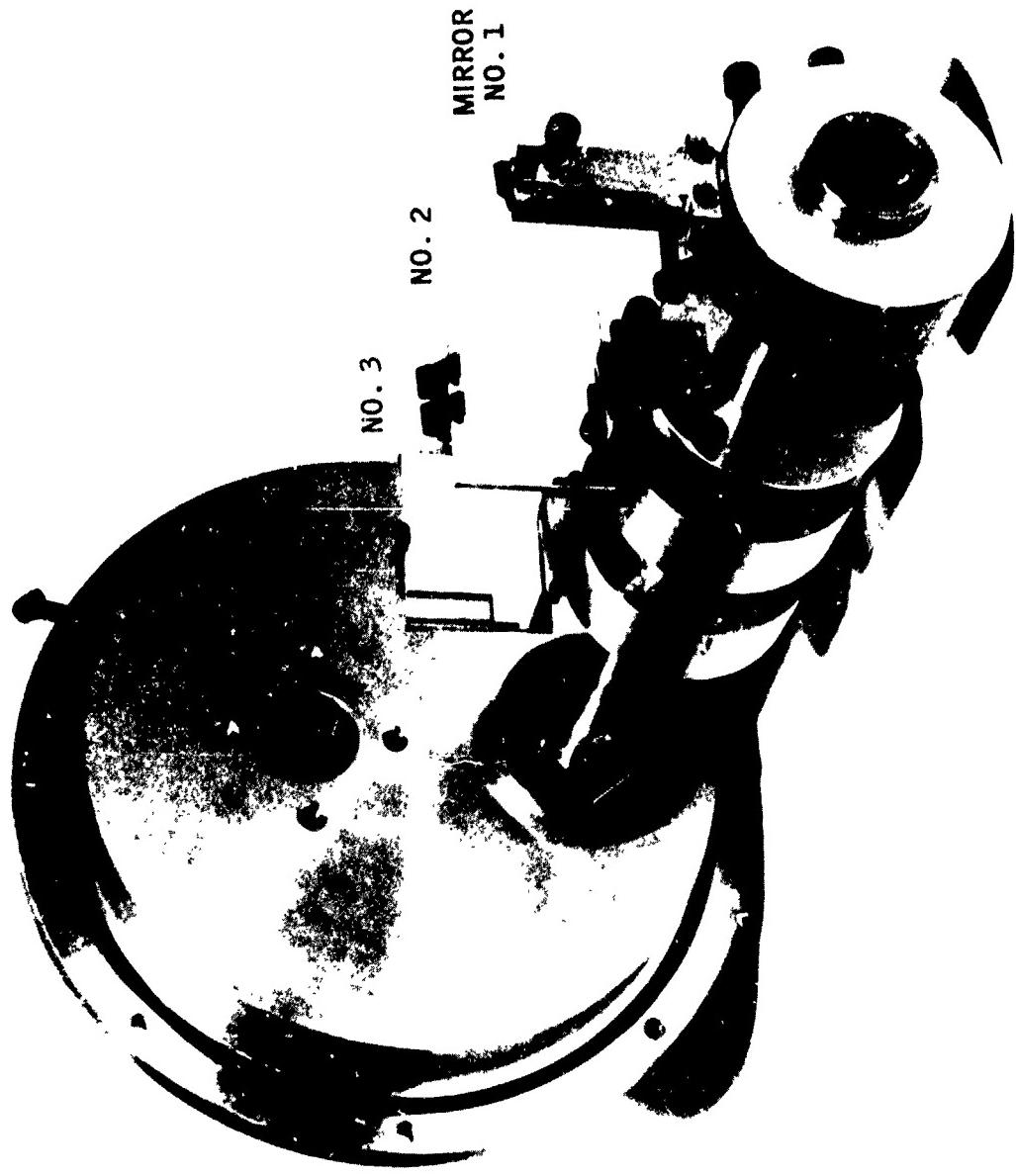


Figure 10 - HEMISPHERE REFLECTANCE MEASUREMENT APPARATUS

mirror was refocused to give a maximum signal each time the mirrors were changed. It will also be noted that the ratio of measured sensor signals appeared under the radical so that any error was further reduced in this manner. Optical alignment was much less critical when measuring the energy reflected from the specimen and diffusers in the reflectometer.

A second problem was that of atmospheric absorption due to water vapor and carbon dioxide. The external light path and the monochromator were flushed with dry nitrogen to reduce absorption effects, but it was difficult to eliminate atmospheric absorption entirely. This was particularly troublesome when measuring the hemisphere reflectance because it was necessary to traverse the spectrum, open the vacuum chamber, change the mirror arrangement, and then make the reference measurement. If the absorption of the external light path changed between the two measurements, an error was introduced. Flushing the light path with dry nitrogen and the use of silica gel appeared to control this problem.

The absorption problem was much less serious when making measurements of the specimens. These measurements were made point by point with the reference measurement being made almost simultaneously. Absorption effects were the same for both the specimen and reference measurements and, therefore, cancelled out in the ratio.

Hemisphere reflectance measurements were made with both mirror configurations described. Reasonably good agreement was obtained between the two techniques. However, since considerably more data were obtained with the 5 reflection case, and the inherent errors in the measurement were decreased by 75% in the reflectance calculation, these data were considered to be more reliable.

The measurements were made in a vacuum of approximately  $5 \times 10^{-6}$  mm Hg. The entire optical path outside of the vacuum chamber was flushed with dry nitrogen to minimize carbon dioxide and water vapor absorption. Three prisms were used to cover the range of 0.38 to 24 microns. Quartz was used from 0.38 to 2.0 microns; NaCl from 0.4 to 12 microns; KBr from 2.0 to 24 microns. The vacuum thermocouple sensor was used for the entire range of 0.38 to 24 microns.

The data are tabulated in Table 1 and are shown graphically in Figures 11 and 12. The theoretical reflectance was calculated from the Hagen-Rubens equation for reflectivity of metals in the infrared (6).

$$D = 1 - 36.05 \sqrt{\frac{r}{\lambda}}$$

where  $r$  = specific volume resistivity in ohm-cm

$\lambda$  = wavelength in microns

The electrical resistivity value of  $2.828 \times 10^{-6}$  ohm-cm for commercially pure aluminum at 20°C was used (7).

The data were plotted as shown in Figures 11 and 12 and the best curve was then drawn. It was found that the theoretical curve gave the best agreement with the measurements beyond 2.5 microns. The reflectance at the wavelengths selected for specimen measurements were then read from the curves. These values are presented in Table 3\* along with the reflectance of the diffusers.

#### Calculation of Loss Factor

In the derivation of the equation for the energy reflected to the spectrometer, it was assumed that the distribution of energy reflected by both the diffuser and specimen follow the cosine law. If the energy initially reflected by the diffuser closely approximates a cosine distribution, then all subsequent reflections will approximate the cosine distribution even more closely since

\* See Page 39

scattering is an irreversible process and the cosine distribution represents the limit for infinite scattering. The loss factor D was then defined as:

$$D = 1 - F_{di} - F_{de} = 1 - F_{si} - F_{se} \quad (21)$$

F = solid angle factor between the diffuser or specimen and the inlet or exit aperture.

Subscript d refers to diffuser

s refers to specimen

i refers to inlet or entrance aperture

e refers to exit aperture

The solid angle factor between two parallel disks is (4):

$$F = 1/2 (x - \sqrt{x^2 - 4(r_a/r_d)^2}) \quad (22)$$

$$x = 1 + (s/r_d)^2 + (r_a/r_d)^2 \quad (23)$$

$r_a$  = radius of aperture

$r_d$  = radius of diffuser

$s$  = distance between diffuser and aperture

The dimensions were:

$$r_a = 0.5 \text{ in.}$$

$$r_d = 0.5625 \text{ in.}$$

$$s = 4.678 \text{ in.}$$

Therefore

$$F = 0.0111 \quad (24)$$

and  $D = 0.978$  (25)

TABLE I  
REFLECTANCE OF ALUMINIZED HEMISPHERE

QUARTZ PRISM			QUARTZ PRISM			NaCl PRISM		
Wavelength, Microns	Reflectance		Wavelength Microns	Reflectance		Wavelength Microns	Reflectance	
	Run 1	Run 2		Run 1	Run 2		Run 1	Run 2
0.375	0.880		2.025	0.953	0.953	11.13	0.986	
0.385	0.871		2.085	0.954	0.955	11.32	0.985	
0.405	0.864		2.16	0.956	0.956	11.66	0.985	
0.425	0.869					12.00	0.985	
0.450	0.870	0.878				12.30	0.985	
0.475	0.872	0.875				12.60	0.981	
0.510	0.873	0.877	1.90	0.966				
0.550	0.874	0.876	2.45	0.967				
0.600	0.872	0.872	3.00	0.974				
0.665	0.867		3.55	0.972		1.6	0.963	0.956
0.750	0.850	0.849	4.06	0.974		2.05	0.969	0.962
0.790	0.842	0.839	4.52	0.969		2.85	0.968	
0.834	0.840	0.831	4.95	0.973		3.45	0.955	0.958
0.880	0.864	0.862	5.35	0.976		4.00	0.981	0.963
0.940	0.893	0.892	7.04	0.982		5.10	0.972	
1.00		0.914	7.25	0.983		6.10	0.968	
1.06		0.923	7.53	0.988		7.00	0.959	
1.13	0.928		7.78	0.976		7.75	0.975	
1.195	0.924	0.934	8.05	0.978		8.50	0.980	
1.26	0.936	0.939	8.26	0.979		9.87	0.980	
1.32	0.935	0.947	8.55	0.978		11.05	0.981	
1.39	0.955	0.947	8.75	0.977		12.2	0.983	
1.46	0.941	0.947	9.05	0.978		13.15	0.984	
1.53	0.945	0.952	9.28	0.981		14.03	0.986	
1.595	0.943	0.950	9.5	0.983		14.82	0.981	
1.66	0.946	0.951	9.73	0.980		15.56	0.986	
1.72	0.945	0.951	9.93	0.983		16.28	0.987	
1.785	0.948	0.951	10.15	0.983		17.67	0.986	
1.85	0.948	0.950	10.35	0.981		18.90	0.988	
1.91	0.946	0.953	10.55	0.981		20.04	0.996	
1.97	0.951	0.951	10.75	0.983		21.15	0.998	

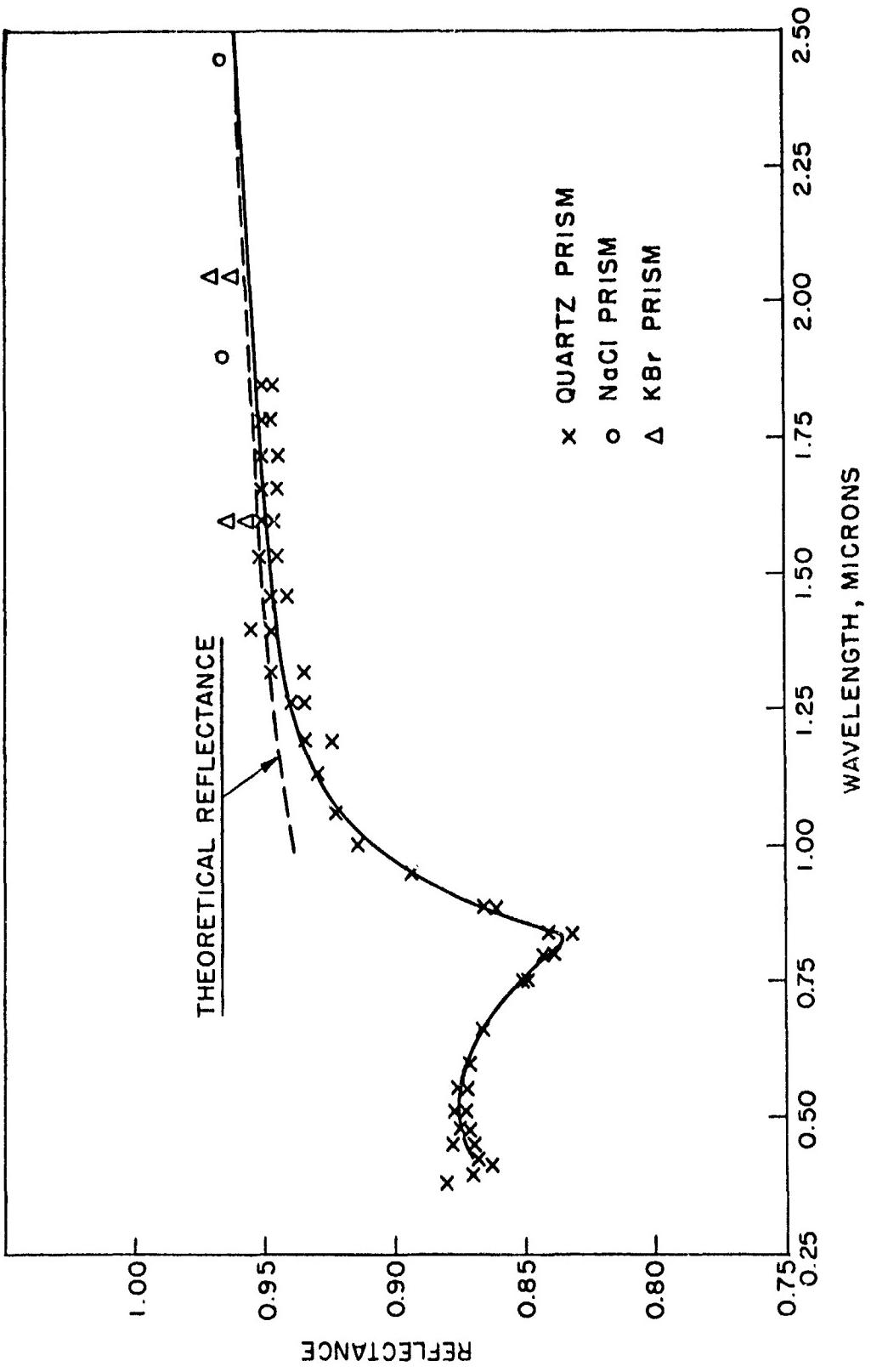


Figure 11 - NORMAL SPECTRAL REFLECTANCE OF EVAPORATED ALUMINUM  
ON GLASS - .38 TO 2.5 MICRONS

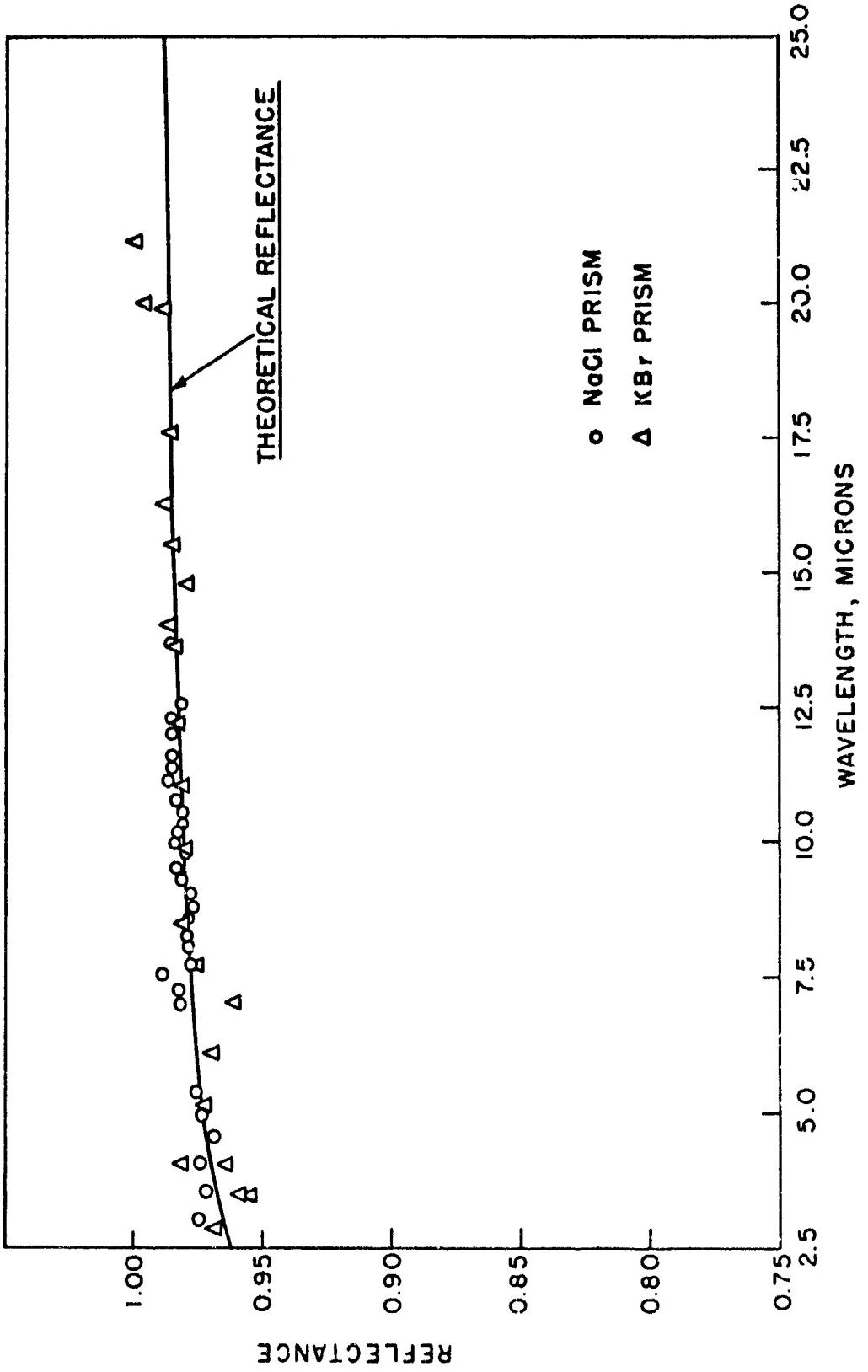


Figure 12 - NORMAL SPECTRAL REFLECTANCE OF EVAPORATED ALUMINUM ON GLASS-2.5 TO 25 MICRONS

## Diffuser Calibration

A diffuser of reasonably high reflectance whose surface properties were stable in the vacuum at room temperature was required. It was also necessary that the diffuser reflect according to the cosine law. Sandblasted, corrugated, surfaces of aluminum, stainless steel, and brass were considered. The reflectance of the stainless steel diffuser was about half the value of the aluminum and would have made a useful diffuser for specimens with a low reflectance. Unfortunately it was quite specular beyond about 10 microns and was rejected. Brass also had a lower reflectance than aluminum in the visible region but was very similar in the infrared and offered no advantages over aluminum.

In an effort to assure better scattering of the energy incident on the diffuser, concentric corrugations 0.008 in. deep by 0.040 in. from crest to crest were machined in the surface. The diffusers were then blasted with coarse silicon carbide, dipped in 35% nitric acid, washed and dried in alcohol. Figure 13 shows the diffuser configuration.

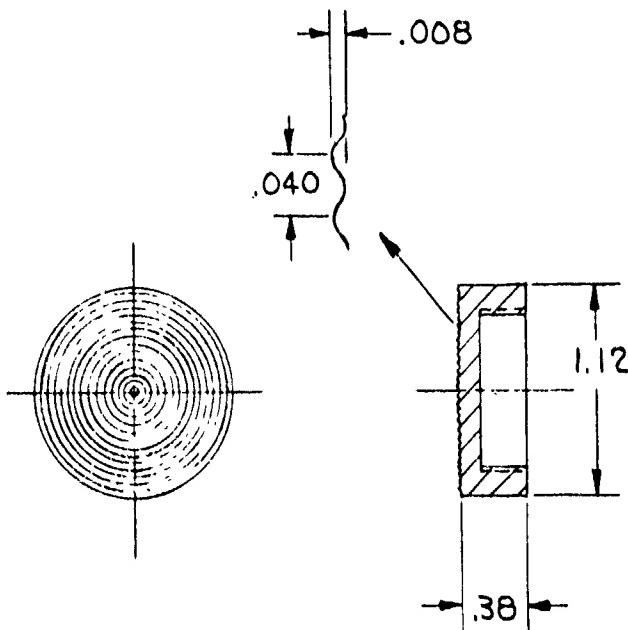


Figure 13 - DIFFUSER

The degree of diffuseness of the diffusers was measured for the reflection of visible light. A tungsten lamp with a projection lens and a photomultiplier were mounted on a goniometer. Light from the lamp was focused on the various diffusers at an angle of  $10^\circ$  from the normal to the diffuser surface, and the relative energy reflected at various different angles was measured. The results are shown in Figure 14.

The curves follow the cosine law quite well and show negligible specular reflection. The experimental set-up was changed for measurements in the negative quadrant, and this probably accounts for the difference in the relative intensities in the two quadrants. The important observation is the absence of large humps in the curve which would indicate specularity in the reflectance of the diffuser.

The reflectance of the diffusers was determined by substituting a second diffuser for the specimen and comparing this to a diffuser and aluminized glass standard. Equation (13) was then written in a slightly different form.

$$\frac{G_{dd}}{G_{dh}} = R_{dh} D \frac{[1 - R_{dh} V]}{[1 - R_{dh}^2 V]} \frac{[1 - (R_{dh}^2 V)^{n+1}]}{[1 - (R_{dh} V)^{n+1}]} \quad (26)$$

where

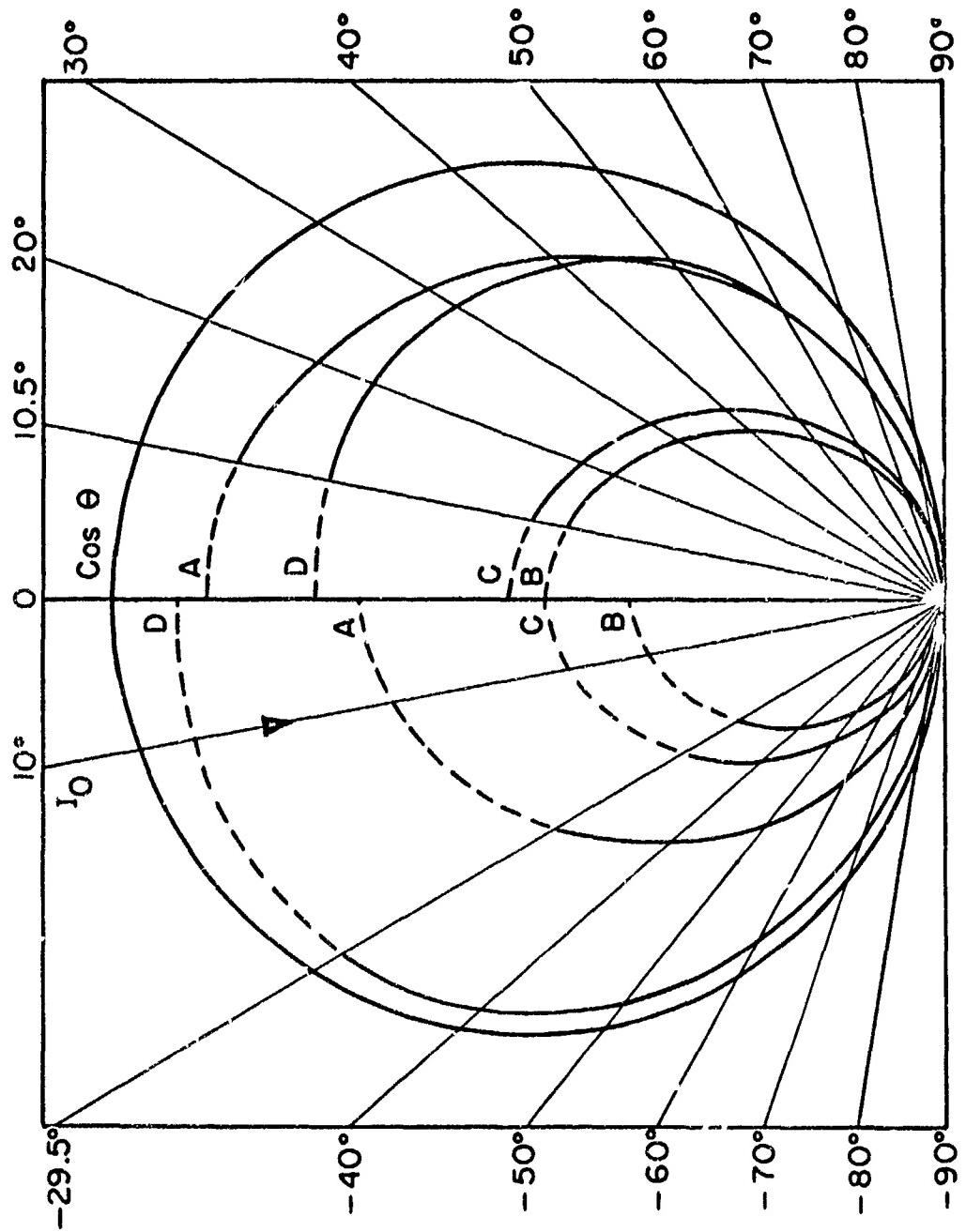
$$R_{dh} = \rho_d / \rho_h$$

$$D = 0.978$$

$$V = \rho_h^4 D^2$$

$n$  = number of internal re-reflections

Since the hemisphere surface reflectance,  $\rho_h$ , was known, the reflectance of the diffuser  $\rho_d$  could be calculated, if  $n$  could be determined.



A — ALUMINUM  
 B — STAINLESS STEEL  
 C — BRASS  
 D — MAGNESIUM OXIDE

Figure 14 - DIFFUSIVITY OF DIFFUSERS

In order to determine  $\rho_d$  and n, an intercomparison of measurements was made on three surfaces: aluminum diffusers, sulphur diffusers, and the evaporated aluminum on glass standard. This involved three unknowns: the reflectance of the aluminum diffusers,  $\rho_a$ , the reflectance of the sulphur diffusers,  $\rho_s$ , and the number of internal reflections, n.

Sulphur was chosen as a diffuser material on the basis of the work reported by Agnew and McQuistan (8) who found sulphur to be the best diffuse reflector in the infrared. Sulphur was found to be a better reflector than the aluminum diffusers out to about 13 microns. However, sulphur has a relatively high vapor pressure and could not be used in the evacuated reflectometer, but it was useful in calibrating the reflectometer.

The sulphur diffusers were formed by first cementing a glass disk to a steel diffuser blank to prevent the steel from reacting with the sulphur. A thin layer of flowers of sulphur was sprinkled on the glass and heated until the sulphur just melted. Sandblasting the glass first helped the sulphur to adhere. A thicker layer of flowers of sulphur was then sprinkled onto the molten sulphur and was tamped to form a solid layer about  $1/16'$  thick. The assembly was then cooled before the tamped layer melted. This procedure gave a sintered surface that could be lightly scraped but would remain intact when held in a vertical position.

When comparing two aluminum diffusers to an aluminum diffuser and the aluminized glass standard, equation (26) became:

$$\frac{G_{aa}}{G_{ah}} = R_{ah} D \frac{\left[1 - R_{ah} V\right] \left[1 - (R_{ah}^2 V)^n + 1\right]}{\left[1 - R_{ah}^2 V\right] \left[1 - (R_{ah} V)^n + 1\right]} \quad (27)$$

where:

$$R_{ah} = \rho_a / \rho_h$$

$$D = 0.978 \text{ (loss constant)}$$

$$V = \rho_h^4 D^2$$

n = number of internal reflections

Subscript "a" refers to aluminum diffuser

"h" refers to hemisphere surface  
(evaporated aluminum on glass)

When similar measurements were made with the sulphur diffuser, the ratio was given by:

$$\frac{G_{ss}}{G_{sh}} = R_{sh} D \frac{[1 - R_{sh} V] [1 - (R_{sh}^2 V)^{n+1}]}{[1 - R_{sh}^2 V] [1 - (R_{sh} V)^{n+1}]} \quad (28)$$

Subscript "s" refers to sulphur diffuser

The third measurement was made with one sulphur and one aluminum diffuser compared with the sulphur diffuser and aluminized glass standard.

$$\frac{G_{sa}}{G_{sh}} = R_{ah} D \frac{[1 - (R_{ah} Z_h)^{n+1}] [1 - Z_h]}{[1 - R_{ah} Z_h] [1 - Z_h^{n+1}]} \quad (29)$$

This is the same as equation (13) except for a change in subscripts. Here subscript s refers to the sulphur diffuser rather than the specimen.

The reflectance of the aluminum diffuser,  $\rho_a$ , was then calculated from equation (27) as a function of n using the values for  $\rho_h$  read from the curve in Figure 11 and 12 and presented in Table 2. Similarly the reflectance of sulphur,  $\rho_s$ , was calculated from equation 28 as a function of n. These values of  $\rho_s$  and n were then used in equation (29) to determine the reflectance of the aluminum diffuser with a sulphur diffuser.

Similar measurements were made over the 0.5 to 2.0 micron region using a magnesium carbonate diffuser in place of sulphur. A 3/16 in. thick by 1-1/8 in. diam. disk of magnesium carbonate was used for this measurement. In all cases the best agreement among the measurements occurred at  $n = \infty$ . This was reasonable since every effort had been made to maximize  $n$ .

Table 2 presents a comparison of the measurements on the aluminum diffuser for  $n = \infty$ . The agreement is quite good out to 2 microns. At 4 microns and beyond the sulphur data was 0.1 higher. This indicated that perhaps the aluminum diffusers were slightly specular at the longer wavelengths. This problem was compensated in the following way. If the specimen to be measured was very specular, then the reference measurement was made with the diffuser and an aluminized glass standard. The specularity of the specimen was then offset by the specularity of the reference. If the specimen was very diffuse, then the reference measurement was made with two diffusers. Since all of the specimens tend to be partially specular, it is believed that the uncertainty was less than indicated in Table 2.

Having determined that an infinite series should be used to describe the inter-reflections in the hemisphere, equations (10) and (13) respectively were reduced to:

$$\frac{G_{as}}{G_{aa}} = R_{sa} \frac{1 - Z_a}{1 - R_{sa} Z_a} \quad (30)$$

and

$$\frac{G_{as}}{G_{ah}} = R_{sh} D \frac{1 - Z_h}{1 - R_{sh} Z_h} \quad (31)$$

$$\text{where } R_{sa} = \rho_s / \rho_a$$

$$R_{sh} = \rho_s / r_h$$

$$Z_a = \rho_a^2 \rho_h^2 D^2$$

$$Z_h = \rho_a \rho_h^3 D^2$$

Subscript s refers to specimen

a refers to aluminum diffuser

h refers to aluminized glass standard

Equations (30) and (31) have the same general form and can both be written:

$$S = R \frac{1 - Z}{1 - RZ} \quad (32)$$

where  $S = \frac{G_{as}}{G_{aa}}$ ,

$$R = R_{sa},$$

$$Z = Z_a = \rho_a^2 \rho_h^2 D^2 \text{ for eq. (30)}$$

or  $S = \left(\frac{1}{D}\right) \frac{G_{as}}{G_{ah}}$ ,

$$R = R_{sh},$$

$$Z = Z_h = \rho_a \rho_h^3 D^2 \text{ for eq. (31).}$$

Equation (32) is plotted in Figure 15 and this set of curves was used for solving equations (30) and (31).

TABLE 2  
COMPARISON OF ALUMINUM DIFFUSER REFLECTANCE MEASUREMENTS

Diffuser Used in Measurement	Reflectance of Aluminum Diffuser		
	Aluminum Diffuser	$MgCO_3$ Diffuser	Sulphur Diffuser
Wavelength (Microns)			
0.5	.353	.357	-----
0.6	.359	.354	-----
0.8	.352	.331	-----
1.0	.403	.406	.398
1.6	.484	.514	.506
2.0	.518	.503	.536
4.0	.607		.701
8.0	.683		.781
13.0	.718		.817

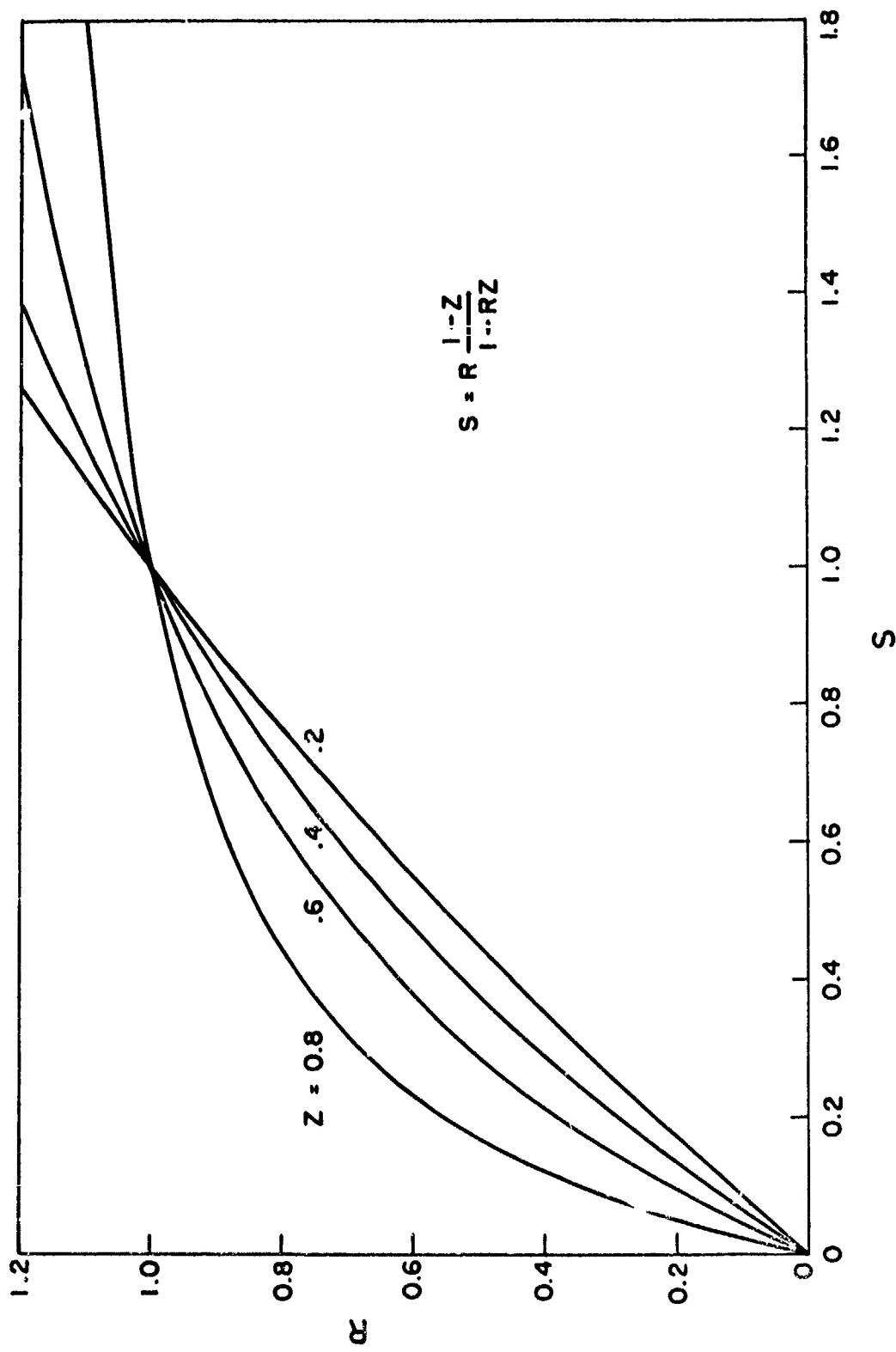


Figure 15 - DIFFUSER CALIBRATION EQUATION

The calibration data used for computing the reflectances of all of the specimens is presented in Table 3 and Figure 16. The values for the aluminized glass standard (and the hemisphere) were read from the curves in Figure 11 and 12. The reflectances of sulphur are as actually measured. The values for the aluminum diffuser were read from a smooth curve drawn through the data as shown in Figure 16.

These data were all computed by hand with the aid of the curves in Figure 15. However, it appeared that data reduction on the large number of specimen measurements was going to be a time-consuming task. Equations (30) and (31) were, therefore, programmed on an IBM 650 computer and all calculations for the specimens were made by the computer.

**TABLE 3**  
**REFLECTOMETER CALIBRATION DATA**

Wavelength (Microns)	Aluminized Glass ( $\rho_h$ )	Sulfur Diffuser ( $\rho_s$ )	Aluminum Diffuser ( $\rho_a$ )	V	$Z_a$	$Z_h$
.38	.873		.330		.079	
.4	.865		.332	.535	.080	.208
.45	.873		.345	.555	.087	.219
.5	.875	.873	.353	.560	.091	.226
.6	.872	.905	.358	.553	.094	.227
.7	.861	.917	.352	.525	.087	.215
.8	.841	.893	.351	.478	.084	.209
.9	.875	.944	.370	.560	.100	.237
1.0	.909	.968	.393	.653	.122	.282
1.2	.937	.961	.423	.737	.150	.333
1.4	.945	.970	.446	.762	.170	.360
1.6	.948	.961	.466	.772	.187	.380
1.8	.952	.957	.484	.785	.203	.399
2.0	.954	.955	.500	.792	218	415
2.5	.960	.958	.533	.812	.250	.451
3.0	.965	.835	.562	.829	.281	.483
3.5	.968	.864	.587	.839	.309	.509
4.0	.970	.899	.608	.846	.333	.531
5.0	.974	.925	.645	.860	.377	.570
6.0	.976	.836	.675	.867	.415	.600
7.0	.978	.820	.700	.875	.448	.626
8.0	.979	.870	.723	.878	.479	.649
9.0	.981	.753	.742	.885	.507	.670
10.0	.982	.895	.760	.889	.532	.688
11.0	.983	.889	.773	.893	.552	.702
12.0	.984	.827	.783	.896	.568	.713
13.0	.984	.877	.792	.896	.581	.722
14.0	.985	.844	.799	.900	.592	.730
15.0	.985	.830	.805	.900	.601	.735
16.0	.985	.832	.810	.900	.609	.740
17.0	.986	.898	.814	.904	.616	.746
18.0	.986	.891	.818	.904	.622	.750
19.0	.986	.838	.821	.904	.626	.752
20.0	.986		.823	.904	.630	.754
21.0	.986		.825	.904	.633	.756
22.0	.987		.827		.637	.760

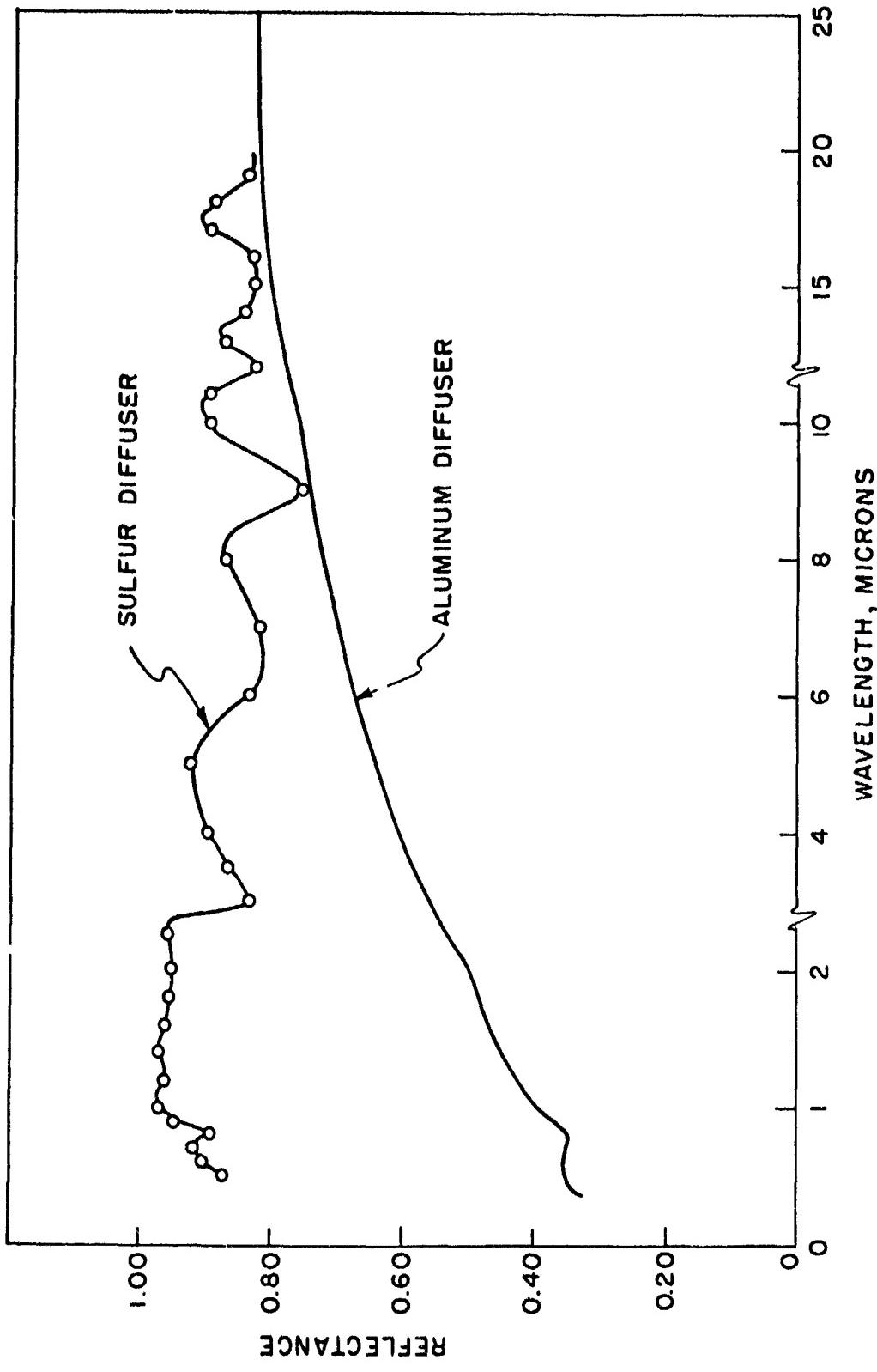


Figure 16 - DIFFUSER REFLECTANCE

## Errors

There were several sources of error in the measurements which were very difficult to predict quantitatively. One of the major uncertainties was due to the spherical aberration produced by the hemisphere. This caused the energy reflected from a point on the diffuser to be spread out non-uniformly along a line on the specimen. This line connected the center of the hemisphere to the point where the energy originated. When the energy falling on this line was reflected back to the diffuser, it produced a longer line. The next reflection back to the specimen produced a still longer line which exceeded the size of the specimen and a small amount of energy was lost from the second and each succeeding reflection from the specimen. Consequently the assumption of a simple geometric series with an infinite number of terms was technically incorrect. The use of a series with a finite number of terms was thoroughly studies by intercomparing measurements on several different diffusers as discussed in the section on diffuser calibration. The studies indicated that limiting the length of the initial image projected on the diffuser to about half the diffuser diameter, orienting the length of the image on a line through the centers of the diffuser, hemisphere and specimen and always using maximum slit opening on the spectrometer reduced the spherical aberration error to a negligible amount.

Scattered radiation produced by the windows in the vacuum chamber also was an unknown source of error. It was possible for energy scattered by surface reflections from the entrance window to fall directly on the specimen without first being reflected by the diffuser. This was particularly troublesome in the case of a specular specimen or the aluminized glass standard. In these cases an appreciable amount of energy could be reflected specularly to the spectrometer without first being diffused by the diffuser. To prevent this, it was necessary to tilt the specimen or standard slightly so that the direct reflection of the entrance window did not enter the spectrometer. The reflectometer was originally designed for vertical, parallel images on the diffuser and

specimen and no such problem then existed. However, when the images were rotated 90° to align them and thus reduce the effects of spherical aberrations, the problem of scattering light was aggravated.

A third area of uncertainty was in the diffusers. In the derivation of the equations it was assumed that the reflectance was constant in all directions; i. e., the diffusers reflected according to the cosine law and gave a Lambertian distribution of reflected energy. The investigation of this problem was covered in the discussion of the diffuser calibration. It was important also that the diffusers did not change with time due to condensation of vapors in the vacuum chamber. The reflectance of the diffusers compared to the aluminized glass standard was, therefore, checked periodically. After some high-temperature runs it was necessary to scrub the diffusers in a detergent, dip in 35% nitric acid, rinse, and dry in alcohol to restore the reflectance to its original value.

Proper focusing of the diffuser and specimen was critical. The hemisphere had a very shallow depth of field and it was necessary to locate the surface of the specimen and diffuser within a few thousandths of an inch of a plane passing through the center of the hemisphere. It was found that the turret could be adjusted with the focusing nut to give a maximum signal and the position of the nut could be marked. Locating the proper position for both the specimen measurement and the reference measurement in this way gave 99% or greater reproducibility in most cases. Good reflectors were more critical in this respect than poor reflectors.

The amplifier and sensor noise also were sufficient to limit measurement accuracy at low energy levels where a high amplifier gain was required. This was especially troublesome when the specimen to reference reflectance ratio was small. When the specimen reflectance was only 10% of the reference reflectance (diffuser or aluminized glass standard) the instrument recording error could be 10% of this ratio. Zero drift in the instrument was also annoying at high gains, necessitating repetitive readings to obtain a reproducible zero.

Measurements at 0.8 microns were awkward. The spectral response of the photo multiplier dropped off sharply in this region and gave readings of questionable accuracy. On the other hand this represented the lower limit for the thermocouple and required maximum amplifier gain along with maximum noise. Disagreement between the two sensors ranged from zero to more than 25% in a few cases. Although both sensors were used at 0.8 microns, the thermocouple data was considered more reliable and is reported in the tables. Both measurements are shown in the curves.

To get an indication of the over-all accuracy, measurements were compared with those of other investigators. Figure 17 shows a comparison of our measurements on magnesium carbonate, with those of Benford (1) and Betz (9). These data agree within 1.5% except for the measurement at 2.0 microns. Disagreement here may have been due to the wide spectrometer slits we were forced to use to compensate for spherical aberrations and the presence of a narrow absorption band at 2.0 microns. Our spectral resolution probably was not as good as in the case of Betz's measurements.

In addition to this, arrangements were made to have the normal spectral reflectance of anodized aluminum specimens measured at the Solar Radiation Laboratory of the University of California, Los Angeles Branch, and at the Thermophysics Laboratory, Dept. 53-11 of the Lockheed Missiles and Space Division, Sunnyvale, California. Both of these laboratories use a Cary integrating sphere (magnesium oxide coated) for measurements in the range of 0.25 to about 2.0 microns and a Gier-Dunkle black-body reflectometer for the 2.0 to 22 micron region. Unfortunately, the UCLA black-body reflectometer was tied up on other work, and they were able to make measurements only in the short-wavelength region.

The UCLA data on chromic acid anodized 1075 aluminum (Specimen No. 55) and sulfuric acid anodized 7075 aluminum (Specimen No. 63) are presented in Table 4. The UCLA data, the Lockheed data, and our data are compared in Figures

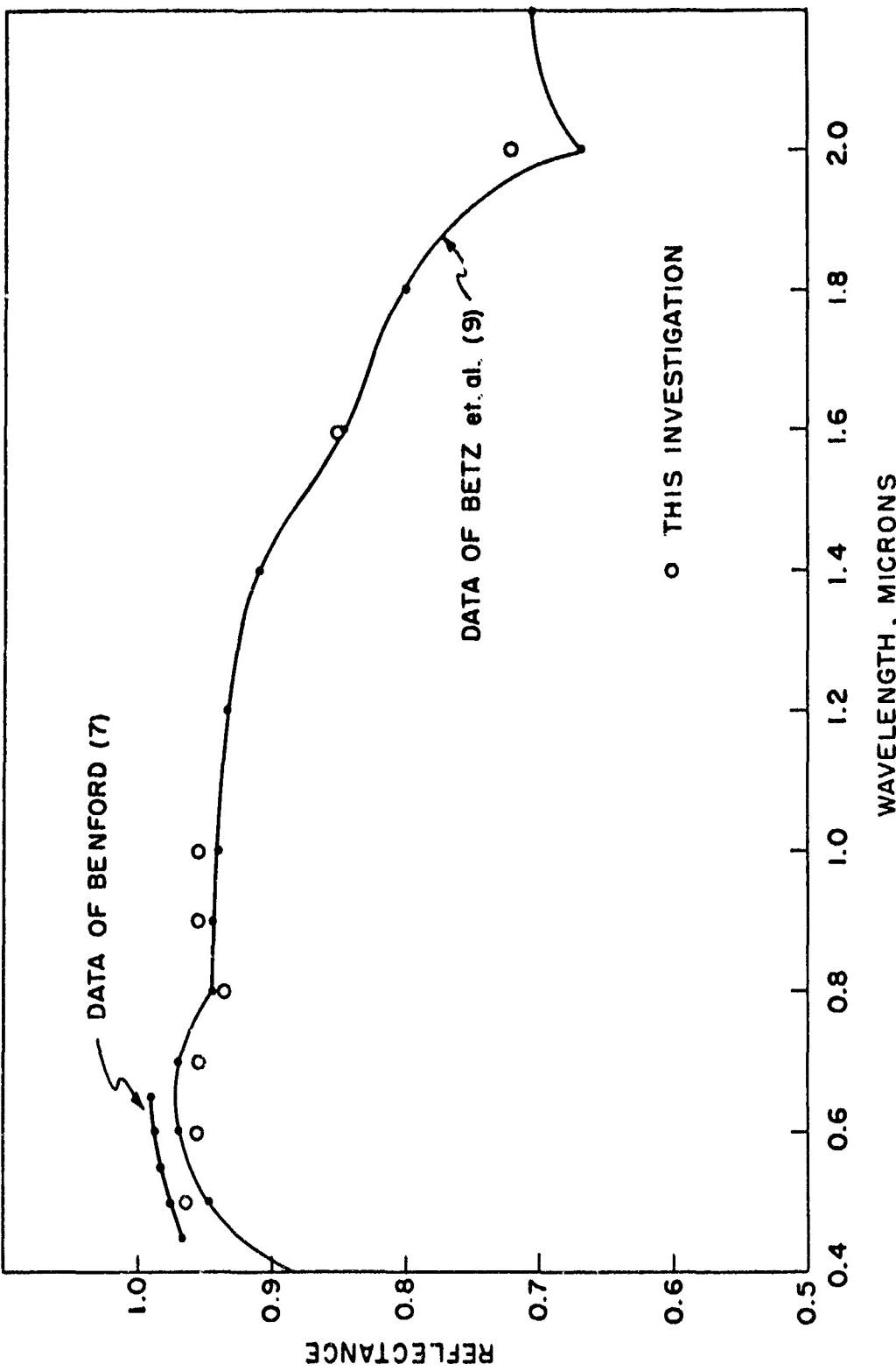


Figure 17 - REFLECTANCE OF MAGNESIUM CARBONATE

TABLE 4  
REFLECTANCE OF ANODIZED ALUMINUM MEASURED AT UCLA

Wavelength Microns	Specimen		Wavelength Microns	Specimen	
	55	63		55	63
.34	.51	.35	.72	.705	.605
.37	.54	.40	.75	.745	.60
.39	.58	.42	.77	.795	.60
.41	.61	.435	.79	.655	.60
.43	.62	.465	.81	.715	.60
.44	.645	.495	.84	.725	.61
.46	.65	.50	.87	.695	.63
.47	.665	.505	.90	.745	.645
.48	.67	.52	.93	.81	.665
.49	.68	.52	.96	.815	.695
.50	.70	.54	1.00	.825	.71
.52	.675	.54	1.04	.875	.715
.53	.71	.55	1.08	.88	.755
.55	.71	.56	1.12	.88	.755
.56	.69	.57	1.17	.90	.78
.57	.735	.58	1.23	.91	.785
.59	.725	.58	1.29	.905	.795
.60	.705	.59	1.36	.915	.80
.62	.735	.59	1.54	.935	.825
.64	.735	.59	1.54	.935	.825
.65	.705	.595	1.67	.925	.82
.67	.725	.595	1.82	.94	.825
.69	.75	.595	2.05	.94	.825
.70	.71	.60			

18 through 22. The specimens measured at UCLA and by us were each punched from the same anodized coupon (No. 55 and 63 respectively). The specimens measured at Lockheed were punched from coupons No. 56 and 64 which we anodized at the same time and should have been identical to 55 and 63 respectively. Lockheed also measured a Martin hard coat anodized 7075 aluminum specimen, No. 88, and we measured a similar specimen No. 87.

The three sets of data are in good agreement in the region below 2 microns. The measurements made at UCLA and LMSD were made with narrower spectrometer slits and at more frequent intervals than ours. The resolution thus achieved revealed absorption bands in the reflectance of specimen 55 and 56 in the region below 1.2 microns. Since these specimens were anodized in a chromic acid bath, the oxide coating was relatively thin and the specimens were quite specular. The absorption bands were undoubtedly due to interference effects. The slight phase displacement between 55 and 56 indicates that the oxide coating on 56 may have been slightly thicker. Although our measurements did not resolve these interference bands they did give an accurate average reflectance in this region.

In the region from 2 to 10 microns our data agreed well with the LMSD data for specimen 56. This specimen was measured in air in both cases. In the case of specimen 64, however, our data for both 63 and 64 was appreciably higher in the 4 to 7 micron region. Since our data for both 63 and 64 were taken in vacuum and the LMSD data were taken in air, it is possible that the increased reflectance was due to a loss of water. The high temperature runs on 63 showed that increasing the specimen temperature drove moisture out of the oxide film and raised the reflectance.

Beyond 10 microns our data was consistently higher. This may also have been partially due to loss of water although specimen 56 which was measured both in air and vacuum did not bear this out. The coating on 56 was more dense and contained less water than either 64 or 88, however.

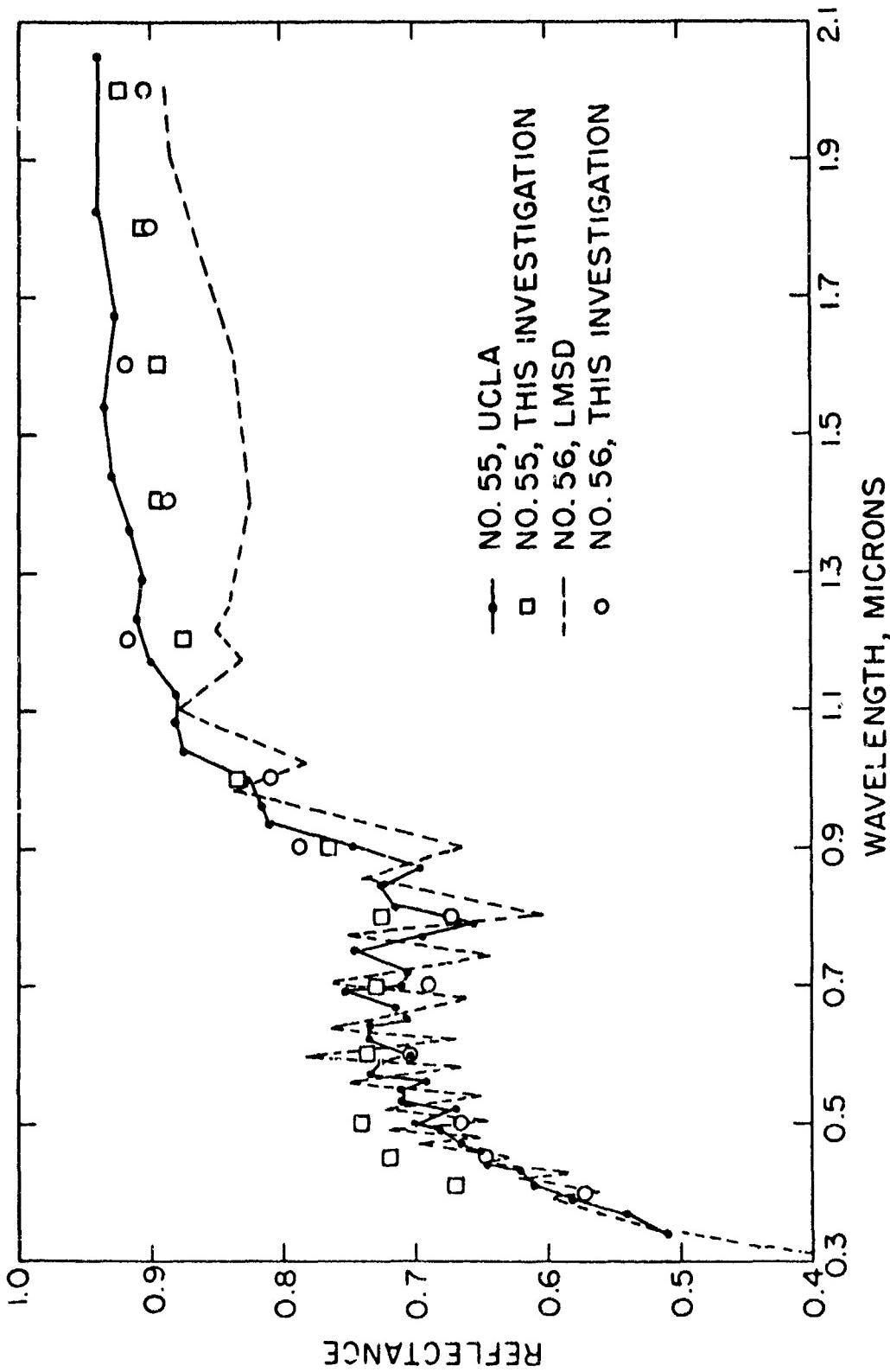


Figure 18 - CHROMIC ACID ANODIZED 1075 ALUMINUM

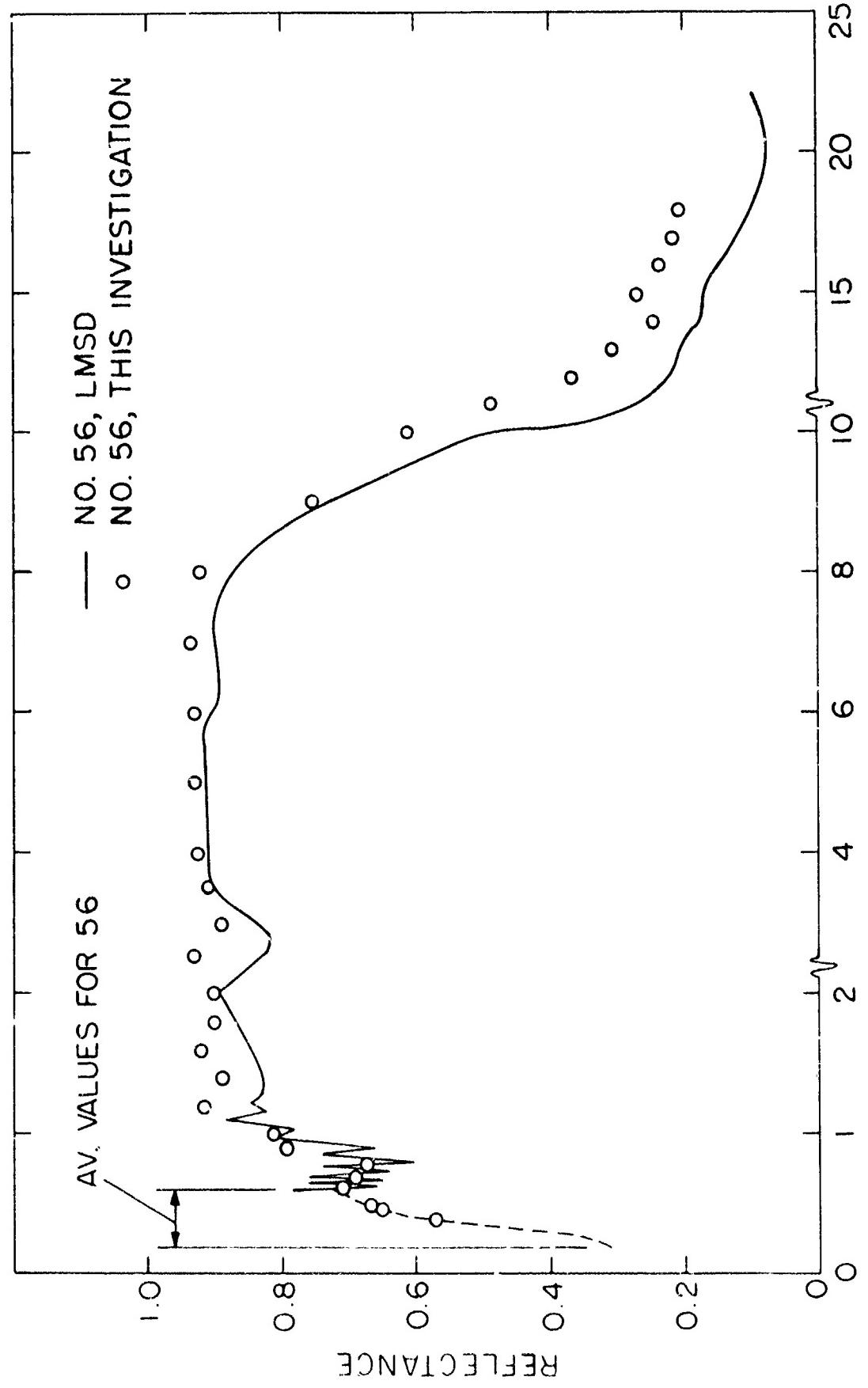


Figure 19 - CHROMIC ACID ANODIZED 1075 ALUMINUM

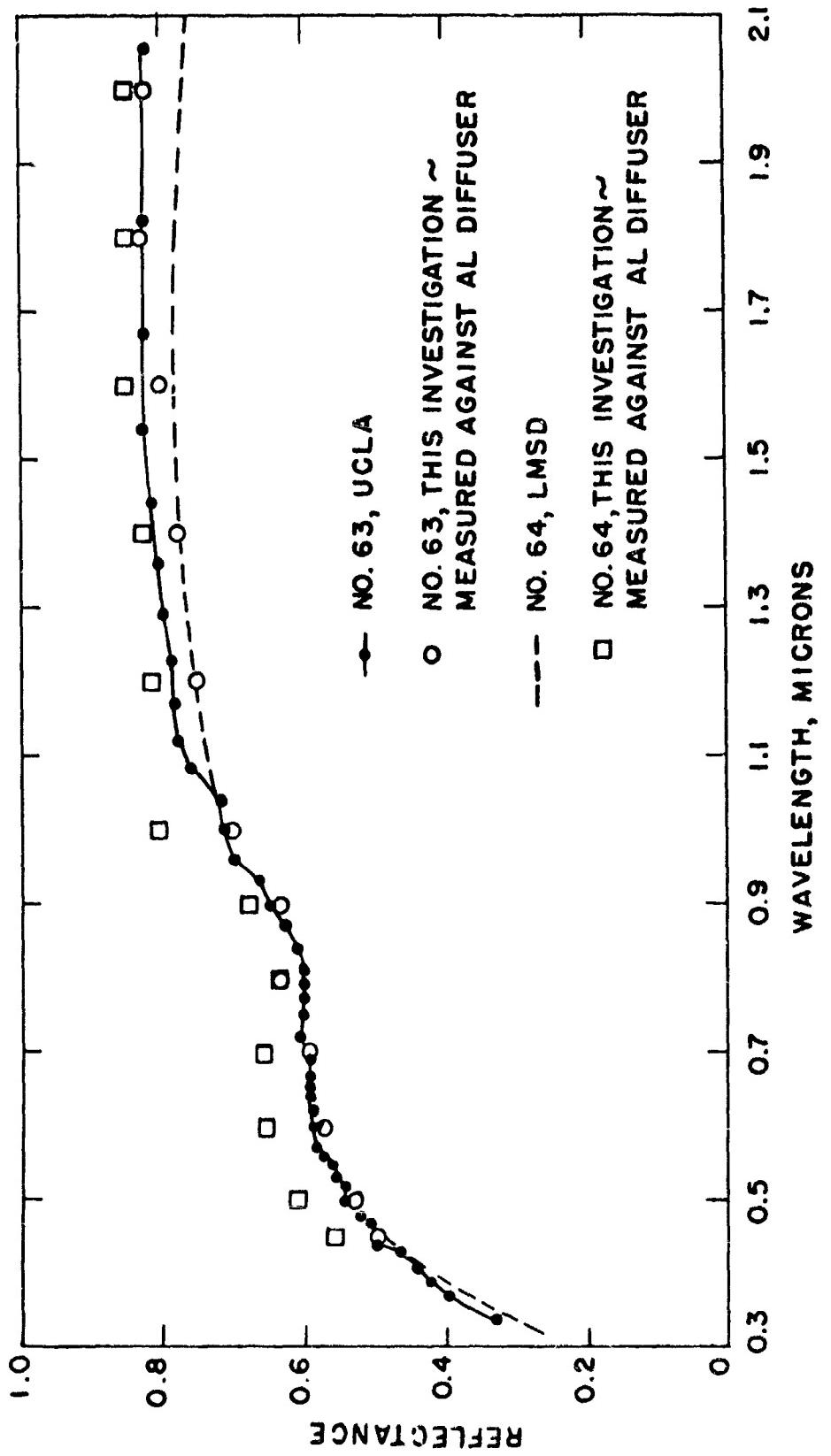


Figure 20 - SULFURIC ACID ANODIZED 7075 ALUMINUM

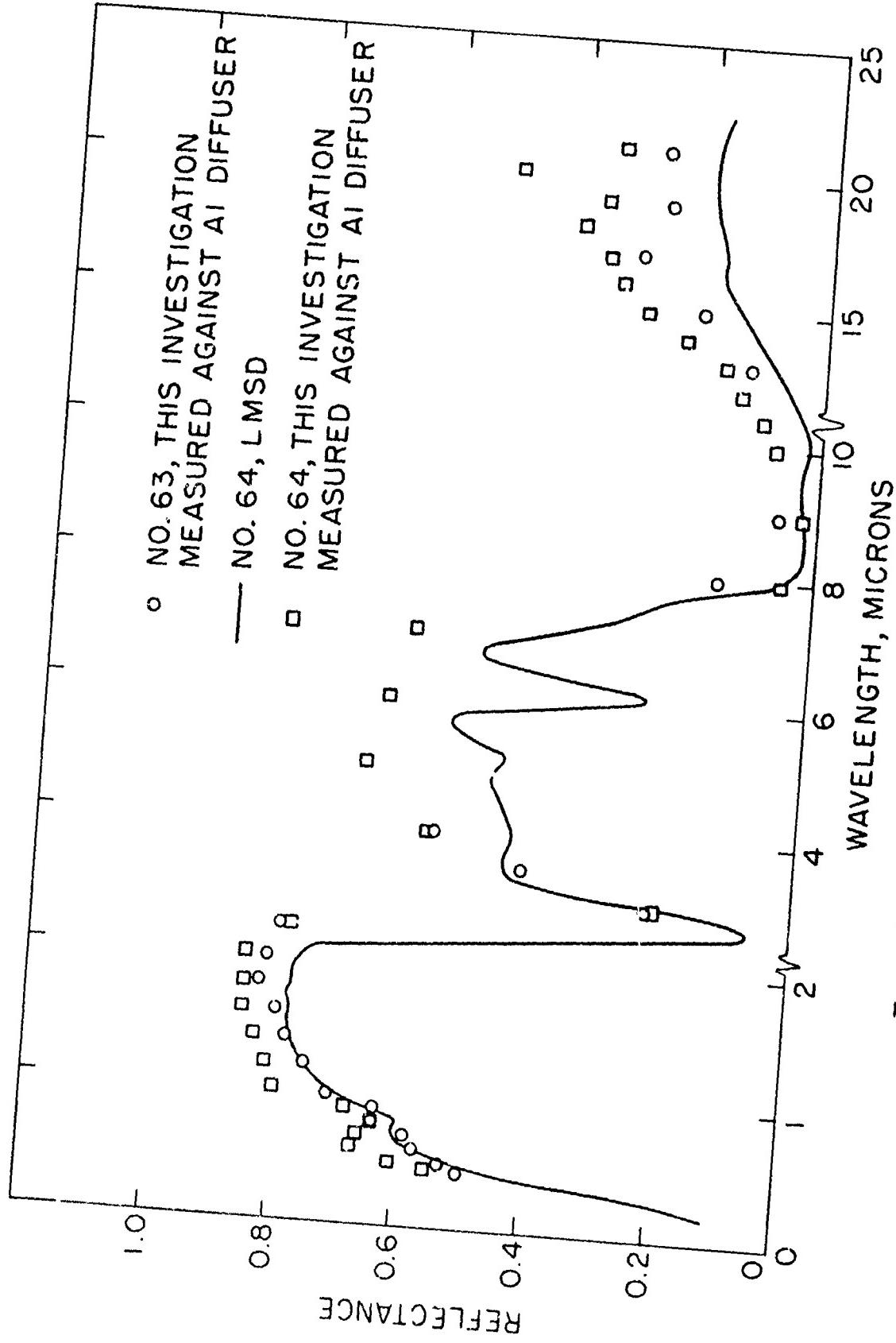


Figure 21 - SULFURIC ACID ANODIZED 7075 ALUMINUM

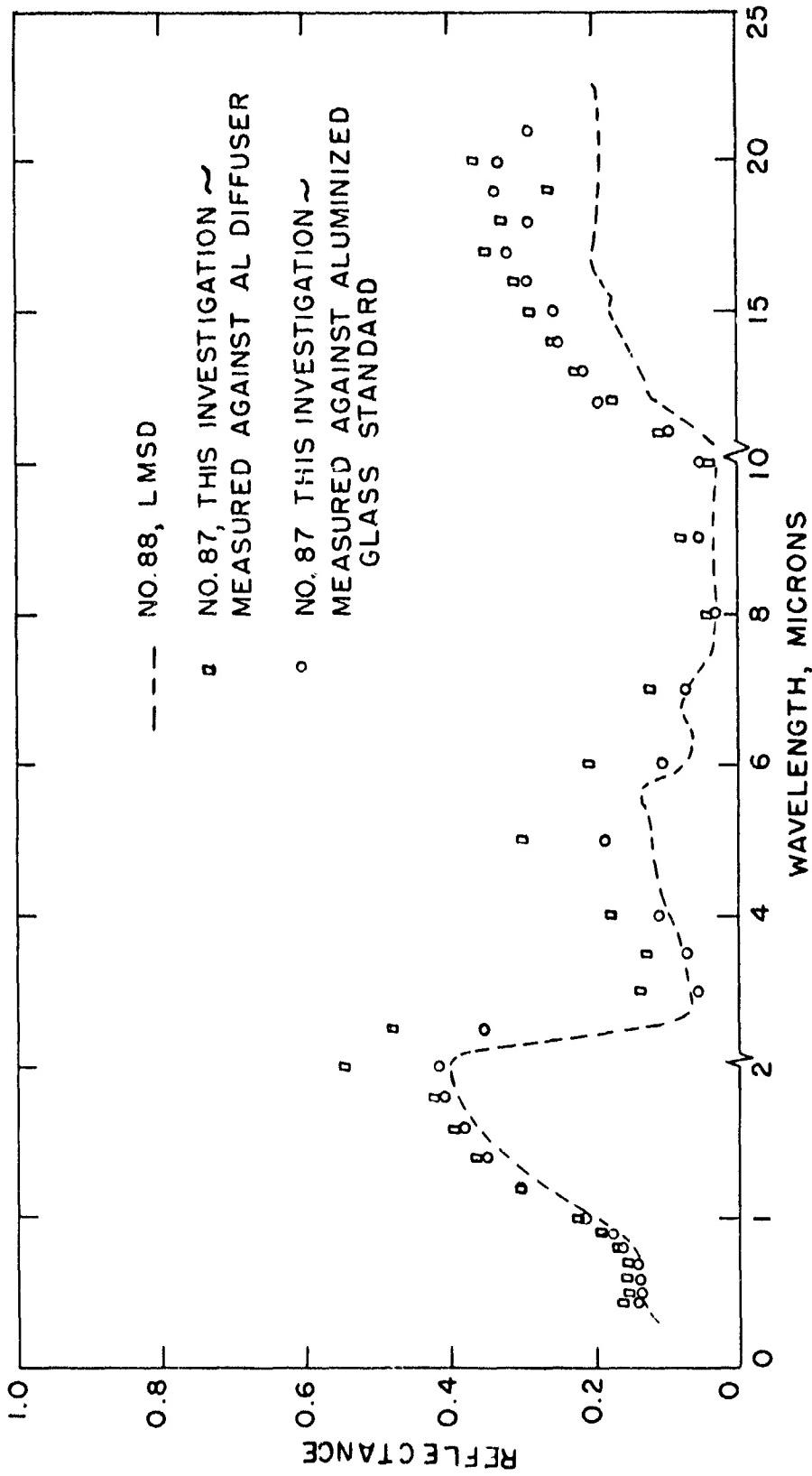


Figure 22 - MARTIN HARD COAT ANODIZED 7075 ALUMINUM

The error problem was probably not quite as severe as it appears. In heat transfer calculations, the engineer is interested in the solar absorptance and infrared emittance values. These are both obtained by subtracting the spectral reflectance values from unity and making the appropriate integrations. An error of even 50% in the measurement of a reflectance value of 10% gives an error of only 5% in the emittance. On the other hand, an error of 1% in the measurement of a reflectance of 90% gives an error in the emittance of 10%. While this error is undesirably high, it should be remembered that in selecting surfaces for space craft, the engineer looks for a high infrared emittance and a low solar absorptance. Solar energy is concentrated in the region below 1.6 microns where it is extremely difficult if not impossible to make direct emittance or absorptance measurements. In general the engineer must resort to reflectance measurements in this region and accept the inherent error problem. The only alternative is to use calorimetric techniques for the measurement of the ratio of solar absorptance to infrared emittance. If the total emittance is then measured at the same specimen temperature, the solar absorptance can be computed from the ratio. There are several sources of error in these techniques also.

## SPECIMEN PREPARATION

### Pretreatment - Polishing

Prior to anodizing, the coupons, from which the specimens were punched, were mechanically polished to establish a uniform and reproducible surface. A Lapmaster 12 was used for this work.

An iron lap was initially used with water-mixed lapping compounds; however, the specimens, especially titanium, had a marked tendency to seize and scratch. A canvas lap with oil-mixed lapping compounds was, therefore, substituted. The lapping compounds were products of the United States Products Company of Pittsburg, Pennsylvania.

The lapping sequence for the various alloys was as follows:

Aluminum 1075 and 7075; magnesium AZ31B

1. No. a-600 alundum (hard) 60 min.
2. No. 38-900 (hard - sharp) 60 min.
3. No. 38-1200 (hard - sharp) 60 min.
4. Green rough (soft) 60 min.

Magnesium HK 31A

1. No. 100 crystalon 30 min.
2. No. 600 alundum 200 min.
3. No. 38-900 A 90 min.
4. No. 38-1200 120 min.

Titanium A-110-AT

1. No. 100 crystolon 150 min.
2. No. 600 alundum 180 min.
3. No. 38-900A 150 min.

Titanium B-120-VAC

1. No. 100 crystolon 300 min.
2. No. 600 alundum 180 min.
3. No. 38-900A 60 min.

Beryllium; QMV and 1% cu

1. No. 100 crystalon 210 min.
2. No. 600 alundum 40 min.
3. No. 38-960A 270 min.
4. No. 38-1200 60 min.

Surface roughness measurements were made with a profilometer on several specimens of both aluminum alloys and the AZ31B magnesium, and the roughness was found to be less than two microinches rms. The other alloys were polished to give the same appearance.

Since the electropolishing and anodizing processes altered the surface of the metals, fine scratches remaining after mechanically polishing were not visible in the finished specimens. Surface defects in unpolished or mill finished specimens were visible after anodizing, however.

### Anodizing

The following anodizing processes were studies:

- Aluminum - sulphuric acid, chromic acid, boric acid and hard coat
- Magnesium - Dow No. 17 and HAE coatings
- Titanium - sulfuric acid, sodium hydroxide, and a proprietary process
- Beryllium - chromic acid, nitric acid, sodium hydroxide, and sulfuric acid

Both machine-polished and mill-finished (machined surfaces for beryllium) coupons were anodized by each process. The effect of electropolishing prior to anodizing was investigated. The anodizing procedures are presented below. Power was dc unless noted.

1. Aluminum 1075 and 7075
  - a. Pretreatment
    - Degrease in trichloroethylene
    - Clean in Oakite 61A (inhibited alkaline cleaner)  
45 gm/L for 5 minutes at 93C
    - Cold water rinse

b. Electropolishing

Patented process, U. S. Patent No. 2338321

c. Anodizing

Sulfuric acid process

Solution:  $H_2SO_4$  - 15% by weight

Temperature: 30°C

Applied potential, 15 vdc with lead cathode

Time: 30 minutes (standard)

Seal: 15 minutes in boiling water where noted.

Chromic acid process

Solution:  $CrO_3$  - 7.5% by weight

Temperature: 35°C

Applied potential: start at low voltage and increase to 40 vdc and hold

Time: 30 minutes (standard)

Steel cathode

Hard Coat process

Solution: 15%  $H_2SO_4$  saturated with  $CO_2$

Temperature: -2°C

Current density: 25 amps/ $ft^2$

Lead cathode

Time: 80 minutes (standard)

Seal: 15 minutes in boiling water where noted.

Boric acid process

Solution: 9%  $H_3BO_3$ , 0.2%  $Na_2B_4O_7$

Temperature: 94°C

Applied potential: 230-250 vdc

Time: 15 minutes (standard)

Steel cathode

2. Magnesium AZ31B and HK31A

a. Pretreatment

Degrease in trichloroethylene

Dip in 180 gm/L  $CrO_3$  at 93C

Clean in Oakite 61A 45 gm/L for 5 min. at 93C

Cold water rinse

b. Electropolishing

Solution: HC1-100ml; Ethylene glycol - 900 ml

Temperature: below 10 C

Time: 10 minutes

Current density: 20 amps/ $ft^2$

c. Anodizing

HAE Process

Solution: The following chemicals were dissolved in the order given, then the bath was dummied with 1 ft<sup>2</sup> magnesium per gallon of solution.

KOH - 163 gm/L

Al (Metal) - 9.25 gm/L

$\text{Na}_3\text{PO}_4$  - 3.33 gm/L

$\text{KMnO}_4$  - 18.3 gm/L

Temperature: 20 C

Current density: 20 amps/ $\text{ft}^2$  ac

For thin tan coating terminate at 64 vac

For thick brown coating terminate at 85 vac

Dow 17 process

Solution:  $\text{NH}_4\text{HF}_2$  - 237 gm/L

$\text{Na}_2\text{CR}$  207. 2H2O-98 gm/L

$\text{H}_3\text{PO}(85\%)$  - 85 ml/L

Heat half the water to 71C add  $\text{NH}_4\text{HF}_2$  slowly.

Add other ingredients and heat to 82 C.

Temperature: 71 C

Current density: 20 amps/ $\text{ft}^2$  a-c

For thin green coating terminate at 64 vac

For thick green coating terminate at 90 vac

d. Post treatment

The coupons were rinsed in cold water.

3. Titanium A110AT and B120 VAC

a. Pretreatment

Degrease in trichloroethylene

Alkaline clean in Oakite 61A, 45 gm/L, 5 min. at 93 C

Pickle in solution of 40% by volume HF (5%) and 30% conc.  $\text{HNO}_3$   
at room temperature those coupons which were not electropolished.

Cold water rinse

b. Electropolishing

Solution: HF (50%) - 160 ml/L; CrO<sub>3</sub> - 500 gm/L

Temperature: 18 C

Current density: 325 amps/ft<sup>2</sup>

Time: 10 minutes

Lead cathode

c. Anodizing

Sulfuric acid process

Solution: H<sub>2</sub>SO<sub>4</sub> - 20% by weight

Temperature: 20 C

Applied voltage: 18 vdc

Time: 20 minutes

Lead cathode

Sodium hydroxide process

Solution: NaOH - 5% by weight

Temperature: 96C

Current density: 50 amps/ft<sup>2</sup>

Time: 20 minutes

Mild steel cathode

Remarks: The B120 vac alloy did not anodize satisfactorily by this method.

Proprietary process

The anodizing procedure for titanium by a proprietary process was supplied by the project engineer.

d. Post Treatment

Coupons were rinsed in cold water, then sealed in boiling water 15 minutes.

4. Beryllium

a. Pretreatment

Degrease in trichloroethylene

Clean in Oakite 61A 45 gm/C 5 minutes at 93 C

Cold water rinse

b. Electropolishing

Identical to that used for aluminum (U. S. Pat. 2,338,321)

c. Anodizing

Chromic acid process

Solution:  $\text{CrO}_3$  - 1% by weight

Temperature: 35C

Current density: 10 amps/ $\text{ft}^2$

Time: 30 minutes

Graphite cathode

Nitric acid process

Solution:  $\text{HNO}_3$  - 50% by volume

Temperature: 25 C

Current density: 30 amps/ $\text{ft}^2$

Time: 30 minutes

Graphite cathode

Sodium hydroxide process

Solution:  $\text{NaOH}$  - 7. 5% by weight

Temperature: 25 C

Current density: 8 amps/ $\text{ft}^2$

Time: 30 minutes

Mild steel cathode

Sulfuric acid process

Solution:  $H_2SO_4$  - 1%

Temperature: 25 C

Current density: 10 amps/ $ft^3$

Remarks: Anodized coating was rinsed off by the solution.

These specimens not sealed.

d. Post treatments

Coupons were rinsed in cold water, then sealed in boiling water 15 minutes.

The anodizing solution variables on all of the coupons used for reflectance measurement are presented in Table 6. \* Similar mill finish coupons were prepared at the same time for coating thickness measurements.

In the anodizing procedure for magnesium, the method of obtaining a light and a dark anodized coating is described. For the other alloys, coating thickness was varied by controlling the anodizing time. Standard/3, therefore, refers to an anodizing time equal to one-third of the time specified in the anodizing procedure. The Martin hard coat anodizing process, the proprietary process, and the patented electropolishing technique used on aluminum and beryllium were supplied by the project engineer.

The coupons were consecutively numbered as they were mechanically polished; therefore, the specimen number has no significance other than identification and location in the filing system.

Heat Treatment

It was desirable to investigate the effect of heat treatment on the reflectance of the specimens. Certain specimens were therefore heated both in air and under vacuum conditions.

\* See Page 76 , Part II

The specimens to be heat treated in air were placed in a furnace heated to the proper temperature, held there for thirty minutes, and then removed. The aluminum and magnesium were heat treated at 800 F; the titanium and beryllium at 1500 F. The reflectance was then measured.

The temperature to which the specimens were heated in the vacuum chamber was limited by the vapor pressure of the material and other reasons as noted in discussion of results. After the reflectance measurements were made at elevated temperatures, in vacuum, a repeat measurement was made at room temperature to determine the extent of irreversible changes in the coatings.

#### ANODIZED COATING THICKNESS

Thickness of the anodized coating was determined for the aluminum, beryllium, and magnesium coupons by dissolving the coupon and determining the weight change of the coupon. Using the handbook density for the oxide, the film thickness could then be calculated. This method, specified by the Project Engineer, is recommended by the American Society for Testing Materials (10).

A solution suitable for selectively dissolving the anodized coating on titanium could not be found. These coatings were, therefore, measured by sectioning the coupon and examining the cross section on a metallograph. The results of the measurements are presented in Table 5.

Coupons of anodized aluminum were also sectioned and measured on the metallograph to check the results obtained with the stripping and weighing technique. This method was tried on magnesium; however, the coating was so soft that a sharp cross section could not be obtained. A comparison of coating thickness data for aluminum by both methods is given in Table 5.

**TABLE 5**  
**ANODIZED COATING THICKNESS**

Material	Solution	Time	Electro-plated	Sealed	Spec. No.	Thickness (in)	Spec. No.	Thickness (cm)	Assumed Oxide Density	
									A1 7075	A1 1075
Aluminum	Sulfuric acid	Std/3	yes	no	122	.000231	94	.000231	3.99 gm/cm <sup>3</sup>	
	"	Std	yes	yes	121	.000554	96	.000711	"	
	"	Std	yes	no	119	.000482	97	.000635	"	
	"	Std/3	yes	yes	123	.000231	102	.00033	"	
	"	Std/3	no	yes	124	.000221	103	.000254	"	
	"	Std	no	no	125	.000635	104	.000584	"	
	"	Std	no	yes	126	.000381	105	.000736	"	
Aluminum	Chromic acid	Std	yes	no	109	.000239	91	.000163	"	
	"	Std/3	yes	no	112	.000022	93	.000046	"	
	"	Std/3	no	no	111	.000019	98	.000066	"	
	"	Std	no	no	113	.000016	99	.000145	"	
Aluminum	Hard coat	Std	yes	yes	130	.00481	0	.0061	"	
	"	Std	yes	no	128	.00559	106	.00508	"	
	"	Std/3	yes	no	151	.00140	107	.00231	"	
	"	Std/3	no	yes	133	.00239	134	.00231	"	
	"	Std	no	no	127	.00508	136	.00532	"	
Aluminum	Boric acid	Std	yes	no	116	.000028	M	.000043	"	
	"	Std	no	no	114	.000038	101	.000038	"	
	"	Std/3	yes	no	117	.000028	135	.000033	"	
Magnesium	A731B					HK31A				
	HAE	Light cast	no	no	148	.000356	219	.000597	3.65 gm/cm <sup>3</sup>	
	HAE	Dark	no	no	145	.01922	216	.02584	"	
	Dow 17	Light	no	no	146	.00027	217	.00496	"	
	Dow 17	Dark	no	no	149	.02639	218	.02152	"	
Beryllium	QMV					1% copper				
	Chromic acid	Std	no	yes	172	.000295	168	.000719	3.02 gm/cm <sup>3</sup>	
	" "	Std/3	no	yes	-	-	170	.000103	"	
	Nitric acid	Std	no	yes	174	.000927	167	.000949	"	
	" "	Std/3	no	yes	176	.000613	-	-	"	
Titanium	Sodium hydroxide	Std	no	yes	173	.000086	166	.000106	"	
		Std	yes	yes	177	.000126	164	.000159	"	
	A 110 AT					B 120 VAC				
	Sulfuric acid	Std	no	yes	43	.00004	207	.00004	none	
	"	Std	yes	yes	38	.00004	203	.00004	"	
Proprietary	"	Std	no	yes	45	.00004	-	-	"	
	"	Std	yes	yes	48	.00004	-	-	"	
	Proprietary	Std	no	yes	178	.00004	-	-	"	
	"	Std	no	yes	41	.00008	199	.00011	"	
	"	Std	yes	yes	42	.00008	201	.00019	"	
A	Sectioning Method					Stripping Method				
	1075	Hard coat	Std	yes	yes	15	.0089	0	.0061	
	"	"	Std/3	yes	no	20	.0032	107	.00231	
	7075	Chromic acid	Std	yes	no	49	.00019	109	.000239	
	"	"	Std/3	yes	no	51	.00005	112	.000021	
	"	Boric acid	Std	yes	no	52	.00010	116	.000028	
	1075	Chromic acid	Std	yes	no	55	.00024	91	.000163	
	"	"	Std/3	yes	no	57	.00010	93	.000046	
	"	Boric acid	Std	yes	no	59	.00010	M	.000043	
	"	"	Std/3	yes	no	60	.0008	135	.000033	
	7075	"	Std/3	yes	no	62	.00005	117	.000028	
	"	Fluor. acid	Std	yes	yes	64	.00012	121	.000559	
	"	"	Std/3	yes	no	65	.00036	127	.000231	
	1075	"	Std/3	yes	no	66	.00036	74	.000241	
	"	"	Std	yes	yes	68	.00011	77	.000711	
	7075	Hard coat	Std	yes	yes	74	.00011	149	.00442	
	"	"	Std/3	yes	no	75	.00019	151	.00140	

In most cases the thicknesses determined by the "stripping" technique was less than that by the "sectioning" method. Handbook density values for the metallic oxide were used to calculate thickness by the stripping method. Lower thickness values by this method, therefore, indicated that the anodized coatings had a lower density than the handbook values and probably were porous. The difference between the two measurements may, therefore, be used as an indication of the porosity of the anodized coatings.

## RESULTS

The reflectance data are grouped according to alloy in Tables 7 through 31 and are presented graphically in Figures 23 through 186. Table 6 serve as an index to the curves.

The large number of variables made it impractical to attempt to measure all possible combinations of variables. Combinations of variables were therefore selected as shown in Tables 7 through 31 to give a good survey of the range of reflectance values that could be expected for each metal. For example, measurements were made at elevated temperatures for only one specimen of each alloy anodized by each specified process. The effect of elevated temperature on specimens measured only at room temperature can be estimated by comparing with changes produced in the similar specimen measured at elevated temperatures.

Several factors limited the maximum temperature at which measurements could be made. Chief among these was the vapor pressure of the metal. Any evaporation from the specimen would damage the hemisphere surface. Magnesium has a vapor pressure of  $10^{-6}$  mm Hg at only  $480^{\circ}\text{F}$ . Measurements on magnesium were therefore limited to  $350^{\circ}\text{F}$ . Aluminum specimens softened and sagged and were severely damaged somewhere between  $700^{\circ}\text{F}$  and  $1000^{\circ}\text{F}$ . The maximum temperature for the aluminum was therefore held to  $825^{\circ}\text{F}$ .

At 1200° F and above the intensity of emitted radiation from the titanium and beryllium specimens with low reflectance was so high compared to the reflected energy that measurement accuracy became very poor. Beryllium temperatures were therefore limited to 1200° F and titanium to 1300° F. A method for eliminating the problem of poor accuracy due to emitted radiation with high specimen temperatures is discussed in the section, "Recommendations for Future Work". Time did not permit an investigation of this technique, however.

Rather severe outgassing from a titanium specimen was observed at about 1350° F and specimens heated to this temperature in vacuum or 1500° F in air showed marked changes in appearance. For these reasons 1300° F was considered a practical temperature limit for titanium.

As was expected, the reflectance of the specimens was strongly influenced by the oxide coating thickness. The boric acid process which produced a very thin coating on aluminum was almost transparent at all wavelengths. On the other hand, the hard coat process produced thick, hard oxide films with relatively low reflectance. It was observed that the hard coat films were crazed; punching the specimens from the anodized coupons and heating to elevated temperatures increased the crazing in these coatings.

The sulfuric acid process produced a very thin, transparent, deep-blue film on titanium. However, it was so fragile that it could be easily wiped off. The proprietary process produced an excellent coating, but it darkened appreciably at elevated temperatures. The sodium hydroxide process worked well on 110AT titanium, but produced no oxide coating at all on B120VAC. The sodium hydroxide anodized titanium specimen heated in air to 1500° F changed so greatly in appearance that it was considered unusable and was not measured.

Elevated temperatures appeared to produce two major effects. As would be expected, if water was present in the oxide film this was driven out at the higher temperatures as evidenced by the disappearance of the absorption

bands - especially at 3 microns. The second effect was a darkening or decrease in the reflectance of some coatings. This frequently had a greater effect in the short-wavelength region than in the long-wavelength region.

Absorptance or emittance values can be obtained by subtracting the reflectance values from unity. The ratio of solar absorptance to infrared emittance ( $\alpha/\epsilon$ ) can then be estimated by comparing the average absorptance in the 0.4 to 1.6 micron region to the average emittance in the 7.0 to 20 micron region. An estimate of this type shows that  $\alpha/\epsilon$  for bare magnesium or boric acid anodized aluminum was about 2. The hard coat process gave a ratio of about 0.5, chromic acid about 0.4, and sulfuric acid about 0.3 on 1075 aluminum. The sulfuric acid process gave this low value with a relatively high infrared emittance. This would be an advantage on a spacecraft where there is appreciable internal heat generation (occupants, etc.) which must be radiated to space.

Since magnesium oxide has a very high reflectance in the visible region, one would expect anodized magnesium to have a low  $\alpha/\epsilon$  ratio. Just the reverse of this was found. Bare magnesium had an  $\alpha/\epsilon$  of about 7. The light Dow 17 process on AZ31B also had a ratio of about 7. The  $\alpha/\epsilon$  for the dark Dow 17 was reduced to about 0.7, however, due to the increase in the infrared emittance. Both solar absorptance and infrared emittance were higher with the HAE process and  $\alpha/\epsilon$  ranged from about 1 to 3.

The  $\alpha/\epsilon$  ratios for titanium were all high and varied from about 3.5 to 7. The sodium hydroxide process on 110 AT gave a solar absorptance of about 0.70 and an infrared emittance of about 0.10 making this an attractive combination for a solar collector.

Anodized beryllium was also found to have relatively high values of  $\alpha/\epsilon$ . The nitric acid process gave values for the ratio of about 1 and the sodium hydroxide process on beryllium plus 1% copper alloy gave a ratio of about 5.

The solar absorptance for this latter combination was about 0.92, however, indicating the blackness of its appearance. The extremely light weight of beryllium coupled with the high  $\alpha/\epsilon$  would make it attractive as a solar collector surface for space craft applications. Here toxicity of the beryllium oxide and high cost would not be serious disadvantages.

The type of anodizing process used was more significant than the effect of alloying elements in the substrate. In the case of magnesium for example, the HAE process produced tan or brown coatings on both alloys whereas the Dow 17 process produced white or green coatings on both alloys depending on whether or not the process was carried to the dark condition. These differences extended into the infrared also.

It was necessary to wash specimens produced by a number of the processes in boiling water to remove traces of the anodizing solution. This had the effect of a hot water seal on those films subject to sealing and produced water absorption bands in the infrared spectrum. As mentioned previously, heating to elevated temperature removed this water and produced changes in the reflection values.

The effect of pretreatment was not consistent. In the case of thin coatings on aluminum - chromic acid and boric acid processes - electro-polishing increased the reflectance, particularly at the short wavelengths. There was no noticeable difference in the case of the sulfuric acid and hard coat processes. In the case of magnesium and particularly in the case of beryllium, electropolishing reduced the reflectance. The electropolished beryllium specimens had a thinner oxide film. In the case of titanium, electropolishing produced slightly higher reflectance with the sodium hydroxide process and slightly lower reflectance with the proprietary process. It appears that no generalization can be drawn regarding the effect of electropolishing prior to anodizing.

## CONCLUSIONS

The results of this study indicate that the thermal radiation properties of the metals investigated can be varied over a wide range by the selection of the anodizing process and bath conditions. Aluminum anodized by the sulfuric acid process gave a low value of the solar absorptance, infrared emittance ratio and might be suitable for spacecraft applications if the surface were protected from high temperatures during the launch. On the other hand, sodium hydroxide anodized titanium might be considered for solar energy collectors due to its high value of  $\alpha/\epsilon$ .

The combination of elevated temperatures and high vacuum definitely removed water from those films which contained water and thus increased the reflectance at the wavelengths where water absorption band occurred. Decreases in reflectance especially in the short-wavelength region, were also observed in some heated specimens.

The effect of electropolishing prior to anodizing was not consistent. In some cases it increased the reflectance and in some cases it decreased reflectance.

The integrating hemisphere reflectometer proved to be capable of measurements under a wider range of conditions than has hitherto been possible in a single apparatus. It is believed that further development can increase both the range, i. e., spectral range, resolution and permissible specimen temperature, and the precision of measurements with the device.

## RECOMMENDATIONS FOR FUTURE WORK

The successful operation of the integrating hemisphere reflectometer depends on the characteristics of the diffusers. Both the reflectance and the diffuseness must be accurately known. It is suggested that a further search be made for better diffusers.

An alternate method which would eliminate the need for diffusers is recommended for consideration. This method involves locating the energy source at one of the conjugate foci. If a small black body could be constructed to operate at approximately 2500 F at  $10^{-6}$  mm Hg, this could be then located at the conjugate focus. This would also eliminate any internal reflections and, therefore, reduce the effect of spherical aberrations.

The hemisphere was used as an approximation to an ellipsoid; however, spherical aberration did present a problem. An ellipsoid would greatly reduce or eliminate spherical aberration permitting the spectrometer slits to be varied to control energy, and give greatly increased spectral resolution.

At the higher specimen temperatures, the energy emitted by the specimen (especially low reflectance specimens) became an appreciable part of the total energy reaching the sensor. This signal at times became so large that the zero balance on the spectrometer could not balance out this signal. It is suggested that under these conditions the specimen itself could become the energy source.

When the specimen is heated to a given temperature it becomes an energy source. The specimens and diffuser positions could then be interchanged so that the sensor receives energy reflected from the diffuser. If the energy reaching the sensor is then added up for an infinite number of internal reflections, the sum of the series will be:

$$G_1 = \frac{w \rho_h \rho_{d_1} D F}{\left[ 1 - \rho_h^2 \rho_s \rho_{d_1} D \right]} \quad (32)$$

If the first diffuser is now replaced by a second diffuser with a different reflectance, a reference measurement can be obtained. The two equations can then be solved for the specimen reflectance

$$\rho_s = \frac{\rho_{d_1} - \frac{G_1}{G_2} \rho_{d_2}}{\rho_h^2 \rho_{d_1} \rho_{d_2} \left( 1 - \frac{G_1}{G_2} \right)} \quad (33)$$

This technique would provide a means of making measurements at high specimen temperatures where the emitted energy level is high compared with the reflected energy in the present schemes. Some research would be required, however, to find a pair of suitable diffusers.

The authors believe that this technique offers interesting possibilities and recommend a further investigation of the method.

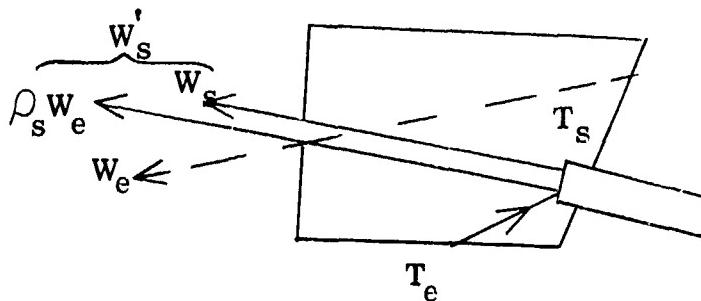
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## APPENDIX TO PART I

### THEORY OF BLACK BODY REFLECTOMETER

The sketch below shows schematically the source of the radiant energies of interest.



The energy,  $W'_s$ , from the specimen consists of emitted energy,  $W_s$ , plus reflected energy,  $\rho_s W_e$ . The energy incident on the specimen is the environment radiation,  $W_e$ , from the hohlraum.

The environmental radiation at any wavelength is given by Planck's equation:

$$W_e = \frac{C_1}{\lambda^5 (e^{C_2/\lambda T_e} - 1)} = \frac{C_1}{\lambda^5 (x_e - 1)} \quad (34)$$

where  $C_1$  = Planck's first constant

$\lambda$  = wavelength

$C_2$  = Planck's second constant

$T$  = absolute temperature

Subscript e refers to environment

Similarly the energy emitted by a specimen with a spectral emittance  $\epsilon_s$  is:

$$w_s = \frac{\epsilon_{\lambda s} c_1}{\lambda^5 (e^{c_2/\lambda T_s} - 1)} = \frac{\epsilon_{\lambda s} c_1}{\lambda^5 (x_s - 1)} \quad (35)$$

A third measurement of a black body at a temperature,  $T_b$ , different from the hohlraum temperature,  $T_e$ , is required. This energy is given by

$$w_b = \frac{c_1}{\lambda^5 (e^{c_2/\lambda T_b} - 1)} = \frac{c_1}{\lambda^5 (x_b - 1)} \quad (36)$$

Making use of the relationship  $\rho_s = 1 - \epsilon_s$ , the total energy from the specimen is:

$$w_s' = \rho_{\lambda s} \frac{c_1}{\lambda^5} \left[ \frac{1}{x_e - 1} - \frac{1}{x_s - 1} \right] + \frac{1}{x_s - 1} \quad (37)$$

Subtracting equation (36) from (34) and (35) and combining these equations gives the following expression for  $\rho_{\lambda s}$ :

$$\rho_{\lambda s} = \left[ \frac{w_s' - w_b}{w_e - w_b} \right] \left[ \frac{\frac{1}{(x_e - 1)} - \frac{1}{(x_b - 1)}}{\frac{1}{(x_e - 1)} - \frac{1}{(x_s - 1)}} \right] - \left[ \frac{\frac{1}{(x_s - 1)} - \frac{1}{(x_b - 1)}}{\frac{1}{(x_e - 1)} - \frac{1}{(x_b - 1)}} \right] \quad (38)$$

If the reference black body measurement is made with the temperature equal to the specimen temperature,

$$T_b = T_s \quad \text{and} \quad x_b = x_s \quad (39)$$

then

$$\rho_{\lambda_s} = \frac{W_s - W_b}{W_e - W_b} \quad (40)$$

Also if the specimen temperature is low compared with the environmental temperature,

$$T_s \ll T_e \quad (41)$$

the energy radiated by the black body at the specimen temperature will be small and can be neglected. Then

$$\rho_{\lambda_s} = \frac{W'_s}{W_e} \quad (42)$$

This is the equation commonly used with the black body reflectometer. Care must be exercised in the use of this equation to make certain that the energy emitted by the specimen can be ignored. Equation (40) was used to estimate the errors in this assumption when using heated specimens and it was found that the effect of specimen emission could not be ignored. Equation (40) was, therefore, proposed for use with the black body reflectometer.

PART II  
PRESENTATION OF REFLECTANCE DATA

CODE FOR DATA:

MF - Mill Finish

MP - Mechanically Polished

EP - Electropolished

Std - Standard Anodizing Time

RT - Room Temperature

RT<sub>f</sub> - Room Temperature after heating in vacuum

HT - Heat Treated in air at specified temperature  
for 30 minutes

INDEXING OF SPECIMEN DATA:

The spectral reflectance data presented in Figures 23-186  
are arrayed in the same sequence as the data presented in  
Tables 7-31

TABLE 6  
INDEX OF SPECIMEN VARIABLES

TABLE 6 (continued)

Specimen Preparation						No. of Specimen Measured								
Pre-Treatment	Anodizing Bath	Time	Sealed	Heat Treat.	no	Alloy AZ31B		Alloy HK31A		RT	350 F	RT <sub>f</sub>	350 F	RT <sub>f</sub>
						RT	350 F	RT <sub>f</sub>	350 F					
MF	none		no	no	no	156								
MP, EP	HAE	light	no	no	no	25	25	25	25	215				
MP, EP	"	"	no	no	no	36								
MP, EP	"	"	no	800	no	25								
MP	"	"	no	no	no	138								
MP	"	"	no	800	no	211	211	211	211	211				
MF	"	"	no	no	no	138								
MF, EP	HAE	dark	no	no	no	155								
MF	"	"	no	no	no	140								
MP	"	"	no	no	no	800	140							
MF	"	"	no	no	no	140								
MP, EP	Dow17	light	no	no	no	31	31	31	214	214	214			
MF, EP	"	"	no	800	no	152								
MF	"	"	no	no	no	152								
MF	"	"	"	"	"									
MC, FP	Dow17	dark	no	no	no	35	35	35	35	35				
MF, EP	"	"	no	no	no	800	35							
MF	"	"	"	"	"	no	158	158	213	213				

TABLE 6 (continued)

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3.00 U.S.A. 3.00 U.S.A.

TABLE 7

## SPECTRAL REFLECTANCE OF 7075 ALUMINUM ANODIZED IN CHROMIC ACID

Specimen No.	50 MP, EP	50 MP, EP	50 MP, EP	51 Std/3	75 Std/3	110 Std
Pretreatment						
Anodizing Time	Std	Std	Std			
Post Treatment	None	None	None	None	None	None
Spec. Temp.	RT	600°F	RT <sub>f</sub>	RT	RT	RT
<b>Wavelength Microns</b>						
.4	.426		392	500	.303	
.45	.448	.387	424		.352	
.5	.464	.401	431	.561	.383	.545
.6	.523	.446	.471	.559	.397	.601
.7	.517	.454	487	.560	.368	.650
.8	.519	.495	492	.554	.309	.617
.9	.592	.510	579	617	.449	.646
1.0	644	.576	584	645	.497	.764
1.2	723	690	730	809	.570	.822
1.4	772	705	756	822	.606	.839
1.6	784	713	760	940	.533	.857
1.8	789	.724	.771	983	.597	.861
2.0	798	738	.771	870	.555	.870
2.5	804	.777	807	903	.647	.886
3.0	807	.777	873	922	.676	.918
3.5	.832	856	891	943	.708	.933
4	854	875	907	959	.721	.966
5	872	896	931	.763	.745	.962
6	870	913	951	.779	.712	.969
7	888	922	950	.884	.776	.981
8	895	938	946	.906	.792	.987
9	905	936	942	.935	.791	.979
10	870	931	927	.972	.779	.963
11	844	929	.939	.977	.755	.958
12	838	.938	.948	.977	.758	.979
13	847	.937	.946	.975	.763	.993
14	856	938	.946	.973	.770	.991
15	857	936	.941	.973	.776	.991
16	859	937	.943	.971	.771	.991
17	854	930	.940	.970	.775	.991
18	855	934	.942	.970	.777	.991
19	856	935	.943	.970	.778	.991

TABLE 8

## SPECTRAL REFLECTANCE OF 1075 ALUMINUM IN CHROMIC ACID

Specimen No.	55	55	55	55	55	56	57	83	84	92
Pretreatment	MP, EP	MP, EP	MP, EP	MP	MP	MF, EP				
Anodizing Time	Std	Std	Std	Std	Std	Std	Std/3	Std/3	Std	Std
Post Treatment	None	None	None	None	None	HT 800°F	None	None	None	None
Spec. Temp	RT	300° F	600° F	825°F	RT <sub>f</sub>	RT	RT	RT	RT	RT
Wavelength, Microns										
.4	.685				.561			550	.477	.78P
.45	.718	.707	.680	.659	.619	.736	.840			
.5	.740	.731	.749	.688	.650	.770	.906	.533	.566	.784
.6	.734	.788	.770	.752	.706	.817	.841	.541	.582	.885
.7	.730	.740	.766	.728	.674	.819	.956	.547	.574	.784
.8	.725	.799	.770	.931	.676	.770	.942	.548	.533	.813
.9	.764	.770	.784	.774	.723	.802	1025	.563	.595	.824
1.0	.832	.823	.806	.780	.819	.896	1.031	.582	.744	.894
1.2	.874	.884	.890	.826	838	.916	.904	.678	.896	.925
1.4	.895	.873	.861	.842	.871	.938	.952	.752	.823	.927
1.6	.895	.896	.886	.856	883	.959	.976	.882	.828	.952
1.8	.909	.908	.894	.870	.893	.968	.984	.836	.847	.957
2.0	.926	.918	.906	885	.900	.974	1.069	.826	.840	.963
2.5	.930	.927	.918	.897	.918	.981	.997	.836	.864	.972
3.0	.928	.923	.928	.899	.922	.947	.976	903	.839	.947
3.5		.937	.936	911	.941	.966	.975	922	.883	.976
4	.959	.947	.940	920	.948	.981	.978	.941	.917	.982
5	.953	.938	.944	902	.933	.971	.989	.938	.913	.971
6	.967	.950	.934	926	.950	.97	1.025	.933	.932	.982
7	.941	.931	.904	897	.932	.956	1.001	.953	.921	.958
8	.906	.895	.865	843	.894	.907	.985	.980	.861	.916
9	.810	.803	.787	752	.770	801	.983	.960		.818
10	.641	.618	.629	604	.625	.625	.950	.893	.678	.643
11	.439	.465	.495	515	.461	.439	.938	.842		.426
12	.343	.362	.427	471	383	.398	.950	.825	.411	.364
13	.332	.340	.362	612	.321	.342	.959	.806	.380	.366
14	.280	.307	.332	316	.278	.293	.765	.793	.348	.389
15	.270	.270	.278	287	.256	.290	.954	.782	.226	.308
16	.228	.264	182	376	.173	.266	.964	.808	.165	.212
17	.187	.182		304	.217	.238	.987	.812	.158	.236
18	.191	.215		129	.153		.782	.864	.242	.190
19	.086	.175			.103			.874	.124	
20		.137			.117			.712		.333

TABLE 9  
SPECTRAL REFLECTANCE OF 7075 ALUMINUM  
ANODIZED IN SULFURIC ACID

Specimen No	63	63	63	63	63	64	65	66
Pretreatment	MP, EP	MP, EP	MP, EP	MP, EP				
Anodizing Time	Std	Std	Std	Std	Std	Std	Std	Std/3
Post Treatment	Sealed	Sealed	Sealed	Sealed	Sealed	Sealed	None	None
Spec. Temp.	RT	300°F	600°F	825°F	RT <sub>f</sub>	RT	RT	RT
Wavelength Microns								
4		.453	332	.309		495	549	
45	499	533	477	395	311	553	512	585
.5	532	.556	514	.417	334	611	567	635
6	575	584	521	454	367	656	617	671
7	.623	573	.523	476	367	658	613	683
8	.637	583	.581	.464	.418	.639	.585	.658
9	637	632	573	548	458	675	632	687
1.0	741	652	595	557	512	804	778	705
1.2	751	689	643	607	582	813	800	.853
1.4	778	720	663	.596	607	827	828	.881
1.6	804	768	684	615	627	850	847	898
1.8	828	745	703	631	638	851	.851	908
2.0	853	801	714	641	657	851	867	922
2.5	799	725	732	689	641	786	857	987
3.0	225	574	718	690	554	217	585	814
3.5	430	689	764	734	620	426	705	889
4	565	743	760	723	677	576	788	916
5	67	784	764	724	743	676	842	940
6	72	777	785	735	730	644	738	881
7	623	673	694	601	699	622	765	886
8	149	056	164	419	419	141	118	450
9	059	022		4.7	385	015	047	2.3
10	064	014		347	334	070	107	269
11		056	159	419	406	094	097	100
12				430	427	132	193	173
13	11t	131	036	479	470	160	173	121
14				363	461	170	535	169
15	20t	211		4.3	362	t	246	23
16				34	33	34	314	314
	4.4			t	3	04	t	t
						t	t	t

TABLE 9 (Continued)

Specimen No.	73	74	77	79	120
Pretreatment	MP, EP	MP	MP	MP	MP, EP
Anodizing Time	Std. 3	Std. 3	Std.	Std.	Std.
Post Treatment	Sealed	Sealed	Sealed	Sealed	Sealed
Spec. Temp.	RT	RT	RT	RT	RT
Wavelength Microns					
4.0	518				
4.5	638	607		484	591
5	677	639	498	519	653
6	703	679	547	565	676
7	692	668	537	626	694
8	694	660	53	576	717
9	754	733	613	653	769
10	788	770	693	732	825
11	816	808	743	787	912
12	848	836	773	805	953
13	863	863	760	819	9608
14	877	853	714	833	933
15	894	873	807	834	954
16	874	859	871	739	710
17	610	586	643	604	7.0
18	736	704	16	503	4.1
19	—	514	790	624	—
20	—	614	76	716	6.0
21	—	514	790	624	—
22	—	514	790	624	—
23	—	514	790	624	—
24	—	514	790	624	—
25	—	514	790	624	—
26	—	514	790	624	—
27	—	514	790	624	—
28	—	514	790	624	—
29	—	514	790	624	—
30	—	514	790	624	—
31	—	514	790	624	—
32	—	514	790	624	—
33	—	514	790	624	—
34	—	514	790	624	—
35	—	514	790	624	—
36	—	514	790	624	—
37	—	514	790	624	—
38	—	514	790	624	—
39	—	514	790	624	—
40	—	514	790	624	—
41	—	514	790	624	—
42	—	514	790	624	—
43	—	514	790	624	—
44	—	514	790	624	—
45	—	514	790	624	—
46	—	514	790	624	—
47	—	514	790	624	—
48	—	514	790	624	—
49	—	514	790	624	—
50	—	514	790	624	—
51	—	514	790	624	—
52	—	514	790	624	—
53	—	514	790	624	—
54	—	514	790	624	—
55	—	514	790	624	—
56	—	514	790	624	—
57	—	514	790	624	—
58	—	514	790	624	—
59	—	514	790	624	—
60	—	514	790	624	—
61	—	514	790	624	—
62	—	514	790	624	—
63	—	514	790	624	—
64	—	514	790	624	—
65	—	514	790	624	—
66	—	514	790	624	—
67	—	514	790	624	—
68	—	514	790	624	—
69	—	514	790	624	—
70	—	514	790	624	—
71	—	514	790	624	—
72	—	514	790	624	—
73	—	514	790	624	—
74	—	514	790	624	—
75	—	514	790	624	—
76	—	514	790	624	—
77	—	514	790	624	—
78	—	514	790	624	—
79	—	514	790	624	—
80	—	514	790	624	—
81	—	514	790	624	—
82	—	514	790	624	—
83	—	514	790	624	—
84	—	514	790	624	—
85	—	514	790	624	—
86	—	514	790	624	—
87	—	514	790	624	—
88	—	514	790	624	—
89	—	514	790	624	—
90	—	514	790	624	—
91	—	514	790	624	—
92	—	514	790	624	—
93	—	514	790	624	—
94	—	514	790	624	—
95	—	514	790	624	—
96	—	514	790	624	—
97	—	514	790	624	—
98	—	514	790	624	—
99	—	514	790	624	—
100	—	514	790	624	—
101	—	514	790	624	—
102	—	514	790	624	—
103	—	514	790	624	—
104	—	514	790	624	—
105	—	514	790	624	—
106	—	514	790	624	—
107	—	514	790	624	—
108	—	514	790	624	—
109	—	514	790	624	—
110	—	514	790	624	—
111	—	514	790	624	—
112	—	514	790	624	—
113	—	514	790	624	—
114	—	514	790	624	—
115	—	514	790	624	—
116	—	514	790	624	—
117	—	514	790	624	—
118	—	514	790	624	—
119	—	514	790	624	—
120	—	514	790	624	—
121	—	514	790	624	—
122	—	514	790	624	—
123	—	514	790	624	—
124	—	514	790	624	—
125	—	514	790	624	—
126	—	514	790	624	—
127	—	514	790	624	—
128	—	514	790	624	—
129	—	514	790	624	—
130	—	514	790	624	—
131	—	514	790	624	—
132	—	514	790	624	—
133	—	514	790	624	—
134	—	514	790	624	—
135	—	514	790	624	—
136	—	514	790	624	—
137	—	514	790	624	—
138	—	514	790	624	—
139	—	514	790	624	—
140	—	514	790	624	—
141	—	514	790	624	—
142	—	514	790	624	—
143	—	514	790	624	—
144	—	514	790	624	—
145	—	514	790	624	—
146	—	514	790	624	—
147	—	514	790	624	—
148	—	514	790	624	—
149	—	514	790	624	—
150	—	514	790	624	—
151	—	514	790	624	—
152	—	514	790	624	—
153	—	514	790	624	—
154	—	514	790	624	—
155	—	514	790	624	—
156	—	514	790	624	—
157	—	514	790	624	—
158	—	514	790	624	—
159	—	514	790	624	—
160	—	514	790	624	—
161	—	514	790	624	—
162	—	514	790	624	—
163	—	514	790	624	—
164	—	514	790	624	—
165	—	514	790	624	—
166	—	514	790	624	—
167	—	514	790	624	—
168	—	514	790	624	—
169	—	514	790	624	—
170	—	514	790	624	—
171	—	514	790	624	—
172	—	514	790	624	—
173	—	514	790	624	—
174	—	514	790	624	—
175	—	514	790	624	—
176	—	514	790	624	—
177	—	514	790	624	—
178	—	514	790	624	—
179	—	514	790	624	—
180	—	514	790	624	—
181	—	514	790	624	—
182	—	514	790	624	—
183	—	514	790	624	—
184	—	514	790	624	—
185	—	514	790	624	—
186	—	514	790	624	—
187	—	514	790	624	—
188	—	514	790	624	—
189	—	514	790	624	—
190	—	514	790	624	—
191	—	514	790	624	—
192	—	514	790	624	—
193	—	514	790	624	—
194	—	514	790	624	—
195	—	514	790	624	—
196	—	514	790	624	—
197	—	514	790	624	—
198	—	514	790	624	—
199	—	514	790	624	—
200	—	514	790	624	—
201	—	514	790	624	—
202	—	514	790	624	—
203	—	514	790	624	—
204	—	514	790	624	—
205	—	514	790	624	—
206	—	514	790	624	—
207	—	514	790	624	—
208	—	514	790	624	—
209	—	514	790	624	—
210	—	514	790	624	—
211	—	514	790	624	—
212	—	514	790	624	—
213	—	514	790	624	—
214	—	514	790	624	—
215	—	514	790	624	—
216	—	514	790	624	—
217	—	514	790	624	—
218	—	514	790	624	—
219	—	514	790	624	—
220	—	514	790	624	—
221	—	514	790	624	—
222	—	514	790	624	—
223	—	514	790	624	—
224	—	514	790	624	—
225	—	514	790	624	—
226					

TABLE 10  
SPECTRAL REFLECTANCE OF 1075 ALUMINUM ANODIZED IN SULFURIC ACID

Specimen No.	67	68	69	69	69	71	71	72	80	81
Pretreatment	MP, EP	MP, EP	MP, EP	MP, EP	MP, EP	MP, EP	MP, EP	MP	MP	MP
Anodizing Time	Std/3	Std	Std	Std	Std	Std/3	Std/3	Std/3	Std	Std
Post Treatment	None	Std. HT 800°F	Sealed	Sealed	Sealed	Sealed	Sealed	Sealed	None	Sealed
Spec. Temp.	RT	RT	RT	600°F	RT	RT	300°F	RT	RT	RT
Wavelength Microns										
.4	.980	.704		.732	.646	.852	.861	.503		
.45		.705	.736	.725	.701			.466	.712	.981
.5	.977	.715	.764	.745	.721	.896	.679	.497	.720	.786
.6	.751	.709	.761	.746	.723	.608	.903	.499	.722	.779
.7	.957	.701	.739	.747	.711	.862	.859	.480	.706	.760
.8	.949	.738	.727	.741	.717	.878	.845	.721	.706	.760
.9	.970	.700	.732	.753	.734	.911	.828	.529	.734	.766
1.0	.957	.773	.833	.764	.769	.898	.945	.564	.775	.815
1.2	.962	.818	.849	.807	.83..		.907	.670	.817	.866
1.4	.960	.834	.848	.837	.852	.944	.899	.667	.849	.874
1.6	.975	.833	.862	.847	.861	.947	.911	.699	.866	.873
1.8	.938	.845	.863	.857	.824	.968	.432	.723	.851	.910
2.0	.976	.849	.843	.861	.859	.944	.931	.754	.864	.932
2.5	.988	.823	.773	.843	.819		.937	.746	.863	.839
3.0	.910	.448	.284	.828	.561	.696	.900	.572	.732	.285
3.5	.951	.658	.340	.872	.702		.916	.766	.827	.582
4	.966	.745	.666	.878	.776	.897	.930	.803	.886	.720
5	.981	.822	.737	.884	.854	.921	.926	.920	.914	.819
6	.891	.776	.724	.848	.806	.912	.921	.855	.896	.762
7	.900	.724	.649	.778	.732		1.031	.908	.829	.704
8	.444	.038	.032	.041	.046	187	22*	.408	.022	.143
9	.283	.068	.046	.037	.054		.062	.203	.045	.021
10	.242	.063	.039	.055	0 ..	.224	.249	.217	.035	.067
11	.141	.106	.089	.143	.129	.100	.108	.121	.203	.114
12	.240	.215				.234	.258	.287	.106	.244
13	.294	.272	.229	.370	.273	.294	.275	.232	.280	
14	.299	.235				.306	.352	.382	.304	.347
15	.354	.313	.315	.396	.328	.390	.375	.338	.365	
16	.394	.330				.326	.373	.458	.363	.393
17	.393	.298	.316	.382	.314	.389	.422	.379	.413	
18	.386	.282				.367	.383	.312	.398	.365
19	.347	.359	.276	.429	.332	.371	.523	.435	.434	
20	.325					.304	.349		.364	.432
21		.342	.432			.379			.371	.473
22						516				

TABLE 11  
SPECTRAL REFLECTANCE OF 7075 ALUMINUM HARD COAT ANODIZED

Specimen No	11	85	86	87	87	87	87	87	88	89	129
Pre-treatment	MP	MP	MP, EP	MP, FP	MP, EP	MP, EP	MF, EP				
Anodizing Time	Std/3	Std	Std	Std	Std	Std	Std	Std	Std	Std/3	Std
Post Treatment	Sealed	None	None	Sealed	Sealed	Sealed	Sealed	Sealed	Std, HT-800°F	None	Sealed
Spec Temp	RT	RT	RT	RT	300°F	600°F	825°F	RT	RT	RT	RT
Wavelength Microns											
4	.099								203		193
45	111	172	160	163	144	070	065	079	210	211	184
5	120	160	161	155	150	070	068	075	214	245	163
6	137	147	149	159	149	071	068	074	214	303	175
7	159	141	135	153	150	076	071	077	202	332	170
8	218	164	120	162	200	135	091	135	243	395	
9	245	141	125	190	188	123	122	141	240	452	202
10	280	156	142	229	217	163	153	159	286	516	235
12	362	203	196	300	296	253	245	252	356	625	285
14	441	251	250	364	347	330	314	318	407	668	335
16	465	290	284	400	389	373	361	369	423	699	362
18	498	326	319	426	420	415	399	411	454	728	379
20	536	347	335	549	445	450	408	441	480	752	387
25	485	311	293	479	402	445		473	411	707	309
30	413	082	074	138	169	351		426	104	450	678
35	133	108	096	127	282	422		497	135	588	664
40	207	162	146	182	368	435		535	206	682	681
5	317	183	247	301	437	439		542	344	755	140
6	229	145	134	208	313	668		399	205	664	690
7	178	085	067	124	137	037		114	114	500	670
8	036	022	017	040	125			015	018	026	023
9	059	048	048	084	135			028	027	050	068
10	085	033	035	041	135			022	035	043	011
11	081	098	095	107	170			092	087	126	116
12	265	194	177	174	207			192		223	183
13	249		216	220	260			216	263	215	234
14	289	268	256	258	292			245	299	249	
15	290		279	292	268			273	270	299	268
16	352	317	284	309	236			290		345	309
17	347		302	349	267			305	300	352	335
18	350	307	321	328	250			286		358	295
19	399		311	362	415			379		394	333
20	415	285	263	367				470		374	280
21											
22											

TABLE 12

## SPECTRAL REFLECTANCE OF 1075 ALUMINUM HARD COAT ANODIZED

Specimen No	13	15	15	15	20	23	87	500
Pretreatment	MP, EP	MP	MP	MP, EP				
Anodizing Time	Std	Std	Std	Std	Std/3	Std/3	Std	Std
Post Treatment	None	Sealed	Sealed	Sealed	None	Sealed	None	Sealed
Spec. Temp	RT	RT	600° F	RT	RT	RT	RT	RT
Wavelength Microns								
4		190	310	381	345	446		236
45	214	242	403	436	421	485	199	289
5	265	292	448	462	.459	511	229	332
6	319	342	483	500	507	542	276	379
7	.341	366	503	.511	514	551	295	404
8	328	389	492	492	541	624	194	443
9	433	442	.584	575	577	.598	368	480
10	.460	494	584	617	635	691	375	511
12	499	562	625	642	677	745	462	580
14	562	593	682	661	706	757	506	603
16	594	618	670	679	732	775	544	601
18	615	639	696	650	730	793	576	644
20	610	645	.709	707	773	816	603	631
25	576	539	628	611	696	752	554	595
30	208	115	298	324	415	301	237	119
35	323	111	571	508	611	483	394	113
40	450	203	547	572	686	617	486	174
5	561	310	580	600	735	724	585	214
6	351	196	342	412	617	639	409	177
7	112	077	162	104	327	442	137	374
8	022	026	175	013	024	046	029	023
9	052	071		058	018	039	018	056
10	016	049	215	035	032	052	058	026
11	093	100		116	063	193	126	067
12	219	251	232	238	128	255	154	219
13	317	367	276	294	1	316		275
14	336	343		298	220	377	364	339
15	330	386	474	283	253	385		342
16	356	379		314	261	370	350	372
17	366	368	403	269	274	361		324
18	409	401		362	270	183	399	493
19	328	384		296	260			40
20	412	177		284	141	190	411	
21	341	314			353		171	
22	436							

TABLE 13  
SPECTRAL REFLECTANCE OF 7075 ALUMINUM ANODIZED IN BORIC ACID

Specimen No.	52	52	52	52	52	54	62	115
Pretreatment	MP, EP	MP,	MP, EP	MF, EP				
Anodizing Time	Std	Std	Std	Std	Std	Std	Std/3	Std
Post Treatment	None	None	None	None	None	None	None	None
Spec. Temp.	RT	300°F	600°F	825°F	RT <sub>f</sub>	RT	RT	RT
<b>Wavelength</b>								
Microns								
.4	.492	.599	.509	.139	.428			.495
.45	.521	.542	.550	.481	.450	.638	.660	.481
.5	.596	.587	.577	.543	.513	.601	.620	.538
.6	.530	.510	.475	.455	.421	.701	.744	.501
.7	.579	.617	.537	.562	.510	.735	.763	.533
.8	.601	.665	.587	.643	.574	.636	.739	.538
.9	.621	.656	.605	.627	.539	.692	.709	.621
1.0	.645	.679	.597	.622	.550	.815	.809	.595
1.2	.753	.768	.735	.748	.710	.793	.884	.767
1.4	.798	.810	.788	.803	.737	.892	.912	.788
1.6	.793	.812	.793	.805	.748	.891	.901	.805
1.8	.803	.840	.793	.792	.771	.891	.902	.812
2.0	.807	.837	.794	.780	.759	.901	.904	.810
2.5	.801	.829	.708	.776	.760	.905	.910	.809
3.0	.775	.819	.799	.785	.784	.909	.903	.790
3.5	.807	.836	.813	.819	.806	.949	.932	.813
4.0	.825	.848	.832	.830	.821	.961	.948	.852
5	.852	.867	.848	.833	.859	.979	.964	.880
6	.857	.874	.817	.853	.871	.986	.981	.888
7	.828	.846	.837	.836	.851	.975	.963	.875
8	.836	.856	.850	.841	.858	.982	.973	.886
9	.846	.871	.864	.867	.874	.985	.976	.907
10	.851	.847	.852	.861	.864	.971	.968	.897
11	.821	.836	.838	.832	.856	.947	.939	.914
12	.808	.830	.824	.818	.849	.954	.949	.919
13	.811	.823	.829	.819	.843	.970	.987	.913
14	.821	.837	.826	.846	.839	.979	.976	.909
15	.825	.828	.845	.843	.847	.984	.982	.922
16	.827	.833	.837	.855	.855	.988	.996	.928
17	.814	.856	.860	.821	.856	1.000	.992	.927
18	.815	.855	.878	.647	.857	1.019	.996	.925
19	.825	.874	.681	.855	.888	1.020	1.007	.927
20	.867	.896	.915	.880	.852	1.012	1.024	.904
21	.863	.919	.912	.851	.874	1.041	1.025	.911
22	.866	.904	.893	.853	.793	1.054	.966	1.017

TABLE 14

## SPECTRAL REFLECTANCE OF 1075 ALUMINUM ANODIZED IN BORIC ACID

Specimen No.	58	59	59	59	59	59	60	79	100
Pretreatment	MP, EP	MP, EP	MP, EP	MP, EP	MP, EP	MP, EP	MP, EP	MP	MF, EP
Anodizing Time	Std	Std	Std	Std	Std	Std	Std/3	Std	Std
Post Treatment	HT 800°F	None	None	None	None	None	None	None	None
Spec. Temp.	RT	RT	300°F	600°F	825°F	RT <sub>f</sub>	RT	RT	RT
Wavelength Microns									
.4	.655	.700	.853	.857	.839	.676		.429	.644
.45	.673	.668	.783	.779	.787	.657		.446	.645
.5	.687	.718	.837	.839	.845	.706	.873	.444	.681
.6	.637	.650	.792	.780	.764	.665	.842	.439	.644
.7	.650	.657	.797	.804	.804	.662	.878	.415	.627
.8	.680	.817	.748	.837	.821	.706	.794	.558	.676
.9	.643	.741	.765	.756	.763	.685	.759	.396	.672
1.0	.724	.688	.792	.767	.759	.691	.857	.470	.687
1.2	.806	.817	.878	.884	.865	.787	.891	.528	.793
1.4	.831	.823	.915	.910	.892	.827	.935	.610	.848
1.6	.835	.820	.928	.915	.906	.834	.932	.654	.870
1.8	.837	.837	.934	.907	.899	.831	.935	.674	.856
2.0	.840	.834	.927	.924	.908	.815	.933	.661	.854
2.5	.899	.827	.939	.922	.907	.814	.931	.673	.850
3.0	.848	.823	.950	.933	.914	.813	.933	.655	.843
3.5	.870	.836	.957	.947	.940	.835	.954	.682	.863
4.0	.889	.858	.967	.958	.944	.855	.972	.708	.890
5	.917	.880	.979	.968	.958	.872	.985	.770	.911
6	.925	.874	.982	.980	.984	.874	.993	.788	.923
7	.926	.878	.962	.973	.970	.870	.981	.791	.922
8	.932	.886	.988	.980	.969	.876	.986	.805	.925
9	.936	.882	.988	.982	.973	.873	.938	.807	.926
10	.914	.853	.966	.963	.958	.843	.961	.788	.903
11	.909	.826	.931	.951	.957	.828	.948	.746	.907
12	.921	.853	.976	.954	.951	.838	.959	.756	.929
13	.921	.842	.962	.985	.975	.846	.977	.733	.922
14	.926	.844	.976	.983	.984	.846	.986	.770	.923
15	.932	.862	.958	.993	.993	.853	.970	.787	.921
16	.943	.863	.984	.995	.997	.864	.997	.801	.924
17	.948	.864	.995	1.006	1.009	.853	1.007	.810	.935
18	.940	.870	1.005	1.026	1.020	.876	1.017	.817	.943
19	.950	.893	.999	1.017	1.025	.876	1.027	.829	.941
20	.945	.912	.999	1.027	1.025	.840	1.026	.741	.935
21	.953	.912	.558	1.026	1.016	.875	1.001	.760	.960
22		706	1.054	1.025	1.038	.853			

TABLE 15  
SPECTRAL REFLECTANCE OF AZ31B MAGNESIUM ANODIZED IN HAE

Specimen No.	25 MP, EP Light	25 MP, EP Light	25 MP, EP Light	25 MP, EP Light	36 MP, EP Light	138 MF Light	138 MF Light	140 MP Dark	140 MP Dark	140 MF Dark	155 MP, EP Dark
Pre-treatment Anodizing Coating	None	None	HT 800°F	None	None	None	None	None	HT 800°F	None	None
Post Treatment Spec. Temp.	RT	350°F	RT	RT	RT	RT	RT	RT	RT	RT	RT
Wavelength Microns											
.4	.229	.265	.168	.209	.241	.282	.279	.061	.076	.072	.072
.45	.256	.277	.175	.282	.302	.298	.246	.073	.082	.076	.072
.5	.308	.304	.190	.309	.334	.346	.378	.077	.083	.078	.071
.6	.398	.377	.227	.361	.424	.304	.420	.090	.092	.096	.105
.7	.433	.424	.260	.425	.475	.264	.467	.111	.087	.109	.132
.8	.549	.484	.408	.406	.453	.514	.494	.134	.169	.275	.166
.9	.530	.525	.350	.510	.552	.566	.539	.157	.147	1.025	.181
1.0	.554	.546	.383	.554	.569	.582	.546	.171	.158	.176	.219
1.2	.553	.556	.449	.541	.560	.585	.582	.205	.202	.216	.271
1.4	.569	.555	.468	.553	.576	.590	.581	.235	.228	.235	.301
1.6	.565	.570	.480	.568	.568	.584	.573	.265	.259	.256	.327
1.8	.553	.568	.484	.544	.556	.593	.571	.275	.281	.276	.346
2.0	.559	.580	.497	.668	.570	.621	.590	.303	.299	.296	.360
2.5	616	.628	.569	.618	.659	.707	.687	.325	.339	.322	.401
3.0	.647	.697	.651	.701	.739	.774	.746	.289	.356	.296	.371
3.5	.721	.761	.737	.773	.800	.831	.808	.386	.437	.382	.445
4	.781	.798	.786	.820	.840	.861	.842	.457	.491	.455	.553
5	.865	.854	.861	.875	.893	.903	.880	.522	.547	.519	.630
6	.858	.848	.880	.867	.905	.911	.894	.521	.587	.523	.642
7	.827	.818	.848	.839	.904	.934	.900	.338	.377	.330	.484
8	.886	.871	.885	.893	.917	.925	.907	.313	.351	.294	.514
9	.660	.681	.654	.679	.743	.756	.742	.035	.036	.044	.036
10	.663	.647	.686	.695	.764	.793	.779	.046	.027	.047	.052
11	.790	.771	.806	.795	.858	.891	.846	.084	.066	.092	.165
12	.781	.763	.796	.788	.843	.911	.885	.052		.060	.085
13	.695	.647	.707	.678	.778	.878	.856	.035	.051	.067	.068
14	.743	.702	.718	.719	.801	.876	.858	.161		.203	.264
15	.713	.691	.700	.751	.786	.866	.861	.376	.254	.323	.403
16	.710	.594	.667	.723	.796	.861	.844	.372	.401	.394	.363
17	.760	.679	.738	.730	.790	.896	.841	.427	.519	.424	.436
18	.740	.616	.802	.800	.807	.848	.849	.515	.530	.513	.526
19	.778	.617	.756	.819	.794	.907	.852	.374	.529	.466	.557
20	.737	.801	.782	.710	.795	.910	.881	.371	.623	.400	.576
21					.198	.785			.569		.387
22					.860	.743			.576		.414

**TABLE 16**  
**SPECTRAL REFLECTANCE OF HK31A MAGNESIUM ANODIZED IN HAE**

Specimen No.	210	211	211	211	211	215
Pretreatment	MP	MP	MP	MP	MP	MP, EP
Anodizing Coating	Dark	Light	Light	Light	Light	Light
Post Treatment	None	None	None	None	HT 800°F	None
Spec. Temp.	RT	RT	350°F	RT <sub>f</sub>	RT	RT
Wavelength Microns						
.4	.087	.287	.321	.306	.178	.188
.45	.080	.308	.325	.327	.176	.204
.5	.083	.345	.355	.357	.209	.35
.6	.108	.438	.413	.448	.258	.338
.7	.134	.500	.497	.500	.294	.407
.8	.134	.454	.533	.571	.381	.418
.9	.235	.558	.553	.605	.382	.501
1.0	.236	.588	.592	.585	.400	.495
1.2	.269	.573	.585	.572	.467	.509
1.4	.307	.565	.596	.572	.423	.491
1.6	.315	.550	.595	.560	.429	.468
1.8	.340	.524	.581	.535	.438	.439
2.0	.361	.550	.592	.547	.459	.443
2.5	.403	.673	.675	.617	.572	.539
3.0	.341	.760	.779	.786	.694	.588
3.5	.418	.826	.832	.838	.767	.685
4	.511	.861	.860	.872	.804	.745
5	.593	.902	.893	.904	.866	.798
6	.576	.906	.905	.909	.886	.787
7	.344	.888	.880	.893	.886	.717
8	.361	.867	.912	.921	.903	.687
9	.044	.749	.750	.752	.749	.382
10	.046	.939	.774	.785	.795	.375
11	.078	.683	.857	.865	.857	.503
12	.038	.850	.836	.860	.846	.453
13	.039	.790	.793	.784	.786	.252
14	.203	.807	.799	.817	.831	.269
15	.302	.781	.757	.794	.803	.226
16	.281	.789	.766	.791	.816	.282
17	.347	.802	.794	.808	.790	.295
18	.420	.829	.816	.811	.805	.294
19	.450	.818	.821	.799	.819	.259
20	.369	.810	.815	.804	.847	.37
21	.436	.776	.764	.779	.806	
22	531				925	

TABLE 17  
SPECTRAL REFLECTANCE OF AZ31B MAGNESIUM ANODIZED IN DOW 17

Specimen No.	31	31	31	35	35	35	35	152	152	158
Pretreatment	MP, EP	MP, EP	MP, EP	MP, EP	MP, EP	MP, EP	MP, EP	MP	MF	MP
Anodizing Coating	Light	Light	Light	Dark	Dark	Dark	Dark	Light	Light	Dark
Post Treatment	None	None	None	HT-800°F	None	None	None	None	None	None
Spec. Temp.	RT	350°F	RT <sub>f</sub>	RT	RT	350°F	RT <sub>f</sub>	RT	RT	RT
Wavelength										
Microns										
.4	.463	.552	.544	.168	.240	.197	.196	.547	.584	.237
.45	.527	.579	.565	.166	.196	.188	.197	.563	.585	.197
.5	.530	.581	.580	.226	.254	.246	.271	.580	.599	.271
.6	.551	.588	.588	.243	.253	.264	.251	.595	.618	.249
.7	.543	.590	.593	.245	.264	.238	.259	.596	.587	.253
.8	.576	.570	.577	.316	.389	.384	.423	.589	.599	.425
.9	.587	.695	.615	.464	.268	.518	.506	.744	.622	.262
1.0	.575	.629	.609	.518	.244	.532	.542	.673	.631	.285
1.2	.613	.635	.613	.503	.592	.569	.613	.661	.659	.572
1.4	.629	.647	.630	.530	.600	.582	.596	.654	.662	.598
1.6	.628	.669	.636	.526	.595	.610	.591	.659	.662	.591
1.8	.617	.662	.616	.524	.585	.606	.586	.659	.659	.579
2.0	.626	.674	.622	.536	.586	.624	.596	.675	.664	.584
2.5	.701	.739	.718	.547	.570	.635	.627	.749	.739	.578
3.0	.766	.807	.801	.456	.220	.619	.529	.818	.802	.248
3.5	.807	.845	.847	.477	.167	.605	.548	.862	.832	.170
4	.831	.856	.879	.536	.254	.596	.615	.881	.852	.254
5	.875	.888	.898	.418	.331	.399	.468	.912	.895	.344
6	.897	.905	.887	.435	.213	404	.412	.919	.902	.242
7	.921	.910	.921	.361	.169	193	.330	.934	.912	.129
8	.928	.921	.925	.014	.038	252	.023	.937	.923	.015
9	.928	.924	.929		.038		.053	.937	.916	.035
10	.940	.938	.931	.052	.036		.050	.947	.931	.055
11	.946	.932	.940	.089	.078	030	.077	.949	.941	.046
12	.945	.944	.949	.084	.034		.077	.956	.941	
13	.940	.954	.949	.083			.035	.956	.948	.046
14	.950	.946	.941	.038	.023			.955	.955	
15	.948	.941	.932				.097	.935	.958	.044
16	.945	.943	.915					.950	.929	
17	.938	.952	.916				.039	.927	.934	.037
18	.942	.935	.928				.122	.935	.927	
19	.945	.940	.921				.179	.952	.935	.074
20	.950	.929	.898				.161	.969	.941	.209
21	.975	.928	.925					.981		
22				888				.978		

TABLE 18  
SPECTRAL REFLECTANCE OF HK31A MAGNESIUM ANODIZED IN DOW 17

Specimen No.	212	213	214	214	214	214
Pretreatment	MP	MP	MP, EP	MP, EP	MP, EP	MP, EP
Anodizing Coating	Light	Dark	Light	Light	Light	Light
Post Treatment	None	None	None	None	None	HT-800°F
Spec. Temp	RT	RT	RT	350°F	RT	RT
Wavelength, Microns						
.4	.601	.183	.564	.616	.683	.274
.45	.607	.184	.634	.663	.880	.310
.5	.639	.241	.657	.671	.674	.350
.6	.665	.227	.673	.677	.713	.382
.7	.673	.219	.668	.703	.688	.403
.8	.600	.356	.585	.613	.630	.444
.9	.644	.512	.724	.673	.646	.285
1.0	.697	.532	.646	.665	.659	.446
1.2	.658	.568	.632	.645	.628	.470
1.4	.638	.572	.618	.632	.619	.476
1.6	.616	.565	.603	.610	.596	.481
1.8	.575	.539	.557	.608	.566	.475
2.0	.577	.546	.565	.613	.569	.484
2.5	.679	.563	.677	.680	.676	.616
3.0	.723	.285	.758	.774	.785	.738
3.5	.791	.222	.811	.817	.824	.792
4	.841	.317	.829	.834	.846	.820
5	.888	.350	.851	.849	.860	.848
6	.854	.259	.855	.852	.859	.858
7	.874	.227	.865	.858	.867	.864
8	.898	.035	.867	.865	.875	.873
9	.736	.050	.843	.840	.851	.852
10	.787	.024	.847	.850	.858	.863
11	.869	.037	.869	.868	.873	.887
12	.899	.084	.863	.864	.868	.882
13	.912	.053	.856	.862	.869	.867
14	.928		.839	.861	.871	.884
15	.928	.069	.854	.858	.865	.893
16	.901		.810	.821	.819	.846
17	.886	.037	.764	.779	.780	.786
18	.868		.789	.808	.808	.827
19	.875	.036	.787	.804	.842	.840
20	.862		.800	.839	.796	.847
21	.855		.791	.832	.844	.895
22	.887			.827		

**TABLE 19**  
**SPECTRAL REFLECTANCE OF A110 AT TITANIUM**  
**ANODIZED IN SULFURIC ACID**

Specimen No.	38	43
Pretreatment	MP, EP	MP, Pickled
Anodizing Time	Std	Std
Post Treatment	Sealed	Sealed
Spec. Temp.	RT	RT
Wavelength Microns		
.4	.406	.210
.45	.324	.155
.5	.207	.085
.6	.059	.055
.7	.100	.124
.8	.211	.173
.9	.328	.268
1.0	.363	.306
1.2	.469	.460
1.4	.527	.487
1.6	.574	.520
1.8	.605	.554
2.0	.633	.575
2.5	.684	.628
3.0	.736	.673
3.5	.782	.710
4	.805	.726
5	.858	.786
6	.874	.811
7	.890	.839
8	.905	.856
9	.915	.868
10	.915	.873
11	.926	.895
12	.932	.902
13	.931	.907
14	.958	.928
15	.946	.931
16	.964	.916
17	.952	.913
18	.993	.918
19	.971	.918
20	.986	.920
21	.985	.910
22	.99*	.998

(3)

TABLE 20  
SPECTRAL REFLECTANCE OF B120 VAC TITANIUM  
ANODIZED IN SULFURIC ACID

Specimen No.	203	207
Pretreatment	MP, EP	MP Pickled
Anodizing Time	Std	Std
Post Treatment	Sealed	Sealed
Spec. Temp	RT	RT
Wavelength Microns		
.4	.279	.245
.45	.237	.198
.5	.158	.132
.6	.038	.043
.7	.076	.063
.8	.206	.185
.9	.347	.131
1.0	.373	.308
1.2	.519	.480
1.4	.594	.528
1.6	.642	.566
1.8	.681	.602
2.0	.708	.626
2.5	.747	.674
3.0	.787	.714
3.5	.802	.736
4	.817	.706
5	.844	.777
6	.861	.804
7	.871	.821
8	.880	.839
9	.991	.858
10	1.068	.857
11	.910	.877
12	.916	.891
13	.928	.895
14	.942	.889
15	.941	.913
16	.947	.938
17	.946	.944
18	.960	.932
19	.970	.935
20	.947	.931
21	.941	.932
22	.964	

**TABLE 21**  
**SPECTRAL REFLECTANCE OF A110AT TITANIUM**  
**ANODIZED IN SODIUM HYDROXIDE**

Specimen No.	45	45	45	45	45	48
Pretreatment	MP Pickled	MP, EP				
Anodizing Time	Std	Std	Std	Std	Std	Std
Post Treatment	Sealed	Sealed	Sealed	Sealed	Sealed	Sealed
Spec. Temp.	RT	300°F	600°F	900°F	RT <sub>f</sub>	RT
<b>Wavelength</b>						
Microns						
.4	.283	.223	.187	.056	.048	.288
.45	.211	.208	.162	.058	.060	.222
.5	.190	.224	.199	.053	.060	.263
.6	.254	.262	.174	.038	.057	.279
.7	.230	.216	.127	.052	.064	.211
.8	.288	.245	.124	.109	.051	.265
.9	.275	.298	.207	.100	.079	.444
1.0	.351	.367	.227	.125	.075	.480
1.2	.387	.335	.232	.168	.171	.439
1.4	.351	.318	.179	.172	.179	.360
1.6	.315	.283	.148	.171	.194	.323
1.8	.281	.277	.155	.227	.210	.300
2.0	.269	.285	.149	.216	.224	.305
2.5	.308	.344	.245	.251	.206	.391
3.0	.425	.469	.390	.375	.340	.538
3.5	.496	.549	.382	.482	.399	.642
4	.616	.595	.551	.438	.423	.687
5	.698	.703	.667	.572	.554	.806
6	.752	.743	.726	.643	.627	.837
7	.803	.804	.751	.619	.680	.870
8	.838	.831	.777	.720	.727	.886
9	.858	.849	.811	.719	.765	.900
10	.878	.861	.831	.777	.785	.908
11	.897	.859	.842	.846	.808	.913
12	.893	.862	.849	.838	.822	.910
13	.904	.866	.871		.830	.921
14	.887	.885	.873	.783	.842	.931
15	.865	.887	.845	.813	.864	.940
16	.902	.890	.857	.789	.853	.950
17	.943	.891	.905	.786	.869	.956
18	.914	.895	.877	.887	.866	.961
19	.892	.900	.866	.839	.862	.962
20	.901	.898	.881		.871	.975
21	.889	.888	.876		.855	.959
22					.834	.953

TABLE 21 (Continued)

Specimen No.	48	48	48	48	48	178
Pretreatment	MP, EP	MP, EP	MP, EP	MP, EP	MP, EP	MP, EP
Anodizing Time	Std	Std	Std	Std	Std	Std/3
Post Treatment	Sealed	Sealed	Sealed	Sealed	Sealed	Sealed
Spec. Temp.	500°F	1000°F	RT <sub>f</sub> (1000 F)	1300°F	RT <sub>f</sub> (1300 F)	RT
Wavelength Mic. s						
.4	.206	.077	.069	152	182	.262
.45	.219	.083	.110	.169	171	.209
.5	.279	.092	.121	.179	178	.250
.6	.210	.122	.142	.220	.219	.297
.7	172	.162	.183	250	.260	.222
.8	.265	.211	.293	.260	.375	.271
.9	.239	.264	.324	.356	.394	.403
1.0	.246	.305	.343	.364	.417	.449
1.2	.294	.351	.433	.401	500	.424
1.4	.247	.381	.481	440	.535	.367
1.6	.208	.419	.502	446	.556	.322
1.8	.231	.449	.542	525	.585	.289
2.0	.266	.508	.568	.464	.609	.290
2.5	.400	.527	.620	.452	.657	.369
3.0	.554	.822	.662	.588	692	.505
3.5	.639	.692	.707	.618	.721	.595
4	675	.644	.724	.641	730	.647
5	761	733	.776	663	.771	751
6	.806	737	807	704	806	787
7	.831	.733	834	710	826	823
8	.852	783	853	736	850	847
9	870	818	874	746	862	856
10	881	.808	885	783	873	860
11	913	882	895	781	891	863
12	910	858	899	808	891	848
13	921	853	903	813	911	853
14	.931	880	922	838	931	854
15	940	901	937	837	932	867
16	950	892	937	842	932	853
17	956	899	951	873	953	854
18	961	927	962	880	952	876
19	962	955	964	818	966	876
20	975	921	955	836	982	894
21	959	896	957	927	94	915
22	953	834	911	838	1.016	847

TABLE 22  
SPECTRAL REFLECTANCE OF A110AT TITANIUM ANODIZED  
BY A PROPRIETARY PROCESS

Specimen No.	37	41	47	47	47	47	47
Pretreatment	MP, EP	MP Pickled	MP, EP				
Anodizing Time	Thin Coat	Std	Std	Std	Std	Std	Std
Post Treatment	Sealed	Sealed	Sealed	Sealed	Sealed	Sealed	Sealed
Spec. Temp.	RT	RT	RT	500°F	1000°F	1300°F	RT <sub>f</sub>
Wavelength Microns							
.4	.145	.144	.158	.156	.067	.124	.117
.45	.156	.157	.157	.162	.084	.108	.118
.5	.160	.164	.158	.161	.088	.109	.123
.6	.165	.174	.165	.167	.078	.116	.135
.7	.171	.183	.171	.175	.081	.128	.168
.8	.208	.248	.211	.206	.112	.269	.224
.9	.206	.224	.235	.212	.095	.240	.262
1.0	.207	.243	.225	.233	.069	.290	.301
1.2	.233	.288	.251	.174	.097	.316	.326
1.4	.261	.306	.290	.284	.090	.338	.334
1.6	.282	.328	.315	.309	.114	.346	.350
1.8	.302	.356	.342	.335	.120	.335	.362
2.0	.323	.374	.354	.358	.163	.321	.374
2.5	.387	.464	.417	.407	.185	.267	.390
3.0	.490	.583	.539	.529	.239	.262	.407
3.5	.544	.636	.630	.625	.199	.279	.423
4	.557	.649	.658	.650	.243	.260	.441
5	.492	.618	.638	.595	.188	.225	.474
6	.559	.671	.651	.614	.243	.589	.502
7	.668	.787	.757	.720	.381	.592	.528
8	.679	.792	.738	.744	.511	.661	.536
9	.526	.654	.587	.581	.544	.669	.571
10	.347	.490	.404	.421	.448	.657	.595
11	.358	.492	.49	.328	.404	.706	.608
12	.486	.640	.554	.542		.720	.627
13	.595	.751	.659	.633	.576		.624
14	.658	.810	.731	.742		.837	.709
15	.696	.858	.799	.759	.772		.709
16	.705	.866	.823	.808			.750
17	.715	.876	.842	.806	.789		.777
18	.745	.909	.874	.847			.772
19	.784	.936	.883	.851	.858		.785
20	.811	.932	.898	.806			.798
21	.799	.927	.880	.909	.869		.837
22	.841	.915	.877	.914			

TABLE 23  
 SPECTRAL REFLECTANCE OF B120 VAC TITANIUM  
 ANODIZED BY A PROPRIETARY PROCESS

Specimen No.	197	199	201	201	201	201	201
Pretreatment	MP, EP	MP, Pickled	MP, EP	MP, EP	MP, EP	MP, EP	MP, EP
Anodizing Time	Thin Coat	Std	Std	Std	Std	Std	Std
Post Treatment	Sealed	Sealed	Sealed	Sealed	Sealed	HT 1500°F	None
Spec. Temp.	RT	RT	RT	500°F	1000°F	RT	RT
<b>Wavelength</b> <b>Microns</b>							
.4	.133	.113	.123	.106	.105	.127	.084
.45	.141	.144	.130	.136	.105	.121	.090
.5	.152	.149	.146	.147	.097	.120	.090
.6	.168	.165	.161	.160	.096	.121	.088
.7	.174	.167	.172	.171	.102	.117	.089
.8	.216	.210	.219	.123	.108	.144	.096
.9	.219	.191	.223	.173	.114	.111	.077
1.0	.242	.229	.327	.176	.113	.134	.087
1.2	.273	.287	.278	.262	.093	.135	.091
1.4	.294	.325	.231	.210	.060	.141	.092
1.6	.316	.344	.326	.296	.087	.139	.096
1.8	.344	.380	.355	.322	.070	.144	.104
2.0	.367	.400	.363	.342	.028	.147	.114
2.5	.409	.452	.418	.379	.053	.156	.127
3.0	.474	.490	.484	.464	.062	.170	.113
3.5	.476	.599	.495	.453	.209	.186	.145
4	.477	.661	.498	.449	.100	.192	.189
5	.631	.716	.571	.622	.274	.184	.308
6	.699	.687	.724	.672	.199	.270	.278
7	.668	.740	.705	.623	.183	.383	.228
8	.432	.695	.485	.416	.252	.286	.354
9	.259	.471	.296	.284	.374	.062	.269
10	.189	.275	.228	.136	.262	.128	.185
11	.172	.238	.196	.269	.418	.161	.159
12	.087	.311	.110	.136		.127	.062
13	.067	473	.119	.098	420	.299	.167
14	.142	629	.174	.273		.398	.312
15	.286	.698	.306	.355	.467	408	.407
16	.317	721	.345	.420		.428	.424
17	.307	.748	.390	.393	494	533	.459
18	.370	.759	.498	.480		580	.505
19	.448	840	.563	.490	572	658	.601
20	.546	852	.608	.569		554	.621
21	.540	845	.619	.562	716	466	.665
22		865	.596	.729			

TABLE 24

SPECTRAL REFLECTANCE OF BERYLLIUM, 1% COPPER,  
ANODIZED IN SULFURIC ACID

Specimen No.	162
Pretreatment	MP
Anodizing Time	Std
Post Treatment	Sealed
Spec. Temp.,	RT
Wavelength	
Microns	
.4	.250
.45	.292
.5	.282
.6	.292
.7	.284
.8	.302
.9	.357
1.0	.350
1.2	.422
1.4	.507
1.6	.563
1.8	.614
2.0	.658
2.5	.732
3.0	.781
3.5	.809
4	.825
5	.843
6	.847
7	.848
8	.858
9	.845
10	.817
11	.797
12	.800
13	.802
14	.806
15	.817
16	.812
17	.823
18	.835
19	.808
20	.809
21	.800
22	.786

TABLE 25

SPECTRAL REFLECTANCE OF BERYLLIUM, 1% COPPER,  
ANODIZED IN SODIUM HYDROXIDE

Specimen No.	164	166	166	166	166	166
Pretreatment	MP, EP	MP	MP	MP	MP	MP
Anodizing Time	Std	Std	Std	Std	Std	Std
Post Treatment	Sealed	Sealed	Sealed	Sealed	Sealed	Sealed
Spec. Temp.	RT	RT	400°F	800°F	1200°F	RT <sub>f</sub>
Wavelength Microns						
.4	.081	.051	.054	.050	.093	.074
.45	.071	.051	.047	.030	.077	.084
.5	.068	.050	.047	.041	.080	.083
.6	.066	.044	.040	.037	.078	.100
.7	.062	.049	.047	.050	.105	.121
.8	.059	.080	.051	.117	.128	.197
.9	.095	.057	.067	.084	.169	.123
1.0	.062	.083	.098	.116	.186	.144
1.2	.085	.104	.100	.125	.198	.230
1.4	.095	.094	.134	.215	.241	.304
1.6	.101	.108	.201	.296	.378	.406
1.8	.102	.183	.291	.396	.501	.542
2.0	.099	.249	.345	.458	.543	.609
2.5	.100	.388	.416	.527	.576	.678
3.0	.123	.372	.424	.573	.628	.703
3.5	.180	.394	.473	.609	.658	.707
4	.215	.410	.495	.639	.668	.731
5	.415	.528	.661	.770	.757	.822
6	.562	.662	.750	.815	.817	.857
7	.636	.782	.805	.842	.878	.875
8	.682	.850	.847	.866	.849	.901
9	.680	.842	.829	.831	.804	.870
10	.660	.818	.803	.814	.768	.851
11	.675	.826	.803	.815	.803	.849
12	.672	.828	.820	.814	.858	.867
13	.681	.807	.807	.834	.871	.871
14	.691	.734	.765	.826	.835	.783
15	.670	.686	.749	.799	.675	.762
16	.713	.825	.108	.837	.817	.891
17	.752	.905	.889	.954	.867	.931
18	.759	.938	.935	.906	.898	.948
19	.773	.945	.943	.934	.901	.960
20	.783	.976	.931	.774	.946	1.029
21	.734	.952	.730	.982	.857	.953
22	.797				.886	.953

TABLE 26

**SPECTRAL REFLECTANCE OF QMV BERYLLIUM ANODIZED  
IN SODIUM HYDROXIDE**

Specimen No.	173	173	173	173	173	177	177
Pretreatment	MP	MP	MP	MP	MP	MP, EP	MP, EP
Anodizing Time	Std	Std	Std	Std	Std	Std	Std
Post Treatment	Sealed	Sealed	Sealed	Sealed	Sealed	Sealed	Std HT-1500°F
Spec. Temp.	RT	400°F	800°F	1200°F	RT <sub>f</sub>	RT	RT
Wavelength Microns							
.4	.208	.259	.217	.144	.201	.075	.271
.45	.217	.229	.237	.160	.192	.057	.285
.5	.220	.235	.240	.154	.119	.054	.294
.6	.219	.229	.230	.116	.068	.052	.295
.7	.223	.240	.231	.124	.167	.050	.300
.8	.271	.265	.225	.116	.241	.058	.365
.9	.225	.261	.160	.101	.284	.037	.340
1.0	.262	.258	.273	.102	.287	.068	.342
1.2	.360	.296	.312	.100	.277	.092	.415
1.4	.383	.362	.255	.114	.252	.059	.488
1.6	.402	.405	.389	.159	.254	.071	.547
1.8	.444	.328	.437	.249	.282	.139	.603
2.0	.465	.462	.461	.260	.314	.203	.640
2.5	.517	.525	.539	.385	.427	.312	.720
3.0	.566	.577	.603	.461	.584	.309	.762
3.5	.602	.630	.634	.593	.671	.298	.796
4	.637	.645	.654	.633	.729	.286	.827
5	.572	.6	.729	.691	.800	.351	.846
6	.708	.741	.787	.739	.829	.514	.827
7	.772	.787	.812	.789	.855	.667	.784
8	.831	.833	.839	.805	.882	.772	.850
9	.870	.854	.857	.832	.874	.771	.789
10	.890	.864	.862	.831	.870	.733	.693
11	.884	.885	.869	.810	.903	.742	.696
12	.899	.894	.886	.909	.909	.725	.647
13	.904	.904	.889	.882	.895	.729	.654
14	.903	.883	.901	.960	.903	.708	.638
15	.881	.881	.898	.831	.810	.606	.242
16	.887	.923	.903	.855	.859	.735	.568
17	.943	.951	.945	.881	.905	.726	.793
18	.959	.950	.953	.996	.953	.865	.859
19	.966	.971	.945	.897	.967	.865	.890
20	.976	.975	.947	.968	.949	.897	.704
21	.959	.968	.951	1.034	.956	.919	.92
22				.977		.963	.911

TABLE 27  
SPECTRAL REFLECTANCE OF BERYLLIUM, 1% COPPER,  
ANODIZED IN NITRIC ACID

Specimen No.	167
Pretreatment	MP
Anodizing Time	Std
Post Treatment	Sealed
S.s.c. Temp.	RT
Wavelength	
Microns	
.4	.018
.45	.009
.5	.013
.6	.013
.7	.015
.8	.016
.9	.011
1.0	.035
1.2	.032
1.4	.043
1.6	.055
1.8	.068
2.0	.088
2.5	.151
3.0	.161
3.5	.242
4	.305
5	.441
6	.415
7	.426
8	.379
9	.137
10	.055
11	.095
12	.142
13	.180
14	.242
15	.099
16	.234
17	.254
18	.518
19	.459
20	.^50
21	.506
22	.535

TABLE 28  
SPECTRAL REFLECTANCE OF QMV BERYLLIUM  
ANODIZED IN NITRIC ACID

Specimen No.	174	176
Pretreatment	"	MP
Anodizing Time	Std	Std/3
Post Treatment	Sealed	Sealed
Spec. Temp.	RT	RT
Wavelength Microns		
.4	.016	.013
.45	.017	.016
.5	.011	.012
.6	.015	.008
.7	.015	.022
.8	.029	.028
.9	.015	.026
1.0	.029	.027
1.2	.036	.053
1.4	.030	.080
1.6	.039	.116
1.8	.052	.142
2.0	.065	.169
2.5	.107	.243
3.0	.104	.231
3.5	.175	.355
4	.225	.423
5	.342	.534
6	.347	.554
7	.317	.452
8	.345	.496
9	.105	.113
10	.121	.033
11	.069	.044
12	.065	.053
13	.175	.070
14	.059	.120
15	.149	.171
16	.127	.269
17	.136	.314
18	.274	.341
19	.303	.436
20	.334	.390
21	.309	.444
22		.534

TABLE 29  
SPECTRAL REFLECTANCE OF BERYLLIUM, 1% COPPER,  
ANODIZED IN CHROMIC ACID

Specimen No.	168	170
Pretreatment	MP	MP
Anodizing Time	Std	Std/3
Post Treatment	Sealed	Sealed
Spec. Temp.	RT	RT
Wavelength Microns		
.4	.169	.341
.45	.164	.354
.5	.166	.367
.6	.162	.367
.7	.162	.377
.8	.133	.462
.9	.160	.438
1.0	.161	.451
1.2	.192	.528
1.4	.218	.580
1.6	.228	.650
1.8	.249	.704
2.0	.269	.679
2.5	.293	.764
3.0	.307	.787
3.5	.335	.789
4	.349	.791
5	.350	.768
6	.361	.756
7	.365	.778
8	.333	.810
9	.317	.839
10	.459	.855
11	.531	.866
12	.613	.870
13	.677	.874
14	.701	.866
15	.678	.856
16	.538	.864
17	.556	.896
18	.462	.909
19	.653	.877
20	.334	.886
21	.504	.889
22		.911

TABLE 30  
 SPECTRAL REFLECTANCE OF QMV BERYLLIUM  
 ANODIZED IN CHROMIC ACID

Specimen No.	172
Pretreatment	MP
Anodizing Time	Std
Post Treatment	Sealed
Spec. Temp.	RT
Wavelength	
Microns	
.4	.172
.45	.184
.5	.187
.6	.179
.7	.173
.8	.168
.9	.210
1.0	.165
1.2	.202
1.4	.232
1.6	.249
1.8	.264
2.0	.280
2.5	.308
3.0	.303
3.5	.318
4	.355
5	.354
6	.335
7	.296
8	.317
9	.461
10	.560
11	.618
12	666
13	712
14	.729
15	673
16	494
17	416
18	.59
19	385
20	494
21	594
22	

TABLE 31

SPECTRAL REFLECTANCE OF UNANODIZED METALS

Specimen No.	18	156	171	46	209	202
Material	Aluminum	Magnesium	Beryllium	Titanium	Titanium	Titanium
Alloy	1075	AZ 31B	1% Cu.	A110AT	A110AT	B120 VAC
Pretreatment	MP, EP	MP	MP	MP, EP	MP	MP, EP
Spec.Temp.	RT	RT	RT	RT	RT	RT
Wavelength Microns						
.4	.708	.586	.356	.468	.365	.522
.45	.740	.592	.372	.534	.417	.512
.5	.737	.593	.377	.579	.443	.517
.6	.731	.623	.371	.569	.458	.524
.7	.710	.638	.374	.573	.475	.515
.8	.711	.647	.396	.589	.479	.543
.9	.740	.709	.386	.625	.528	.552
1.0	.783	.696	.425	.629	.551	.591
1.2	.852	.736	.521	.679	.571	.616
1.4	.863	.751	.613	.670	.588	.662
1.6	.853	.762	.646	.689	.608	.686
1.8	.870	.761	.673	.716	.626	.703
2.0	.861	.767	.703	.732	.640	.726
2.5	.883	.814	.728	.760	.678	.750
3.0	.865	.873	.758	.796	.716	.777
3.5	.879	.912	.781	.831	.738	.815
4.0	.902	.922	.777	.840	.754	.811
5	.917	.943	.764	.883	.798	.831
6	.982	.953	.779	.905	.818	.863
7	.912	.956	.816	.918	.839	.873
8	.803	.958	.818	.934	.855	.886
9	.703	.960	.834	.949	.870	.890
10	.831	.956	.843	.950	.879	.903
11	.807	.975	.817	.964	.902	.907
12	.769	.975	.853	.968	.910	.916
13	.770	.979	.855	.973	.908	.927
14	.790	.984	.845	.990	.904	.946
15	.808	.980	.854	.989	.933	.969
16	.839	.979	.847	1.002	.906	.968
17	.844	.978	.867	1.003	.939	.959
18	.863	.987	.859	1.011	.934	.974
19	.874	.993	.858	1.014	.934	.977
20	.876	.981	.856	1.019	.954	.976
21			884	1.021		.980
22			861	1.034		.981

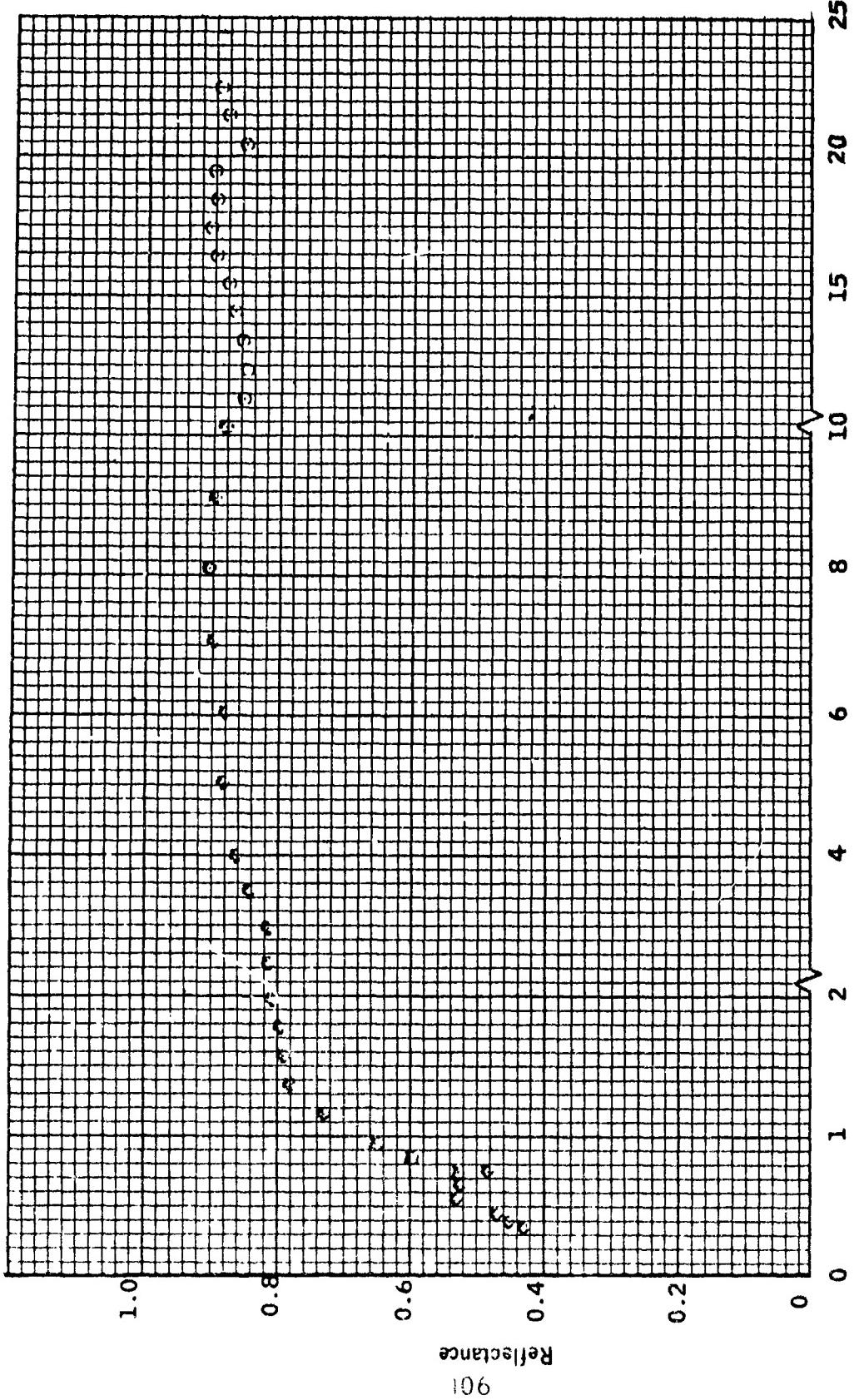


Fig. 23 Normal Spectral Reflectance of Specimen No 50 Temperature RTT



Fig. 24 Normal Spectral Reflectance of Specimen No. 50 Temperature 600 F

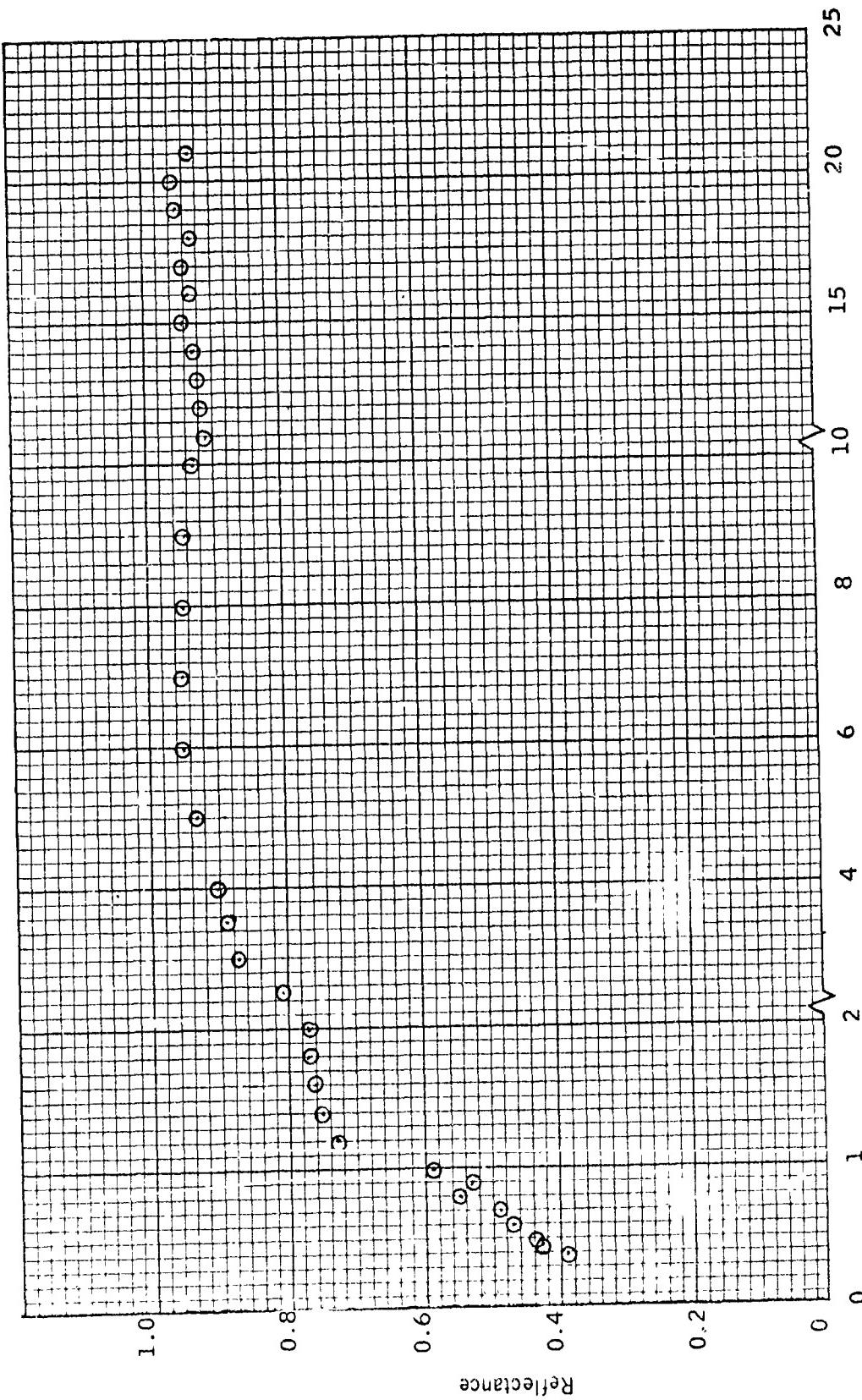


Fig. 25 Normal Spectral Reflectance of Specimen No 50 Temperature  $RT_f$

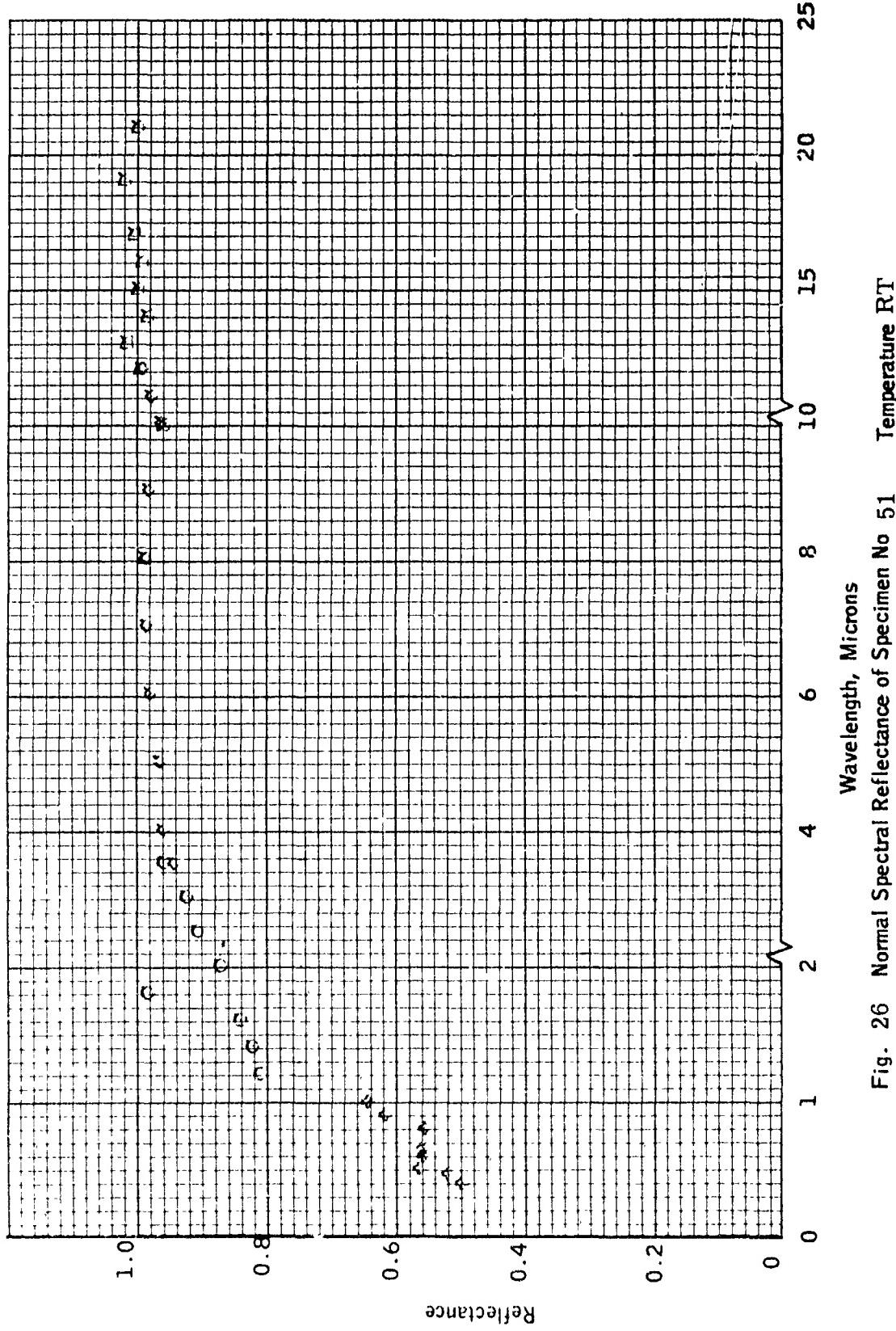


Fig. 26 Normal Spectral Reflectance of Specimen No. 51 Temperature RT

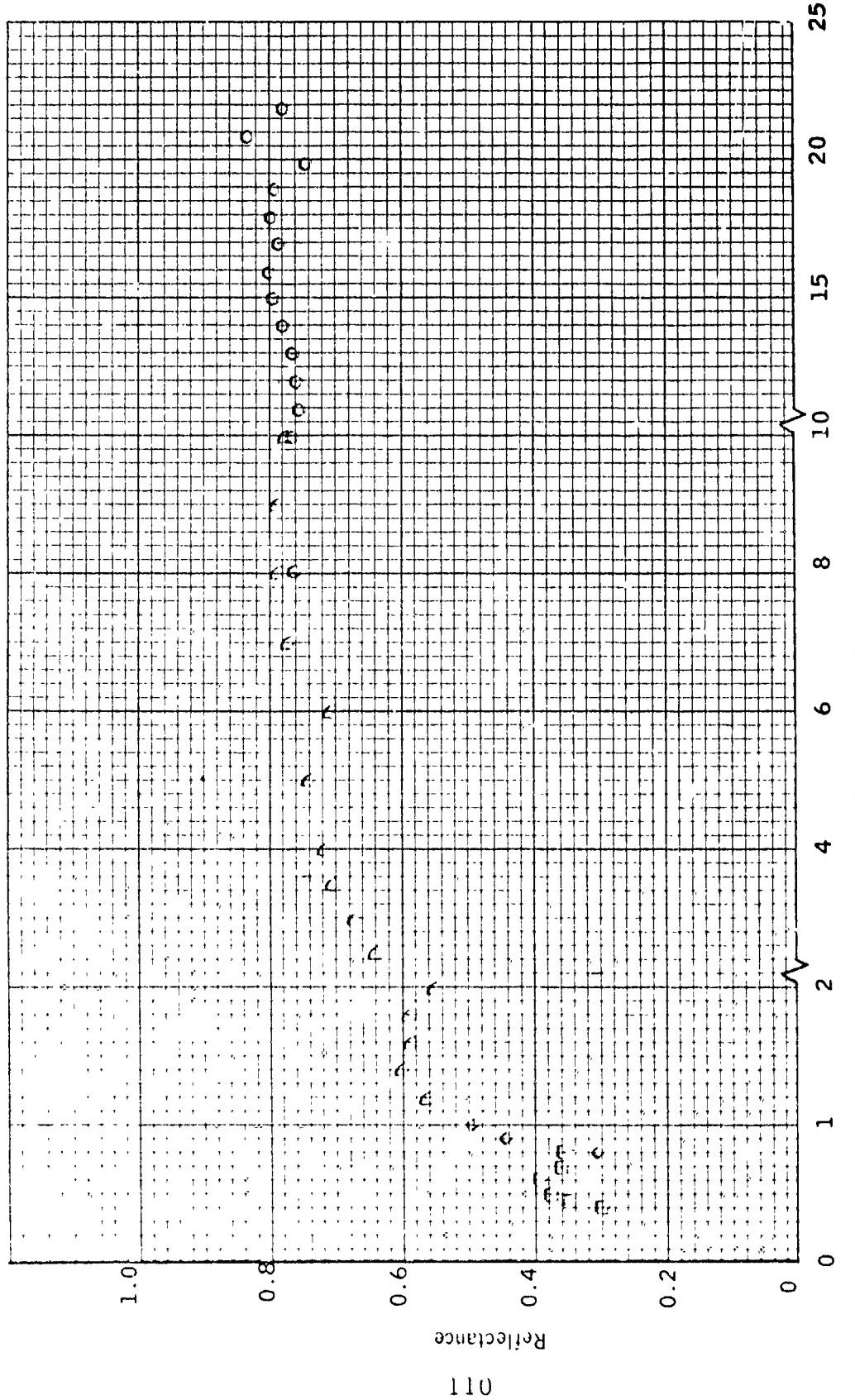


Fig. 27 Normal Spectral Reflectance of Specimen No. 75 Temperature RT

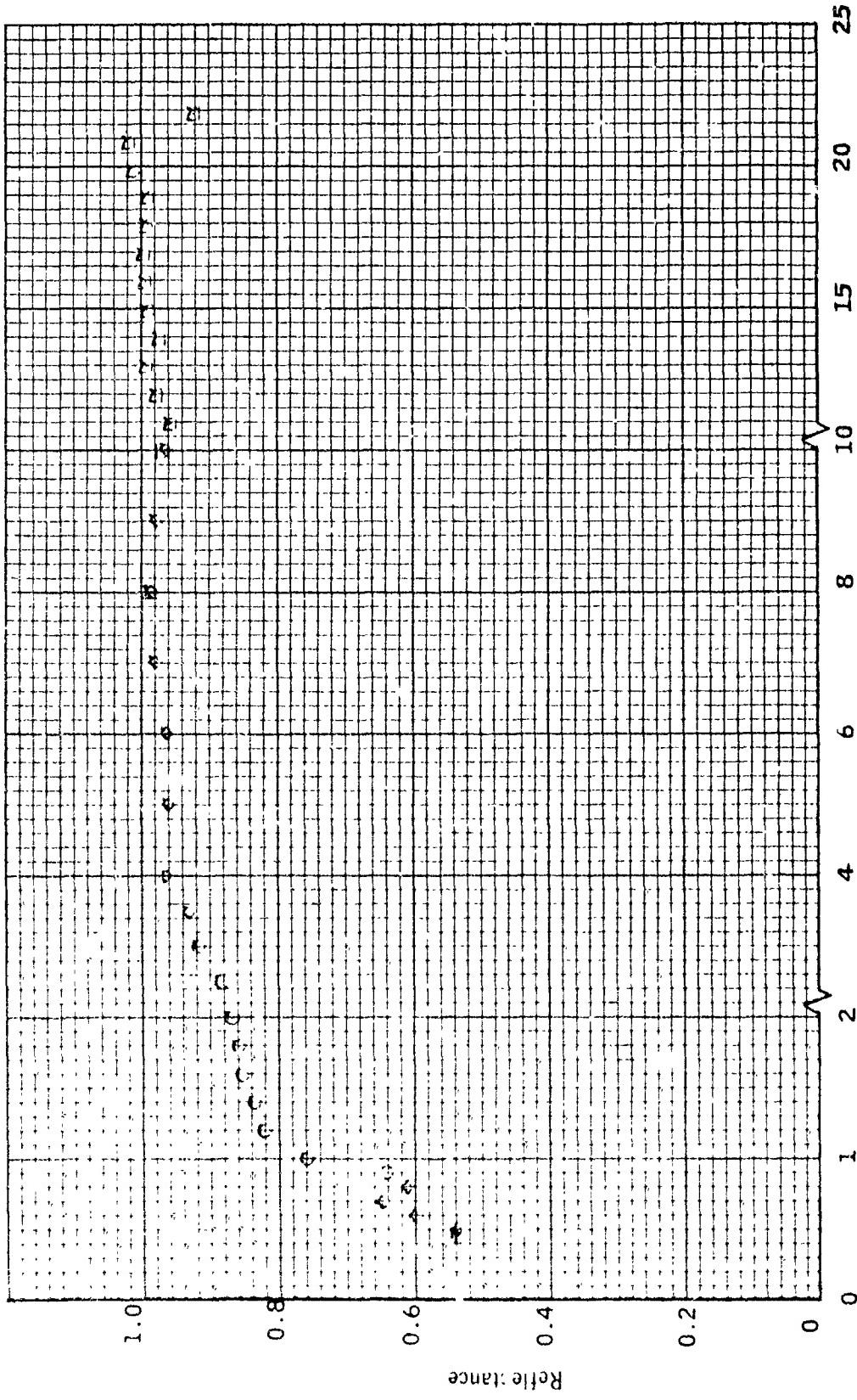


Fig. 28 Normal Spectral Reflectance of Specimen No. 110 Temperature RT

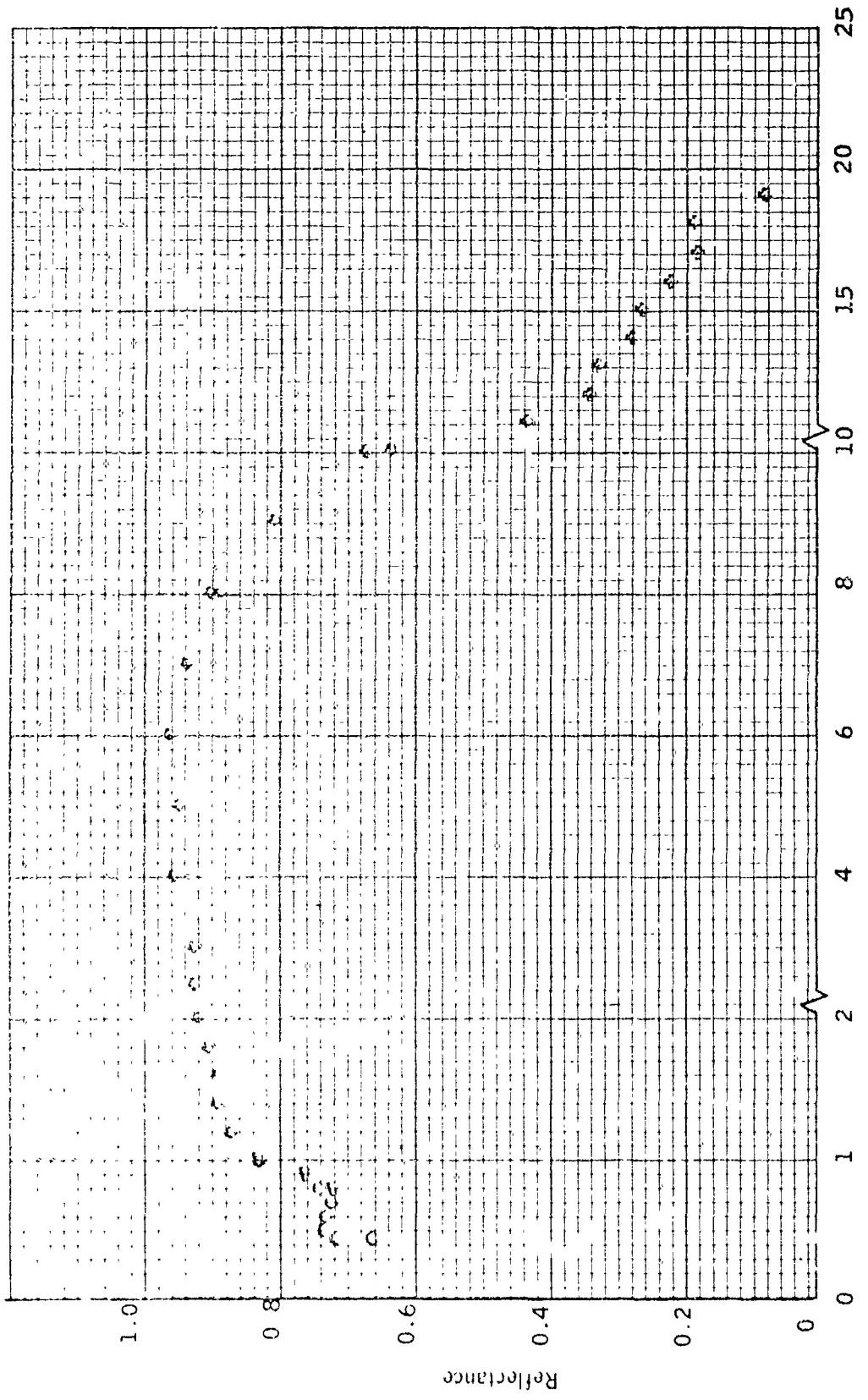


Fig. 29 Normal Spectral Reflectance of Specimen No 55 Temperature RT

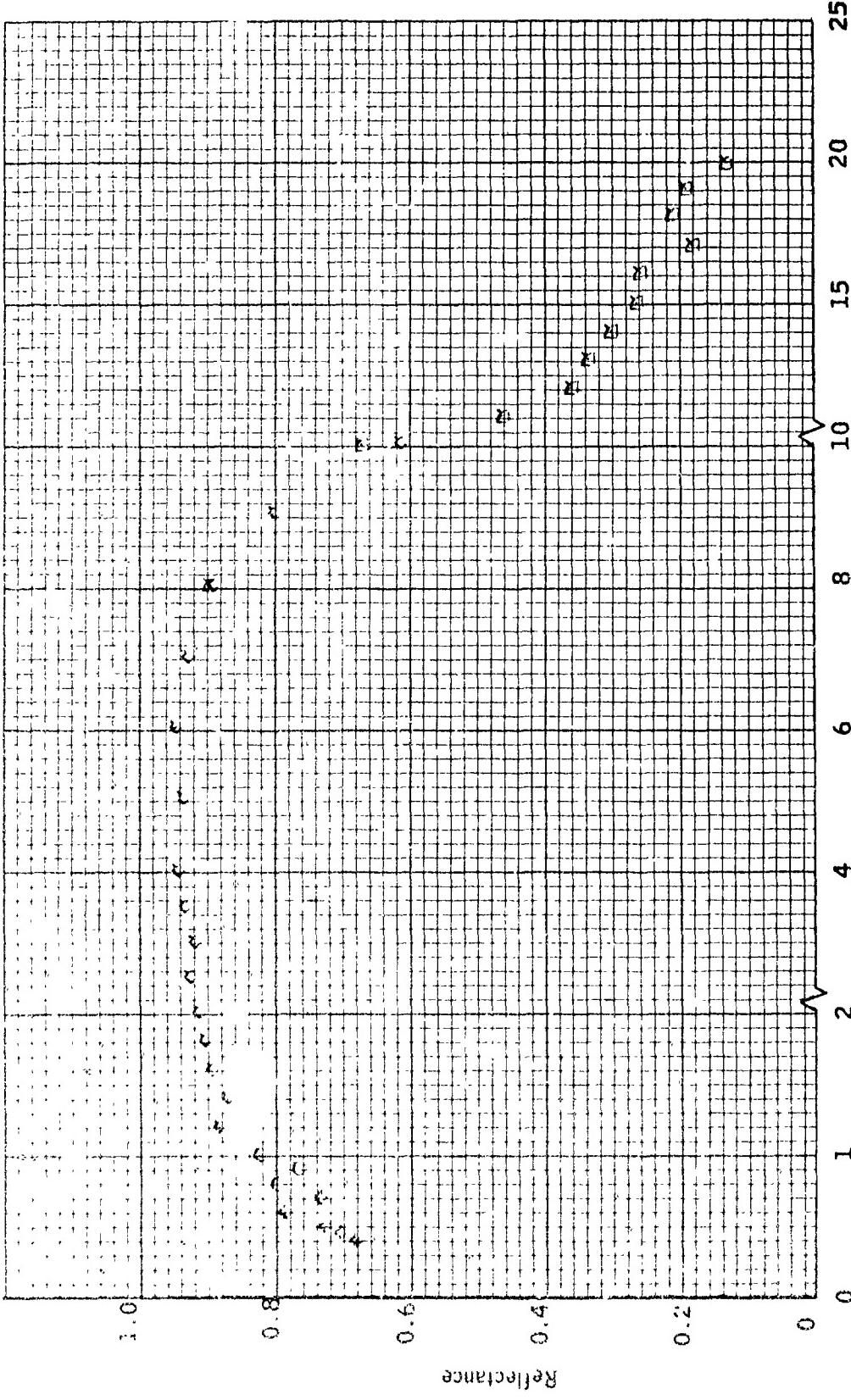


Fig. 30 Normal Spectral Reflectance of Specimen No. 55 Temperature 300°F

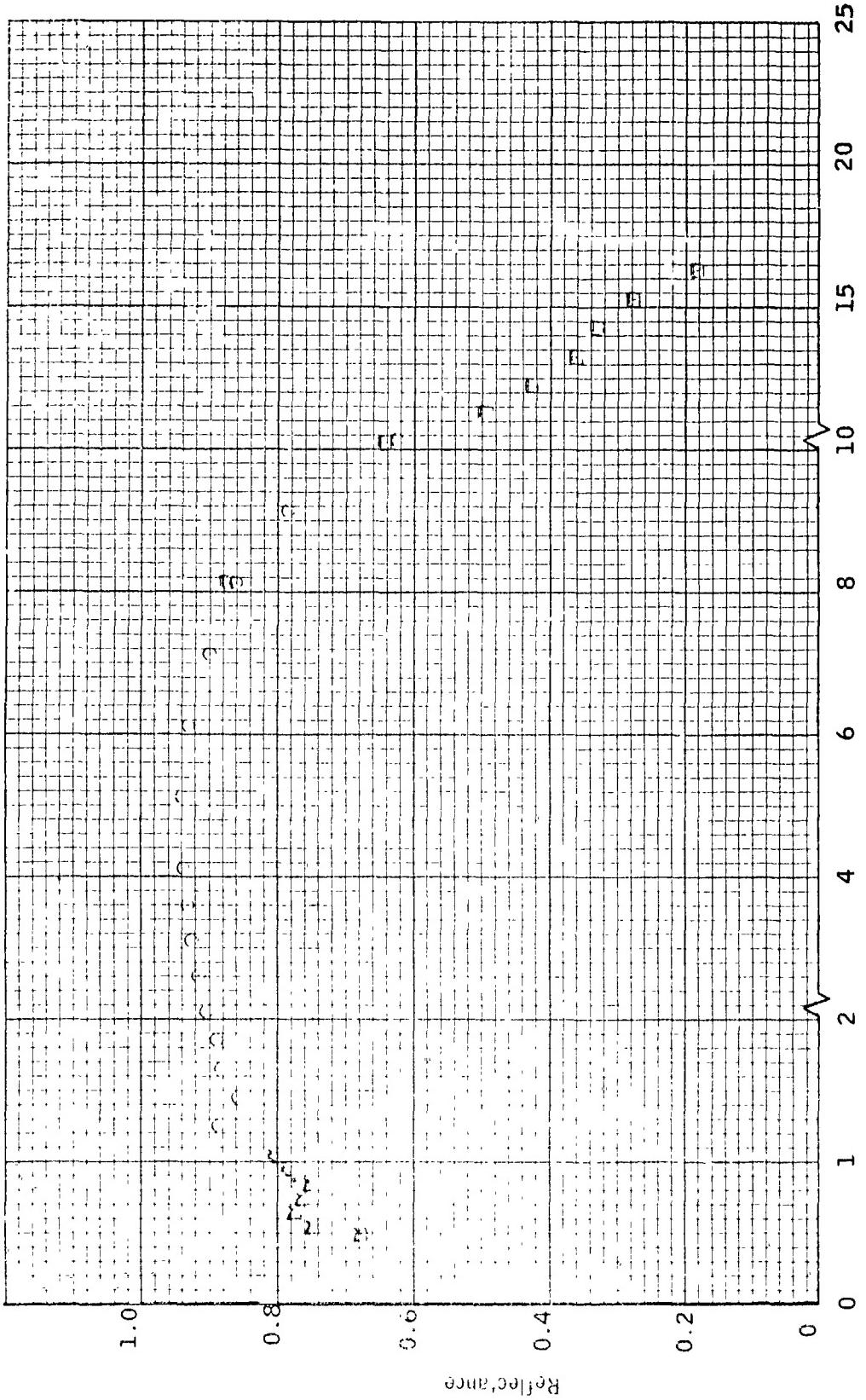


Fig. 31 Normal Spectral Reflectance of Specimen No. 55 Temperature 600 F

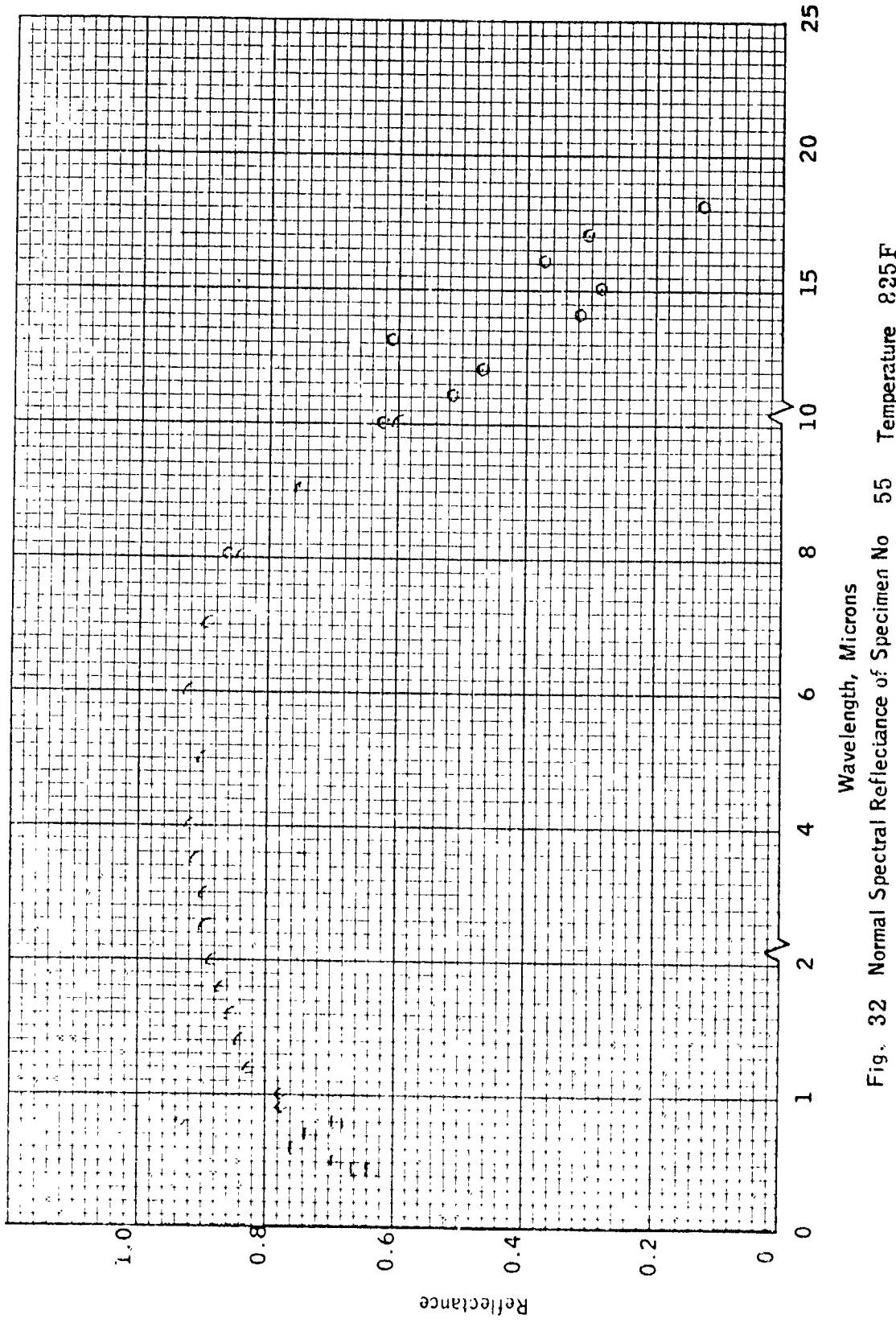


Fig. 32 Normal Spectral Reflectance of Specimen No. 55 Temperature 25°F

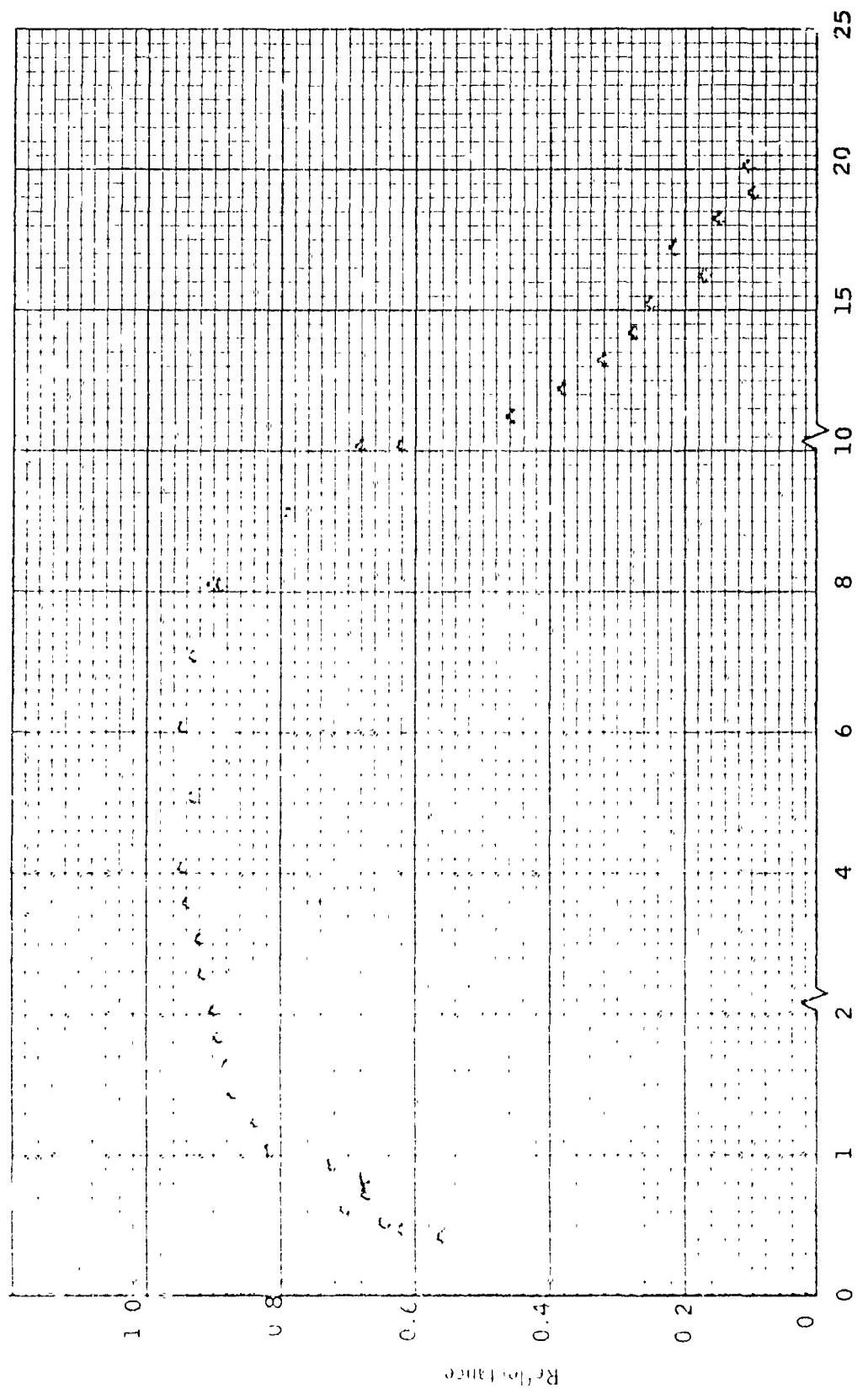


Fig. 33 Normal Spectral Reflectance of Specimen No. 55 Temperature  $RT_f$

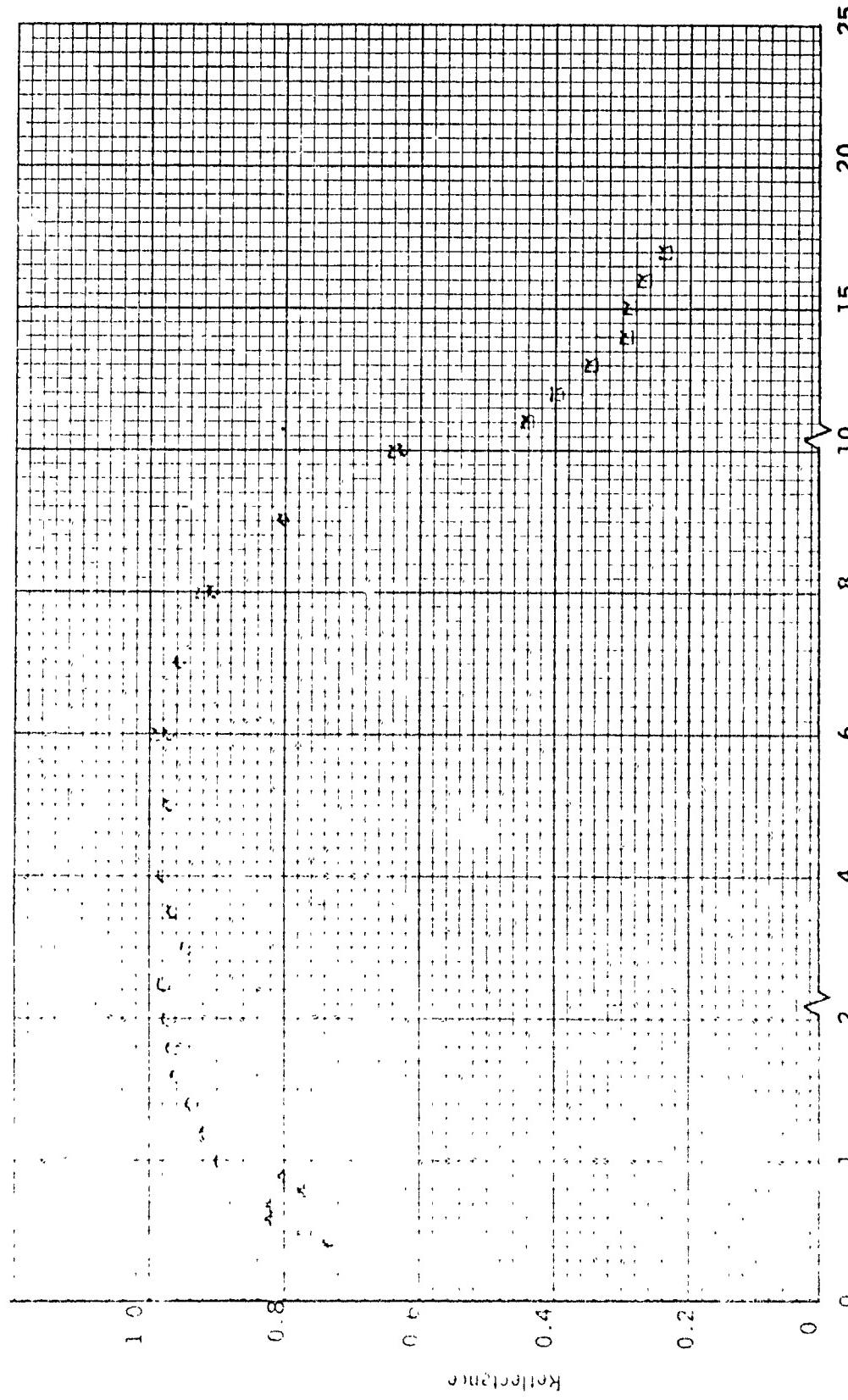


Fig. 34 Normal Spectral Reflectance of Specimen No. 56  
Temperature RT, HT-800°F

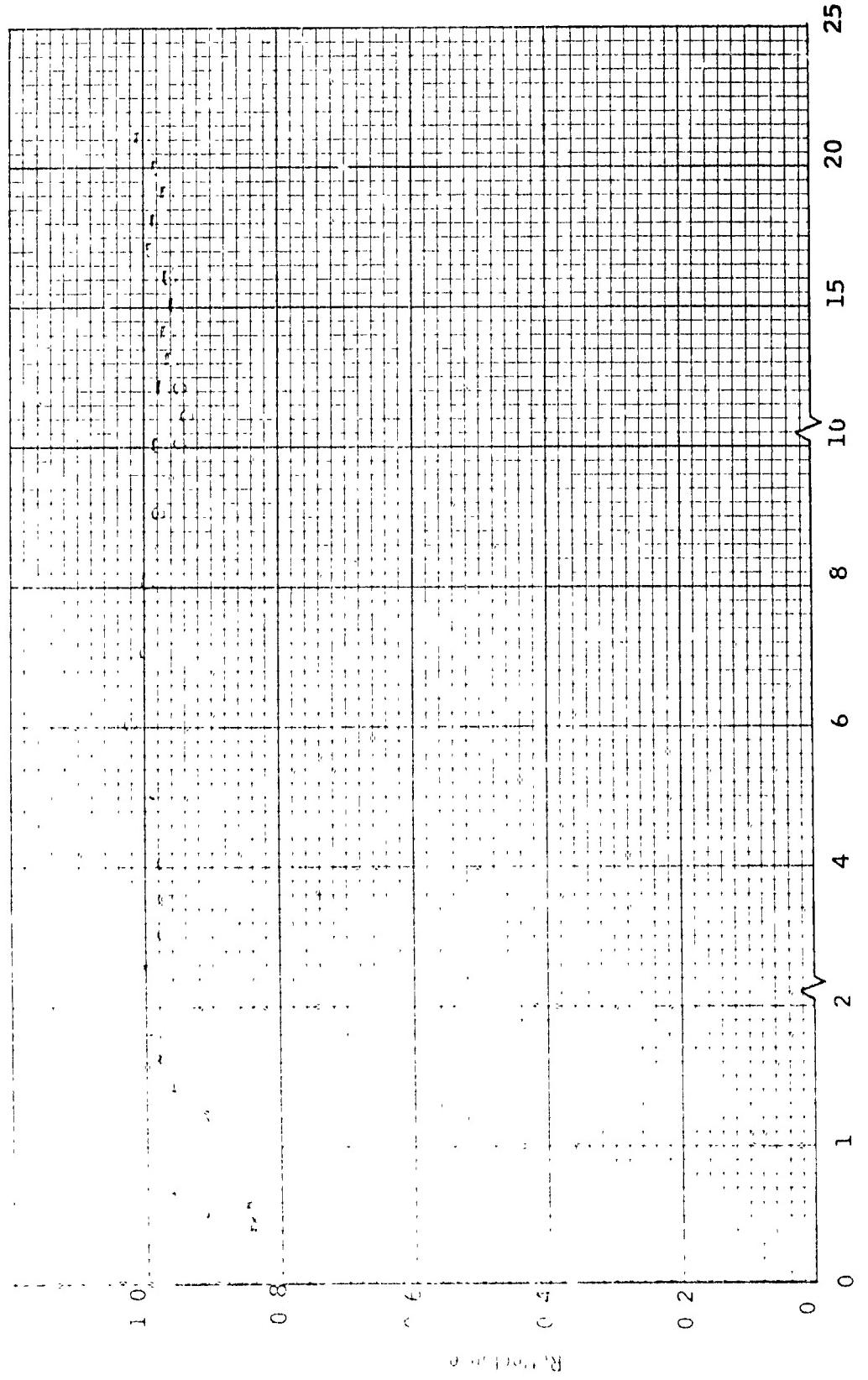


Fig. 35 Normal Spectral Reflectance of Specimen No. 57 Temperature RT

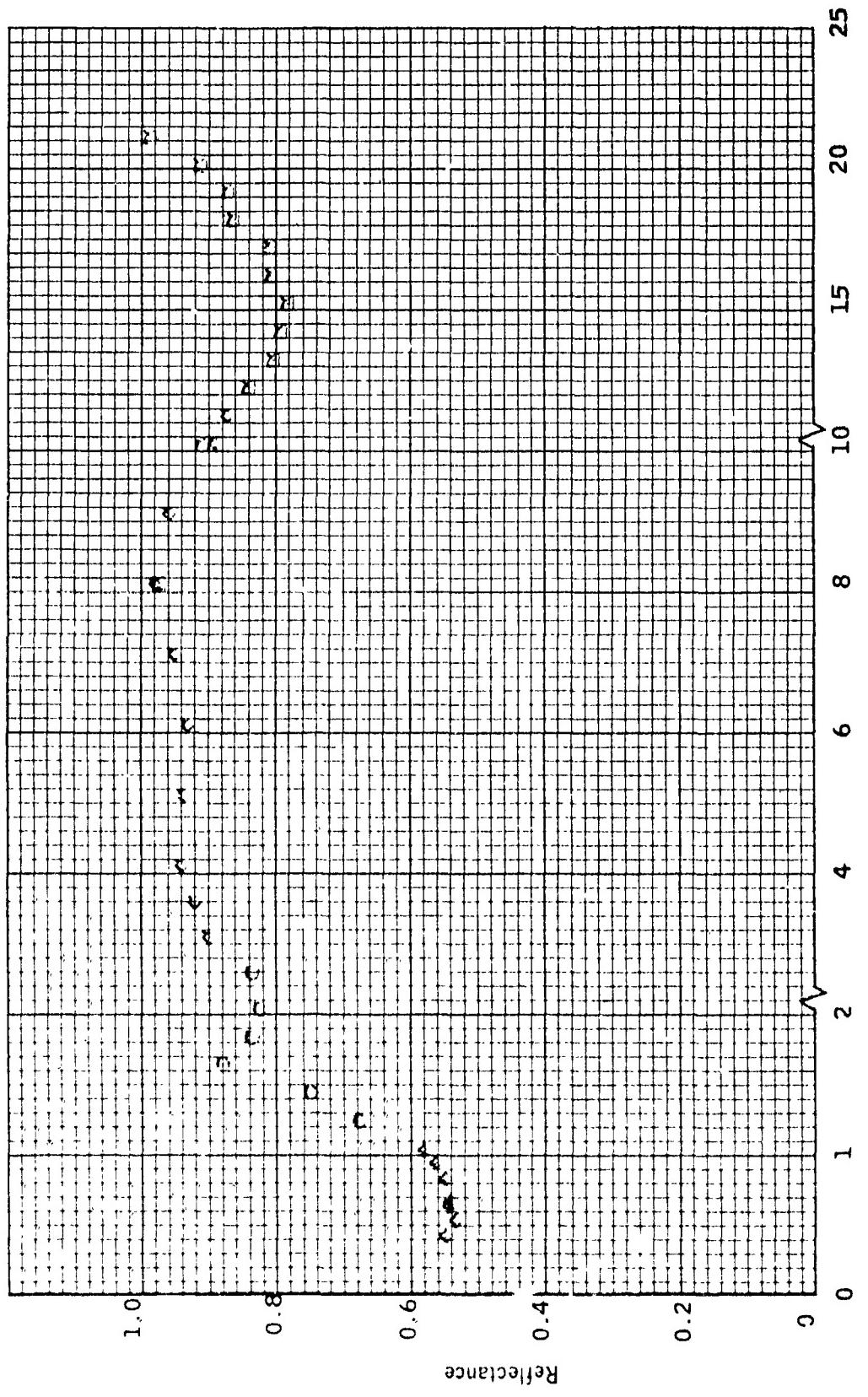


Fig. 36 Normal Spectral Reflectance of Specimen No. 83 Temperature RT

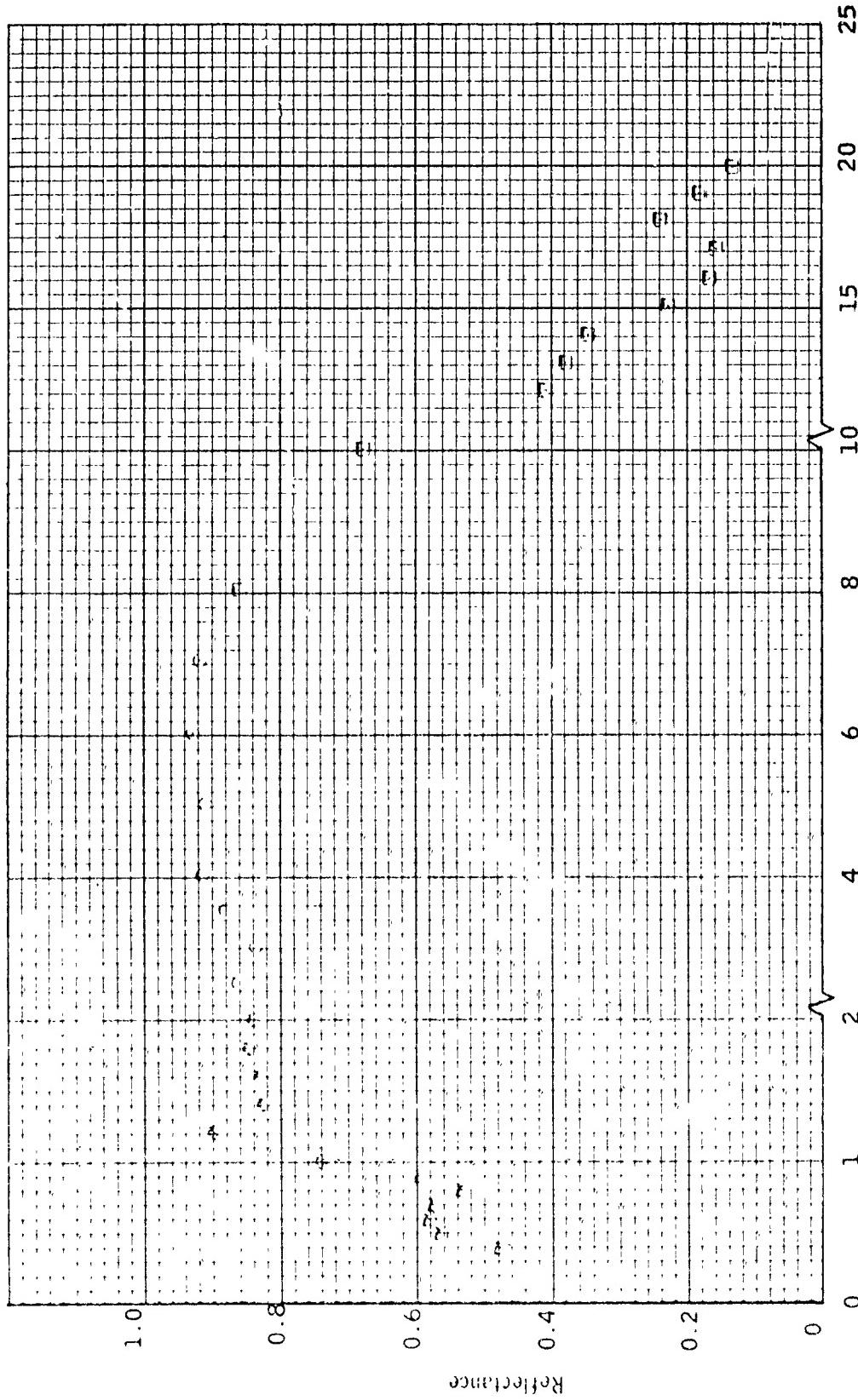


Fig. 37 Normal Spectral Reflectance of Specimen No 84 Temperature RT

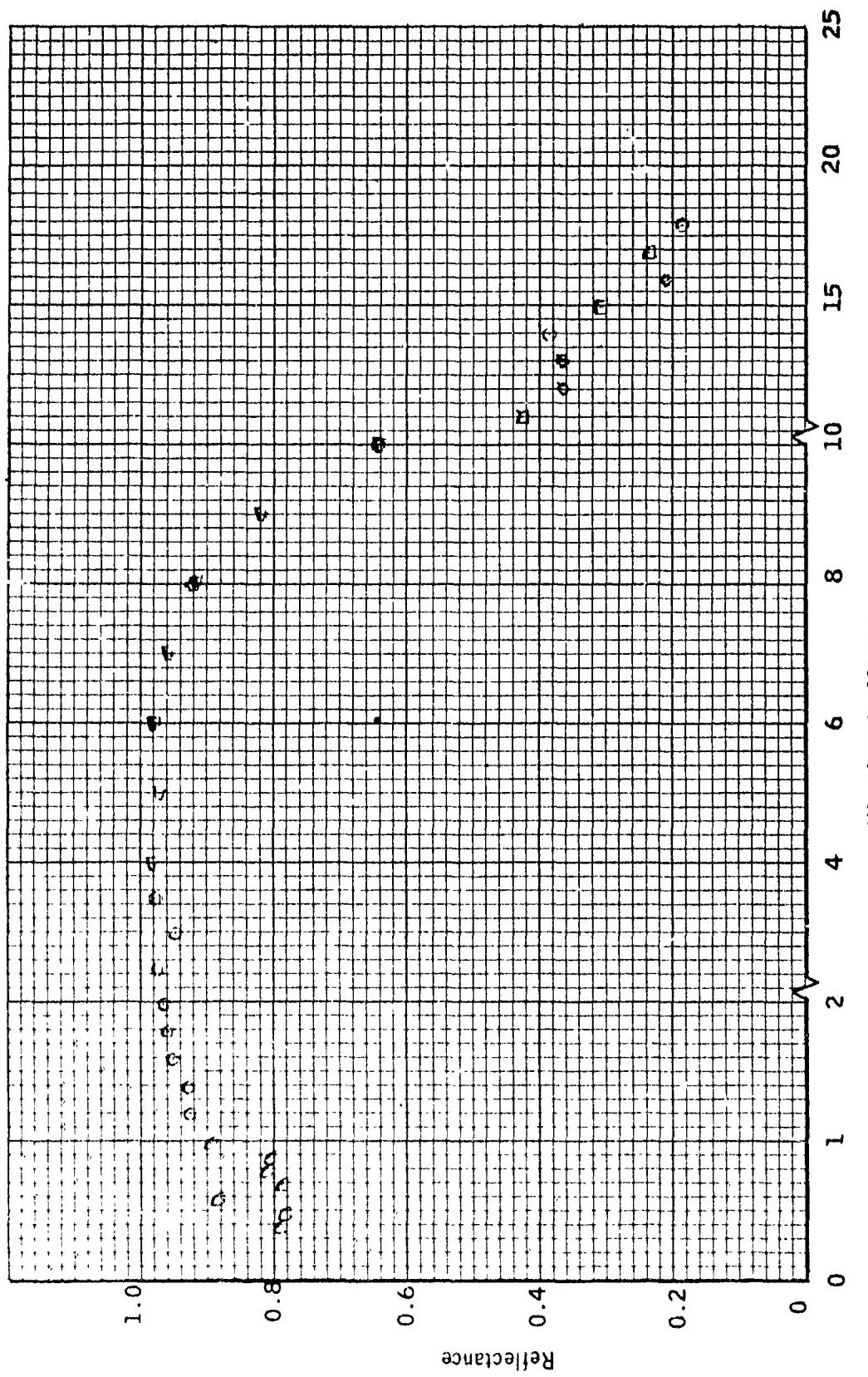


Fig. 38 Normal Spectral Reflectance of Specimen No. 92 Temperature RT

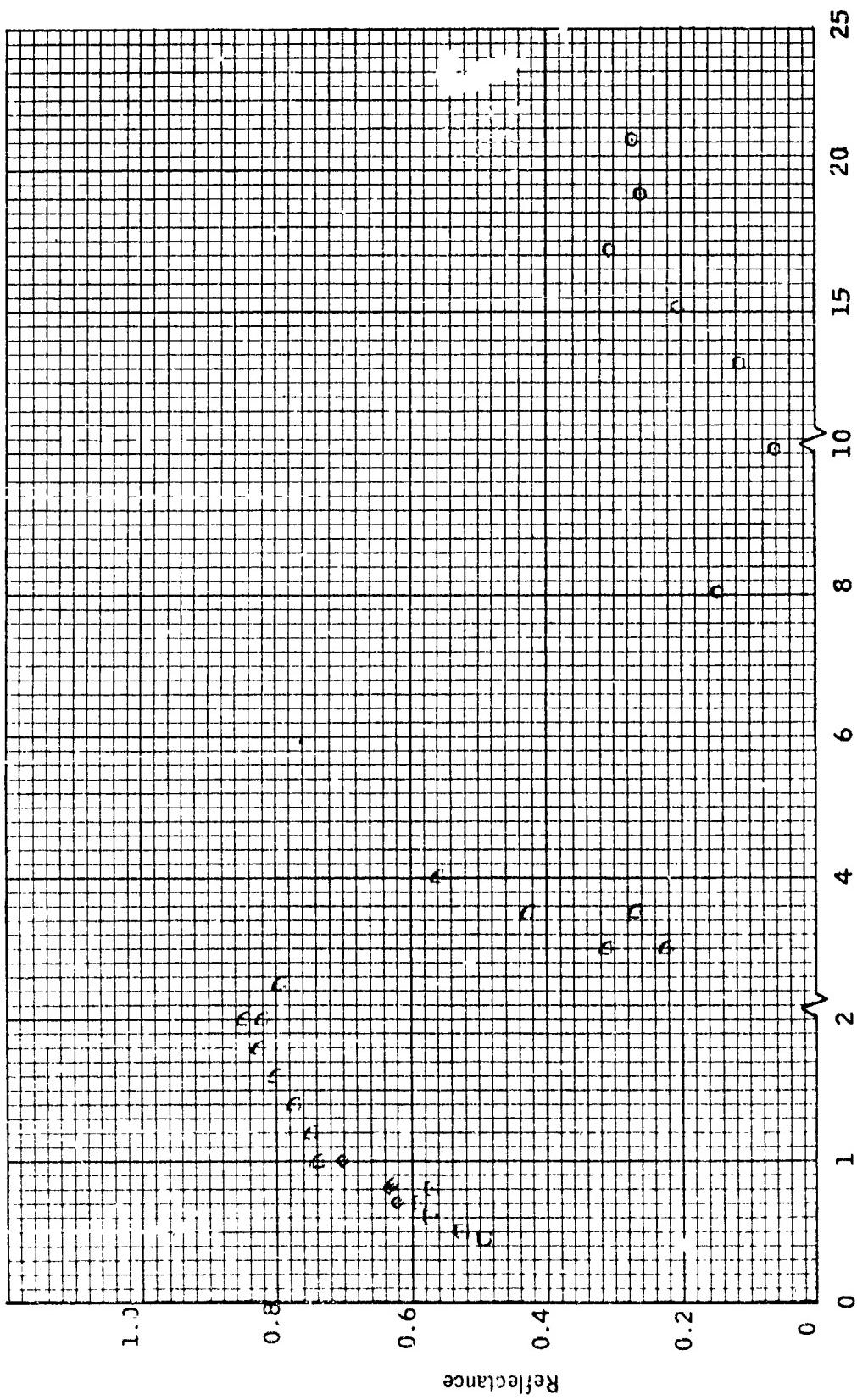


Fig. 39 Normal Spectral Reflectance of Specimen No 63 Temperature RT

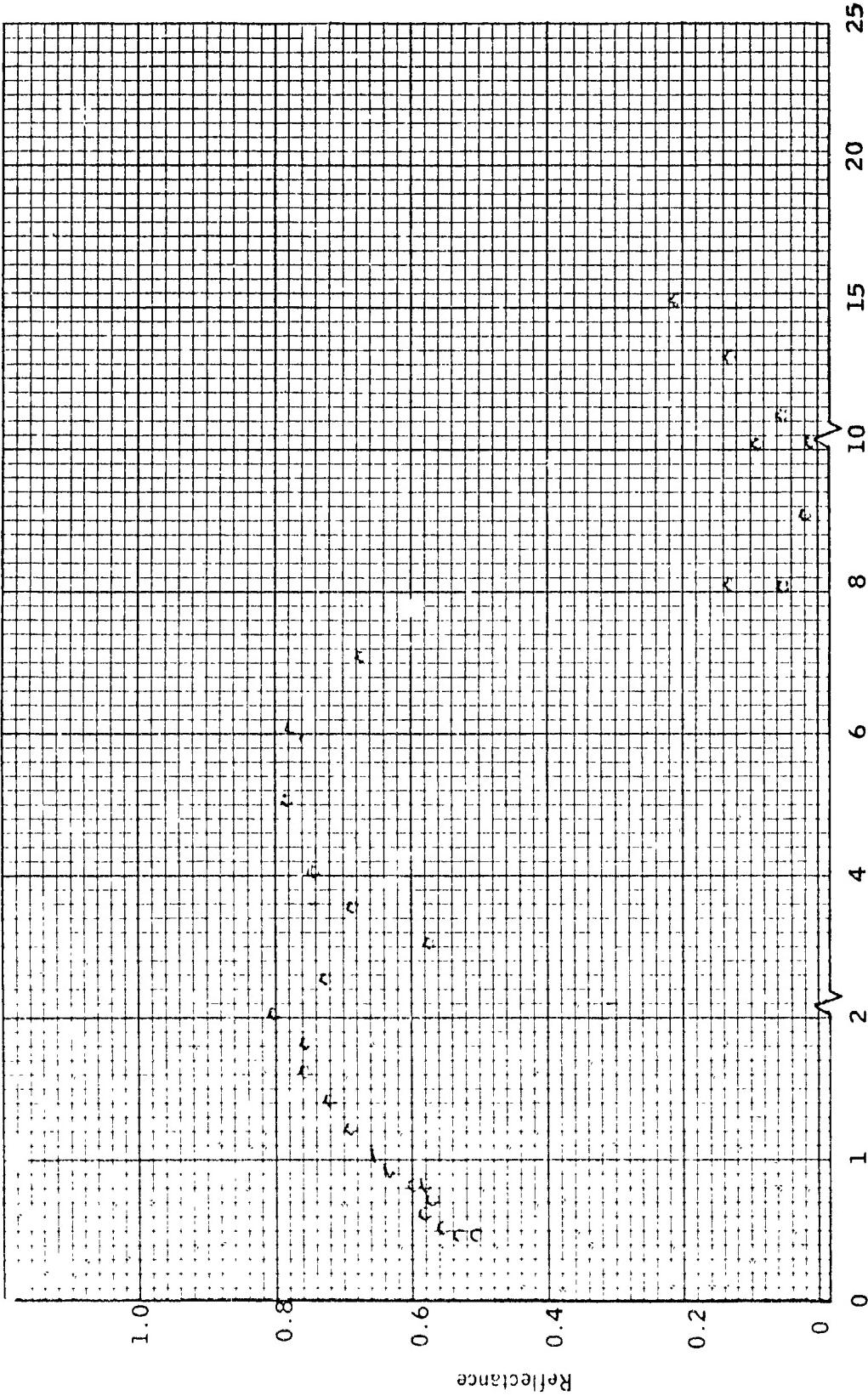


Fig. 40 Normal Spectral Reflectance of Specimen No. 63 Temperature 300°F

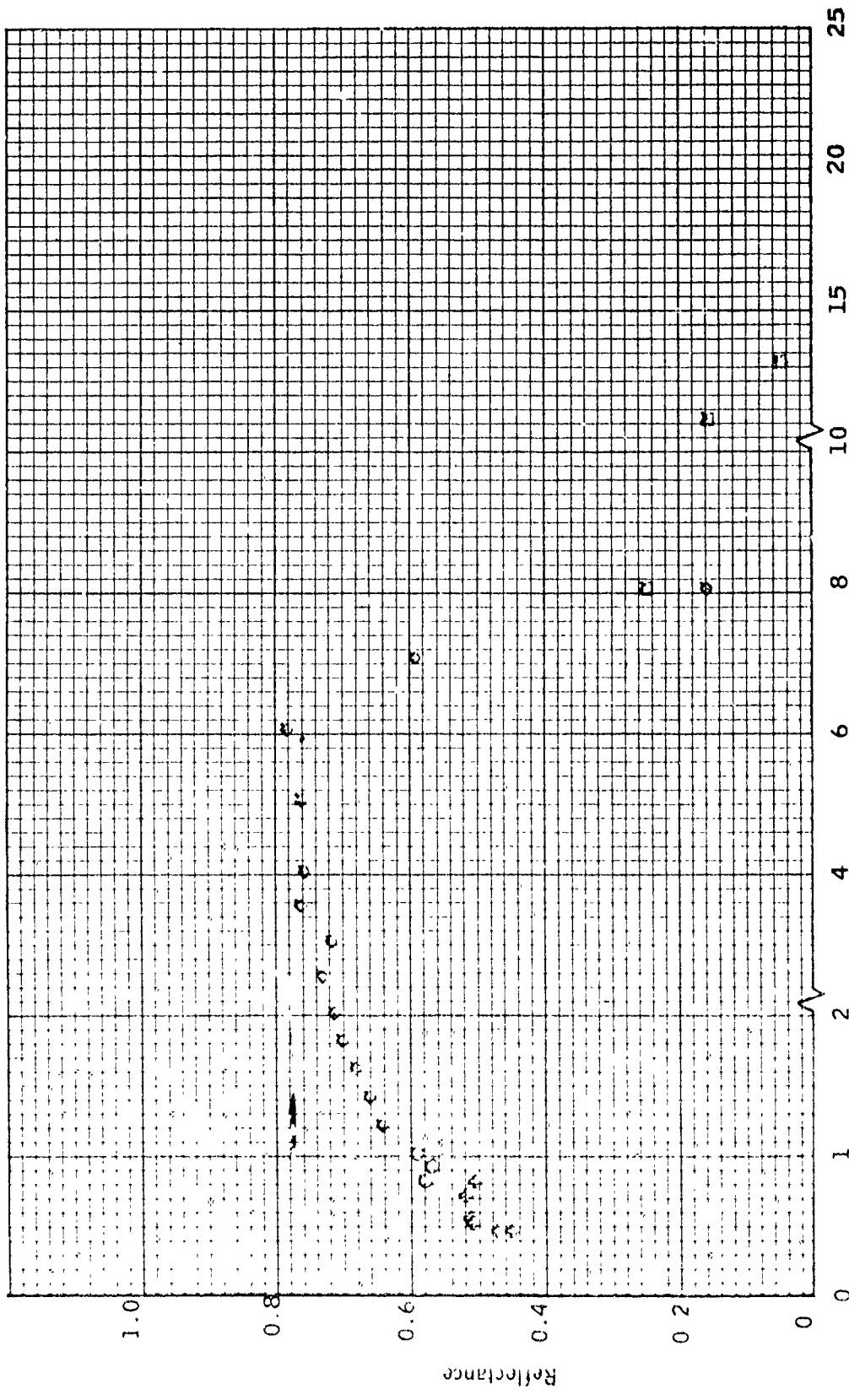


Fig. 41 Normal Spectral Reflectance of Specimen No. 63 Temperature 600 F

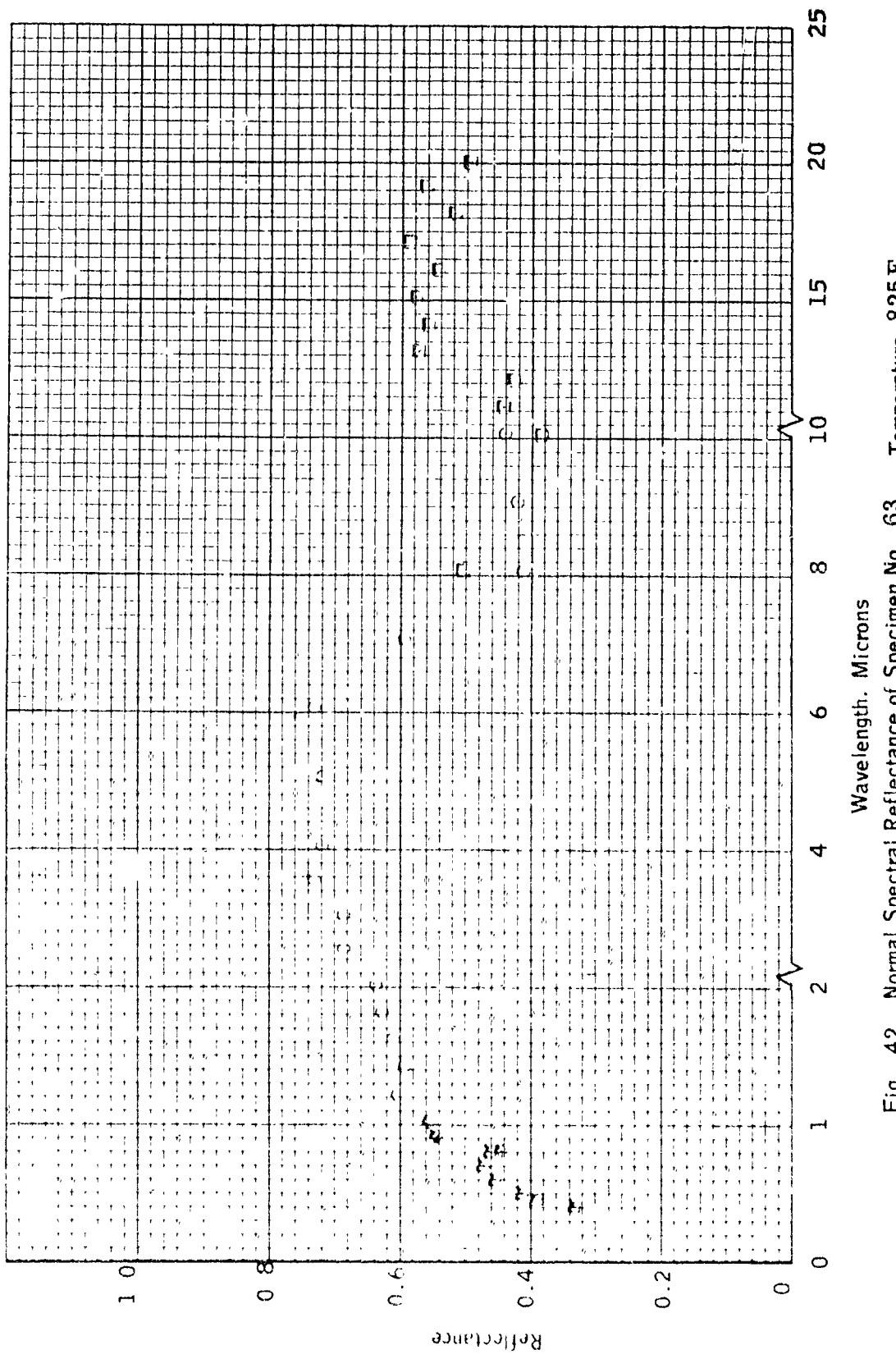


Fig. 42 Normal Spectral Reflectance of Specimen No. 63 Temperature 825°F

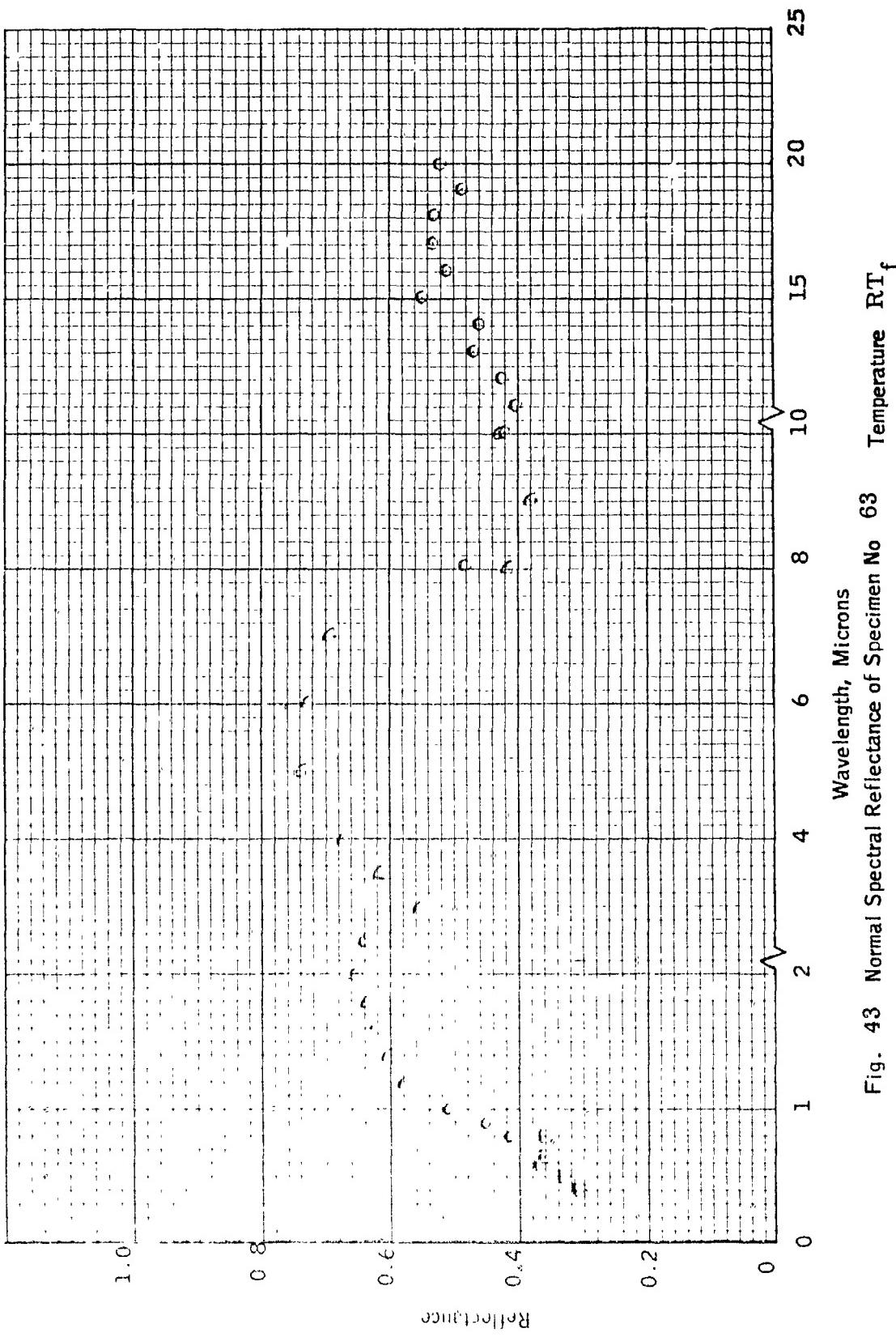


Fig. 43 Normal Spectral Reflectance of Specimen No. 63 Temperature  $RT_f$

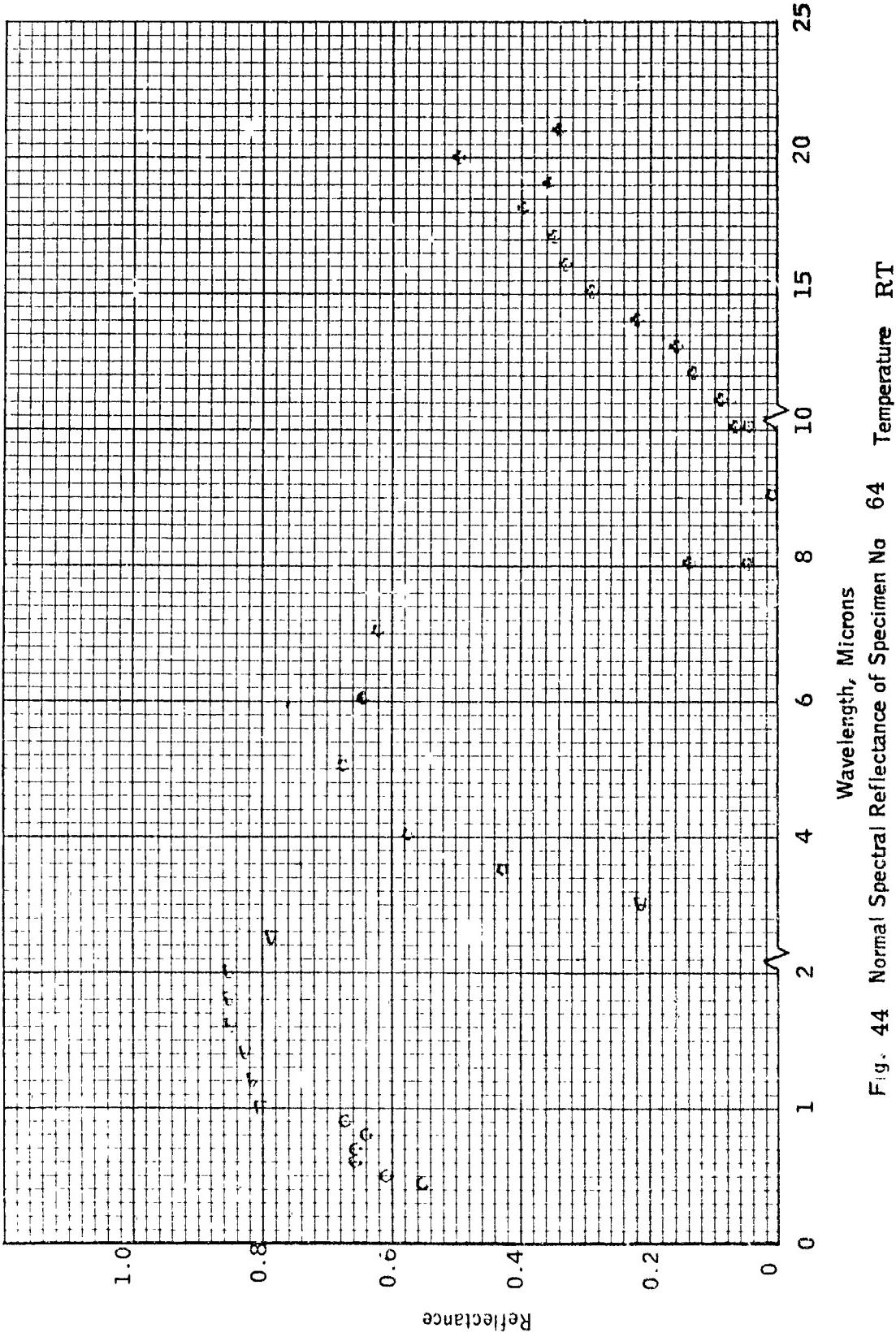


Fig. 44 Normal Spectral Reflectance of Specimen No 64 Temperature RT

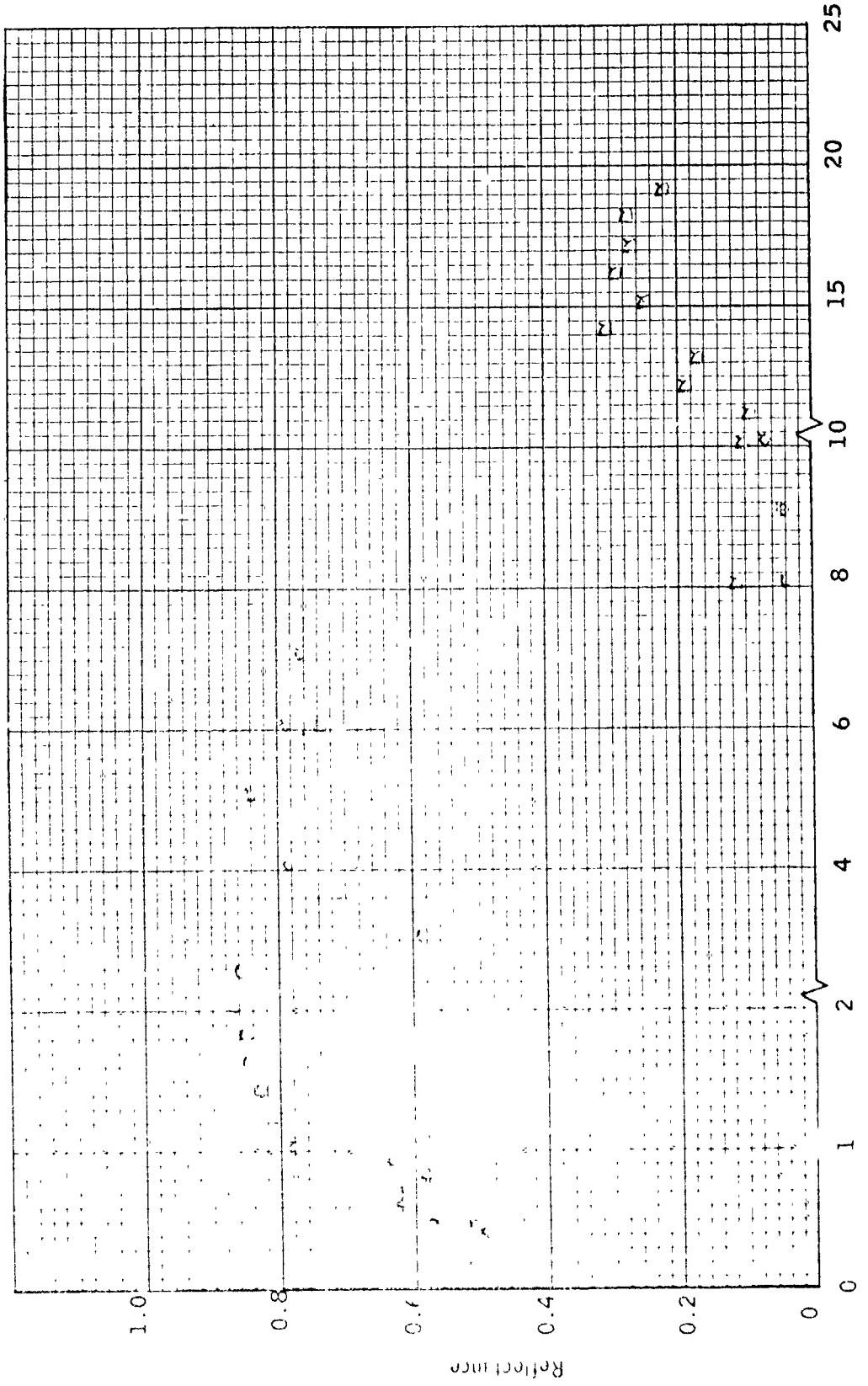


Fig. 45 Normal Spectral Reflectance of Specimen No 65 Temperature RT

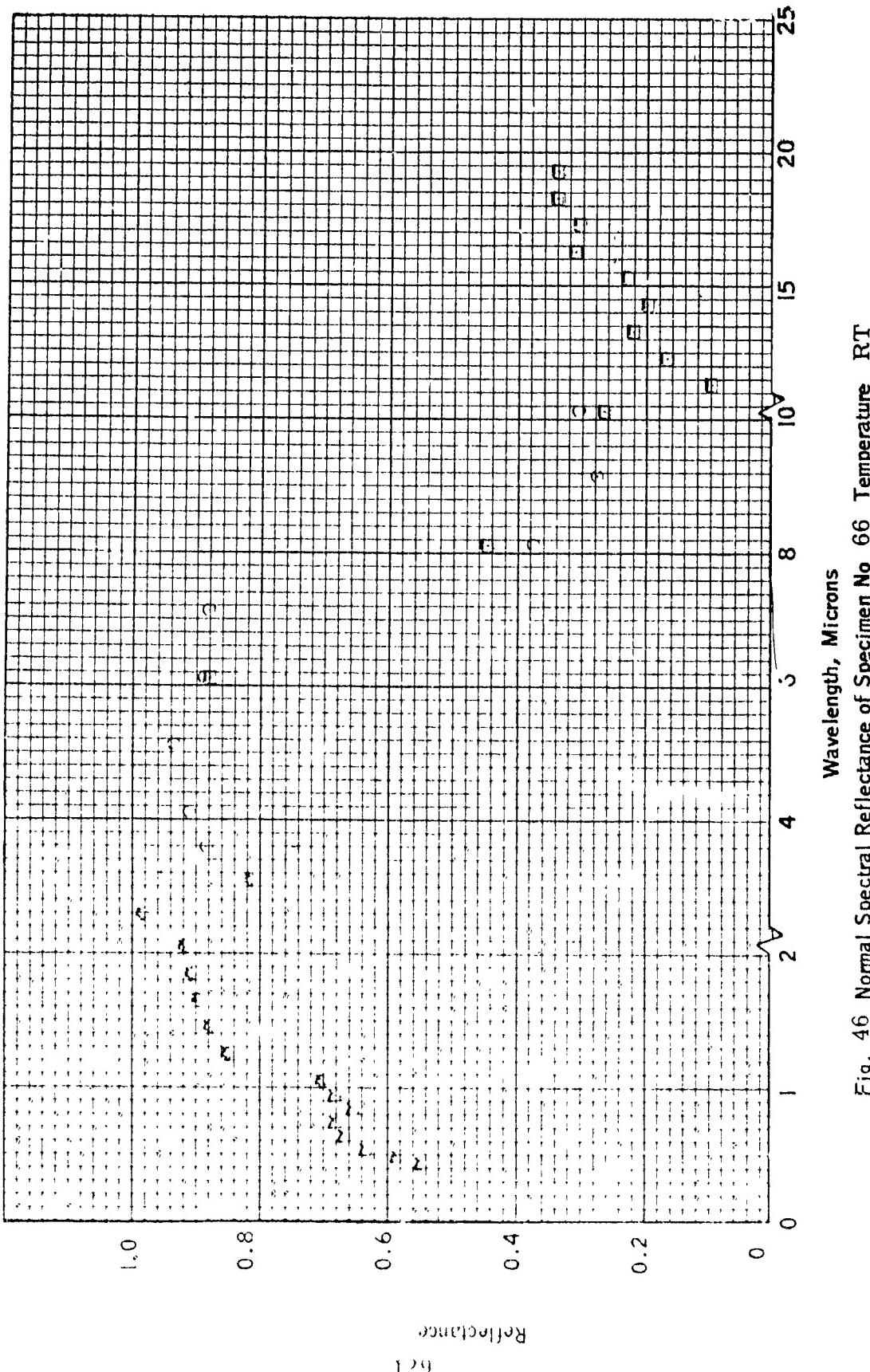


Fig. 46 Normal Spectral Reflectance of Specimen No. 66 Temperature RT

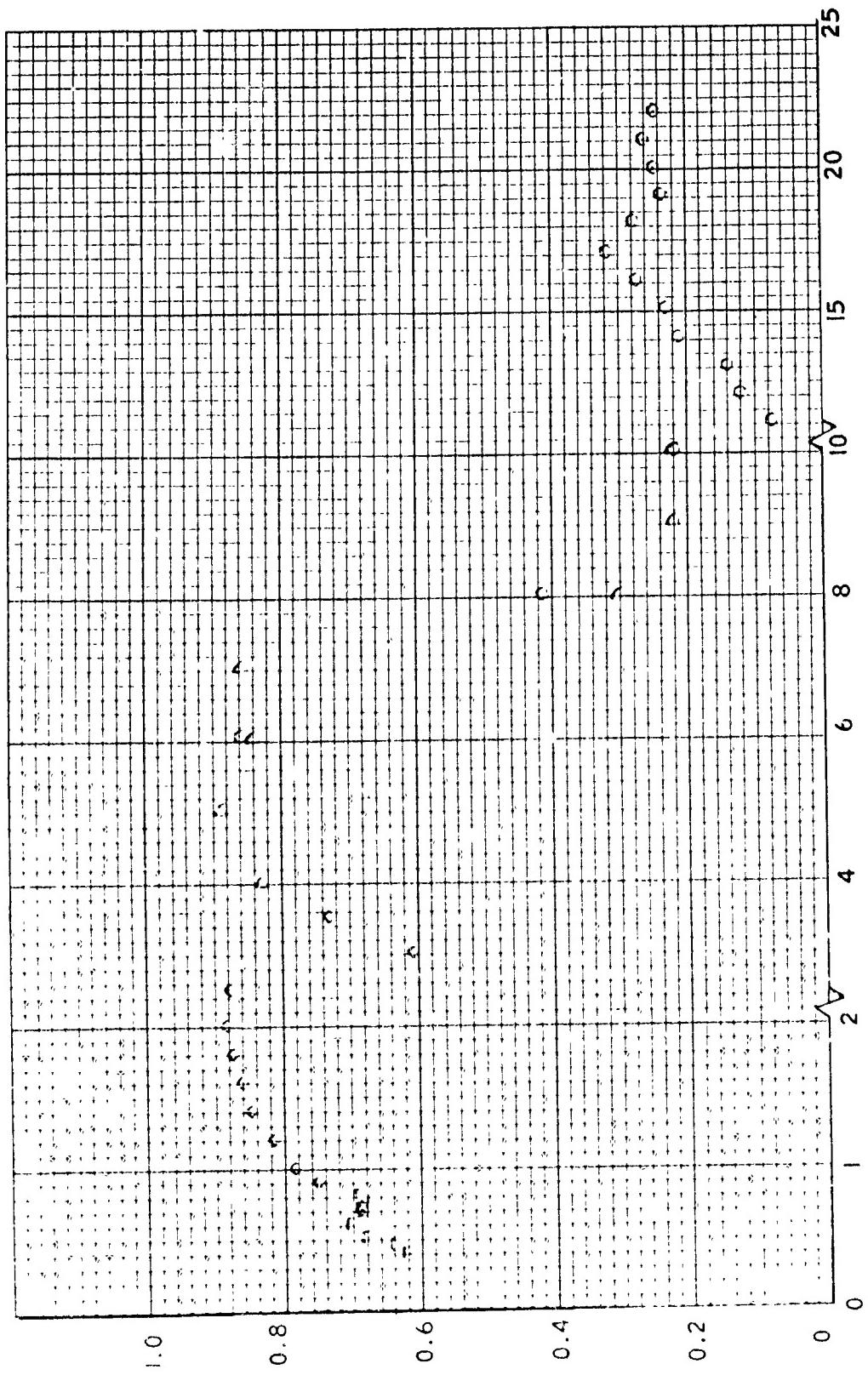


Fig. 47 Normal Spectral Reflectance of Specimen No 73 Temperature RT

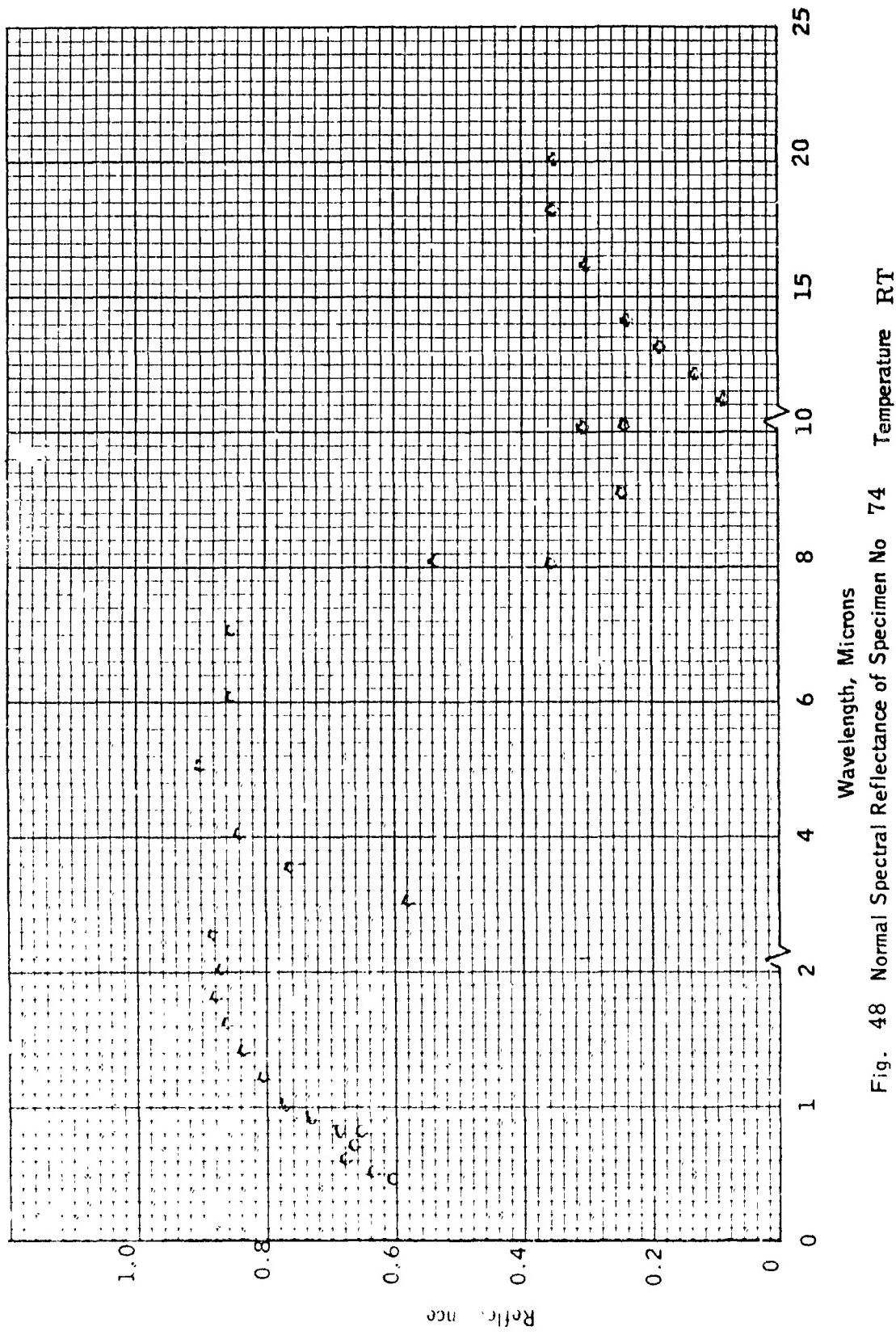


Fig. 48 Normal Spectral Reflectance of Specimen No 74 Temperature RT

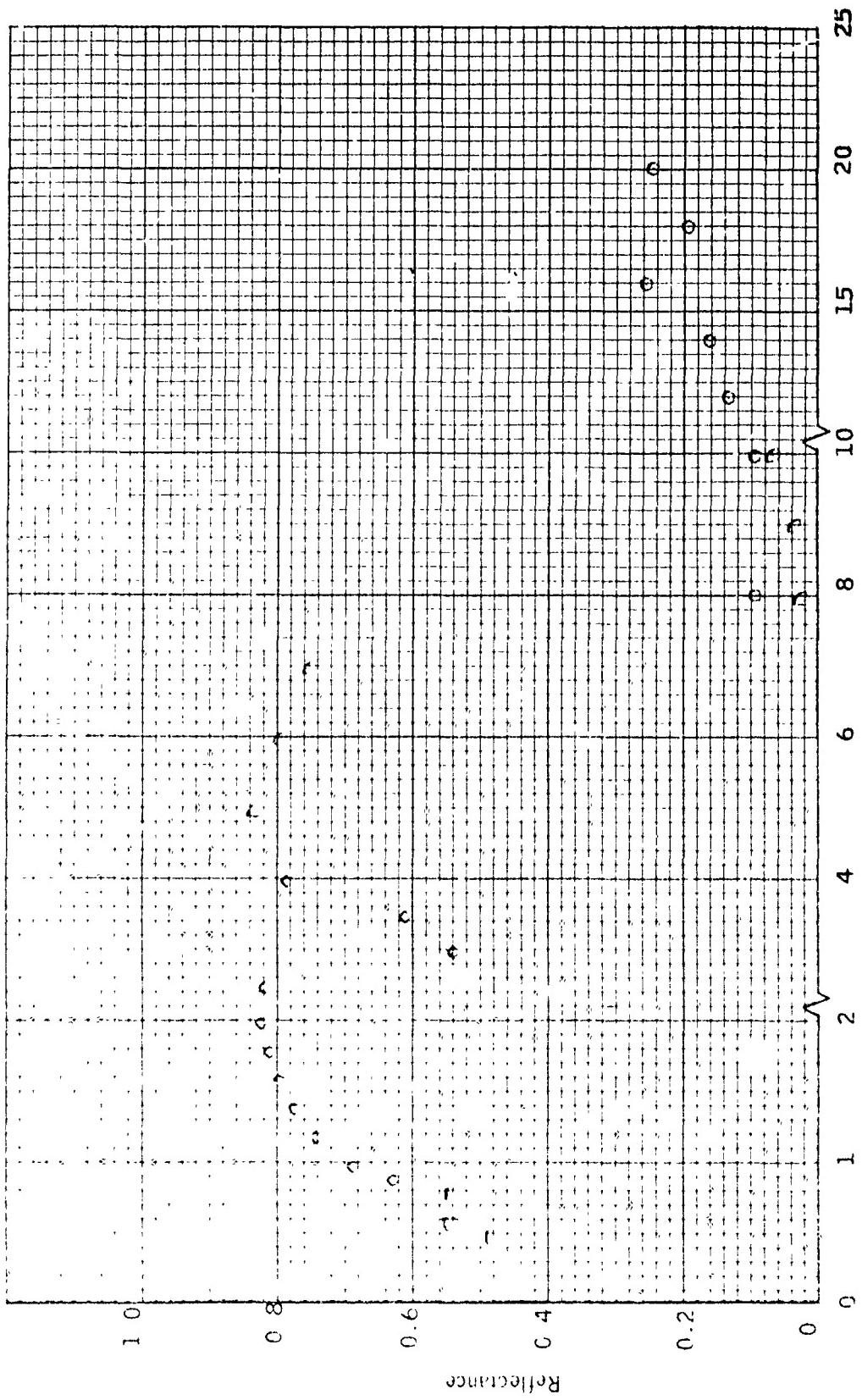


Fig. 49 Normal Spectral Reflectance of Specimen No. 77 Temperature RT

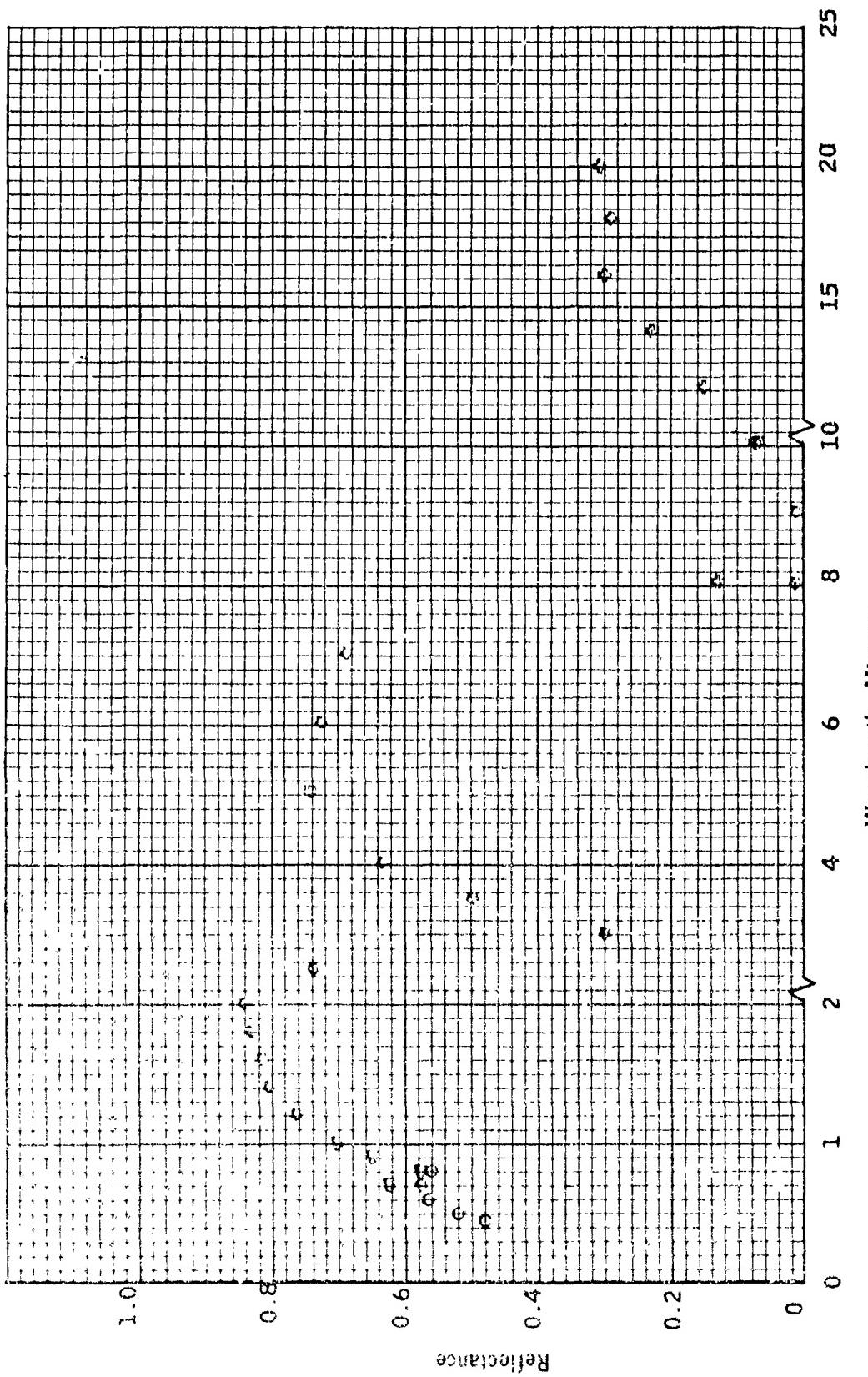


Fig. 50 Normal Spectral Reflectance of Specimen No. 78 Temperature RT

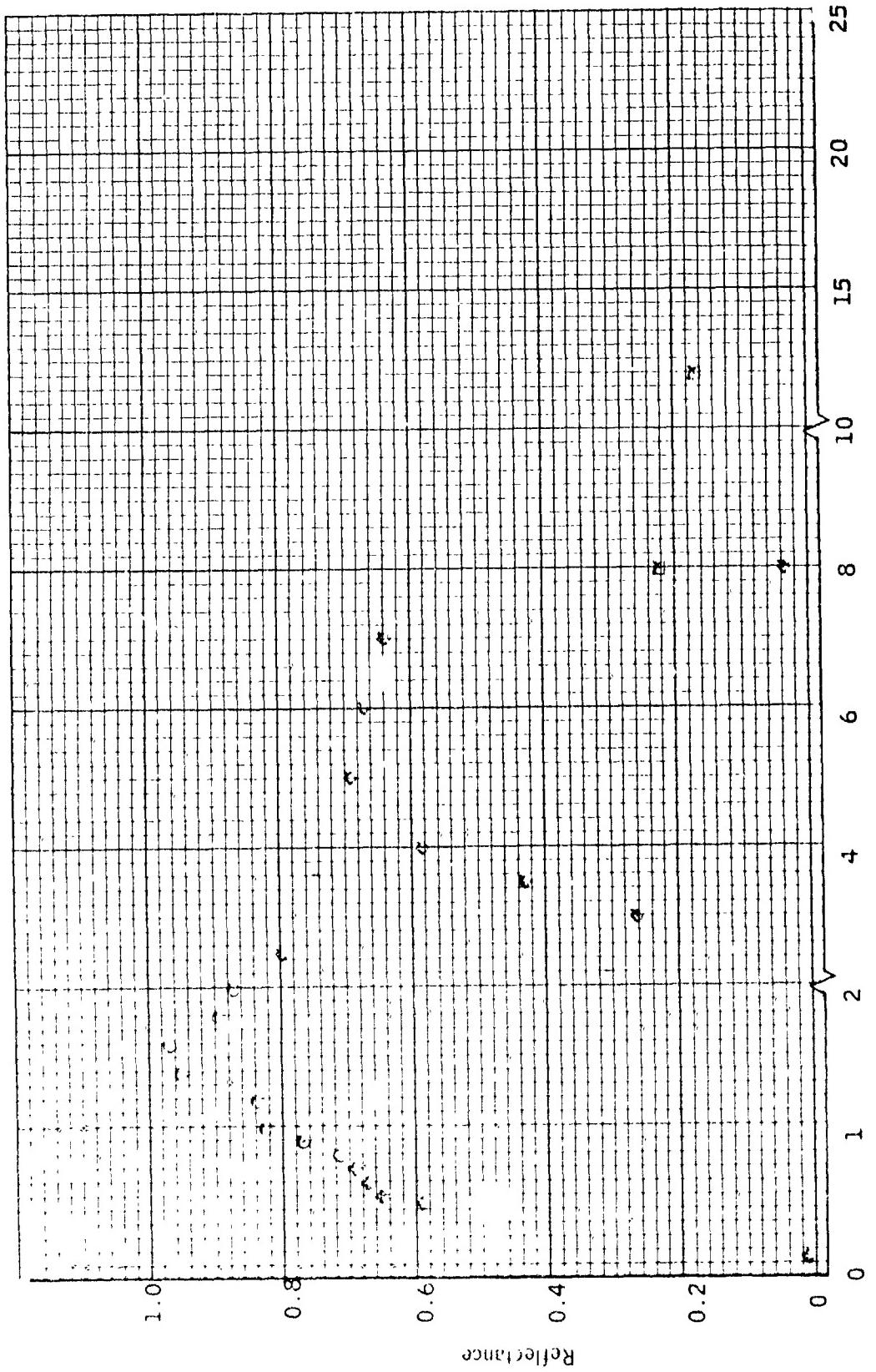


Fig. 51 Normal Spectral Reflectance of Specimen No 120 Temperature RT

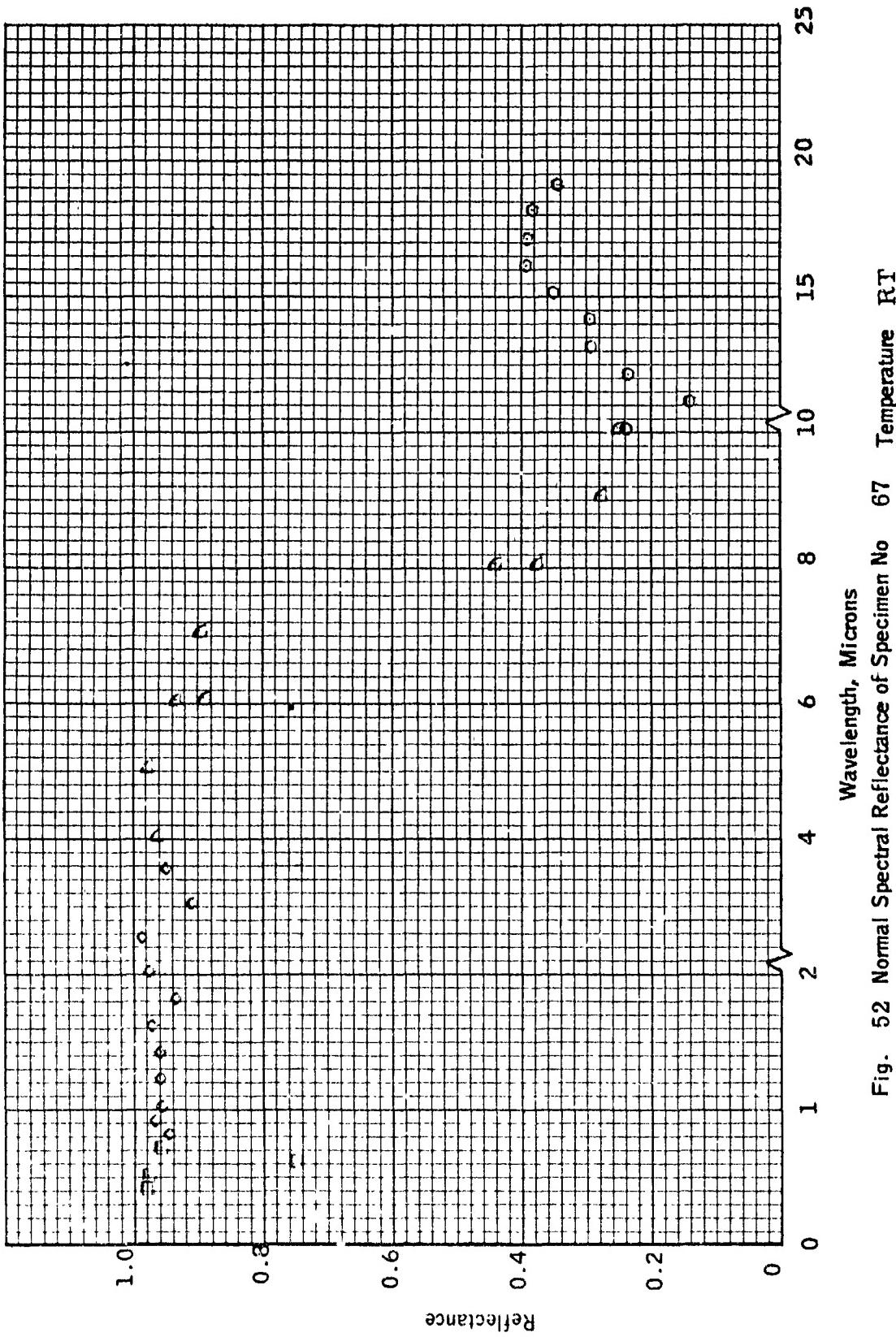


Fig. 52 Normal Spectral Reflectance of Specimen No. 67 Temperature RT

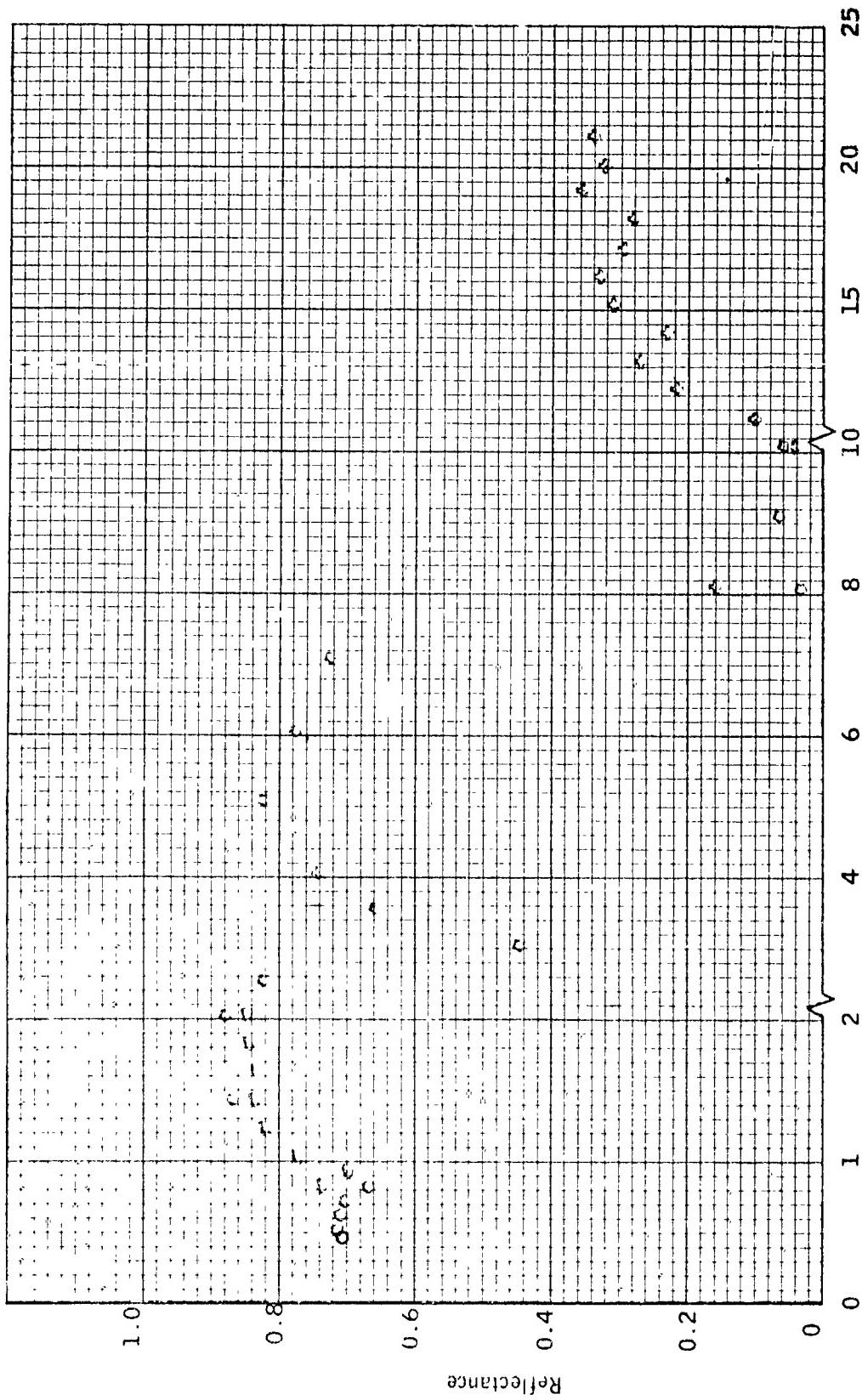


Fig. 53 Normal Spectral Reflectance of Specimen No. 68 Temperature RT, HT-800 F

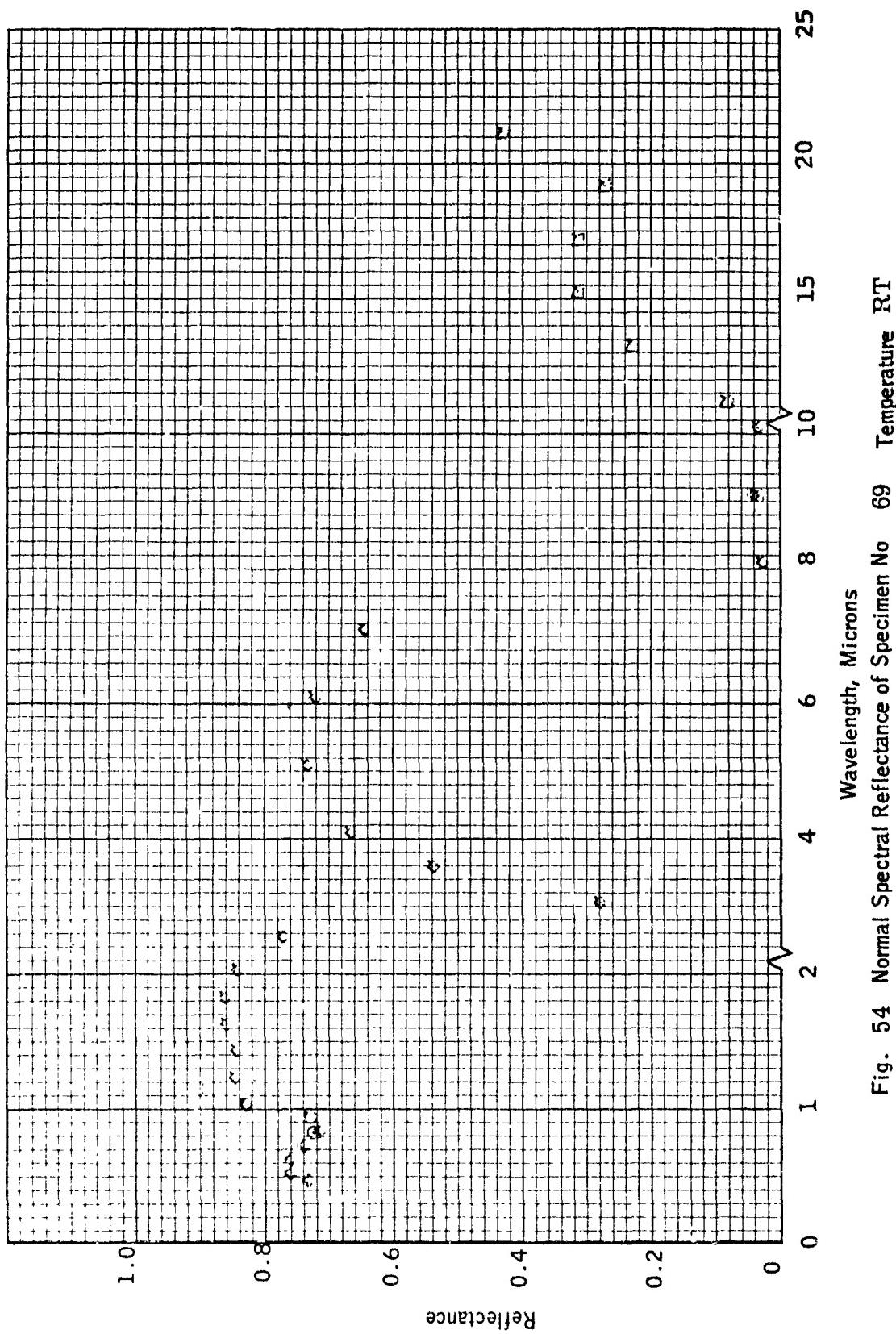


Fig. 54 Normal Spectral Reflectance of Specimen No. 69 Temperature RT

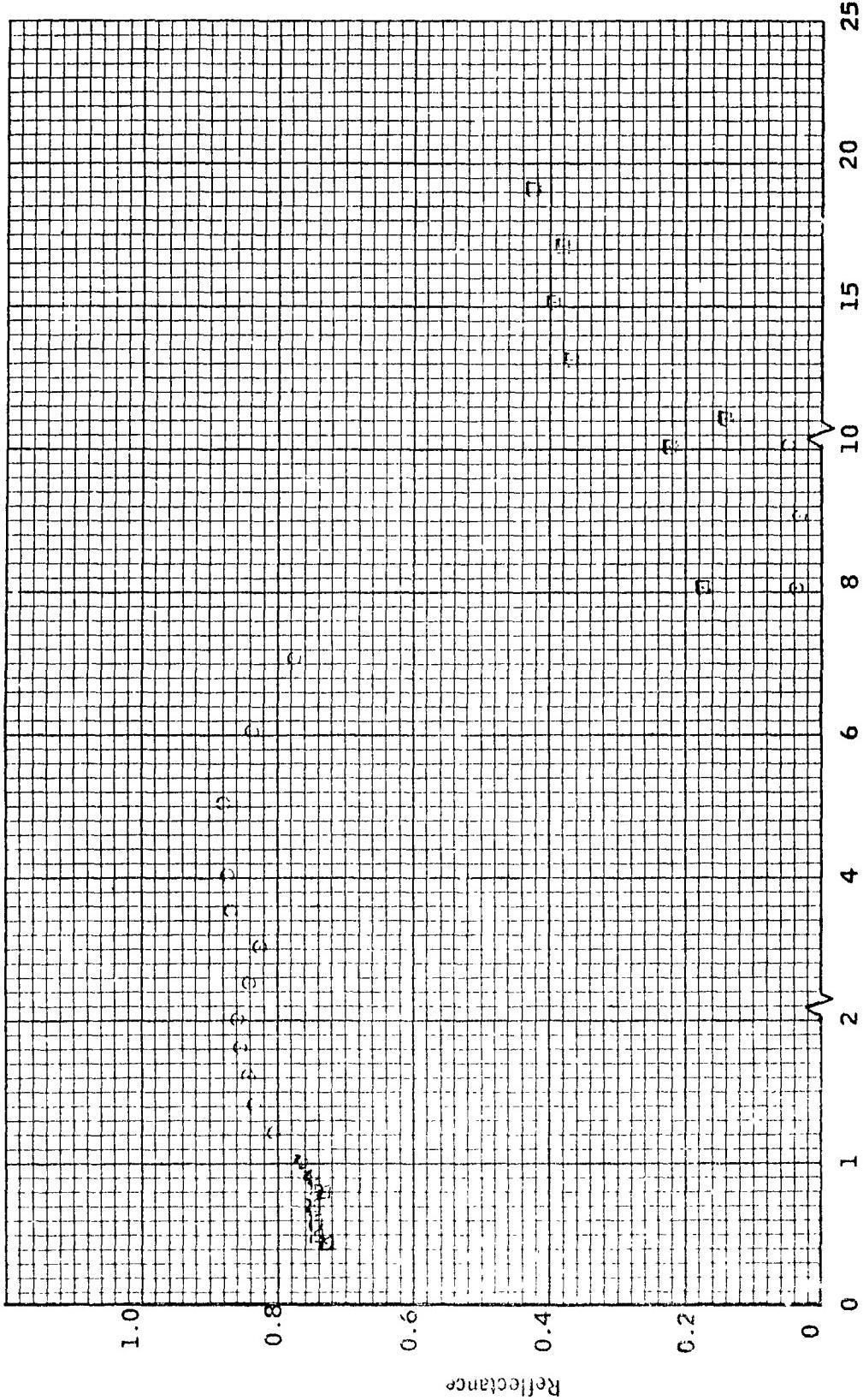


Fig. 55 Normal Spectral Reflectance of Specimen No. 69 Temperature 600°F

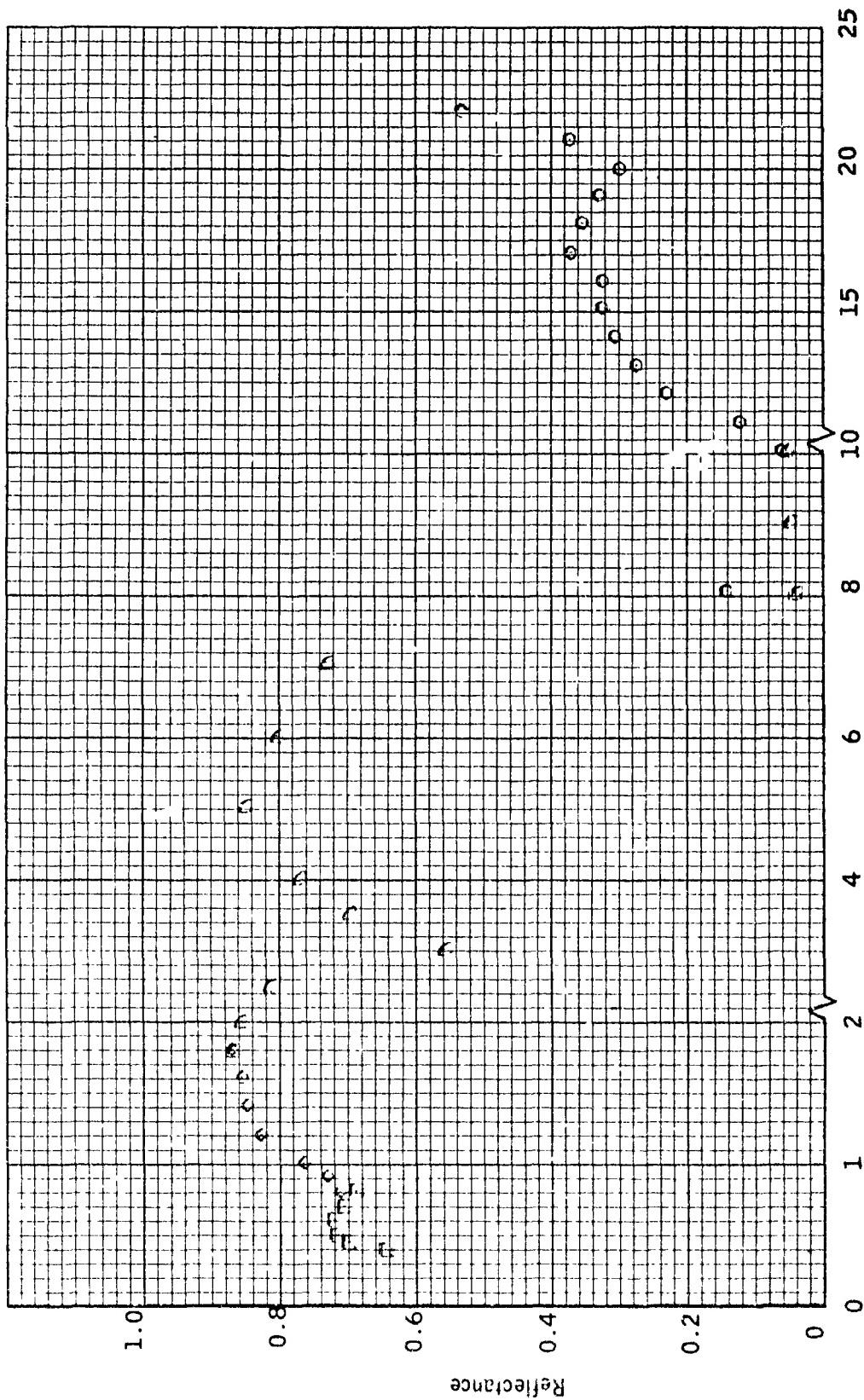


Fig. 56 Normal Spectral Reflectance of Specimen No. 69 Temperature RT<sub>f</sub>

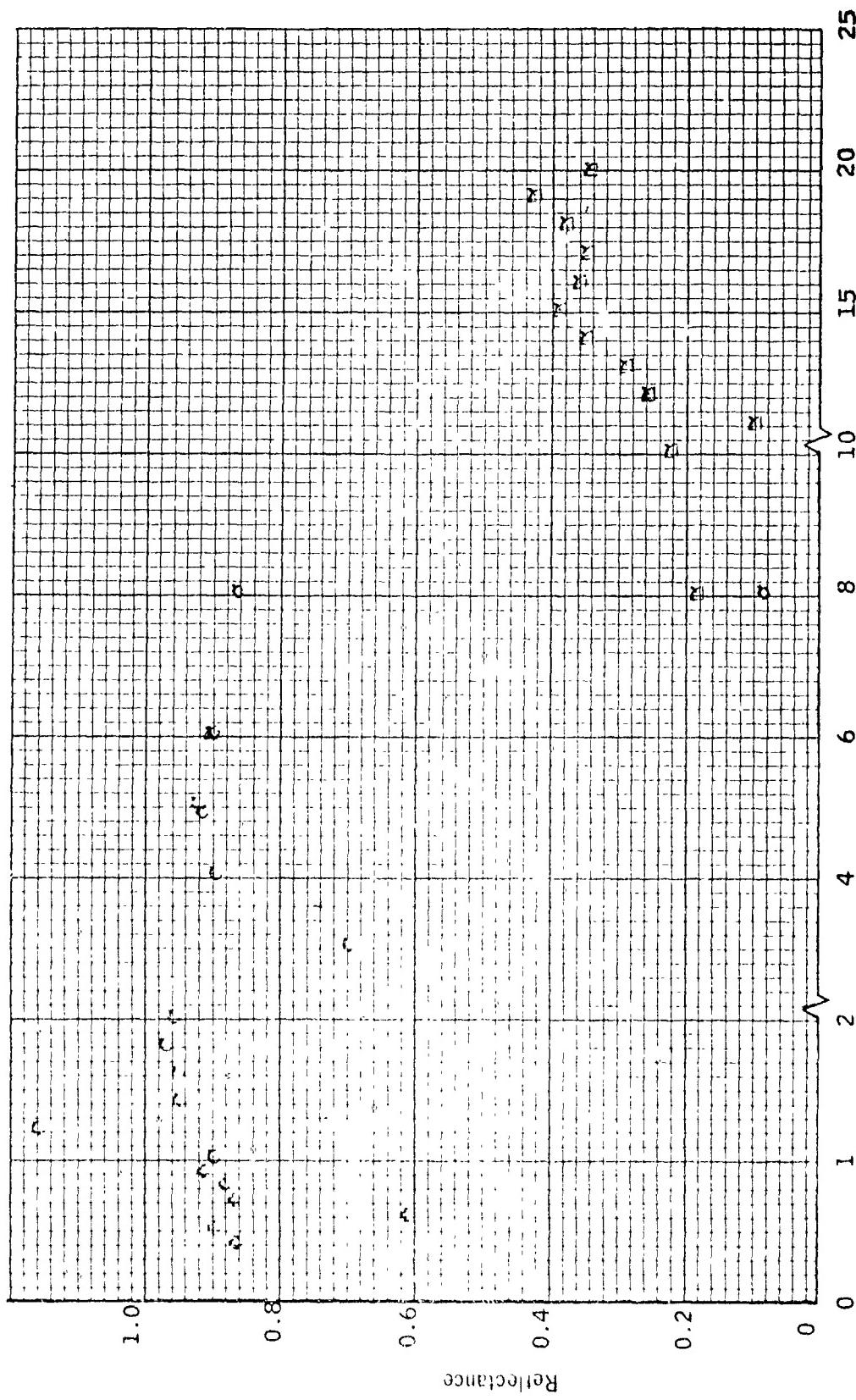


Fig. 57 Normal Spectral Reflectance of Specimen No. 71 Temperature RT

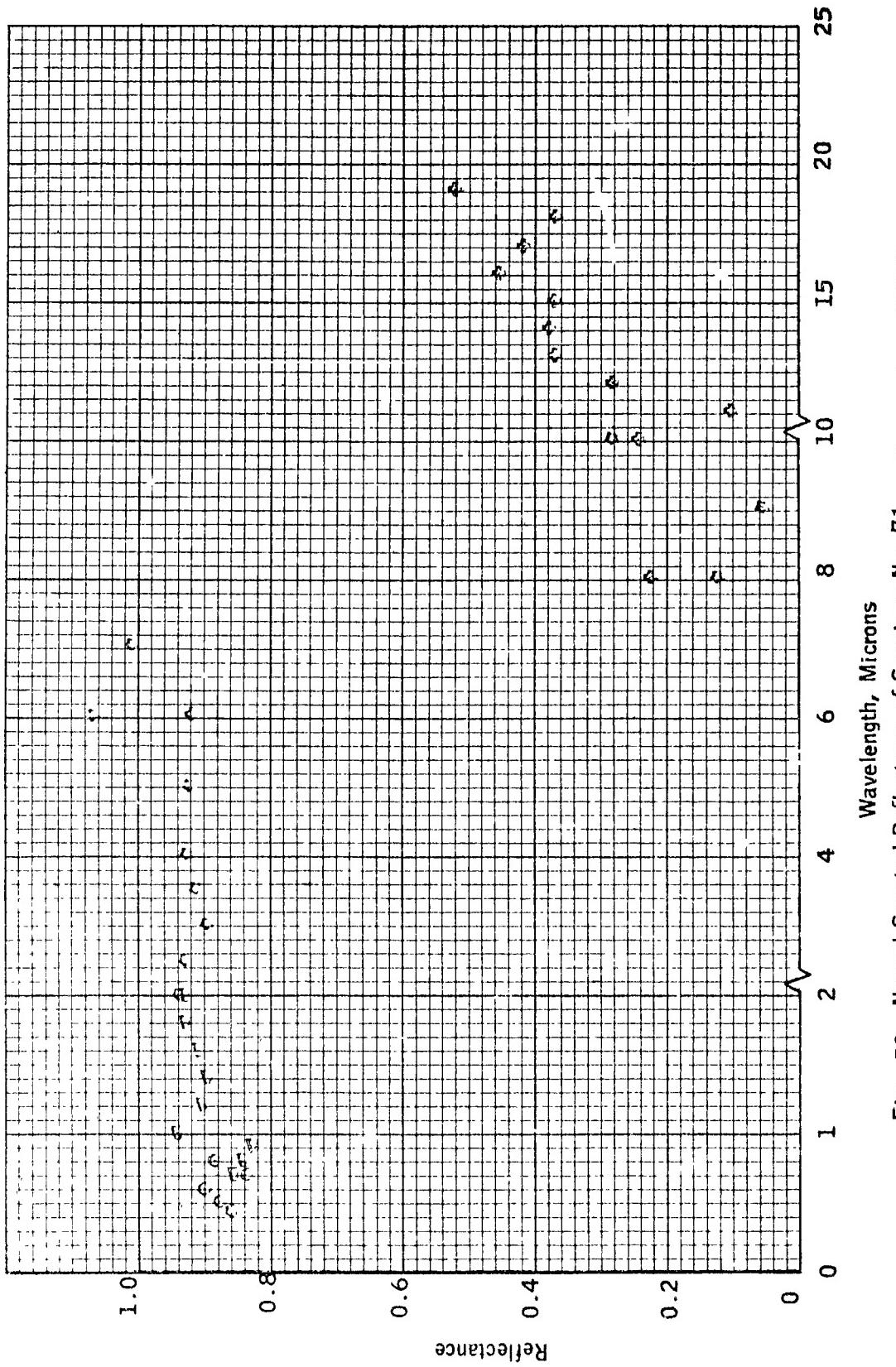


Fig. 58 Normal Spectral Reflectance of Specimen No. 71 Temperature 300°F

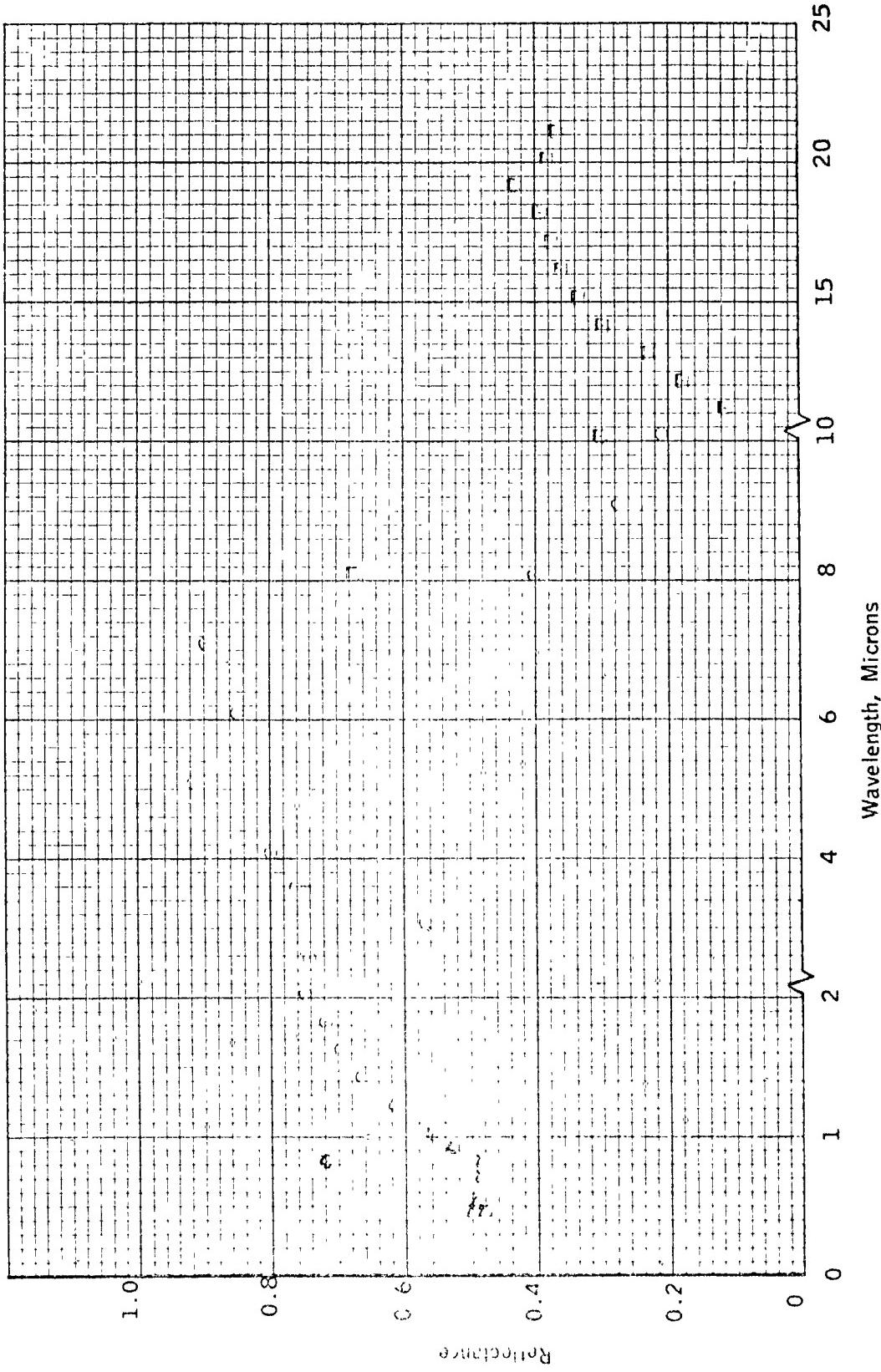


Fig. 59 Normal Spectral Reflectance of Specimen No. 72 Temperature RT

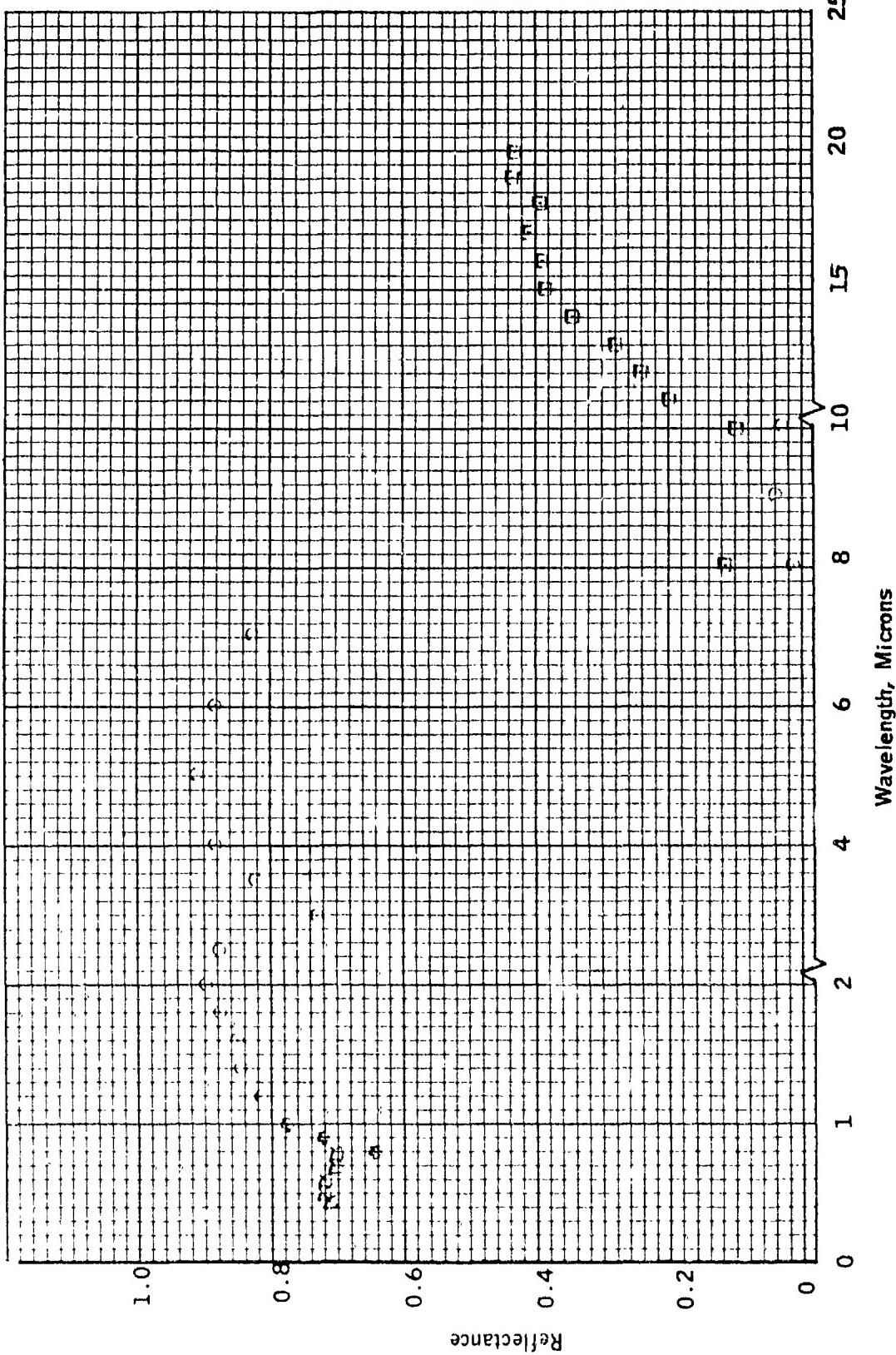


Fig. 6C Normal Spectral Reflectance of Specimen No. 80 Temperature RT

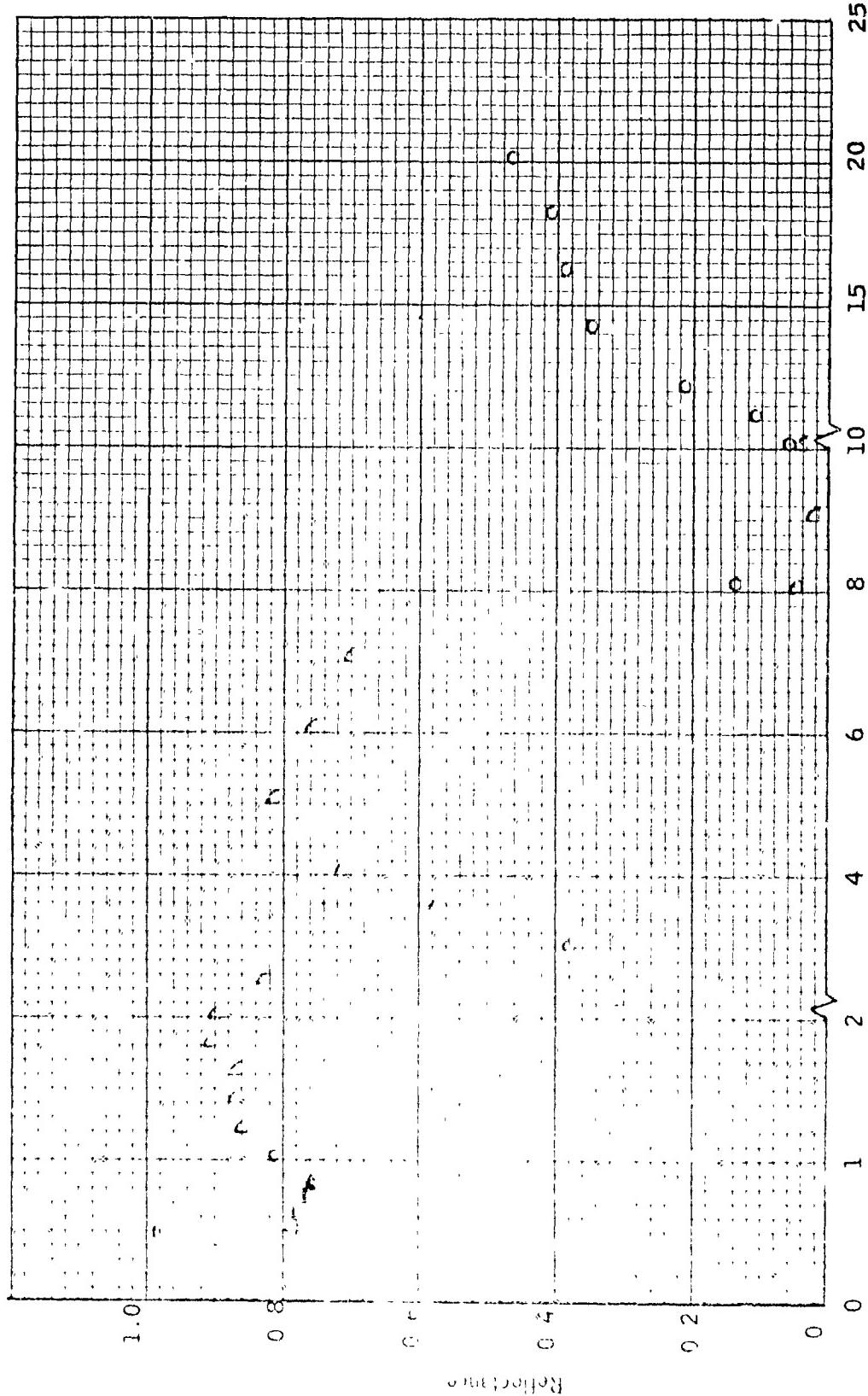


Fig. 1 Normal Spectral Reflectance of Specimen No. 31 Temperature RT  
Wavelength, Microns

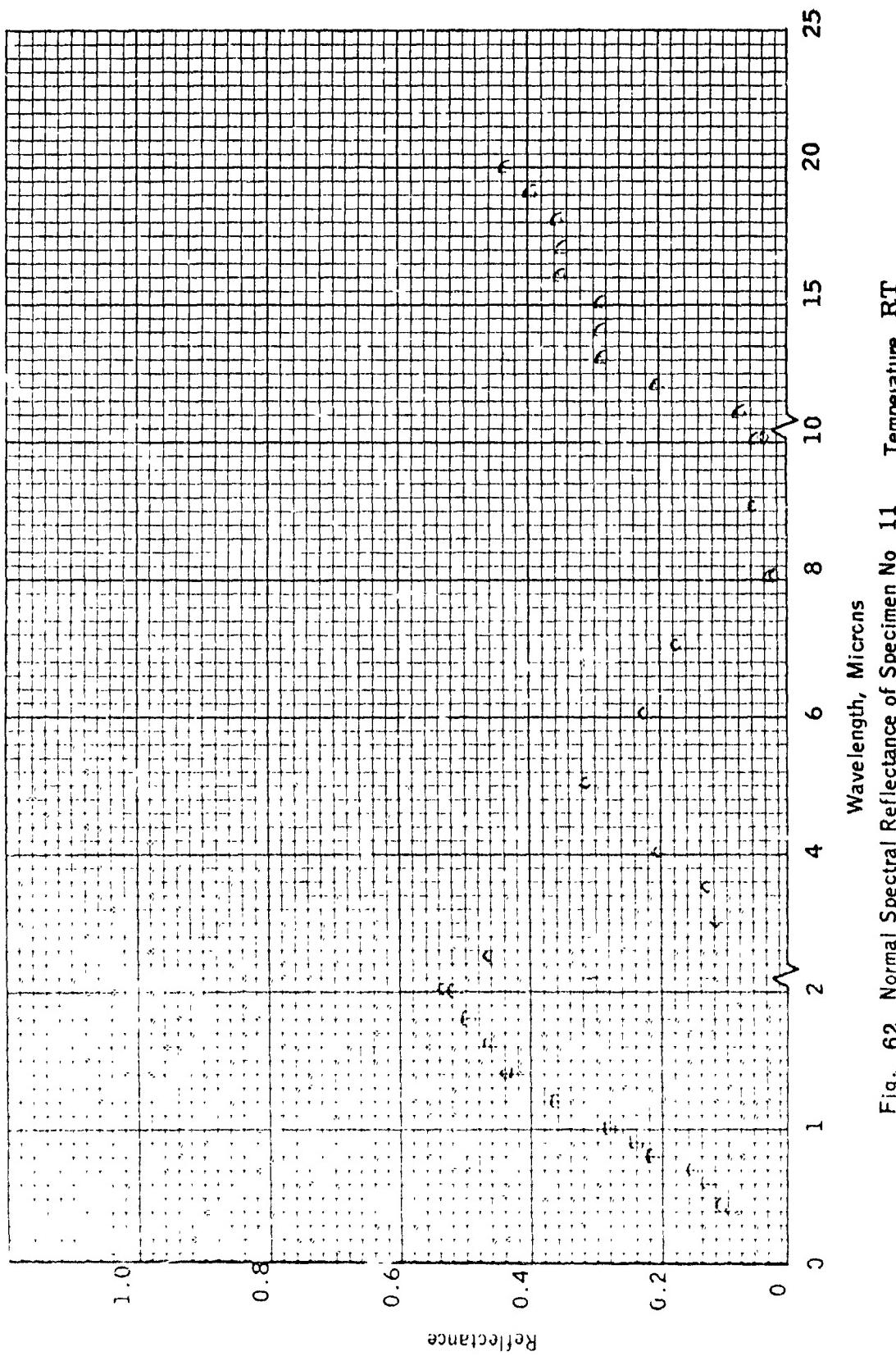


Fig. 62 Normal Spectral Reflectance of Specimen No. 11 Temperature RT

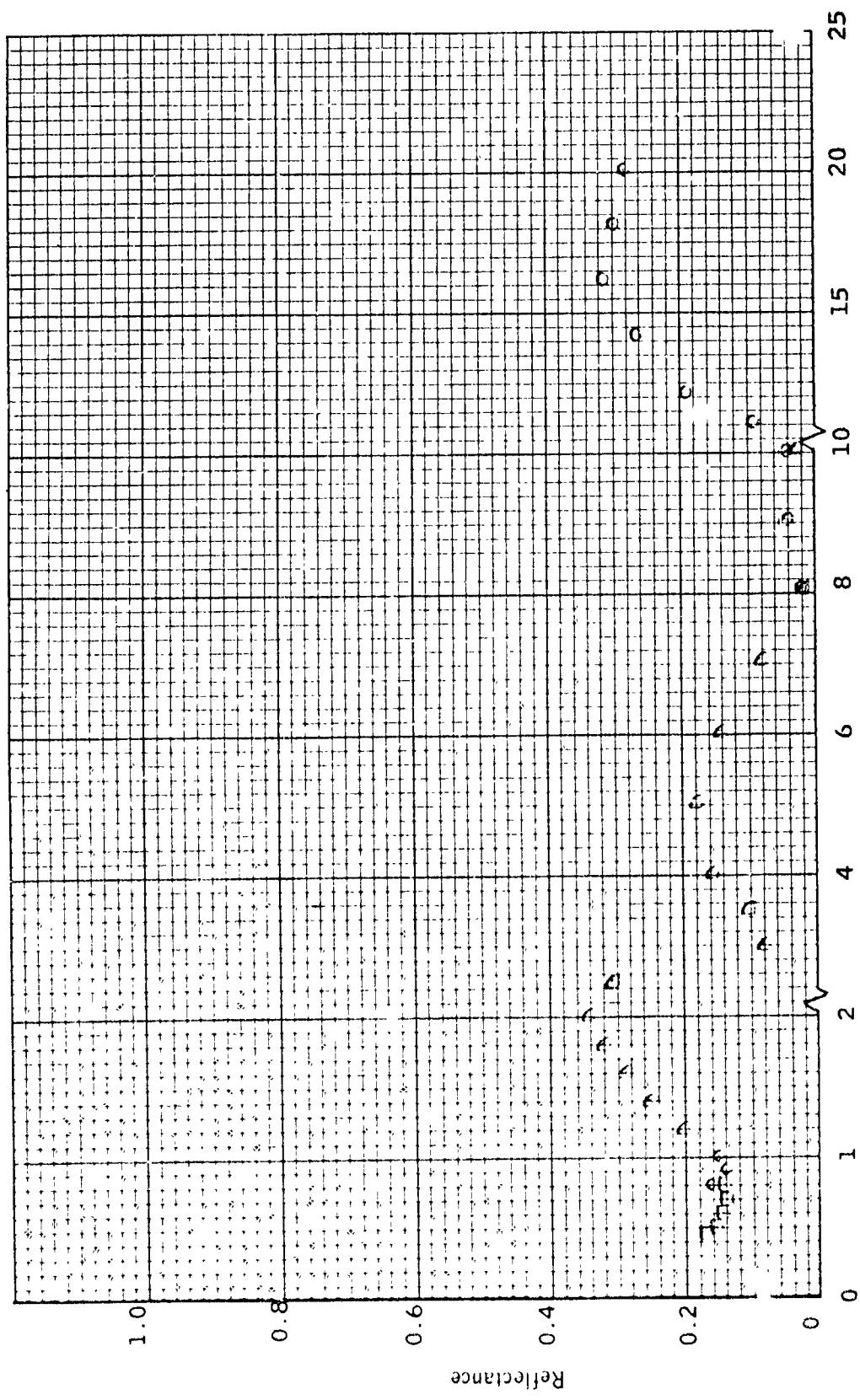


Fig. 63 Normal Spectral Reflectance of Specimen No. 85 Temperature RT

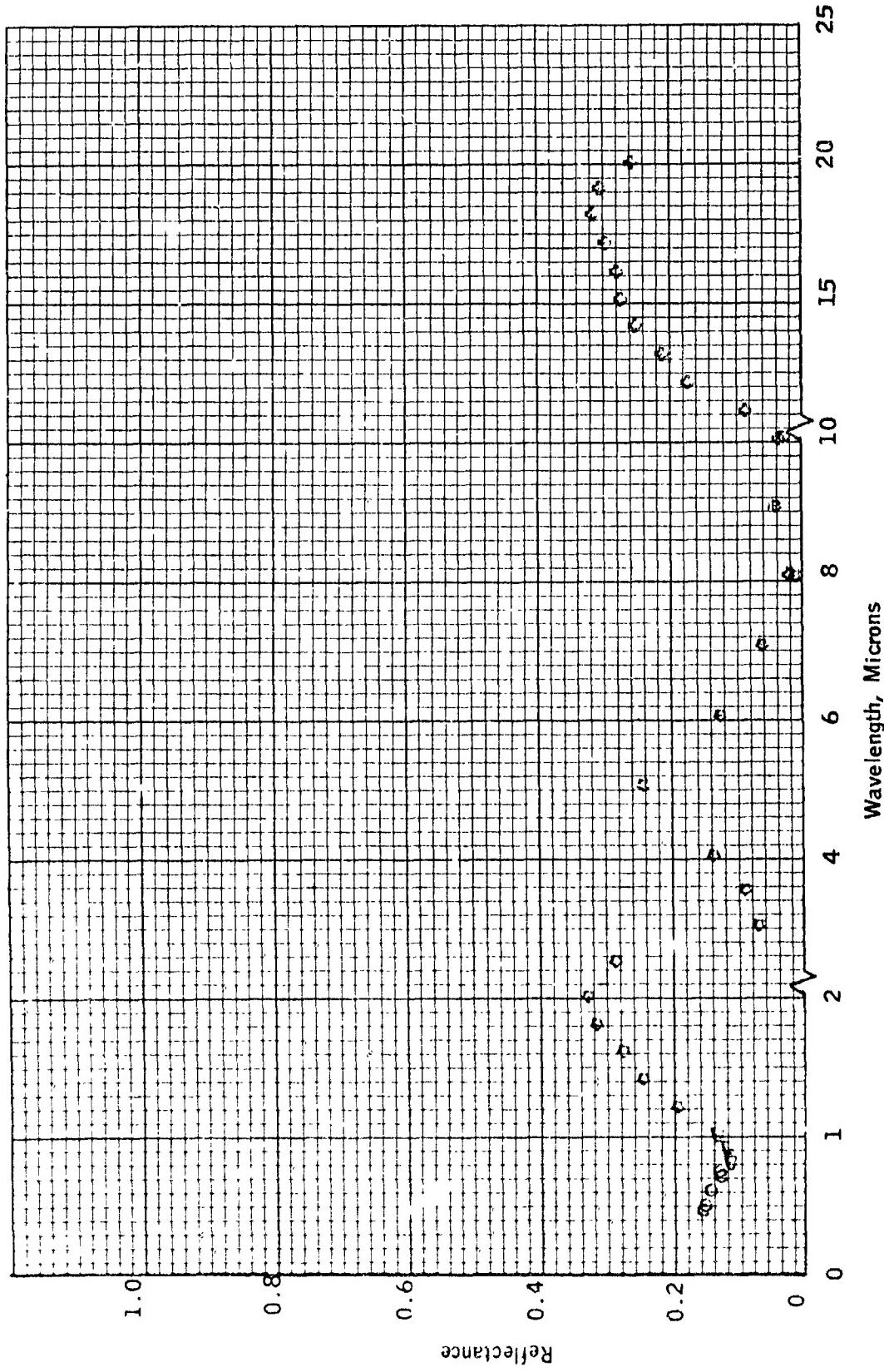


Fig. 64 Normal Spectral Reflectance of Specimen No. 86 Temperature RT

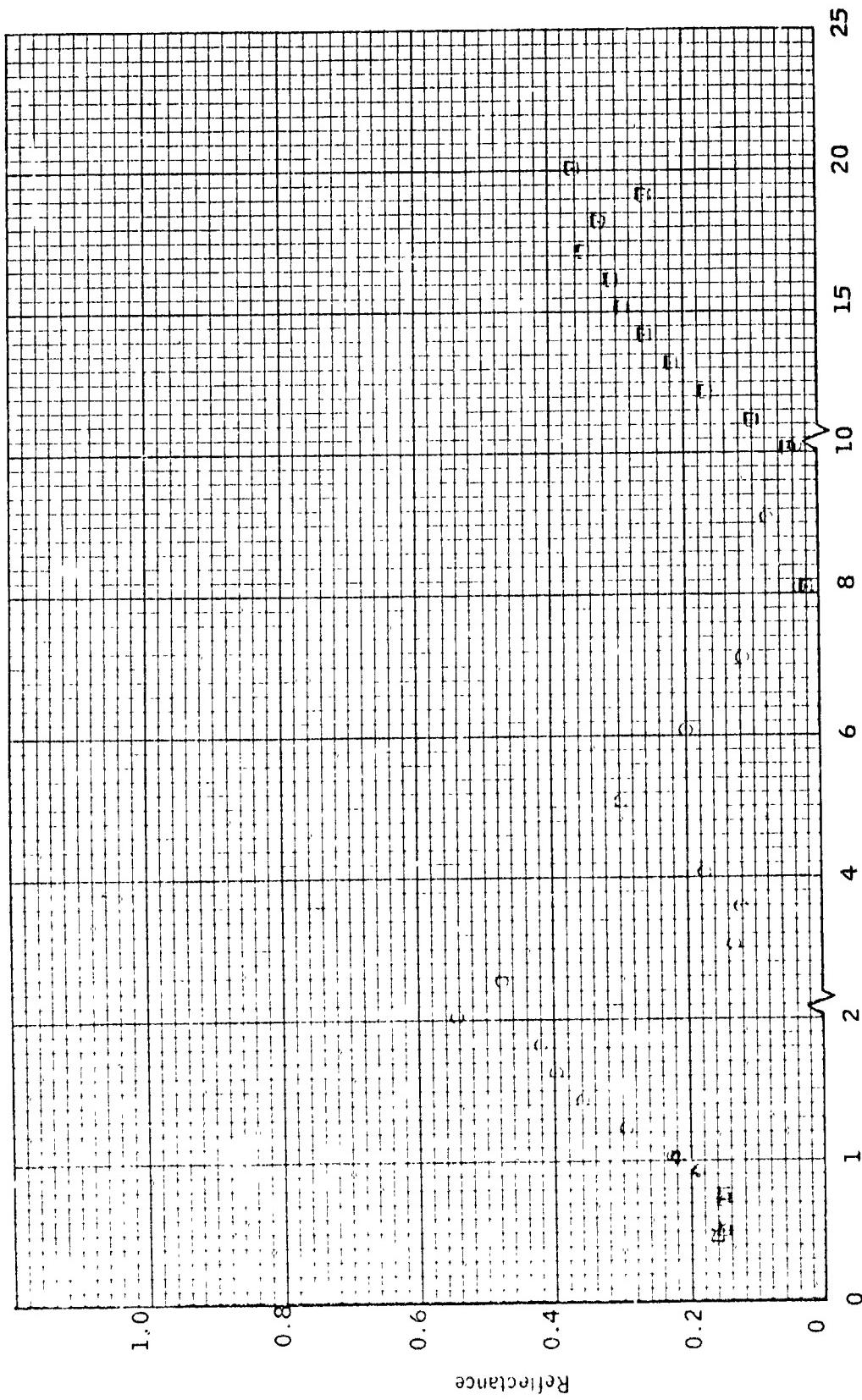


Fig. 65 Normal Spectral Reflectance of Specimen No. 87 Temperature RT

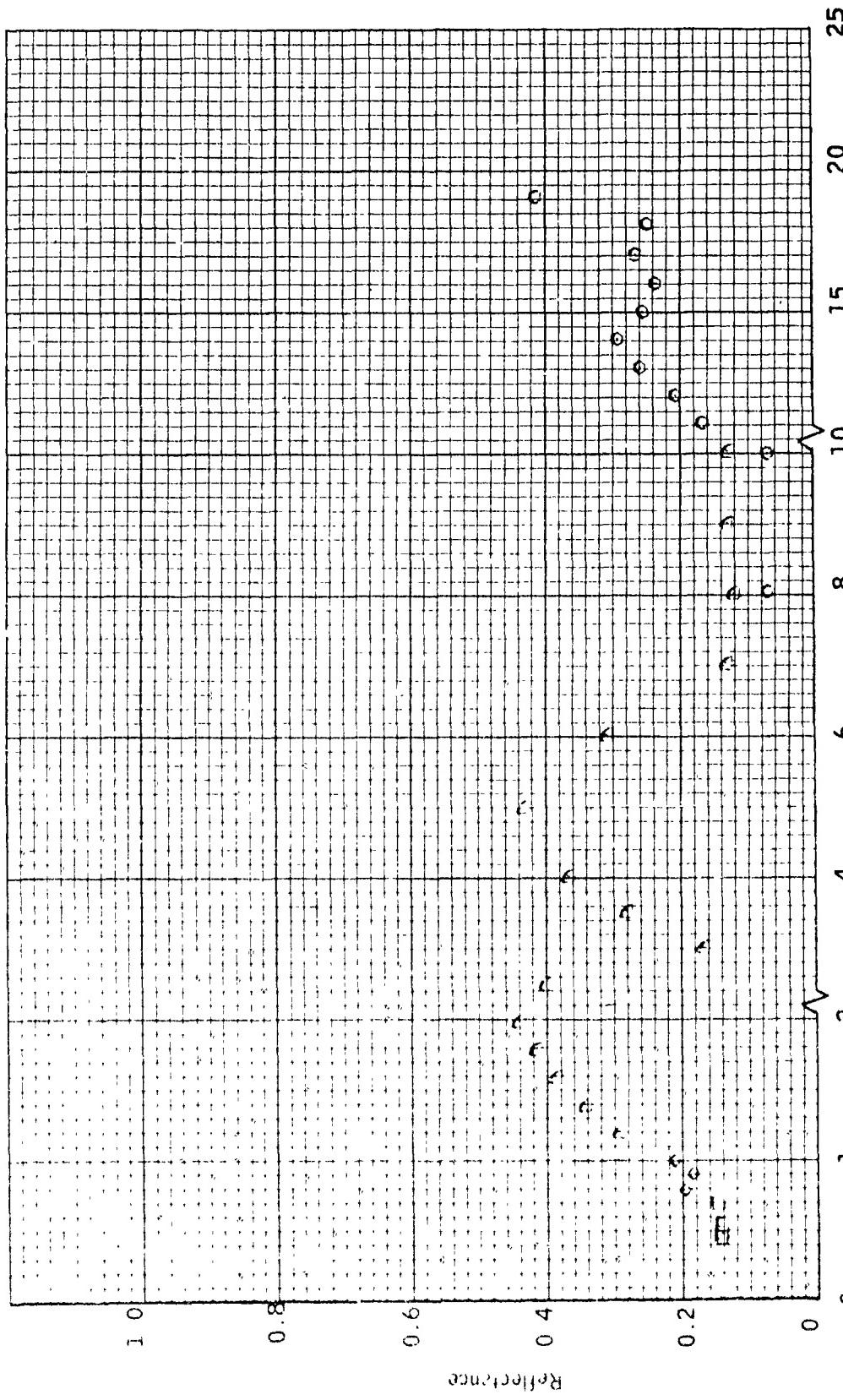


Fig. 66 Normal Spectral Reflectance of Specimen No. 87 Temperature 300°F

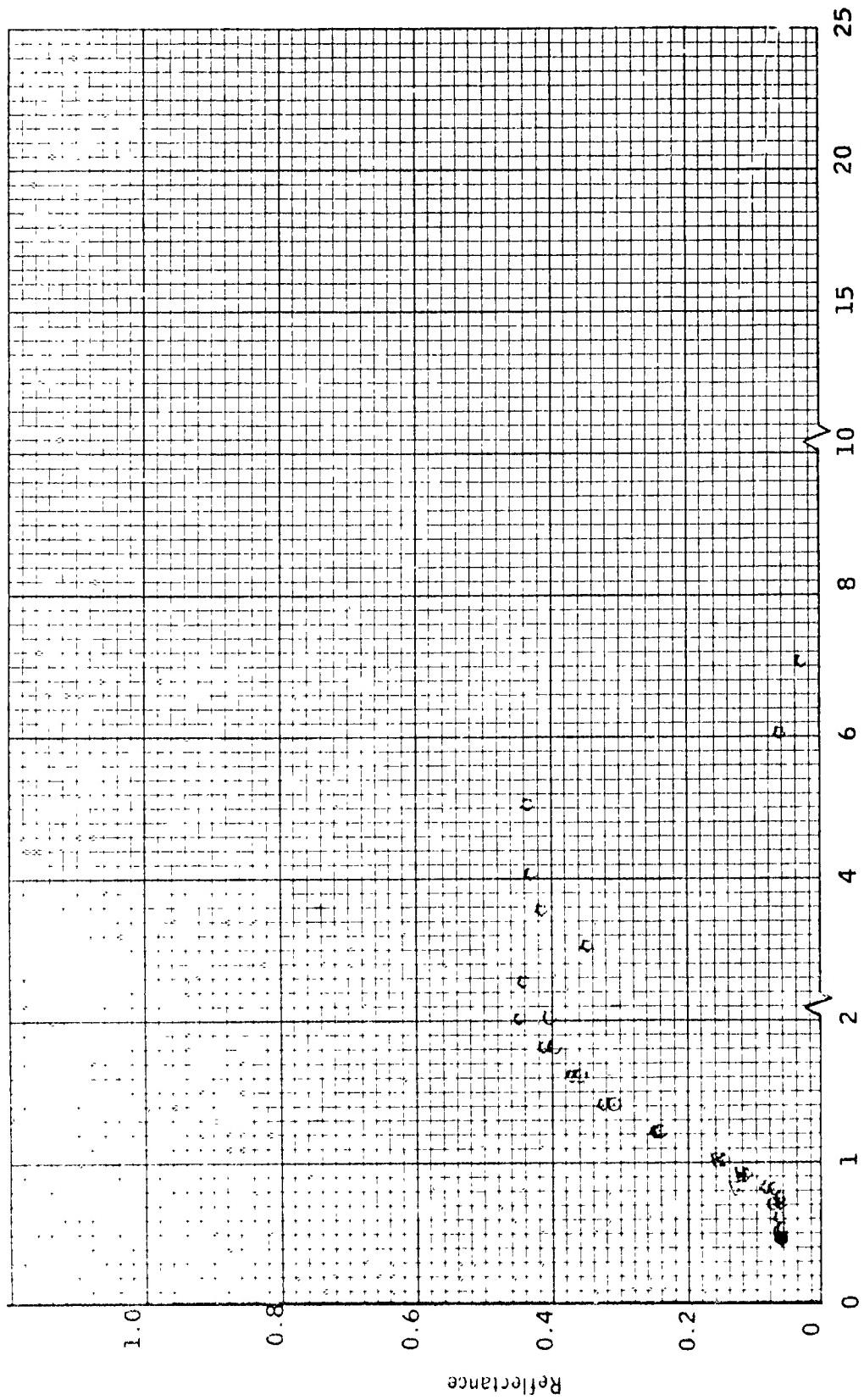


Fig. 67 Normal Spectral Reflectance of Specimen No. 87 Temperature 600°F

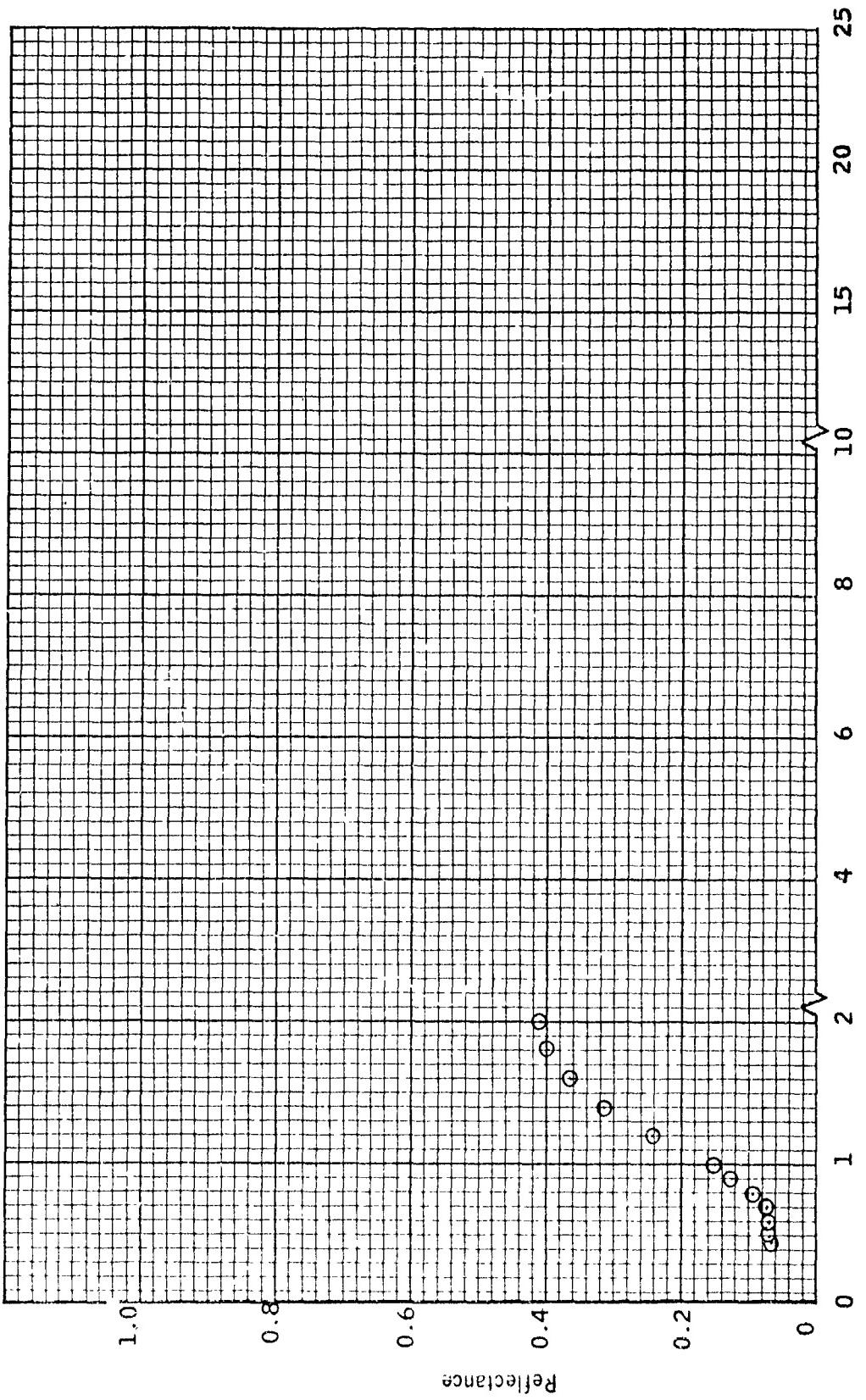


Fig. 68 Normal Spectral Reflectance of Specimen No 87 Temperature 825°F

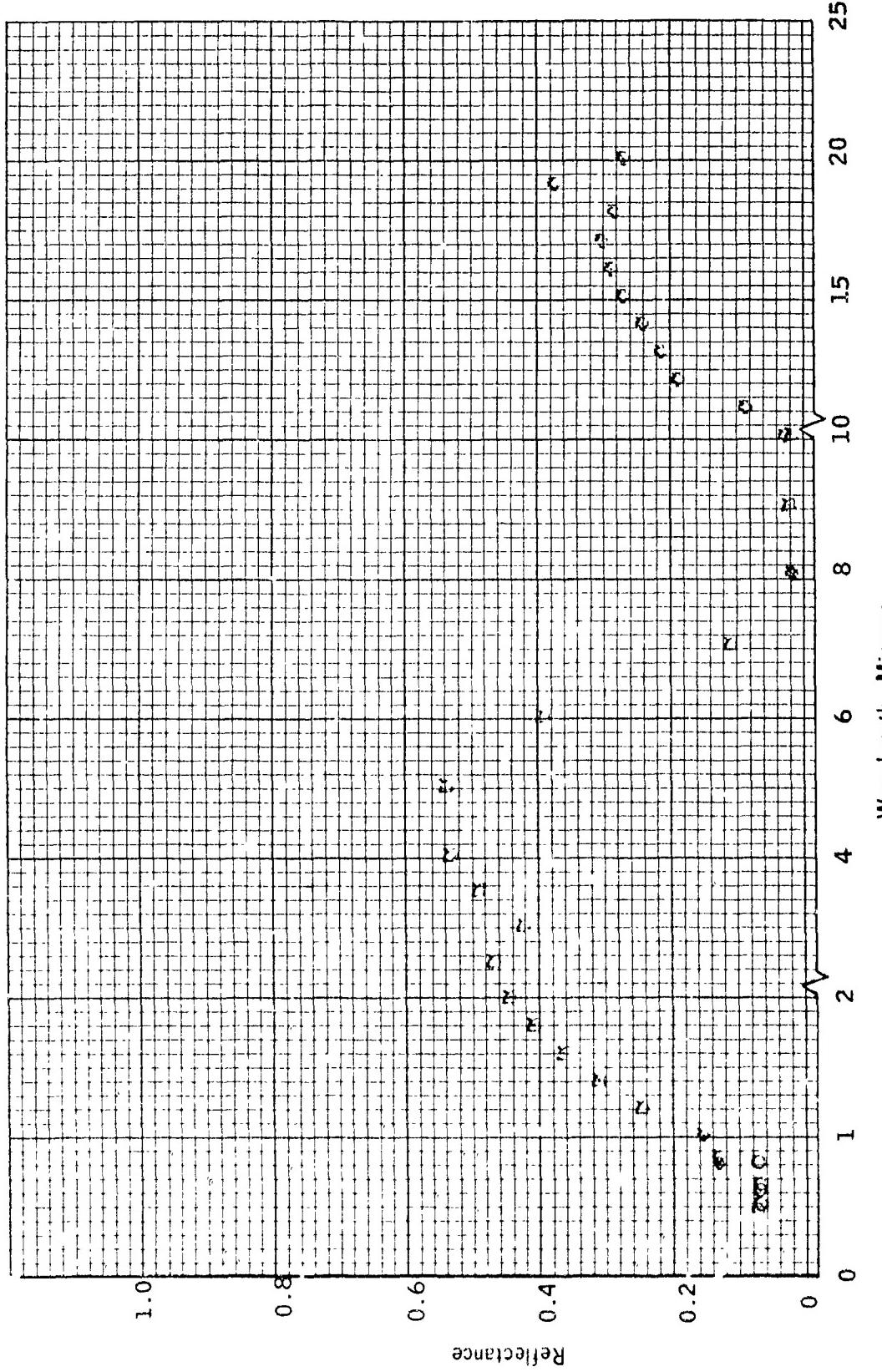


Fig. 69 Normal Spectral Reflectance of Specimen No 87 Temperature  $RT_f$

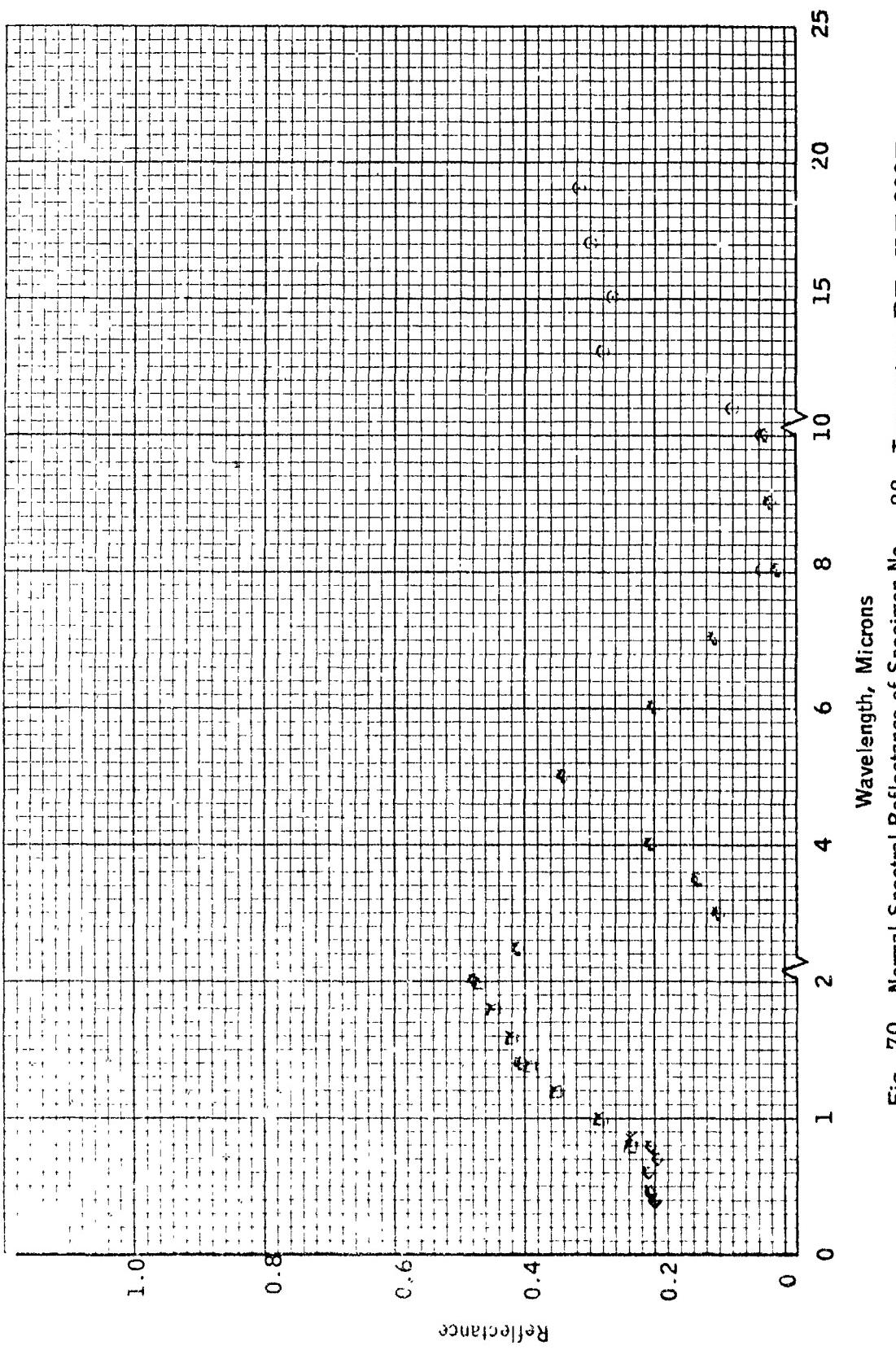


Fig. 70 Normal Spectral Reflectance of Specimen No 88 Temperature RT, HT-800F

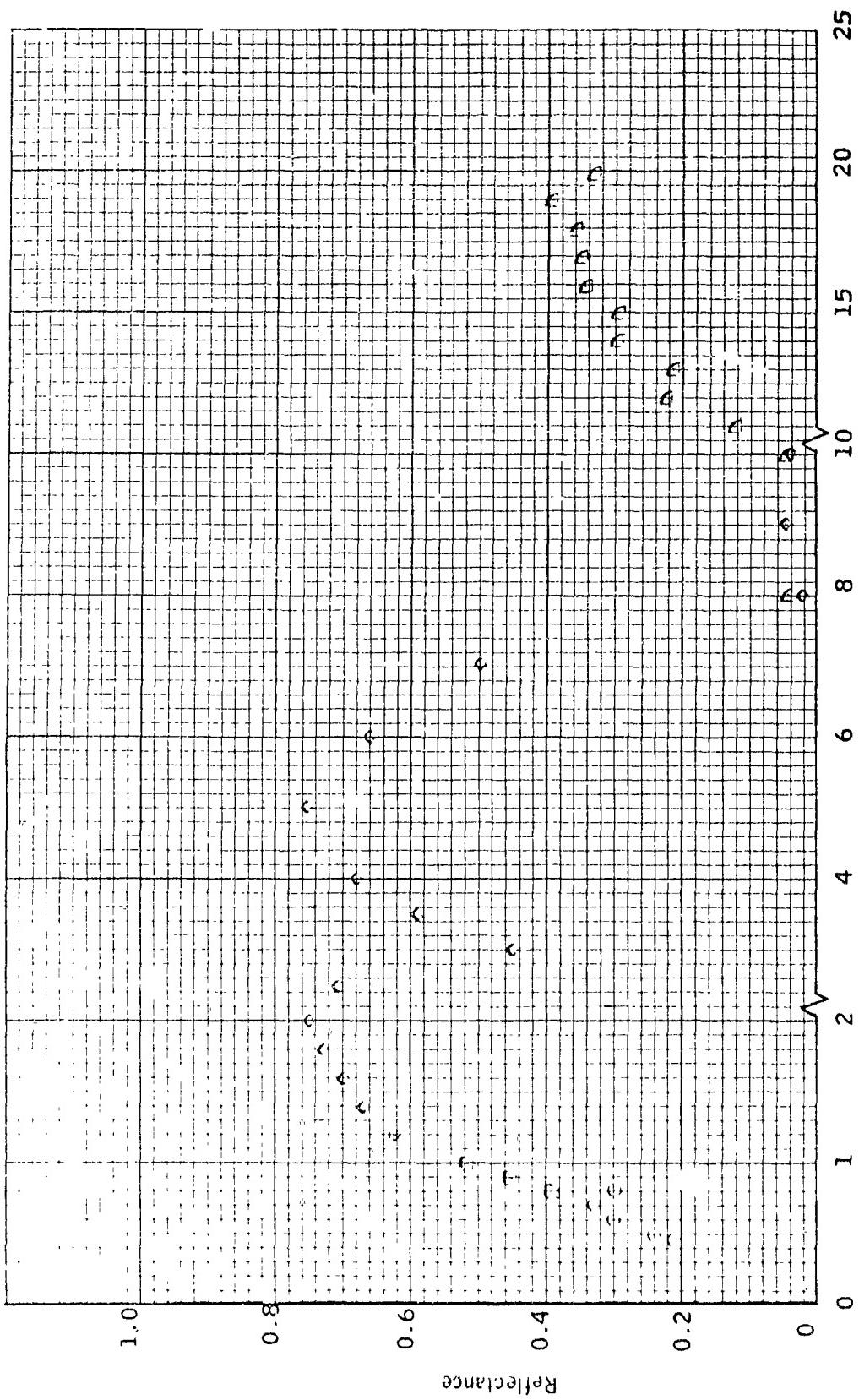


Fig. 71 Normal Spectral Reflectance of Specimen No. 89 Temperature RT

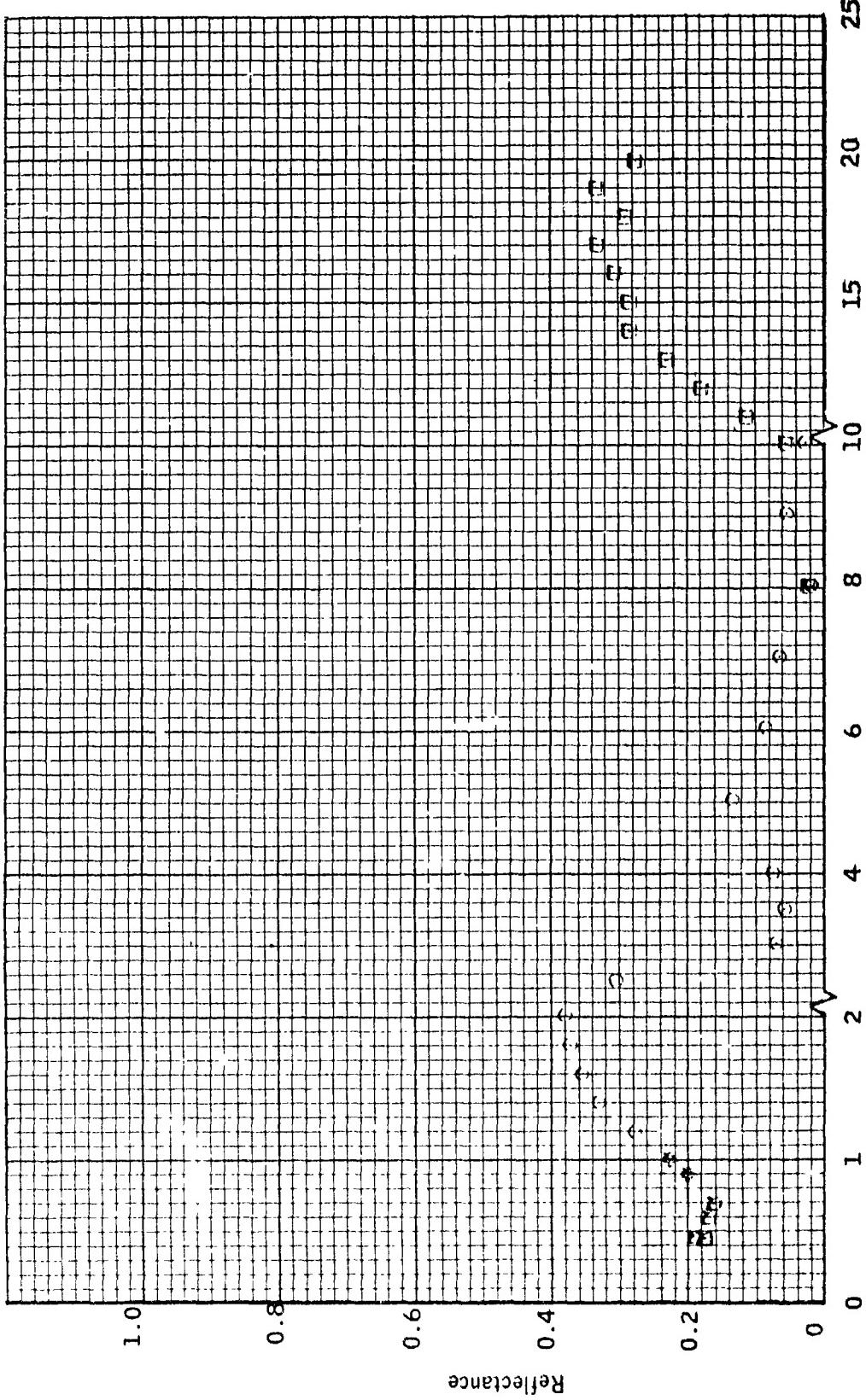


Fig. 72 Normal Spectral Reflectance of Specimen No 129 Temperature RT

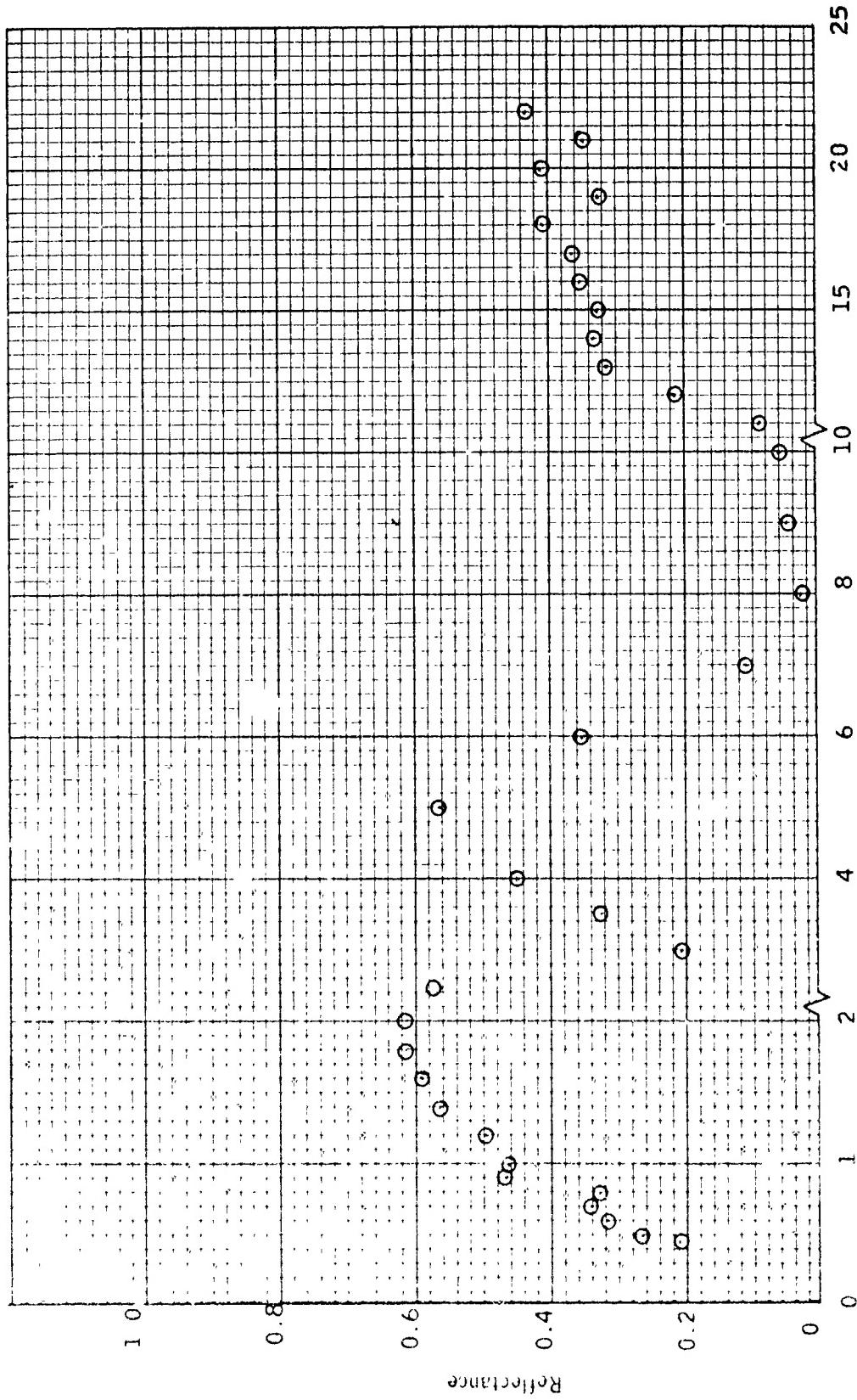


Fig. 73 Normal Spectral Reflectance of Specimen No. 13 Temperature RT

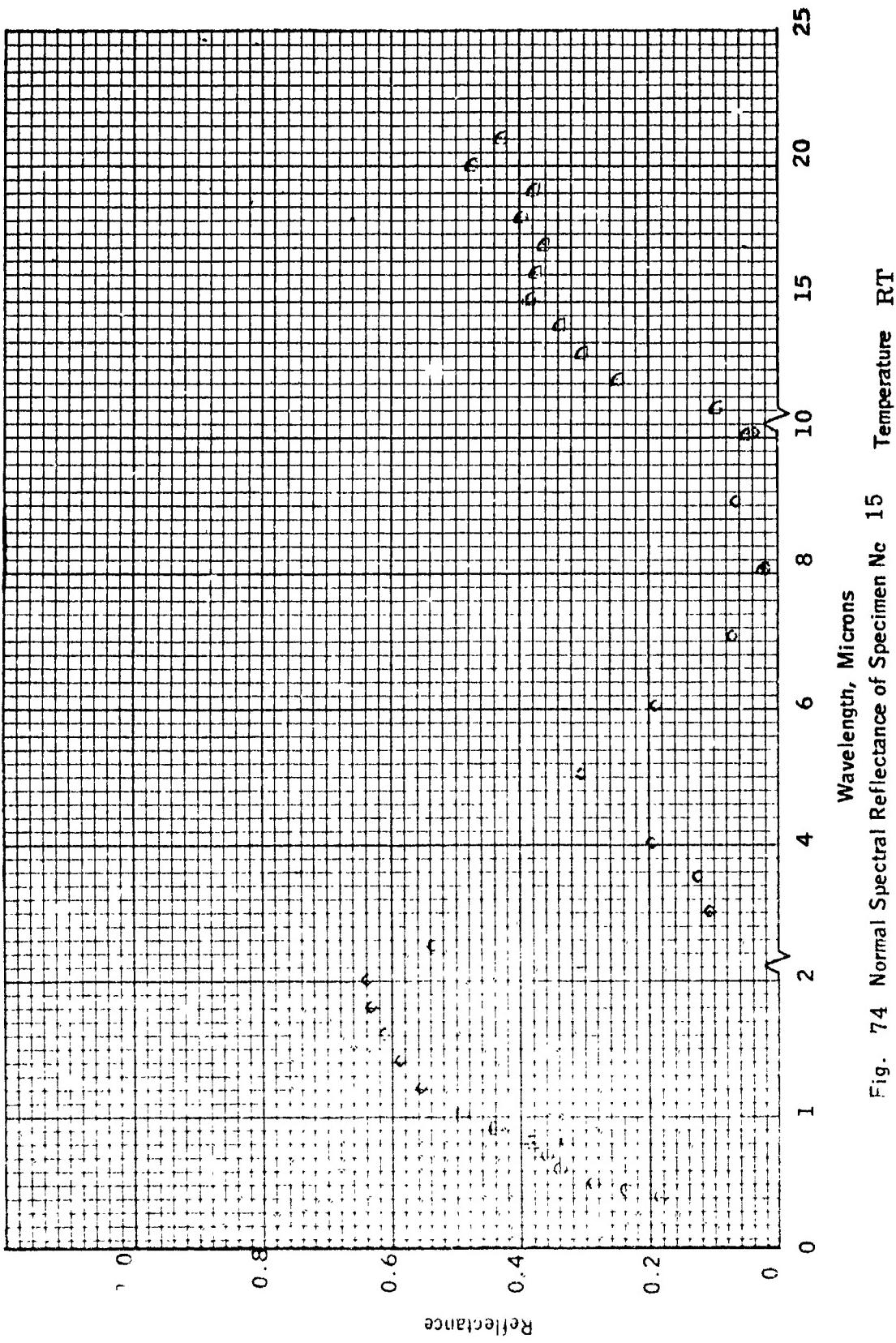


Fig. 74 Normal Spectral Reflectance of Specimen Nc 15 Temperature RT

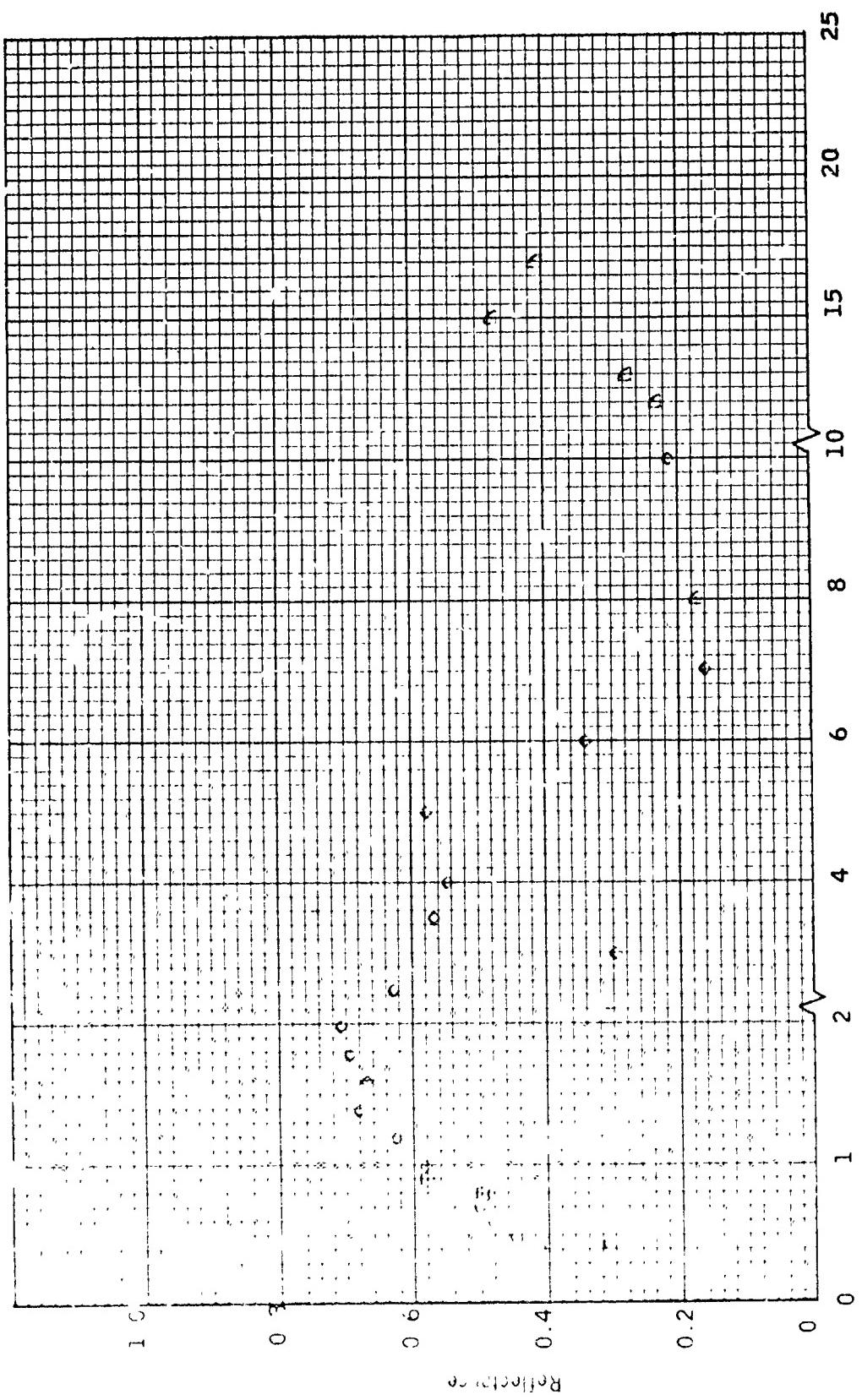


Fig. 75 Normal Spectral Reflectance of Specimen No. 15 Temperature 600°F

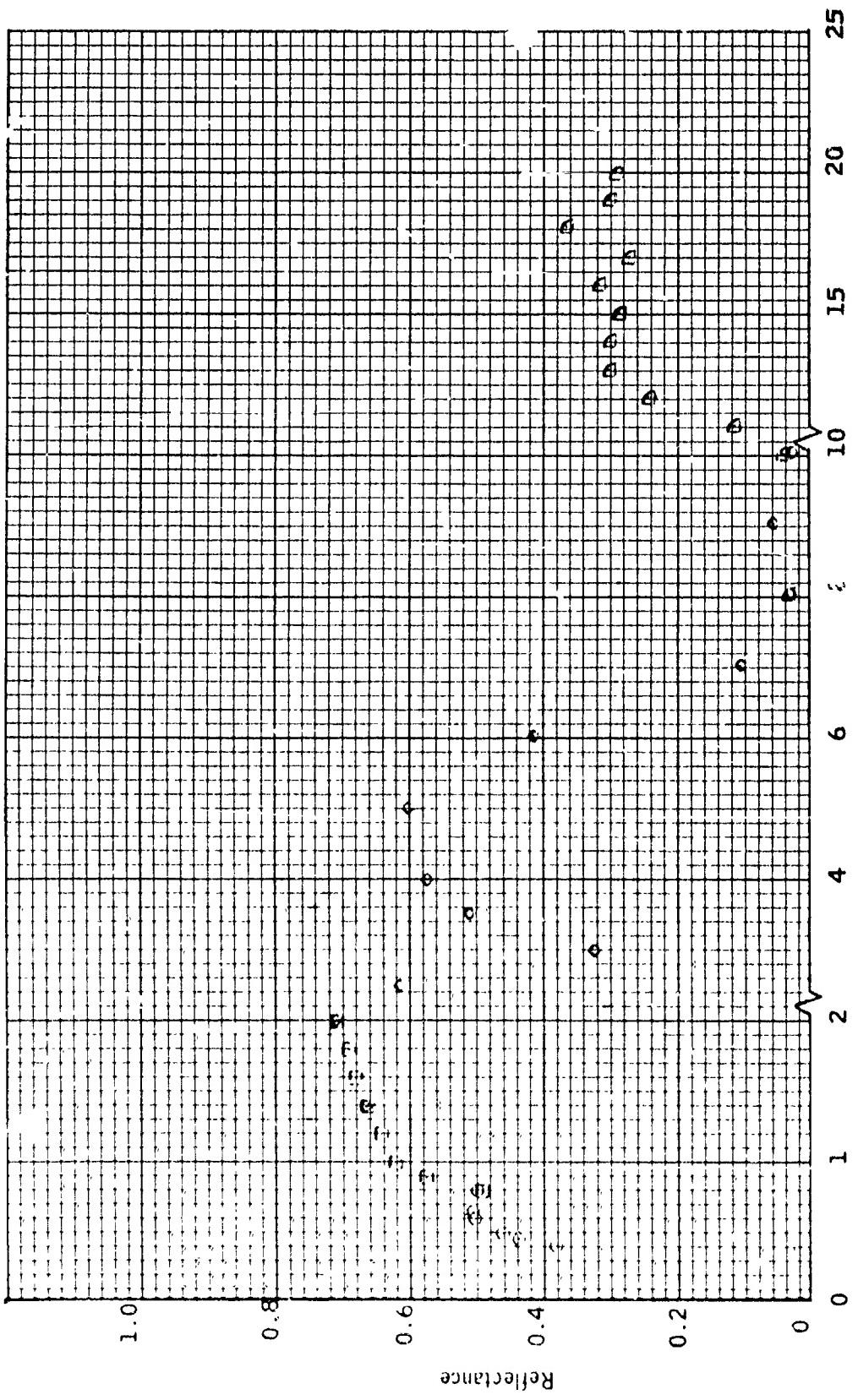


Fig. 76 Normal Spectral Reflectance of Specimen No 15 Temperature RT

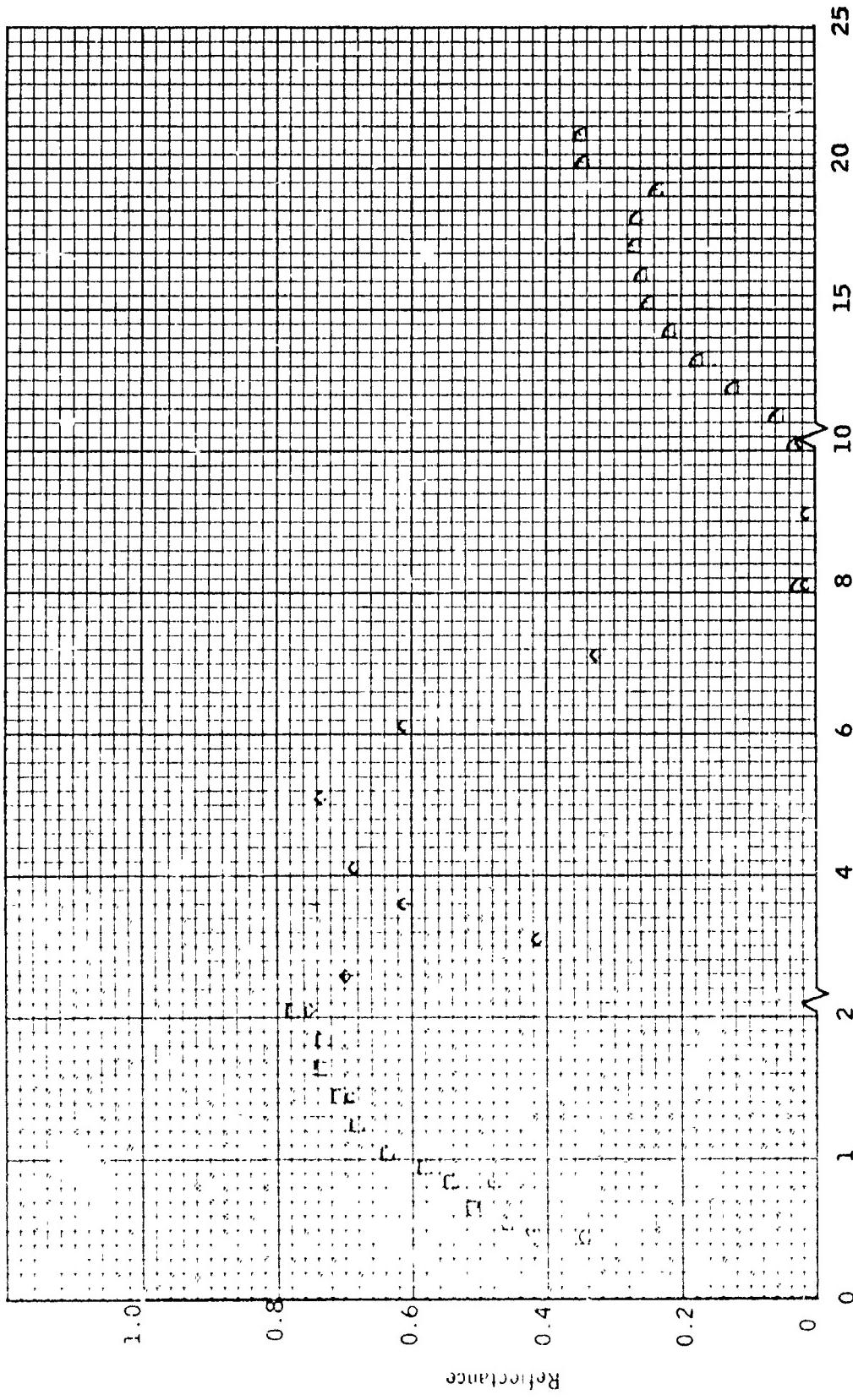


Fig. 77 Normal Spectral Reflectance of Specimen No. 20 Temperature RT

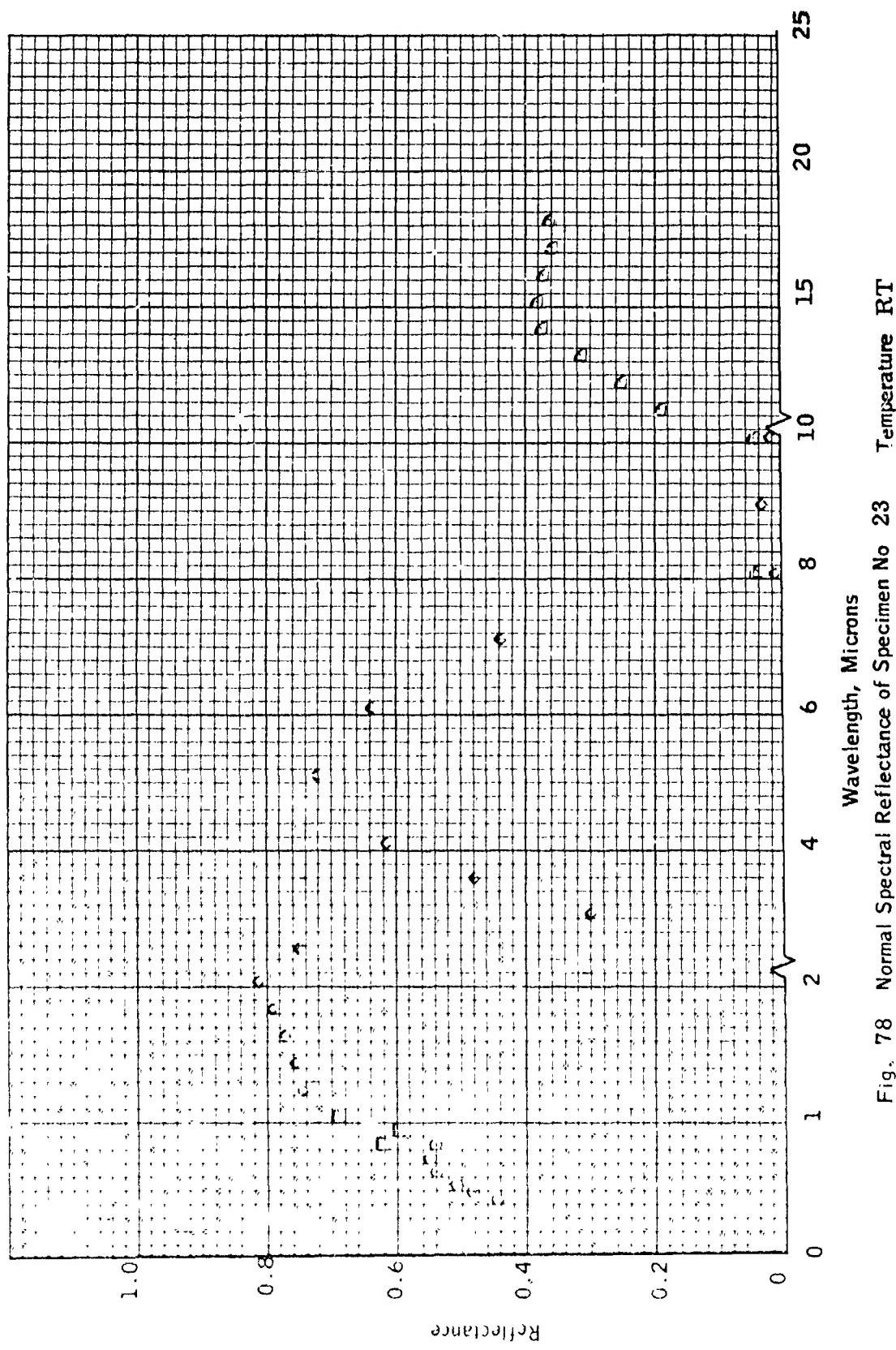


Fig. 78 Normal Spectral Reflectance of Specimen No. 23 Temperature RT

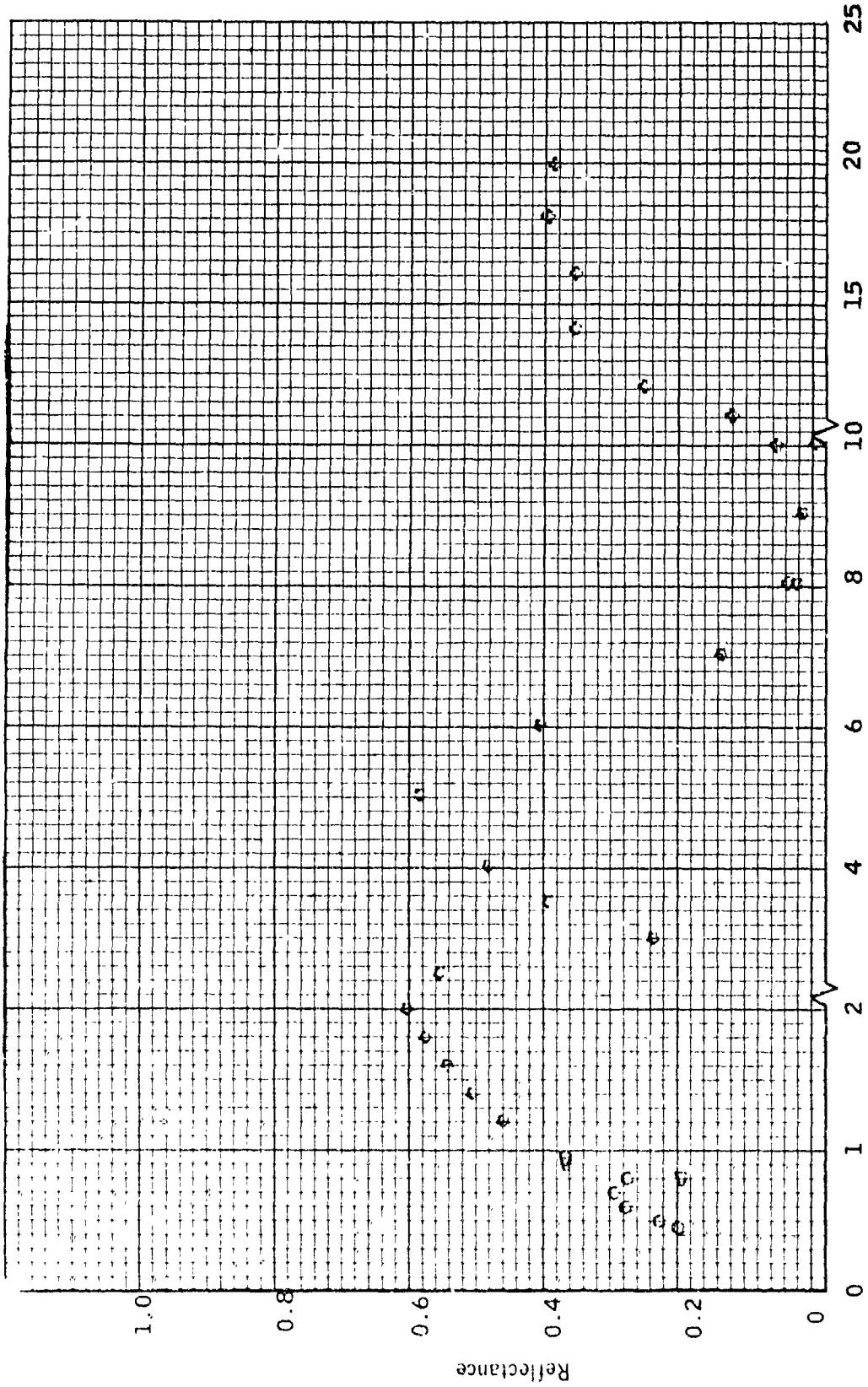


Fig. 79 Normal Spectral Reflectance of Specimen No. 82 Temperature RT

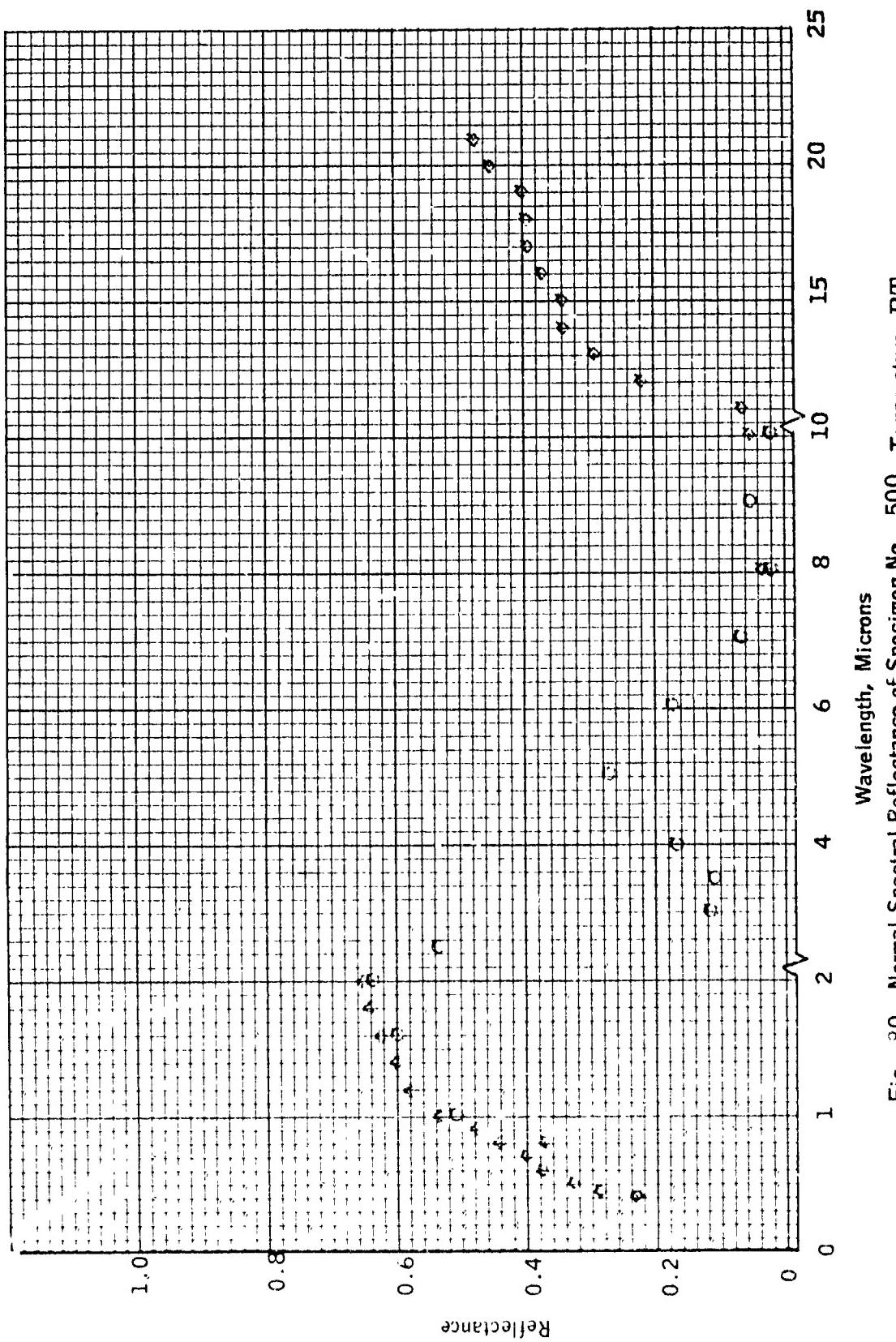


Fig. 80 Normal Spectral Reflectance of Specimen No 500 Temperature RT

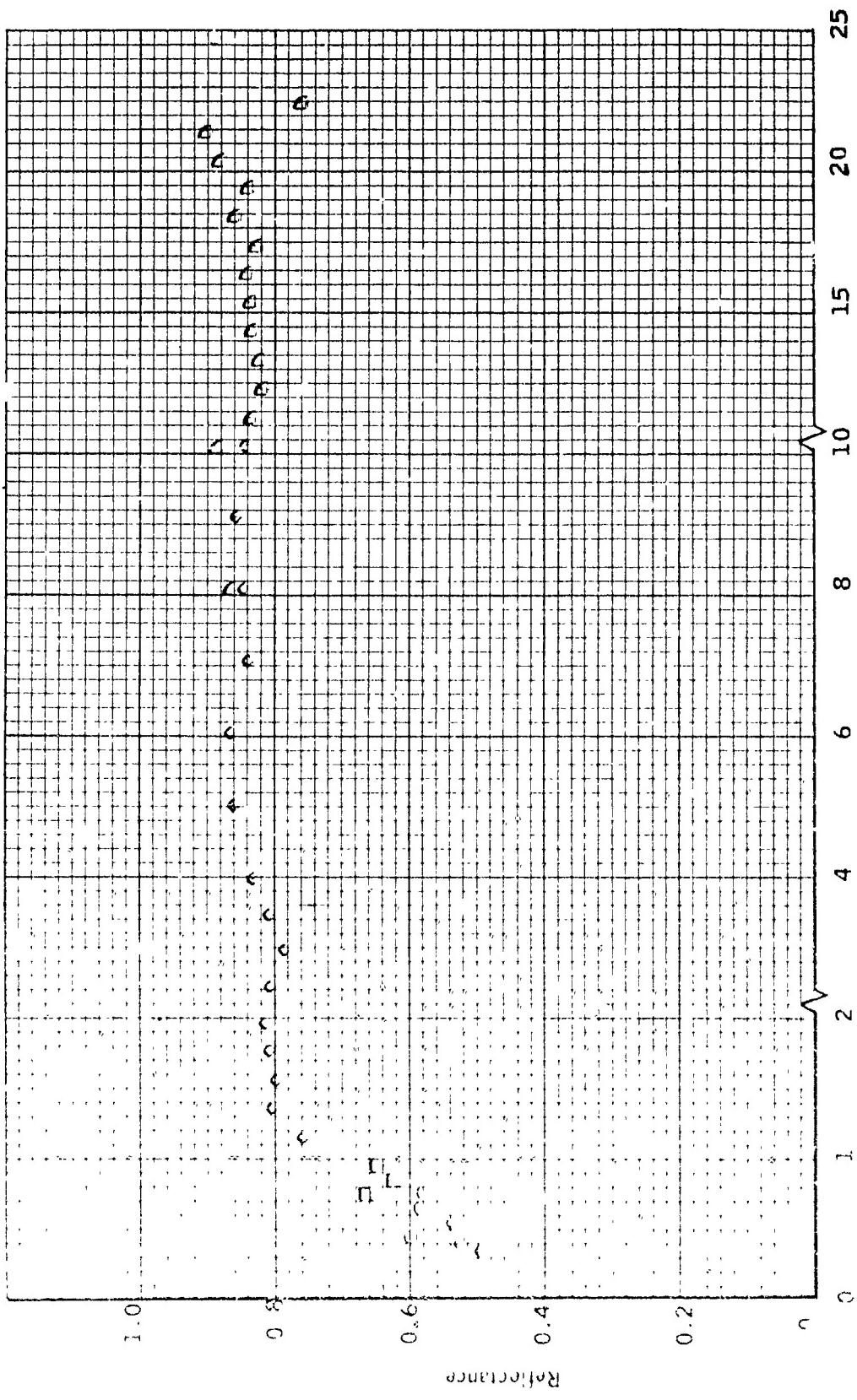


Fig. 81 Normal Spectral Reflectance of Specimen No. 52 Temperature RT

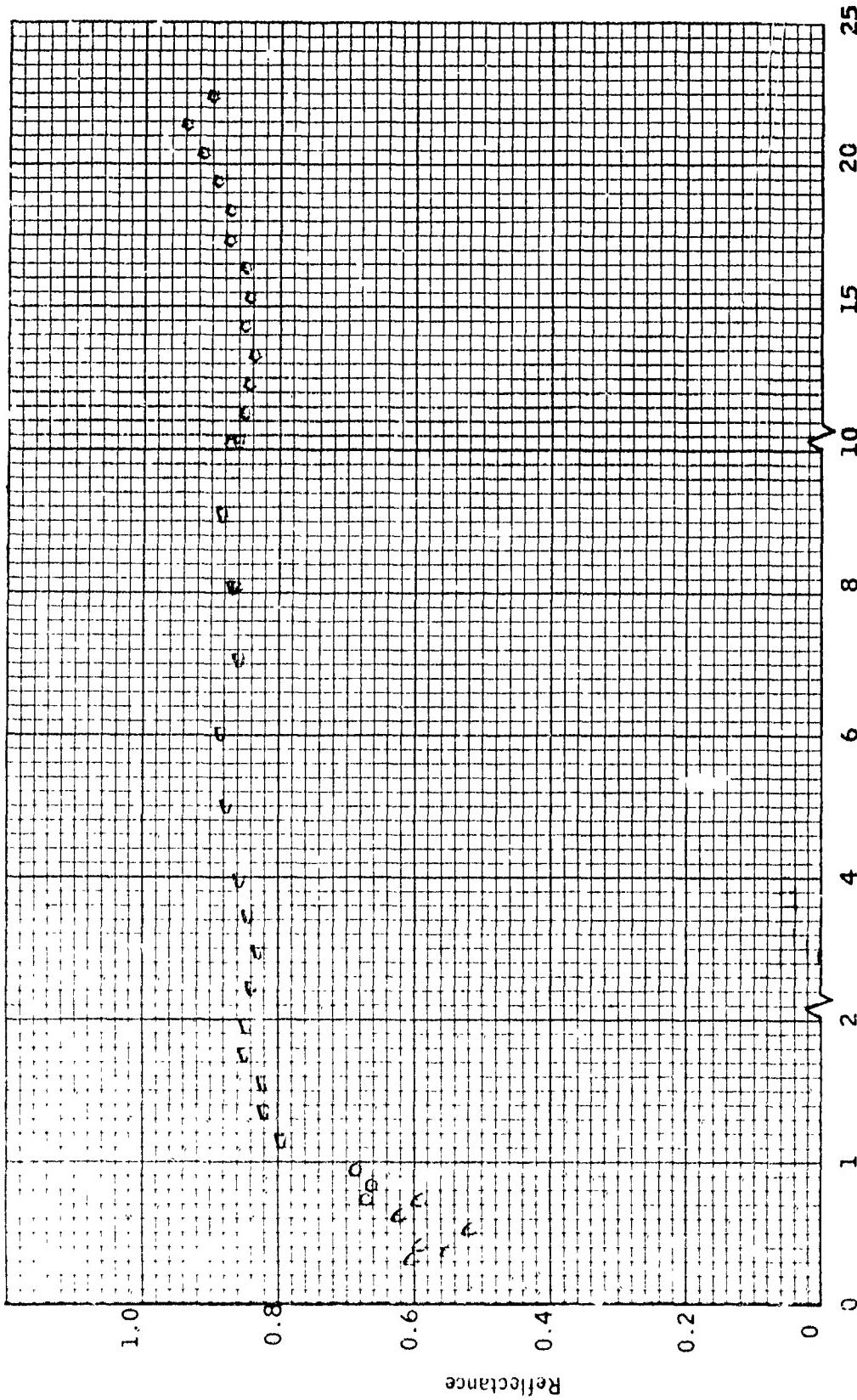


Fig. 82 Normal Spectral Reflectance of Specimen No. 52 Temperature 300°F  
Wavelength, Microns

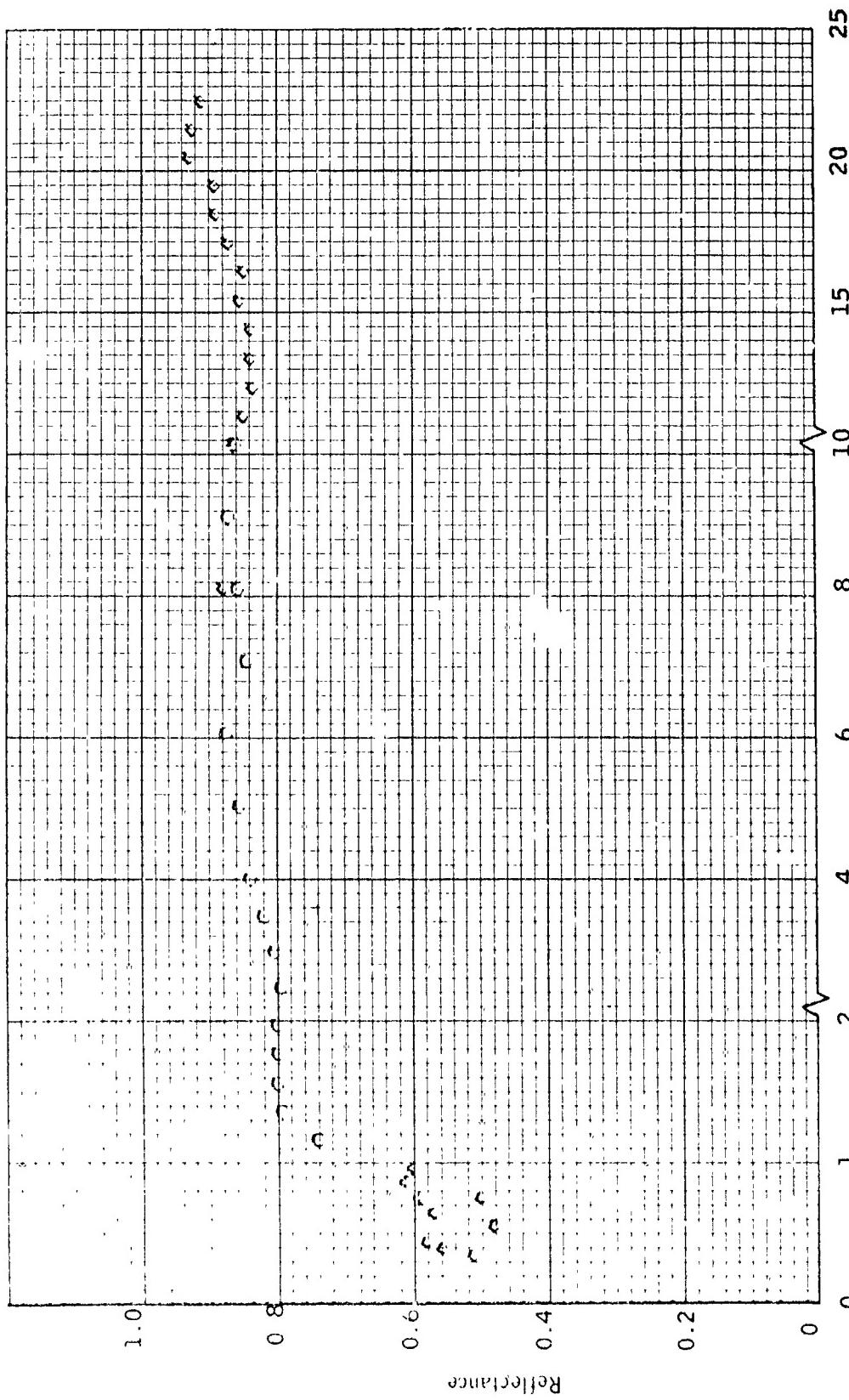


Fig. 83 Normal Spectral Reflectance of Specimen No. 52 Temperature 600°F

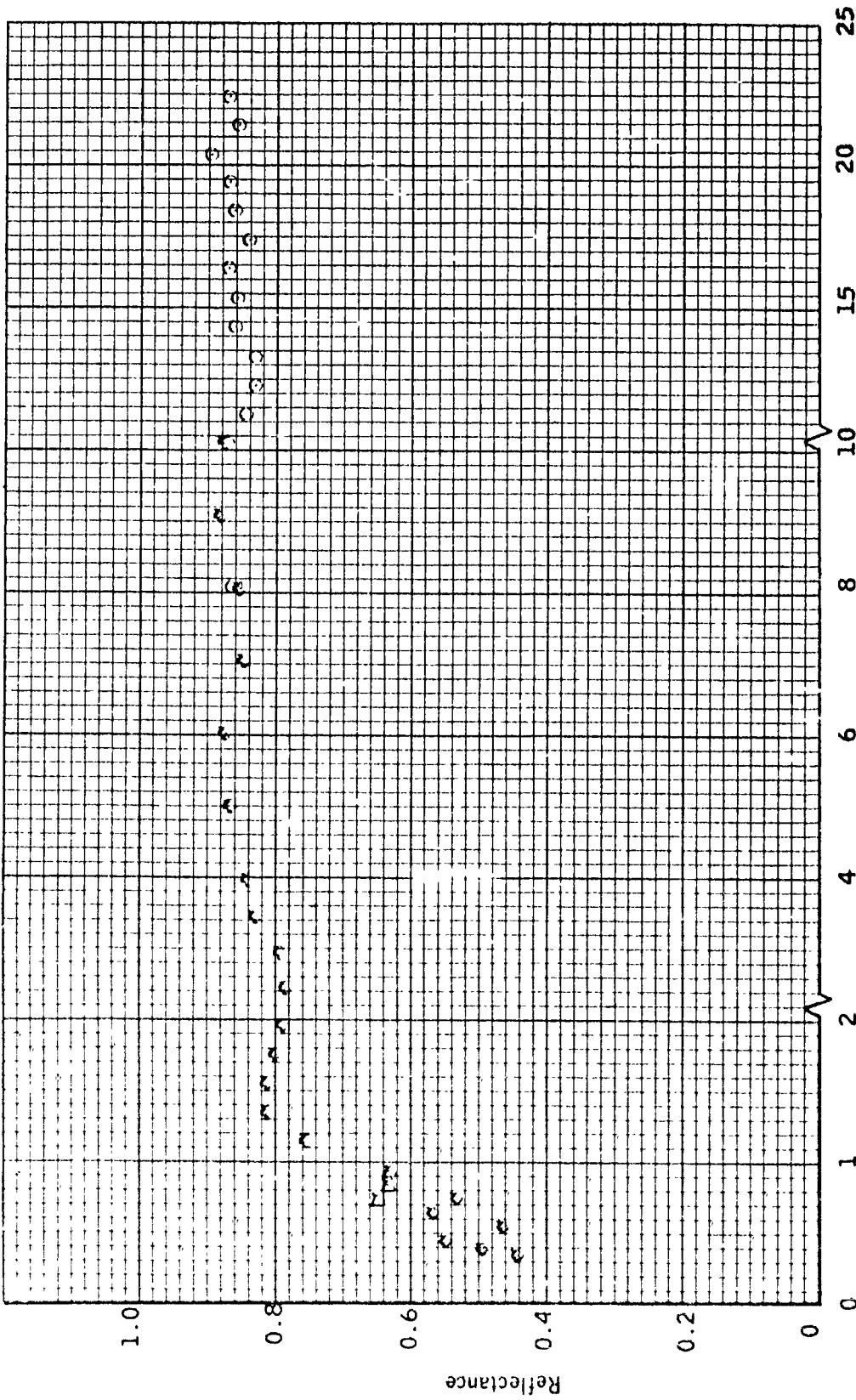


Fig. 84 Normal Spectral Reflectance of Specimen No 52 Temperature 825°F

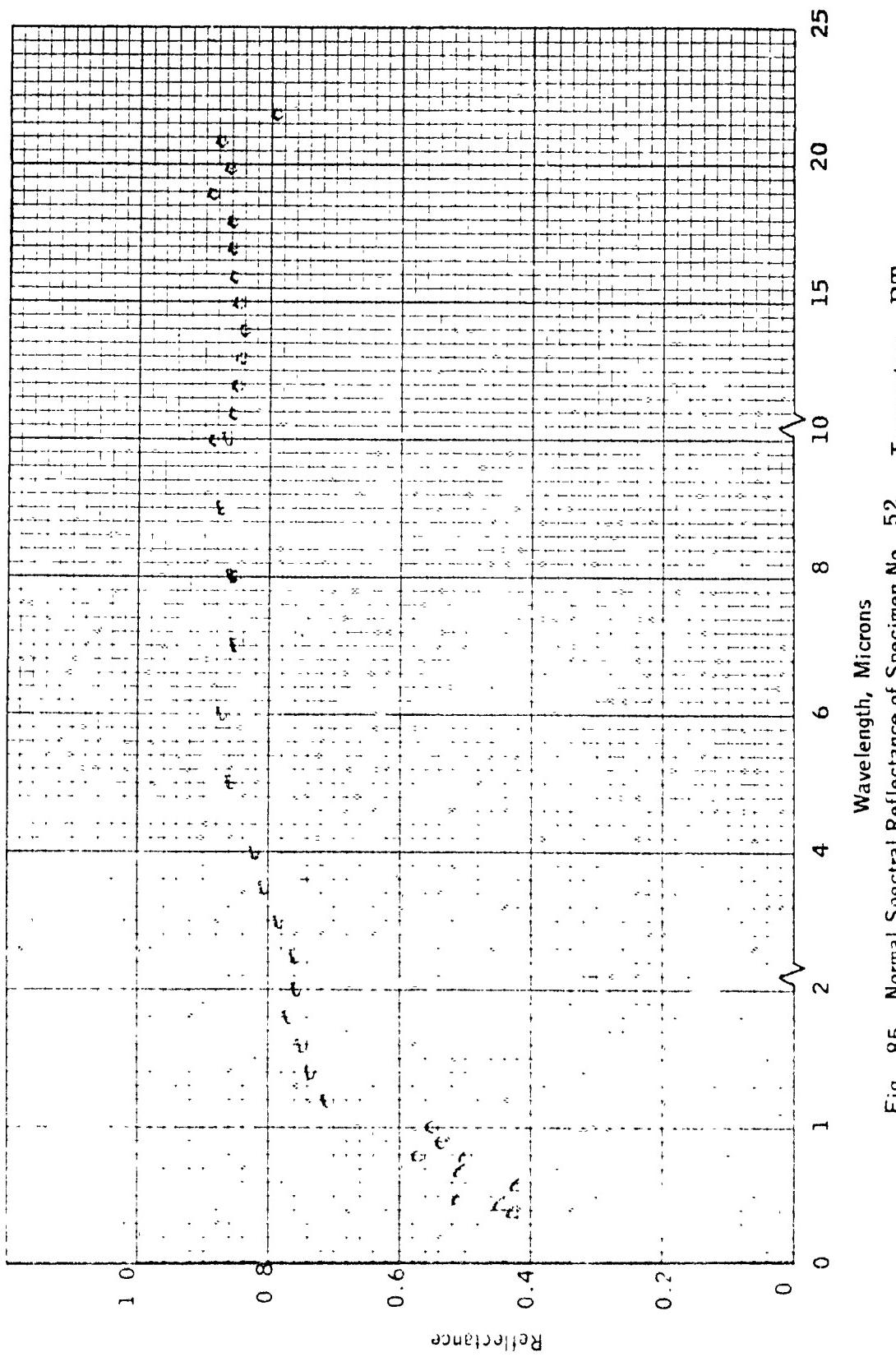


Fig. 85 Normal Spectral Reflectance of Specimen No. 52 Temperature  $RT_f$

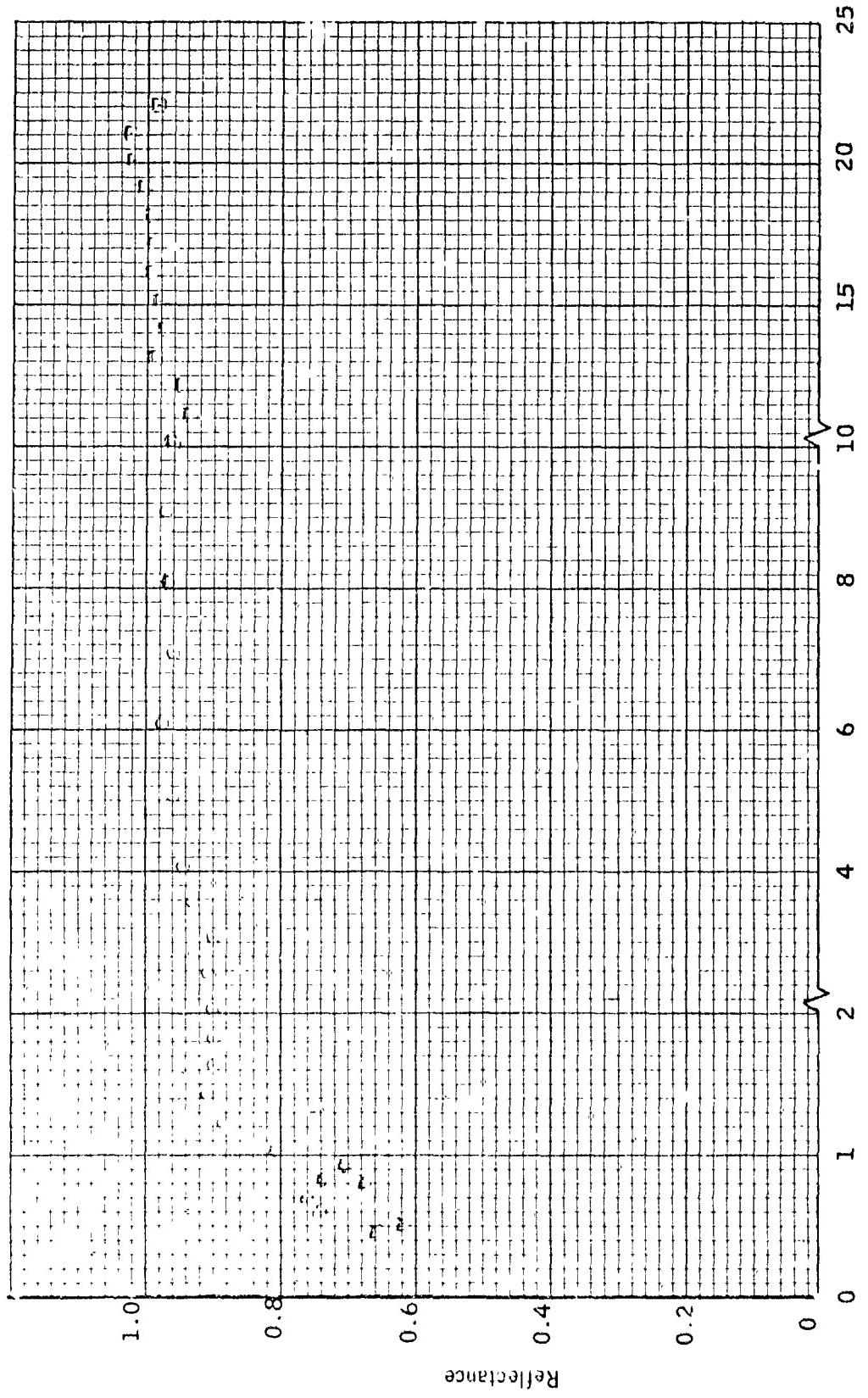


Fig. 87 Normal Spectral Reflectance of Specimen No. 62 Temperature RT

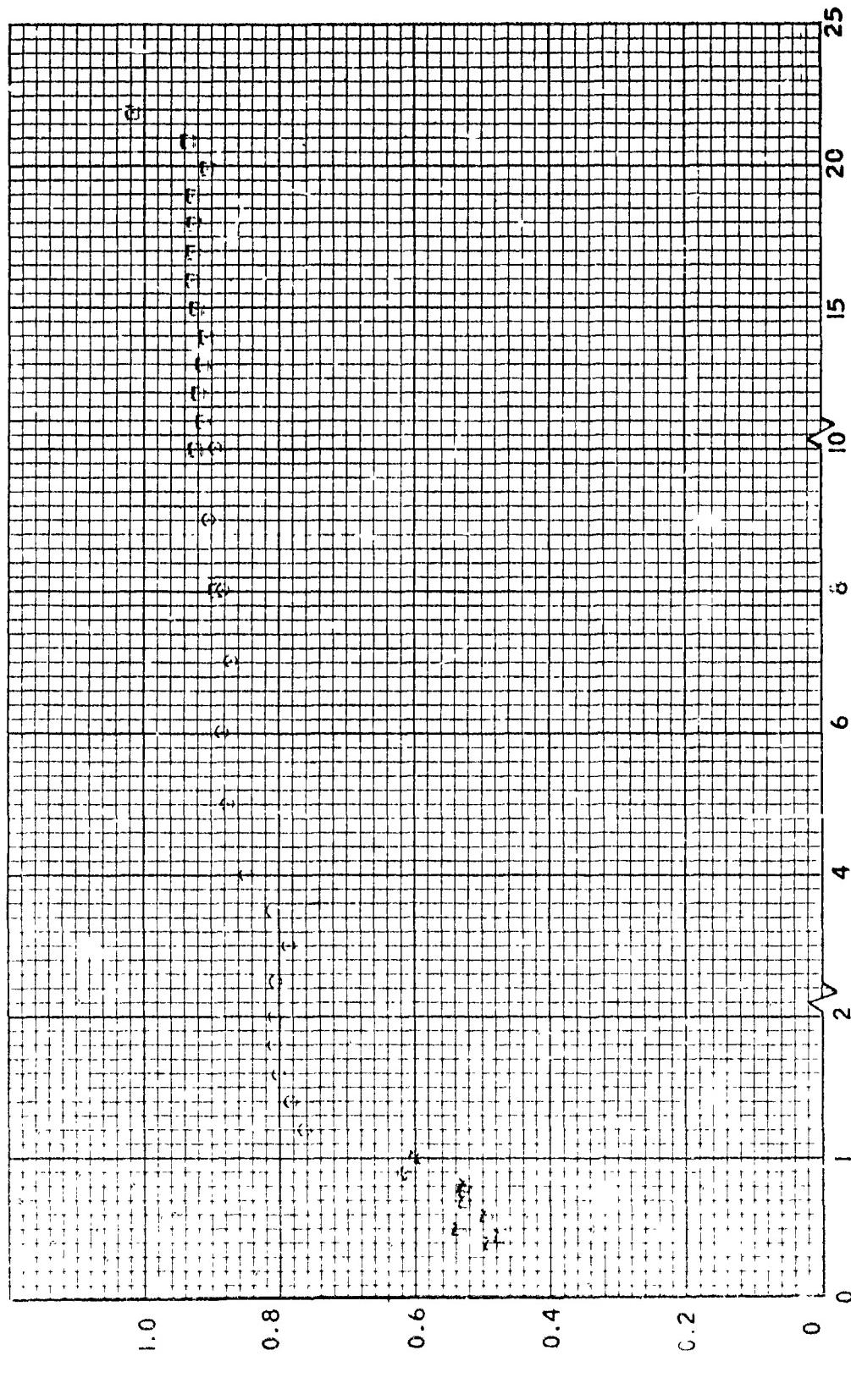


Fig. 88 Normal Spectral Reflectance of Specimen No 115 Temperature RT

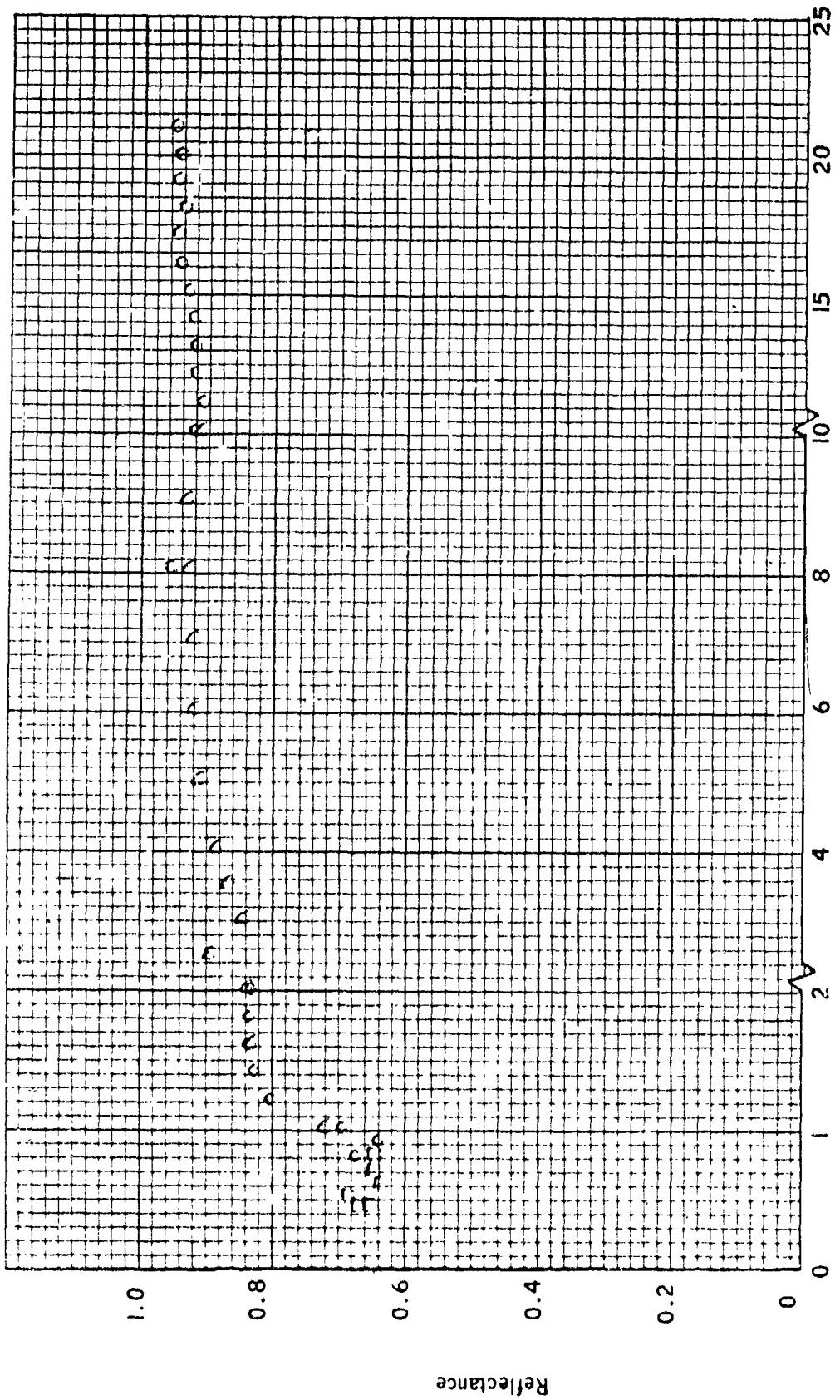


Fig. 89 Normal Spectral Reflectance of Specimen No 58 Temperature RT, HT-800F

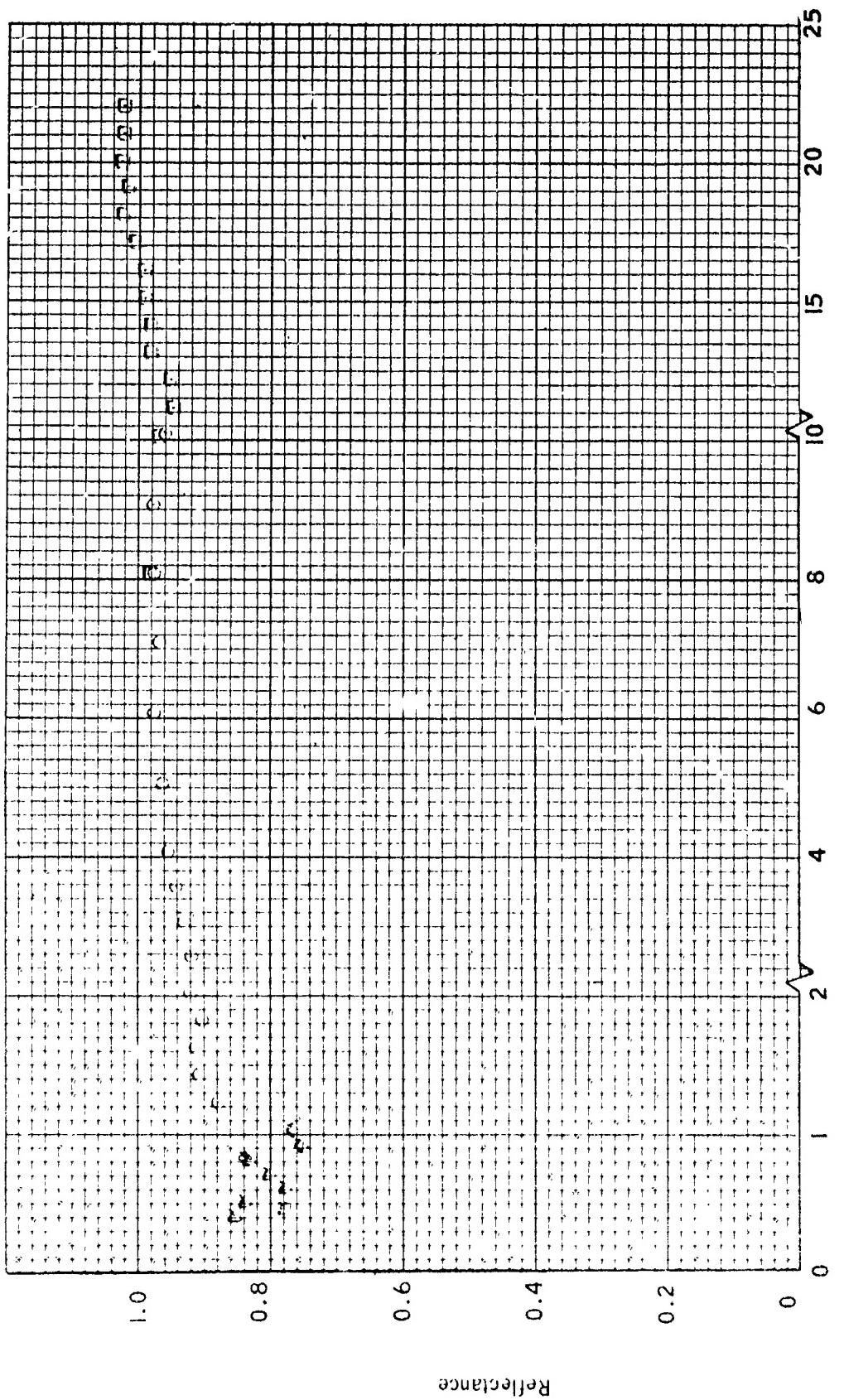


Fig. 90 Normal Spectral Reflectance of Specimen No 59 Temperature RT

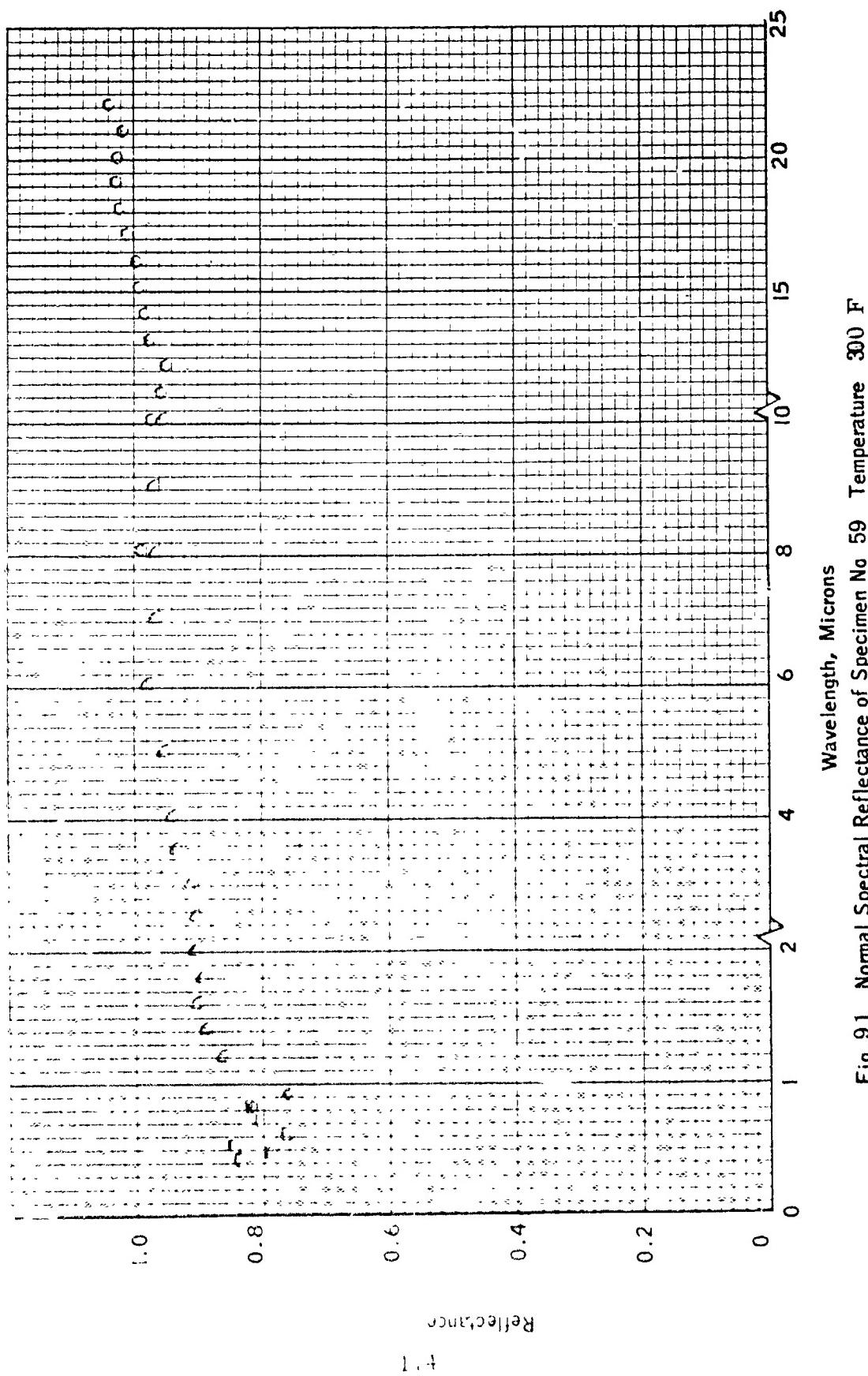


Fig. 91 Normal Spectral Reflectance of Specimen No 59 Temperature 300 F

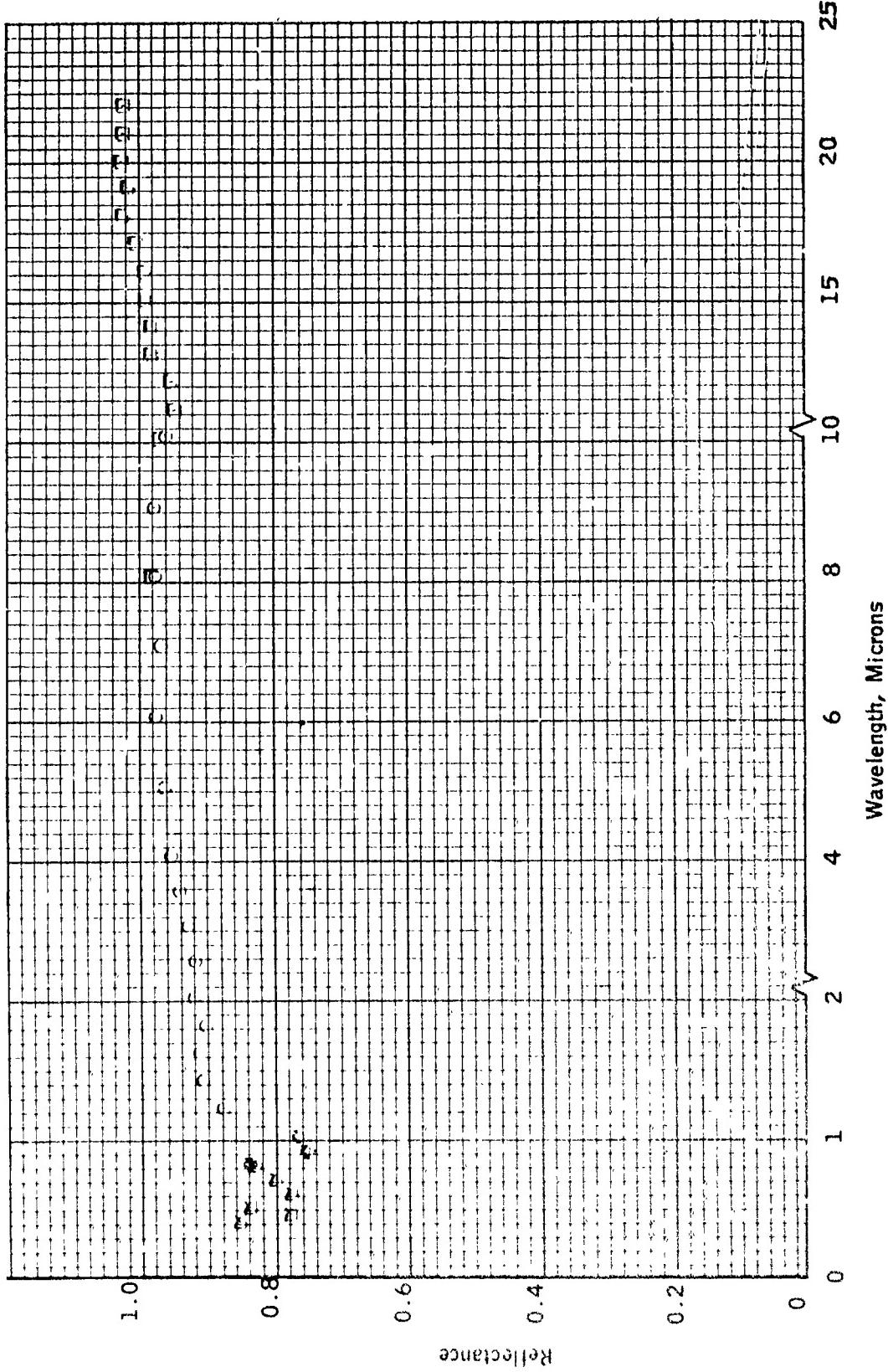


Fig. 92 Normal Spectral Reflectance of Specimen No. 59 Temperature 600°F

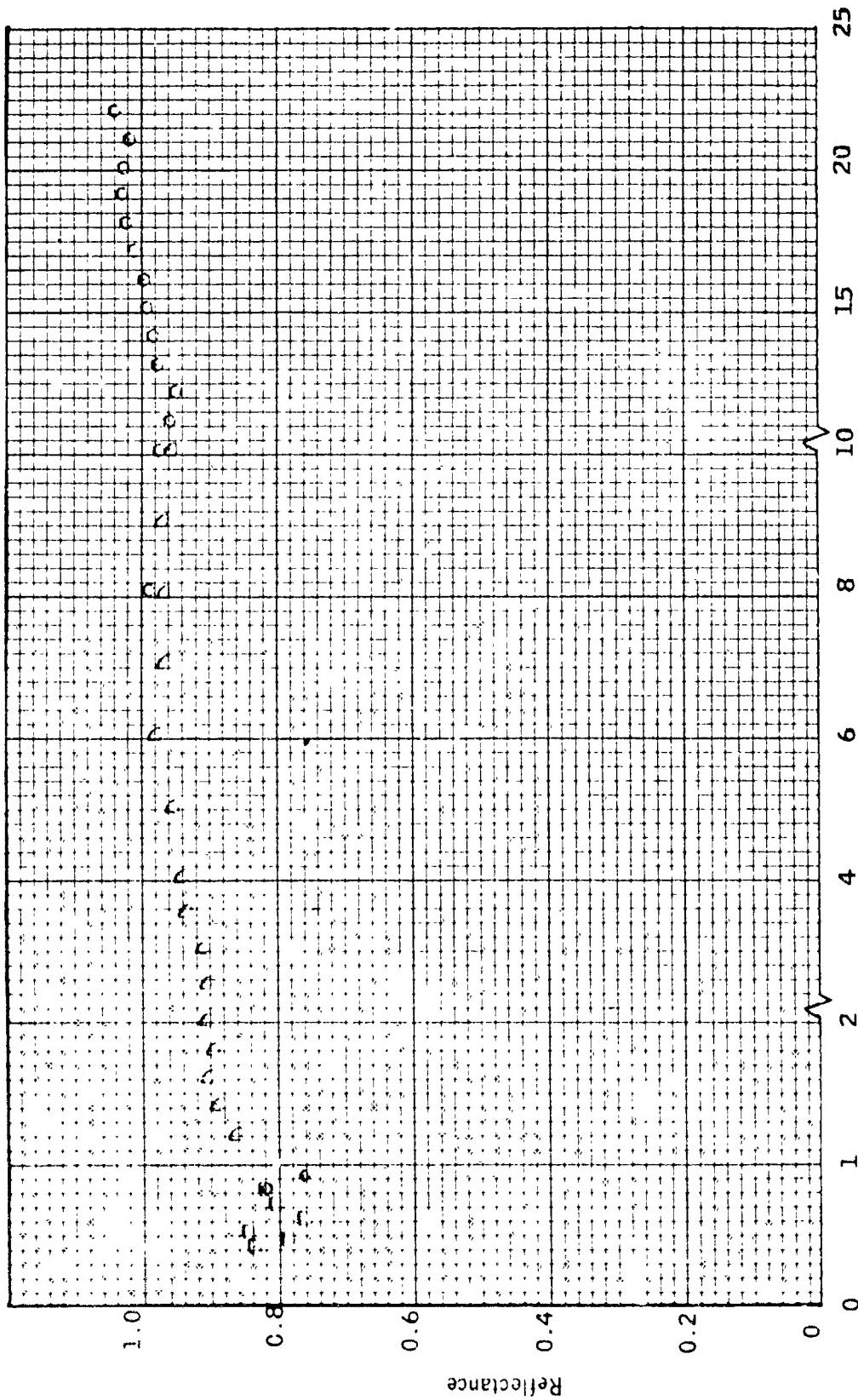


Fig. 93 Normal Spectral Reflectance of Specimen No. 59 Temperature 82.5°F

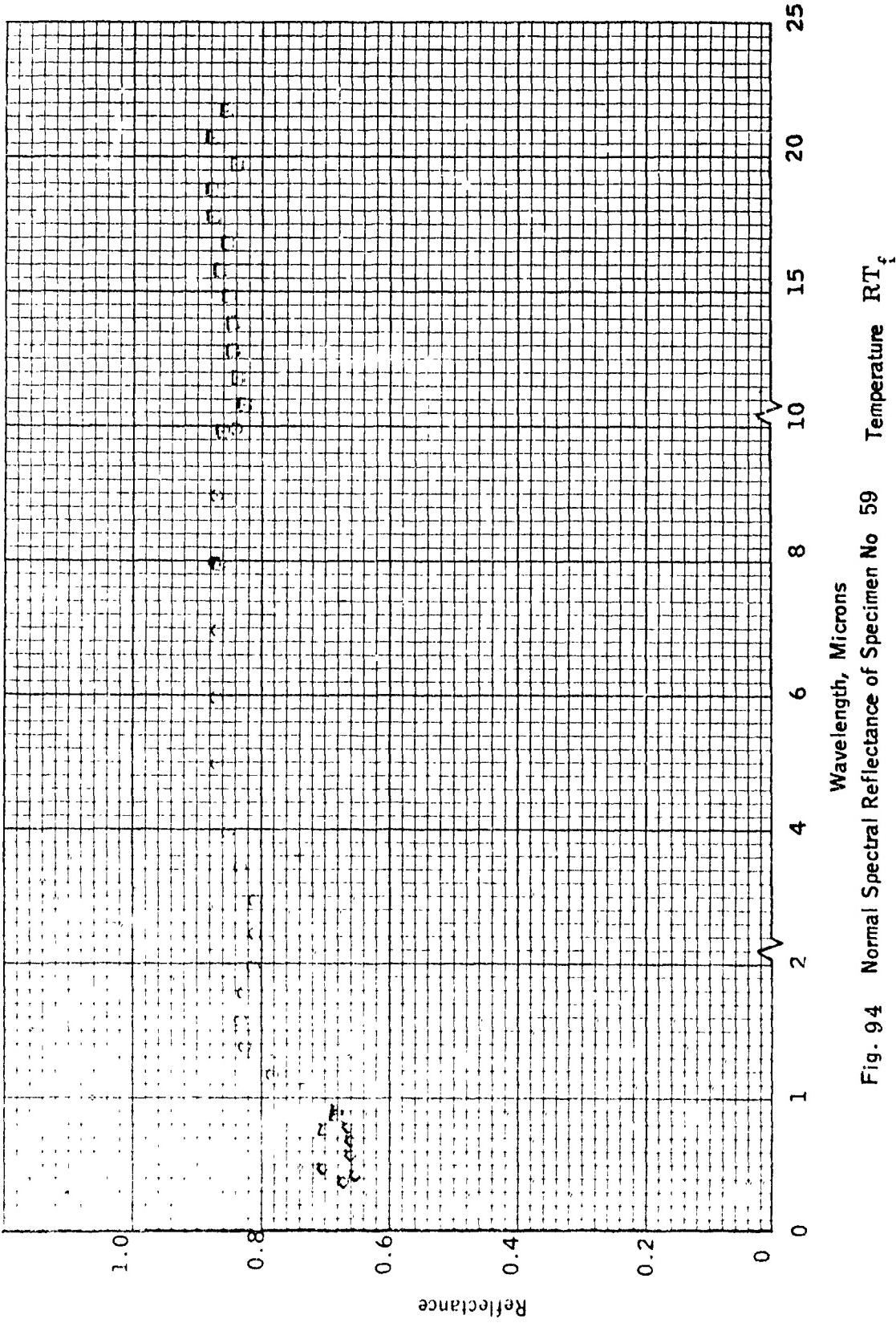


Fig. 94 Normal Spectral Reflectance of Specimen No 59 Temperature  $RT_f$

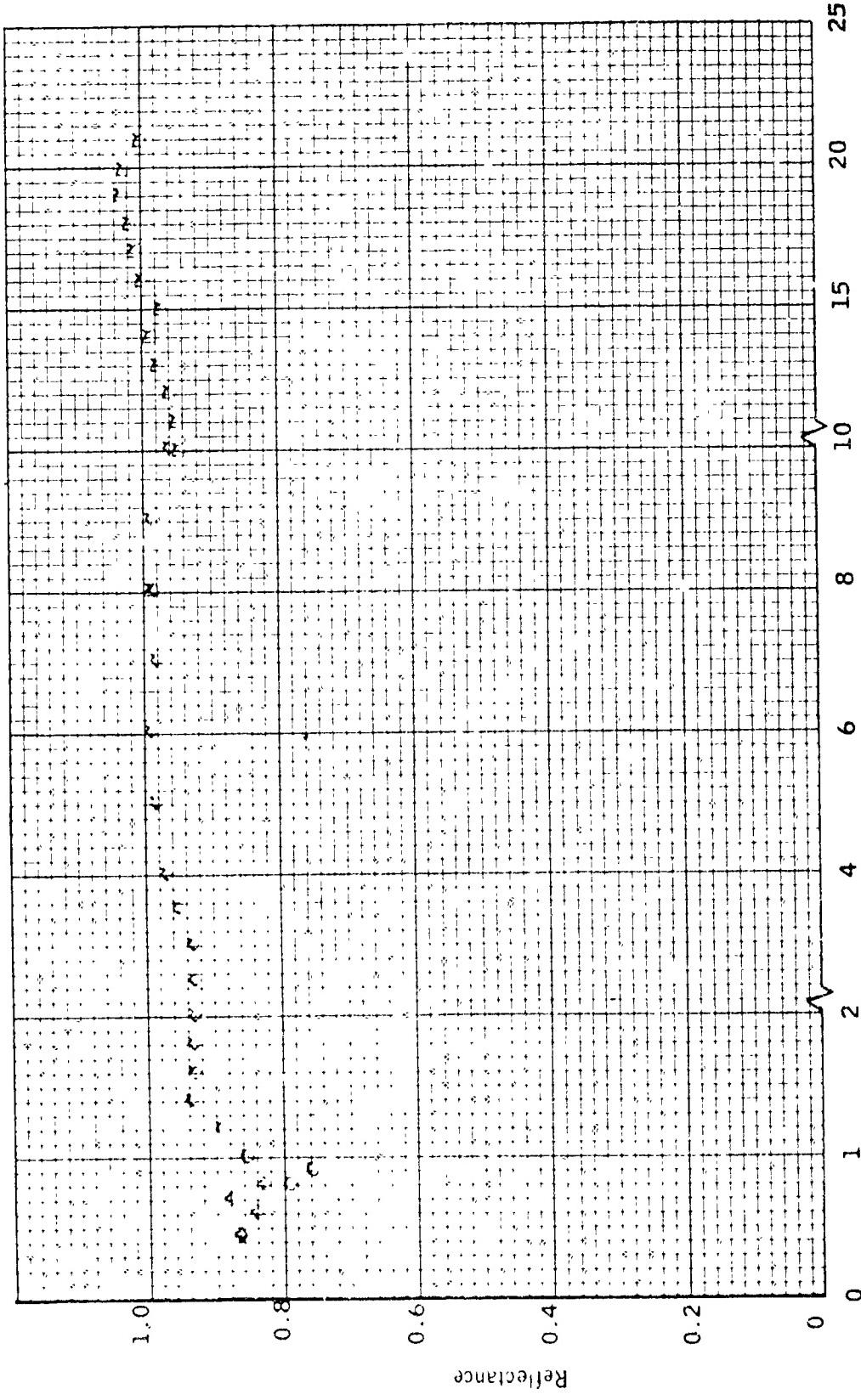


Fig. 95 Normal Spectral Reflectance of Specimen No. 60 Temperature RT

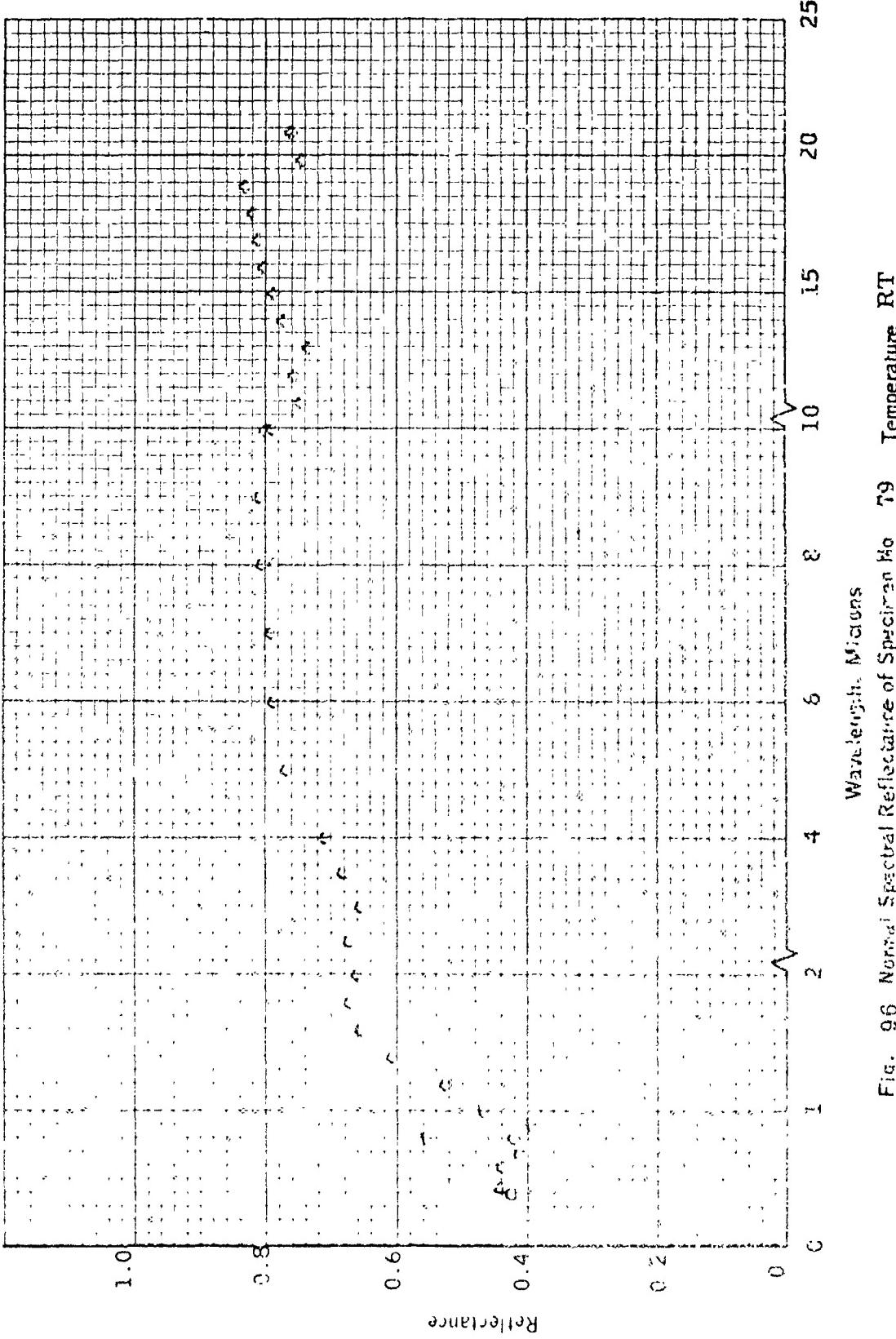


Fig. 96 Normal Spectral Reflectance of Specimen No. 79 Temperature RT

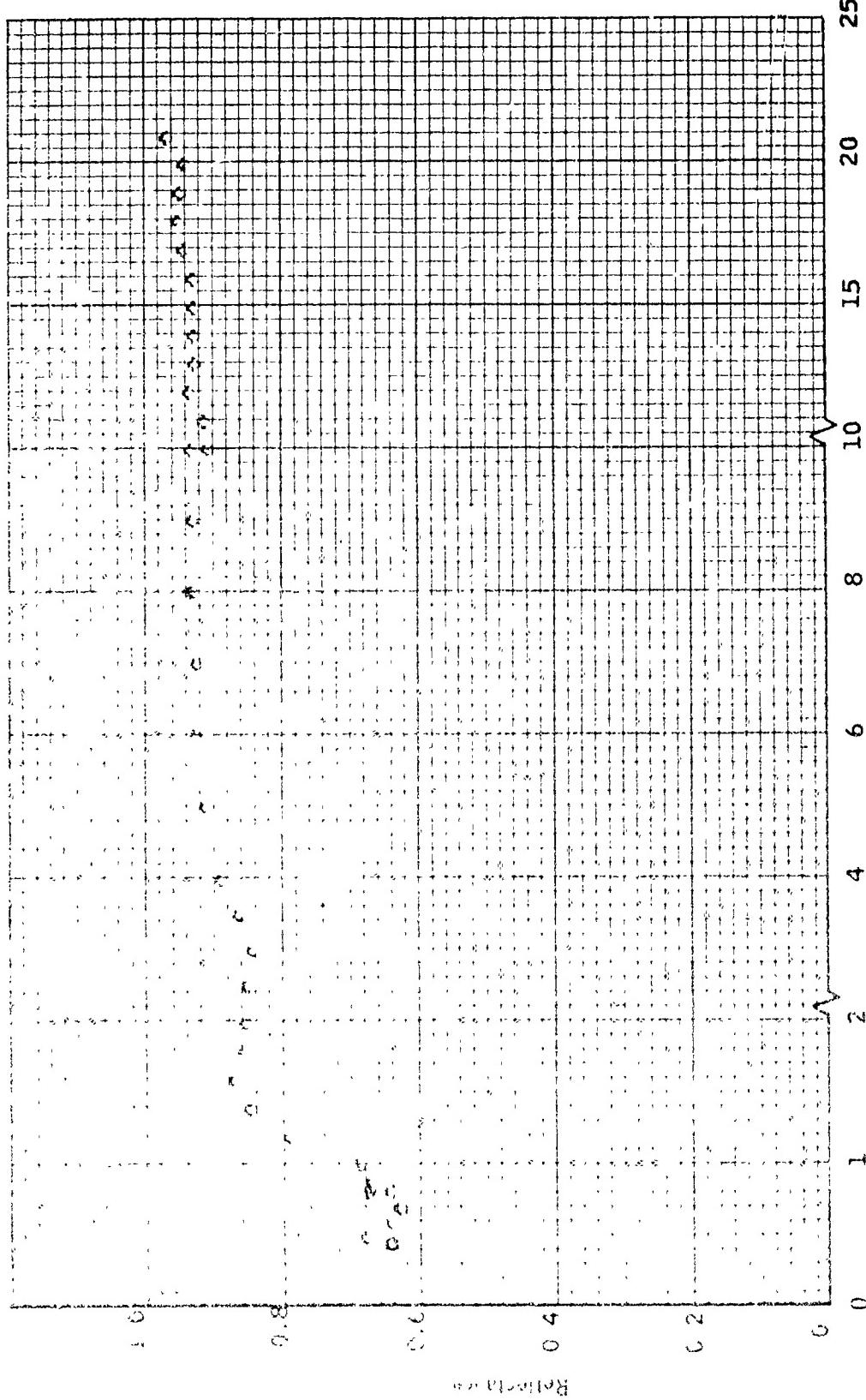


Fig. 97 Normal Spectral Reflectance of Specimen No. 100 Temperature RT

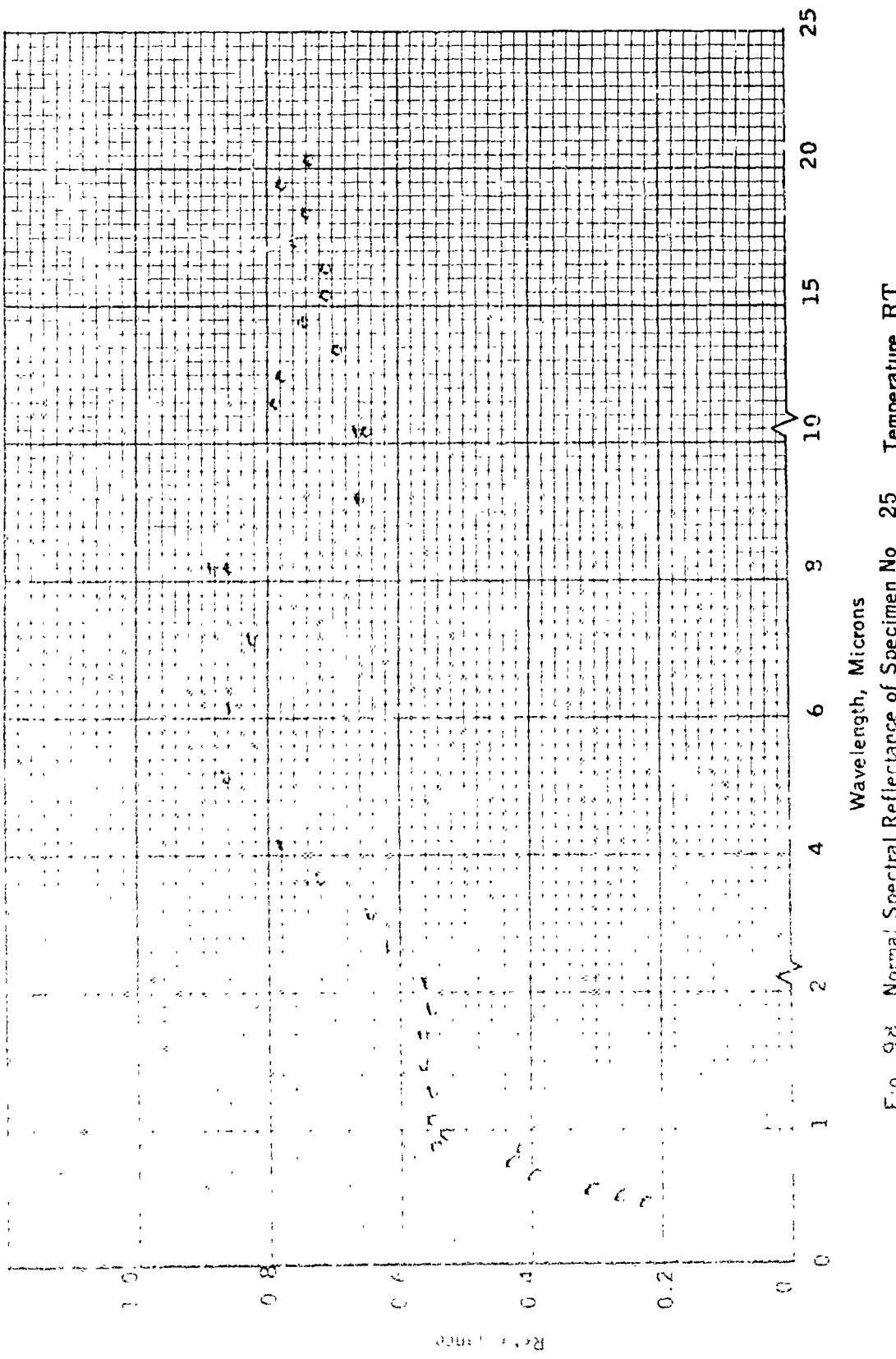


Fig. 92 Normal Spectral Reflectance of Specimen No. 25 Temperature RT

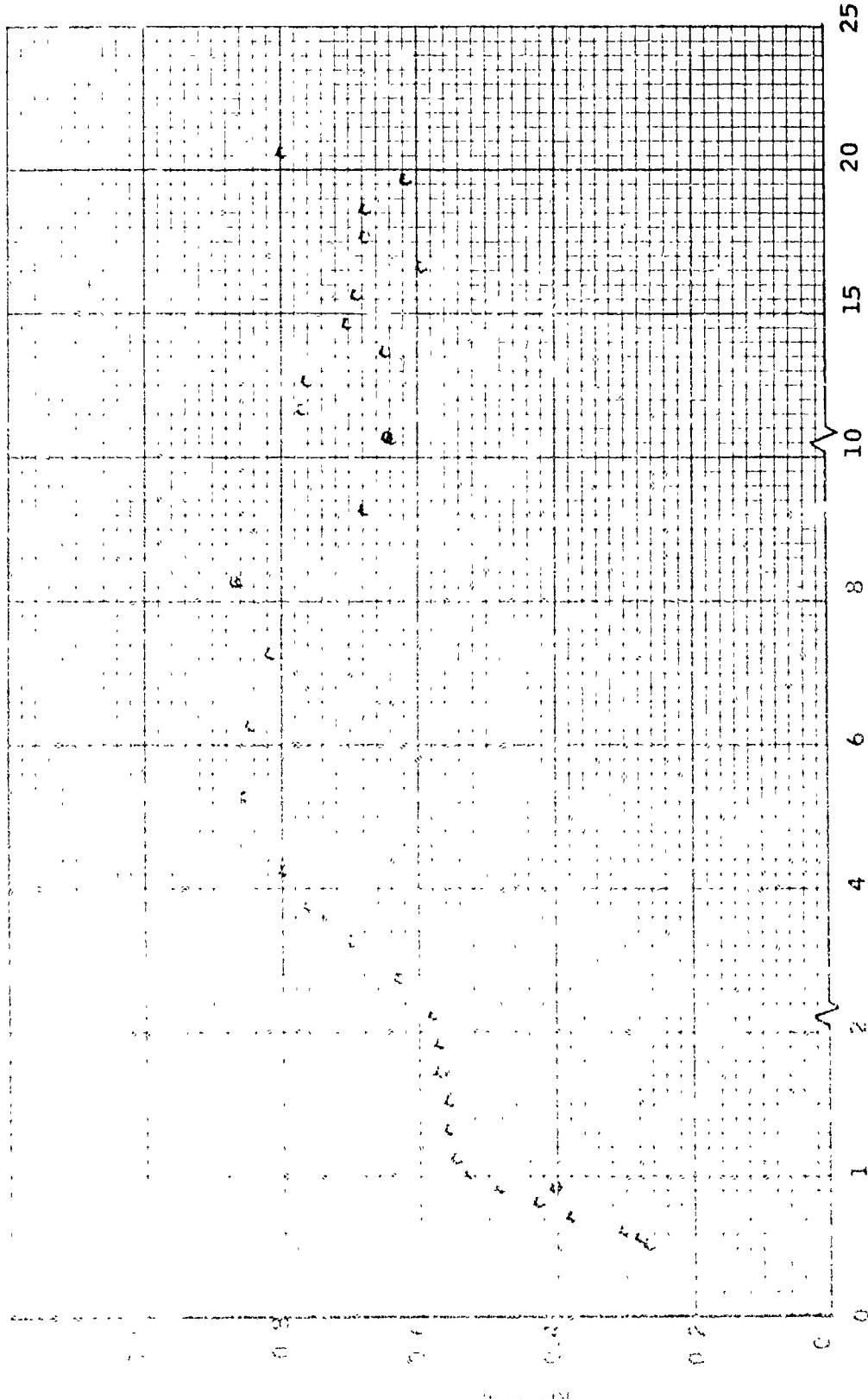


Fig. 99 Normal Spectral Reflectance of Specimen No. 25 Temperature 350°F

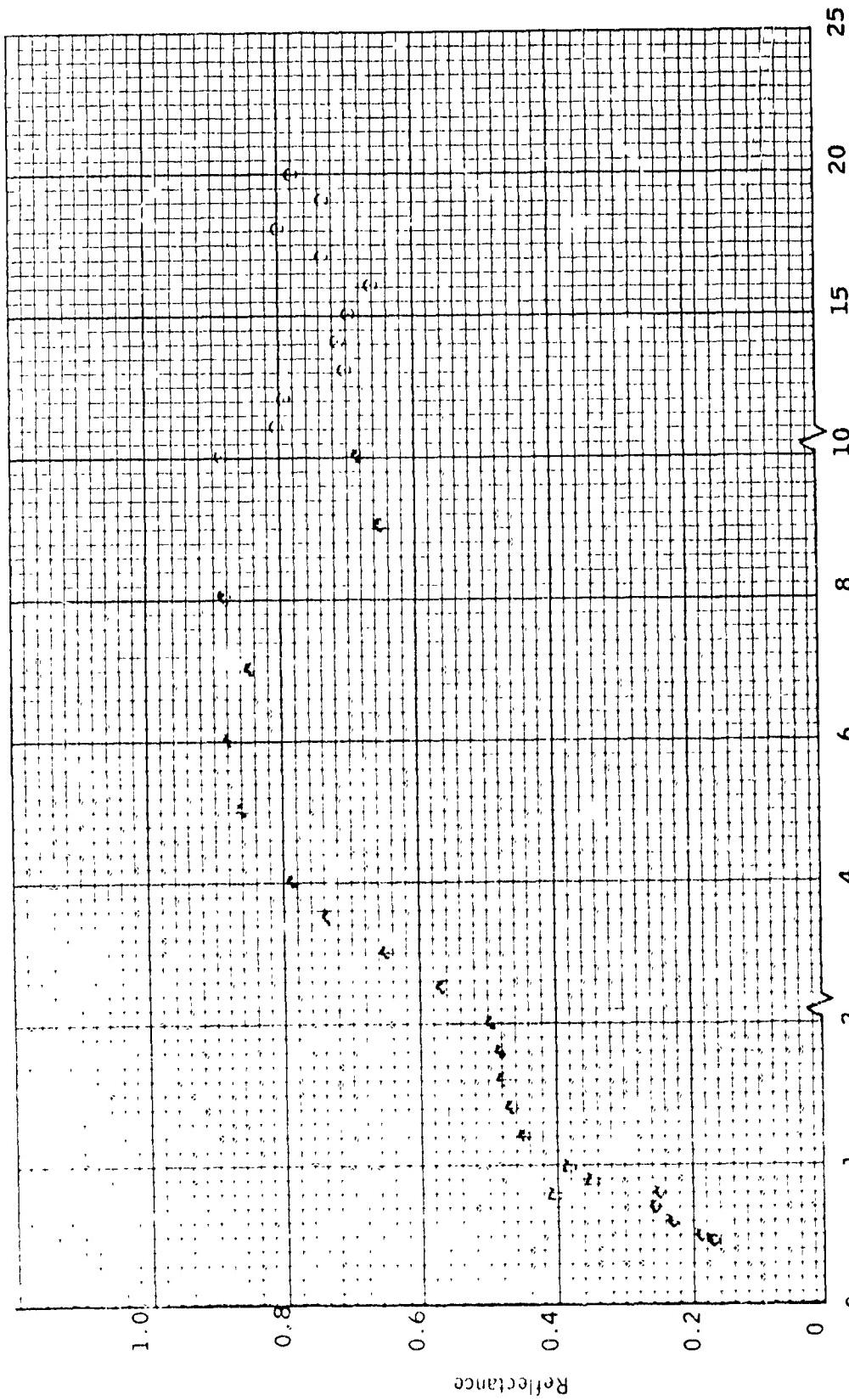


Fig. 100 Normal Spectral Reflectance of Specimen No. 25 Temperature RT HT-800F

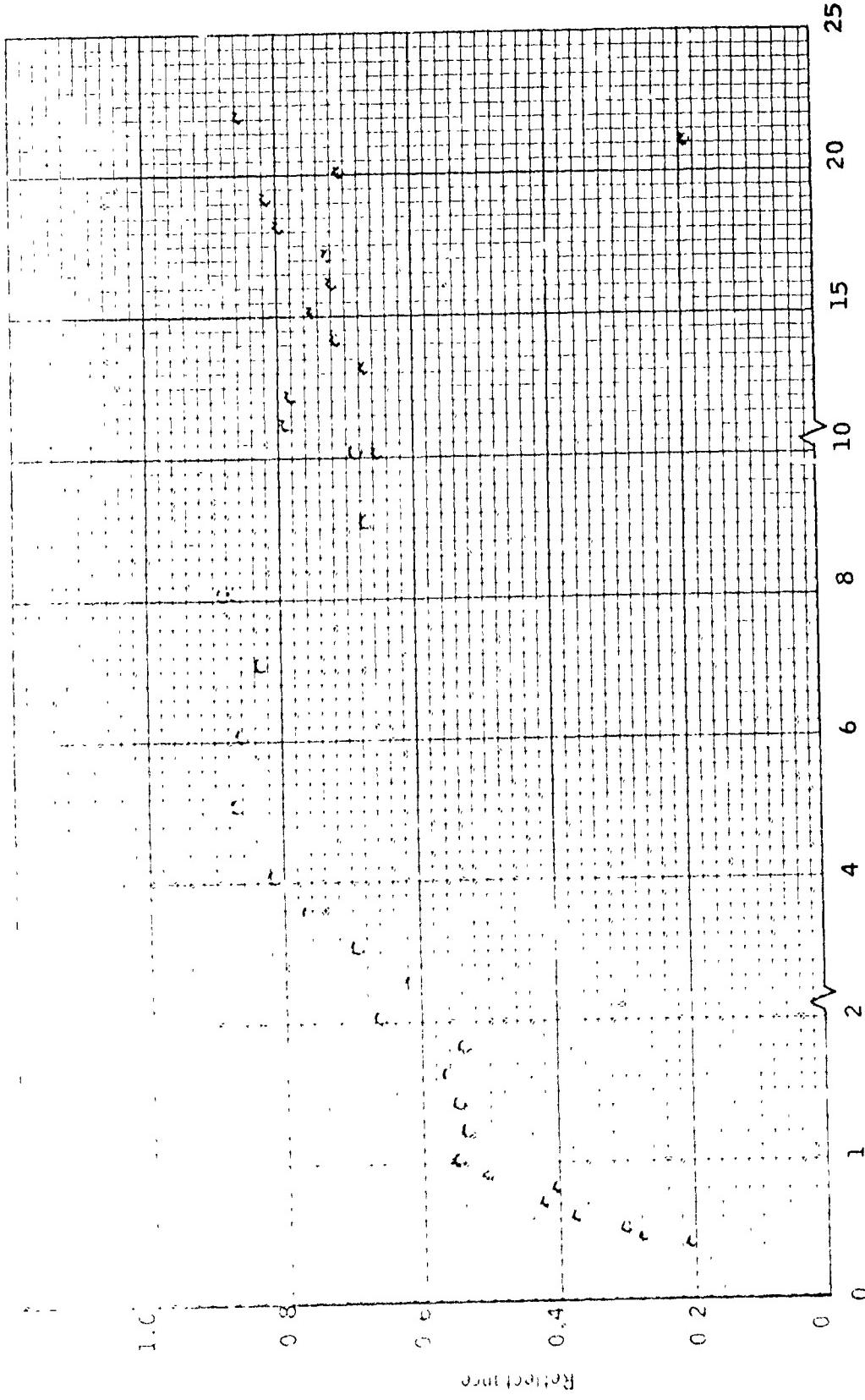


Fig. 101 Normal Spectral Reflectance of Specimen No. 25 Temperature  $RT_f$

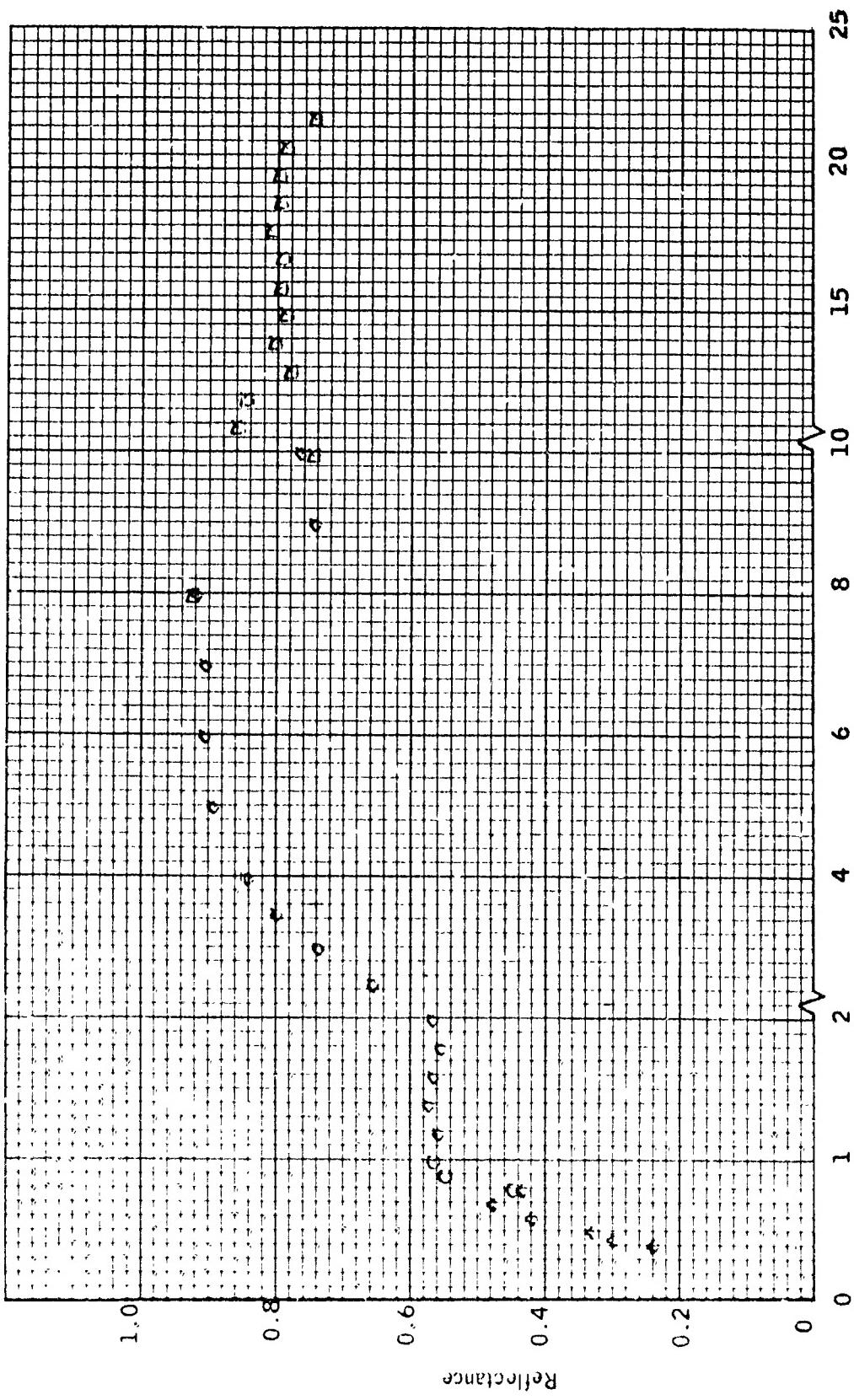


Fig. 102 Normal Spectral Reflectance of Specimen No. 36 Temperature RT  
Wavelength, Microns

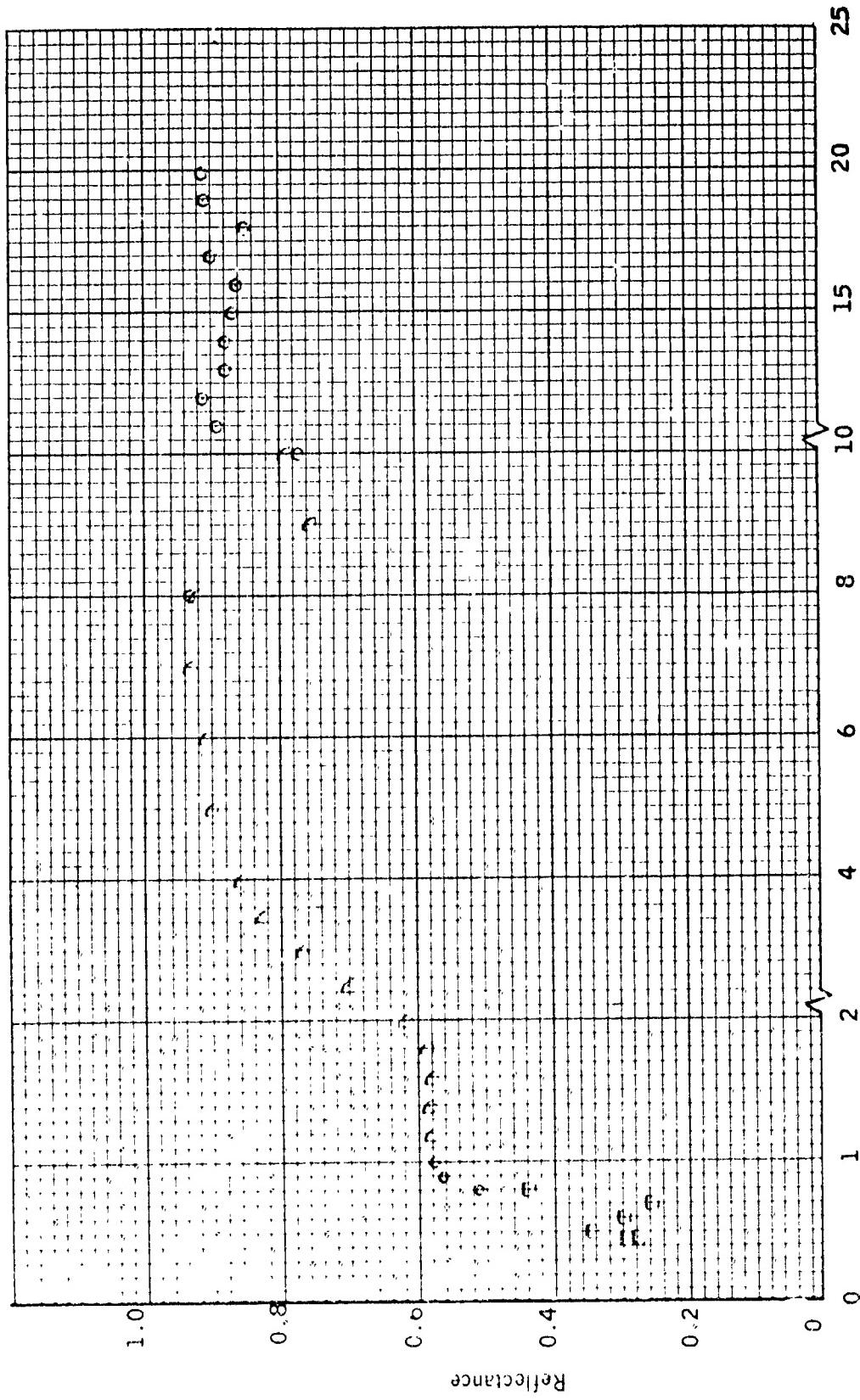


Fig. 103 Normal Spectral Reflectance of Specimen No. 138 Temperature RT

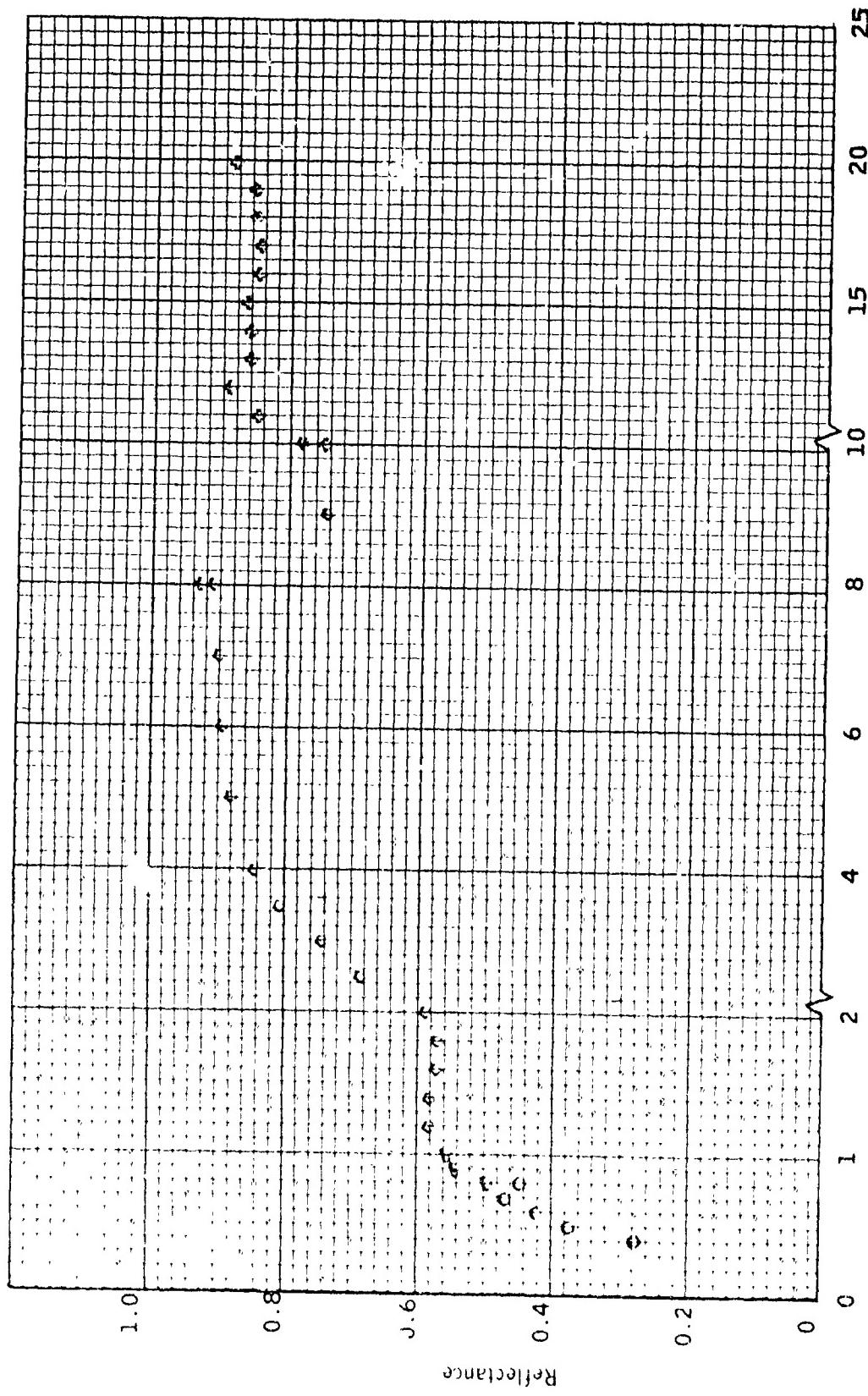


Fig. 104 Normal Spectral Reflectance of Specimen No. 138 Temperature RT, MF

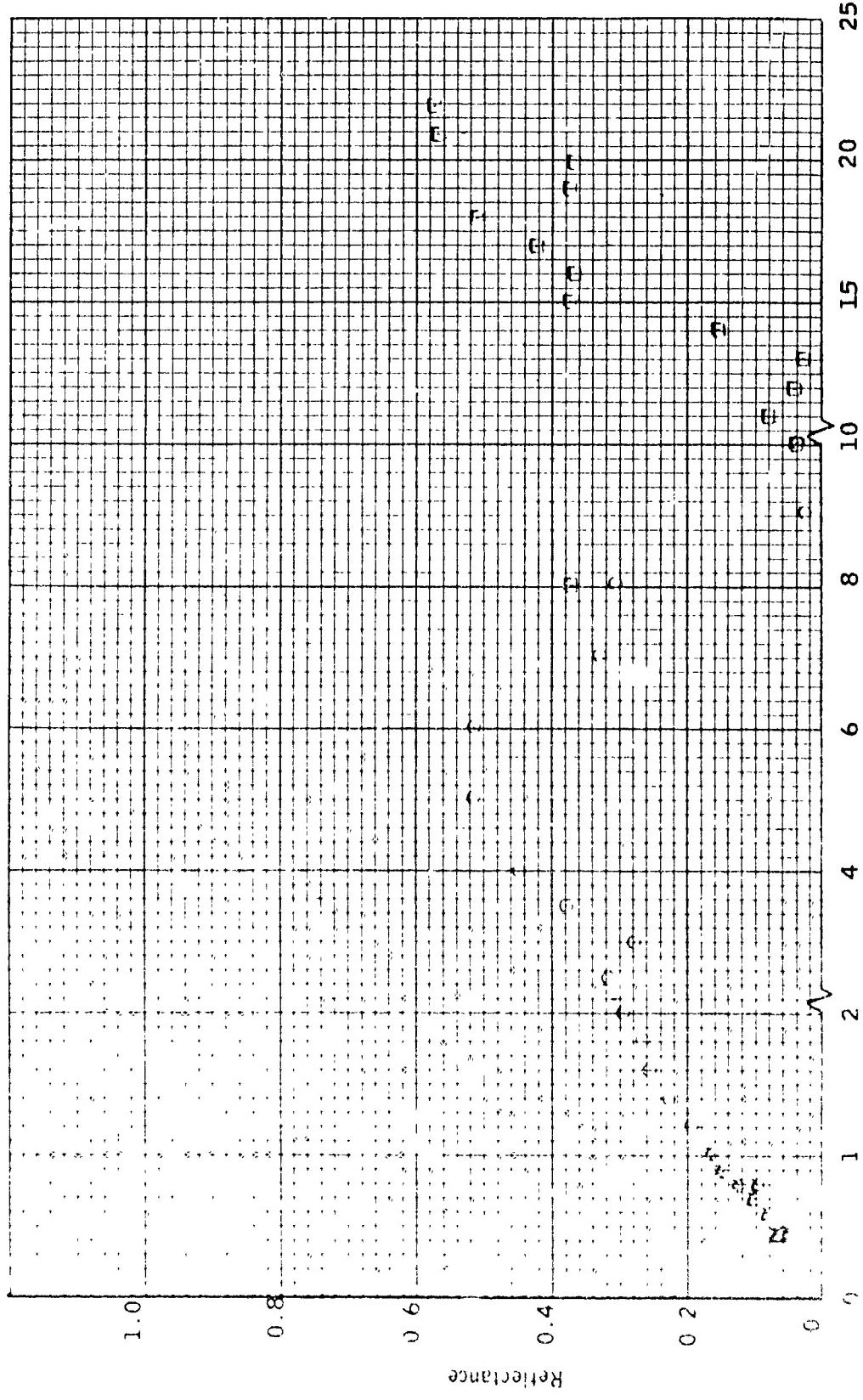


Fig. 105 Normal Spectral Reflectance of Specimen No 140 Temperature RT

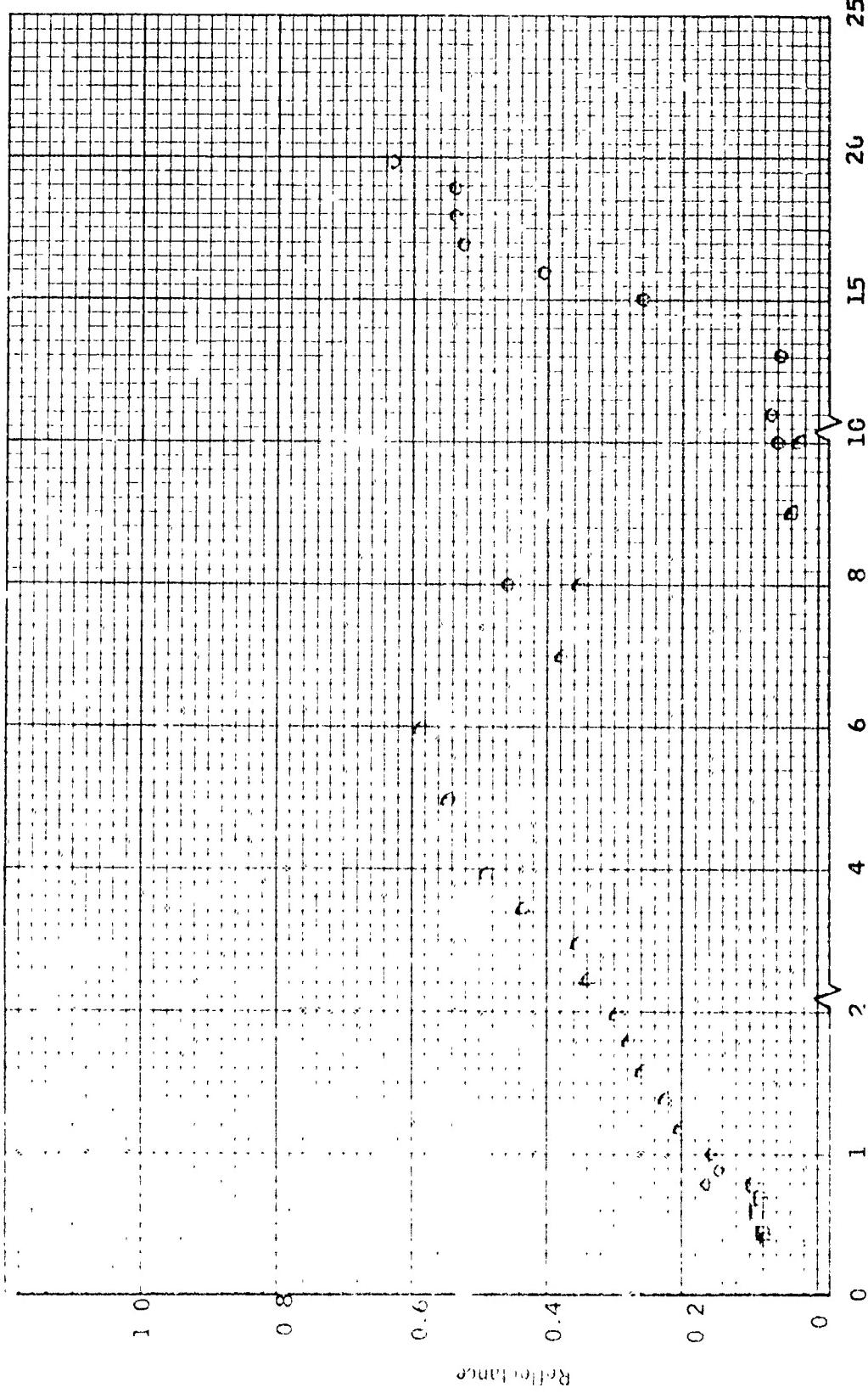


Fig. 106 Normal Spectral Reflectance of Specimen No 140      Temperature RT, HT-800 F  
Wavelength, Microns

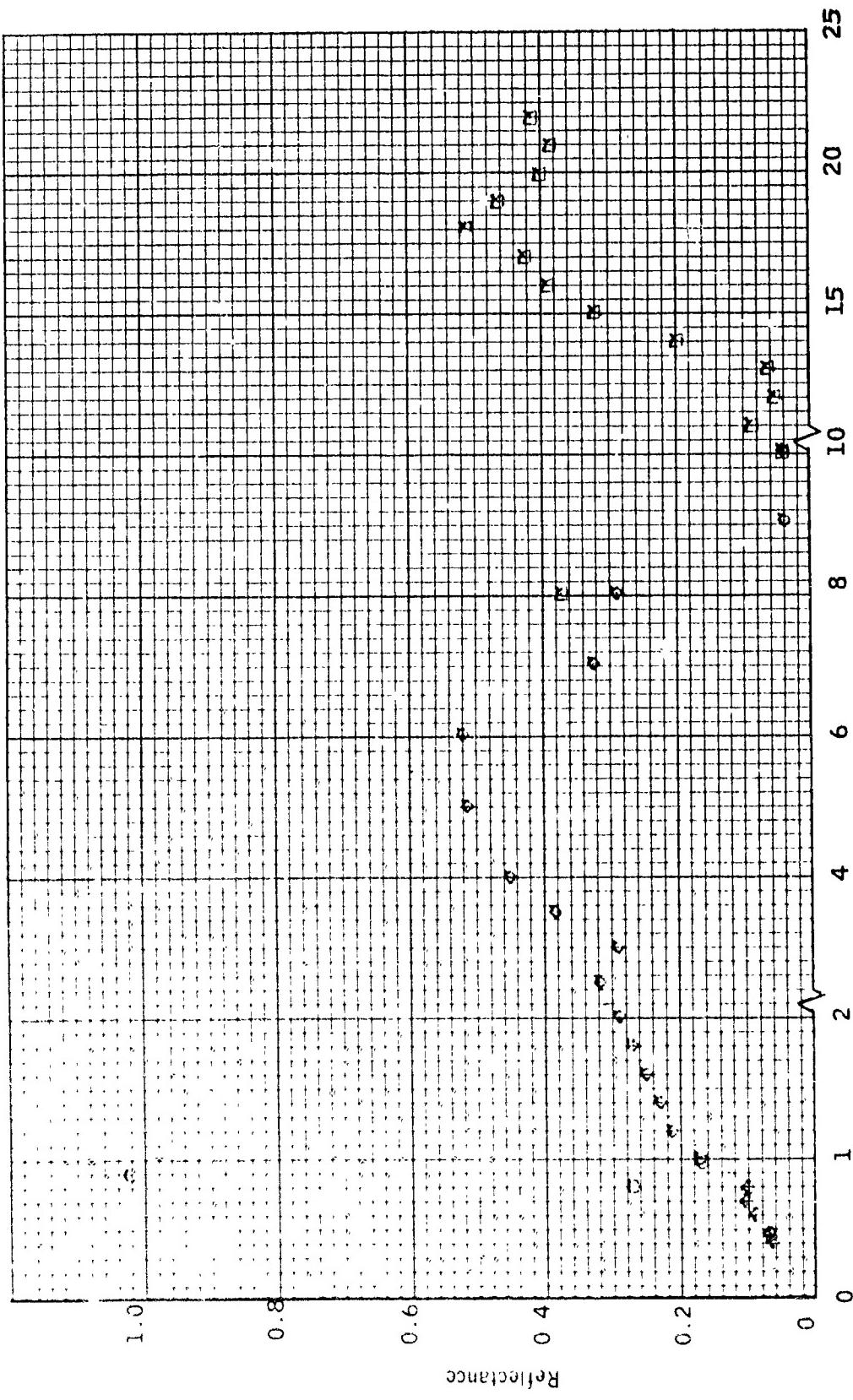


Fig. 107 Normal Spectral Reflectance of Specimen No 140 Temperature RT, MF

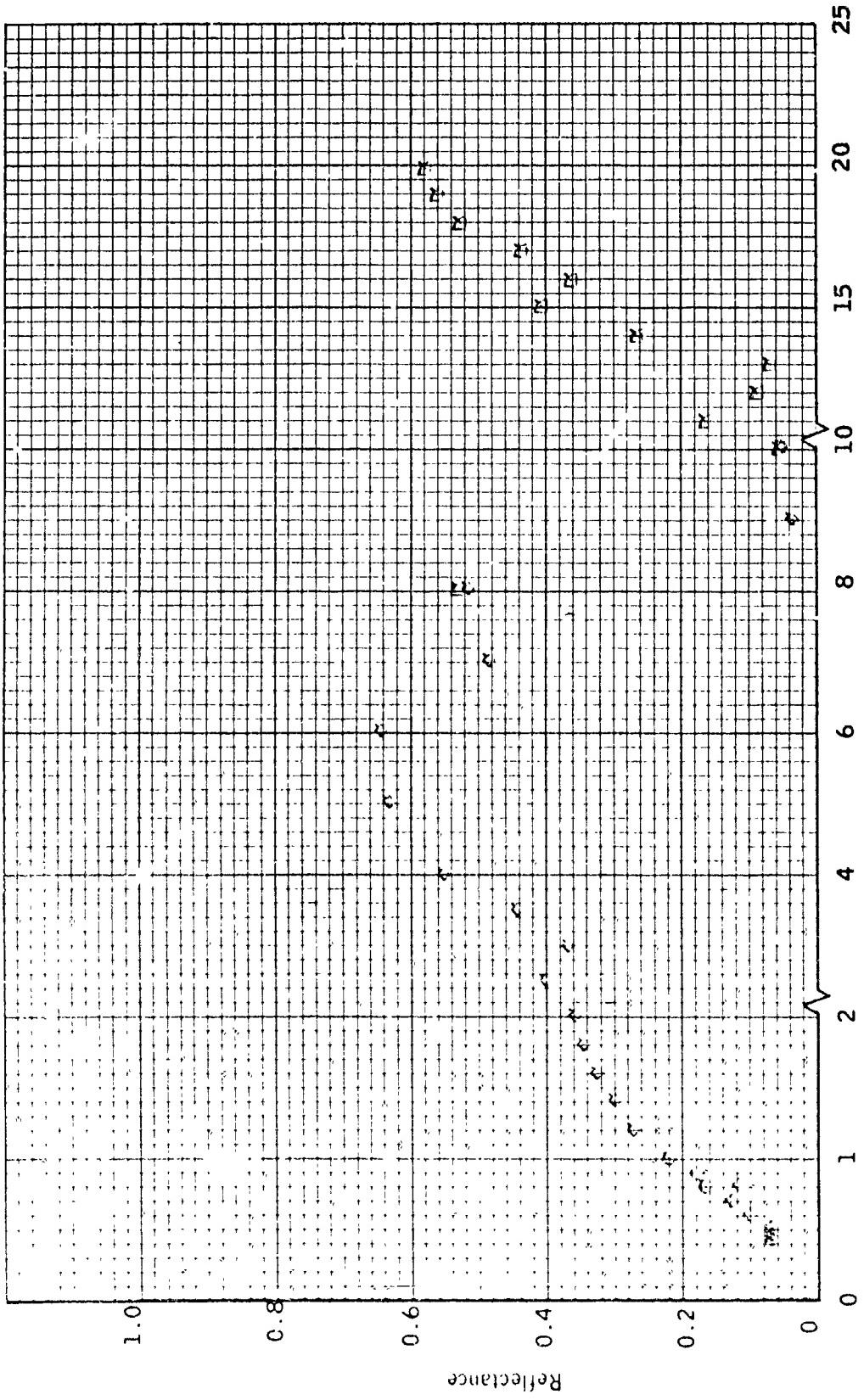


Fig. 108 Normal Spectral Reflectance of Specimen No. 155 Temperature RT

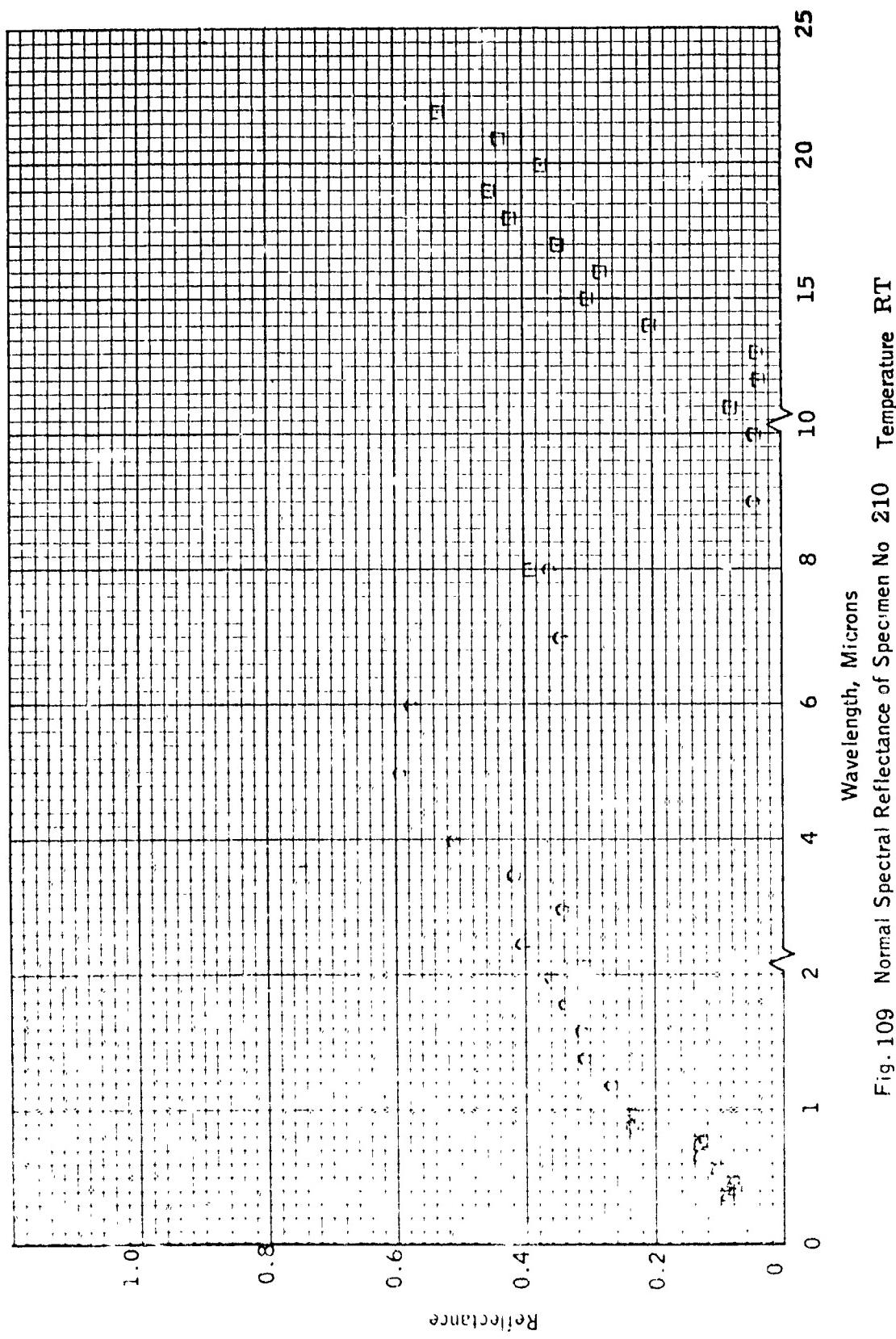


Fig. 109 Normal Spectral Reflectance of Specimen No. 210 Temperature RT

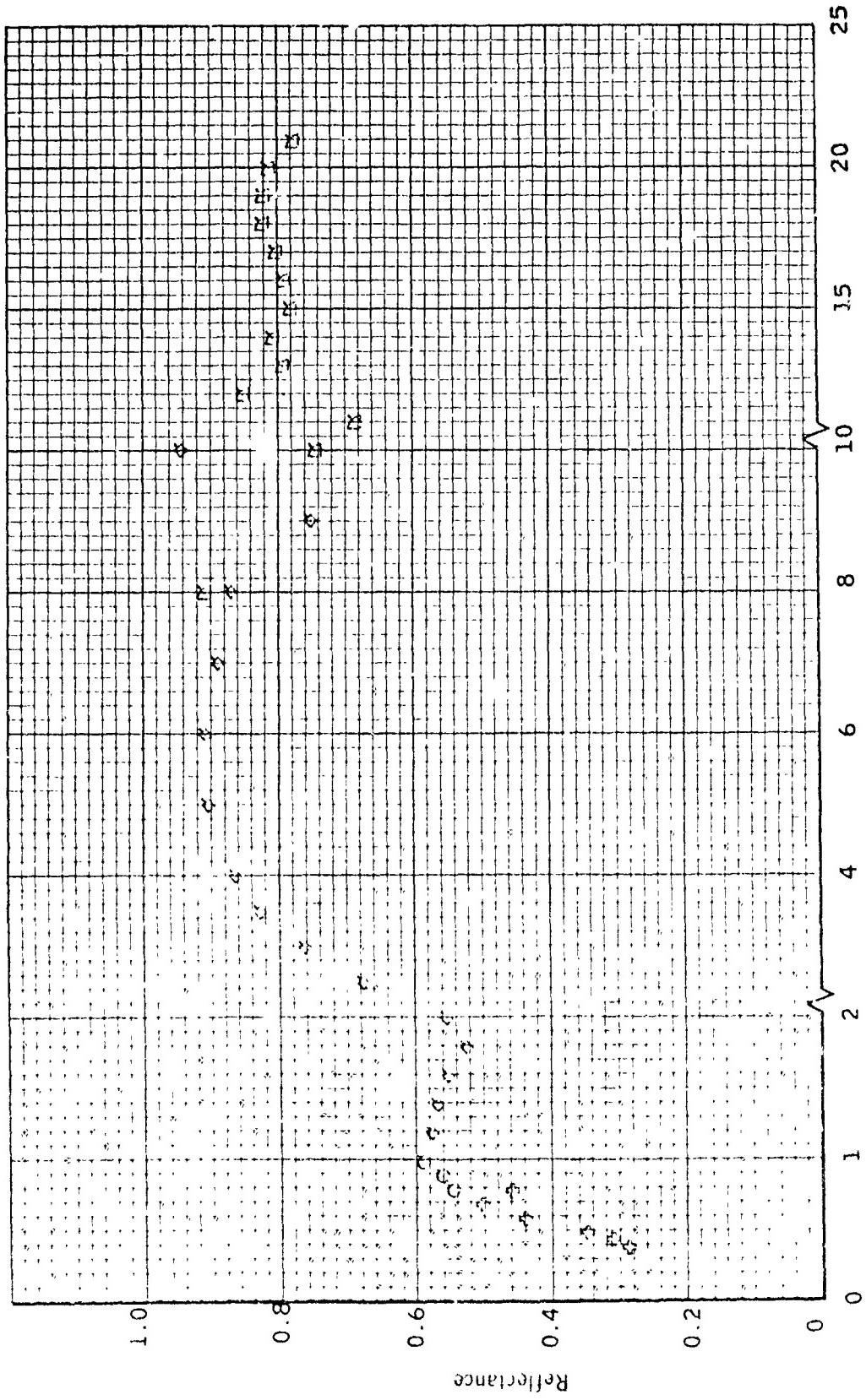


Fig. 110 Normal Spectral Reflectance of Specimen No 211 Temperature RT

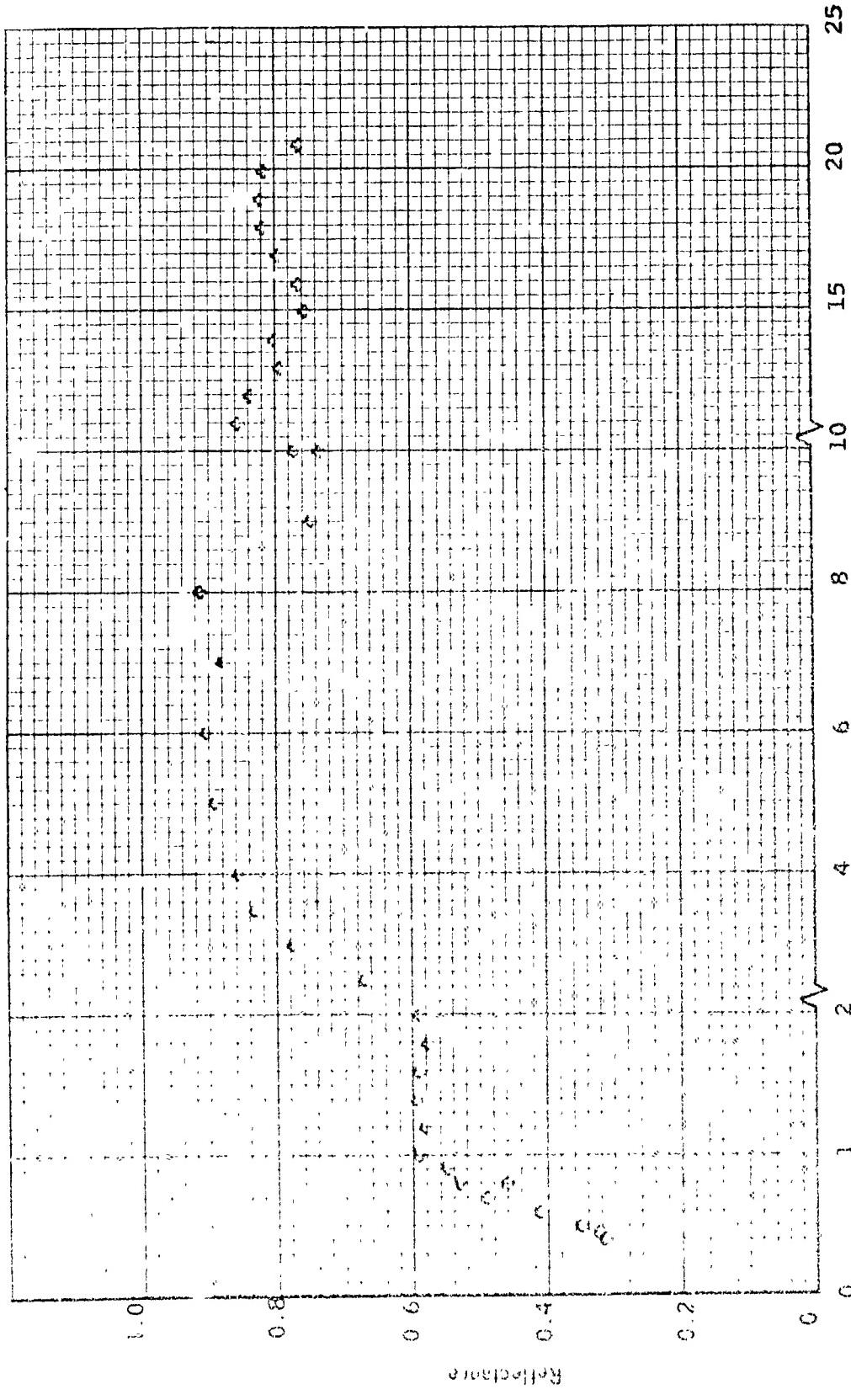


Fig. 111 Normal Spectral Reflectance of Specimen No. 211 Temperature 350 K

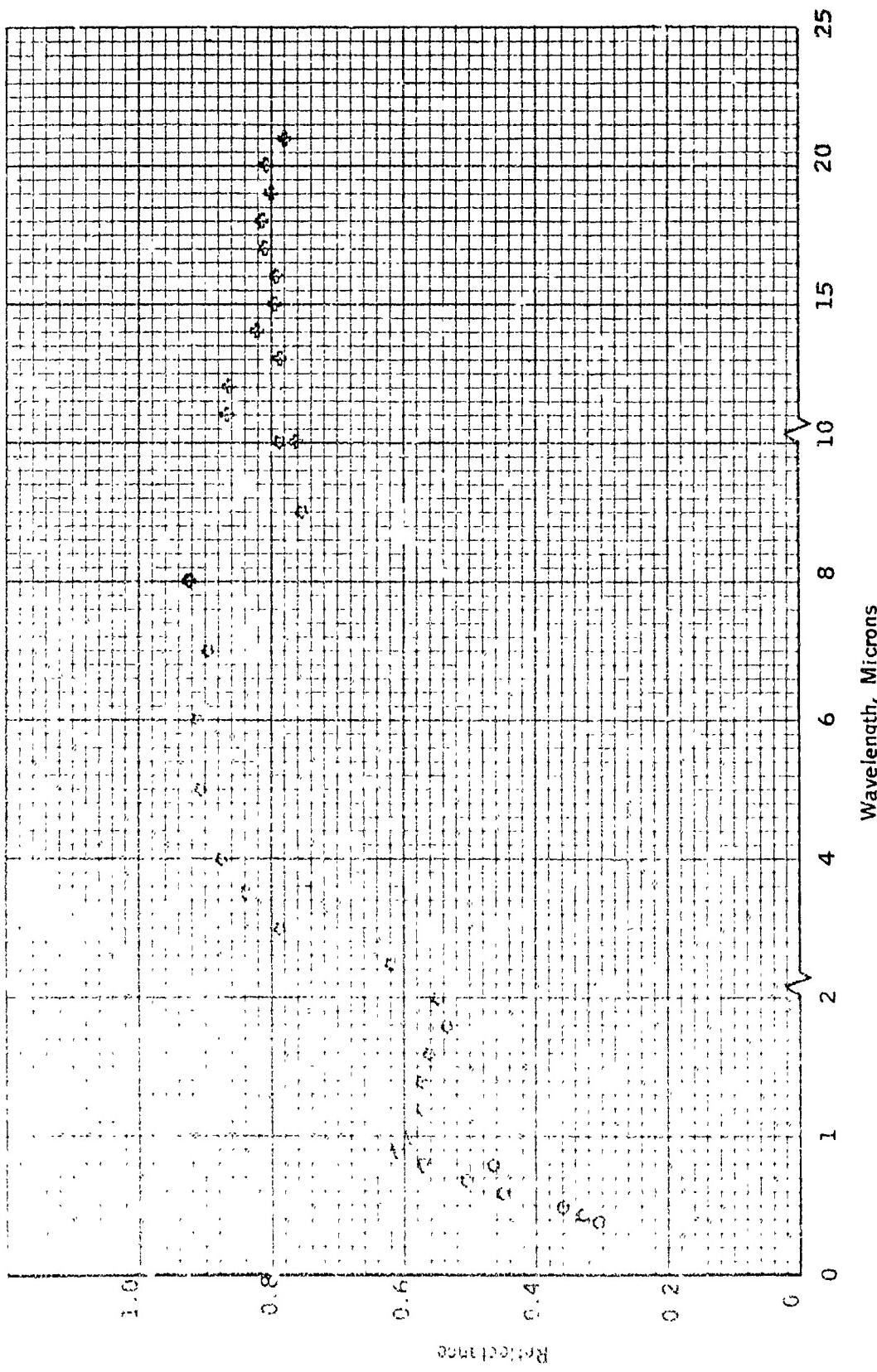


Fig.1112 Normal Spectral Reflectance of Specimen No. 211 Temperature  $RT_f$

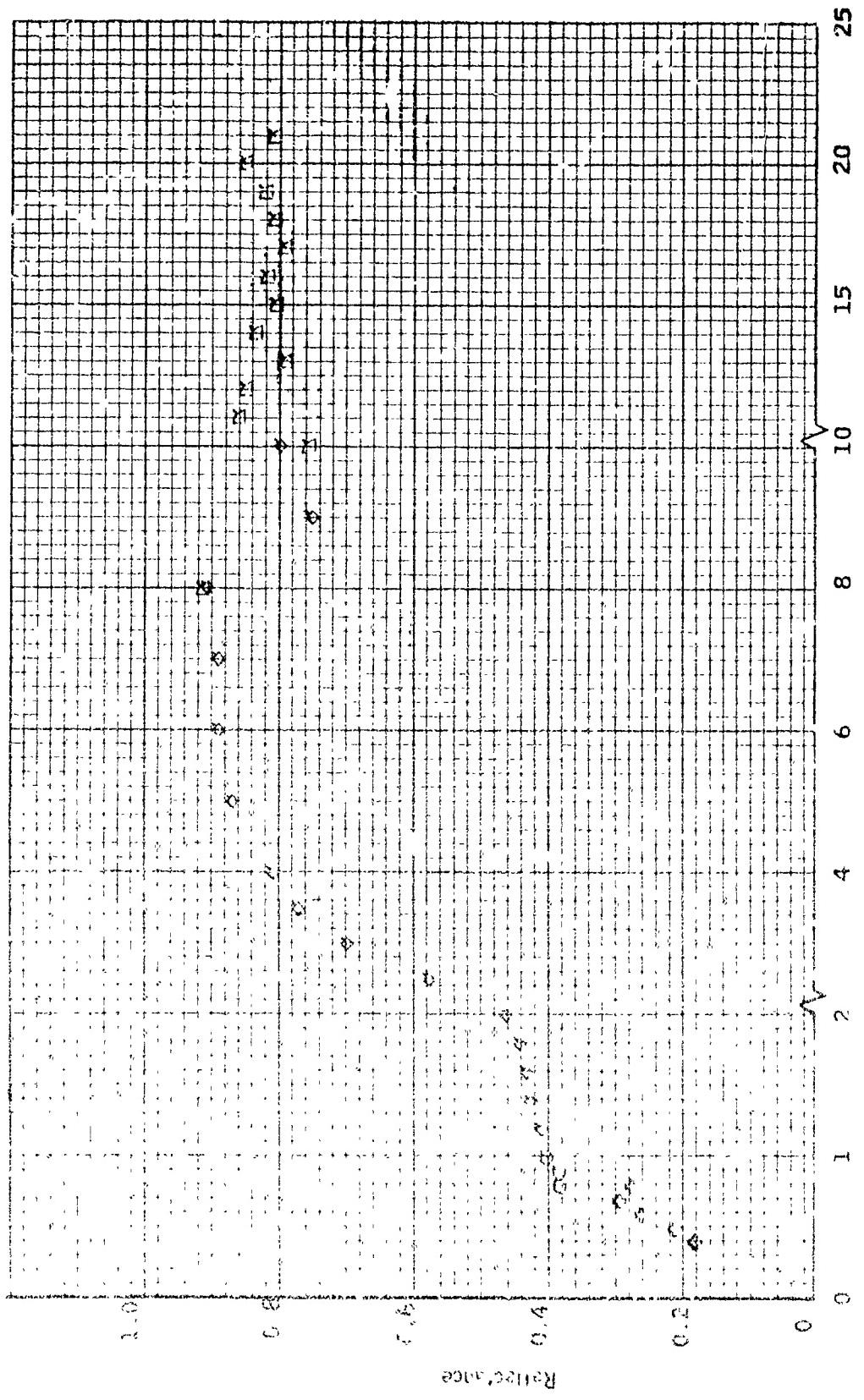


Fig 113 Normal Spectral Reflectance of Specimen No 211 Temperature RT, HT-800F

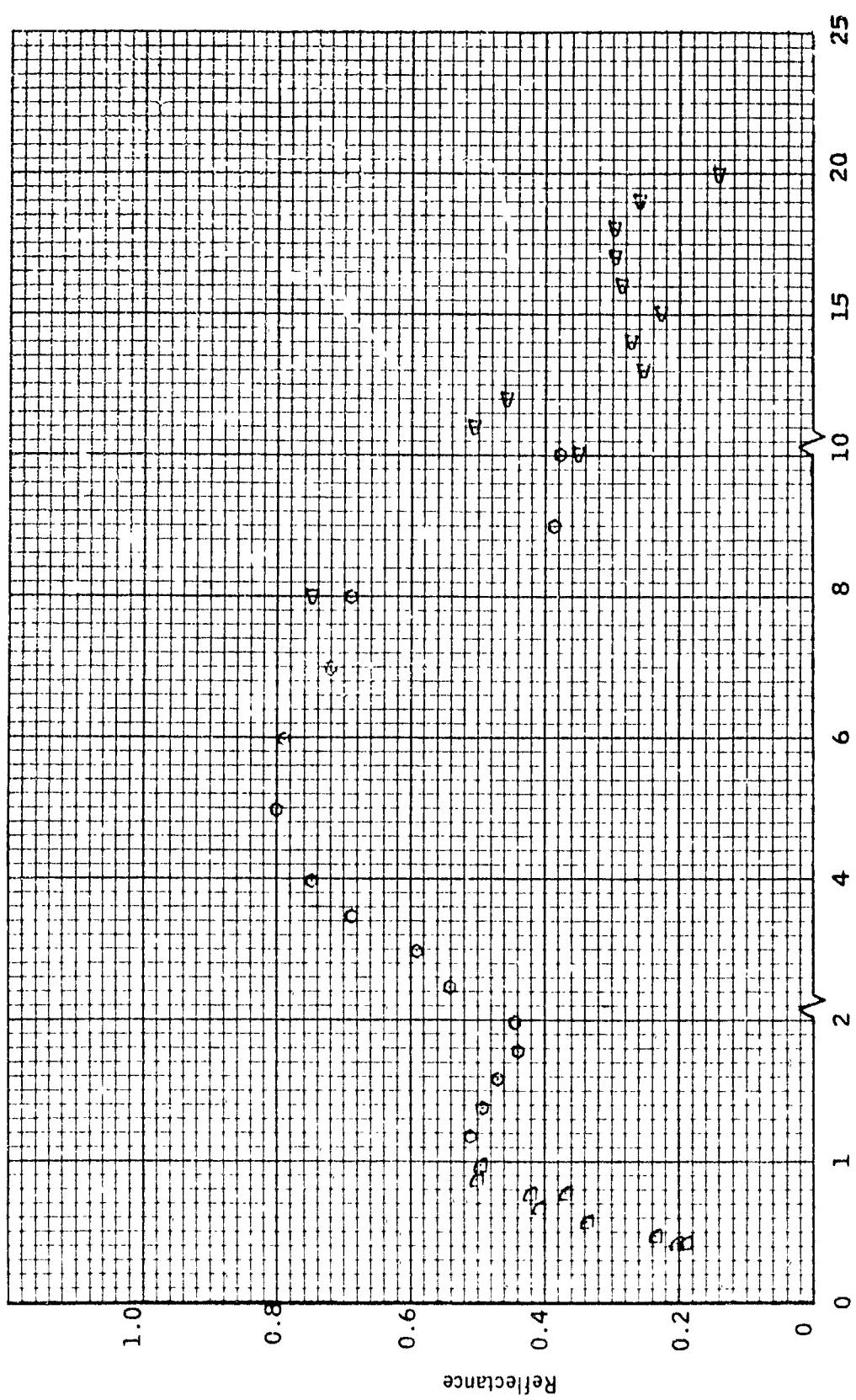


Fig. 114 Normal Spectral Reflectance of Specimen No. 215 Temperature RT  
Wave length, Microns

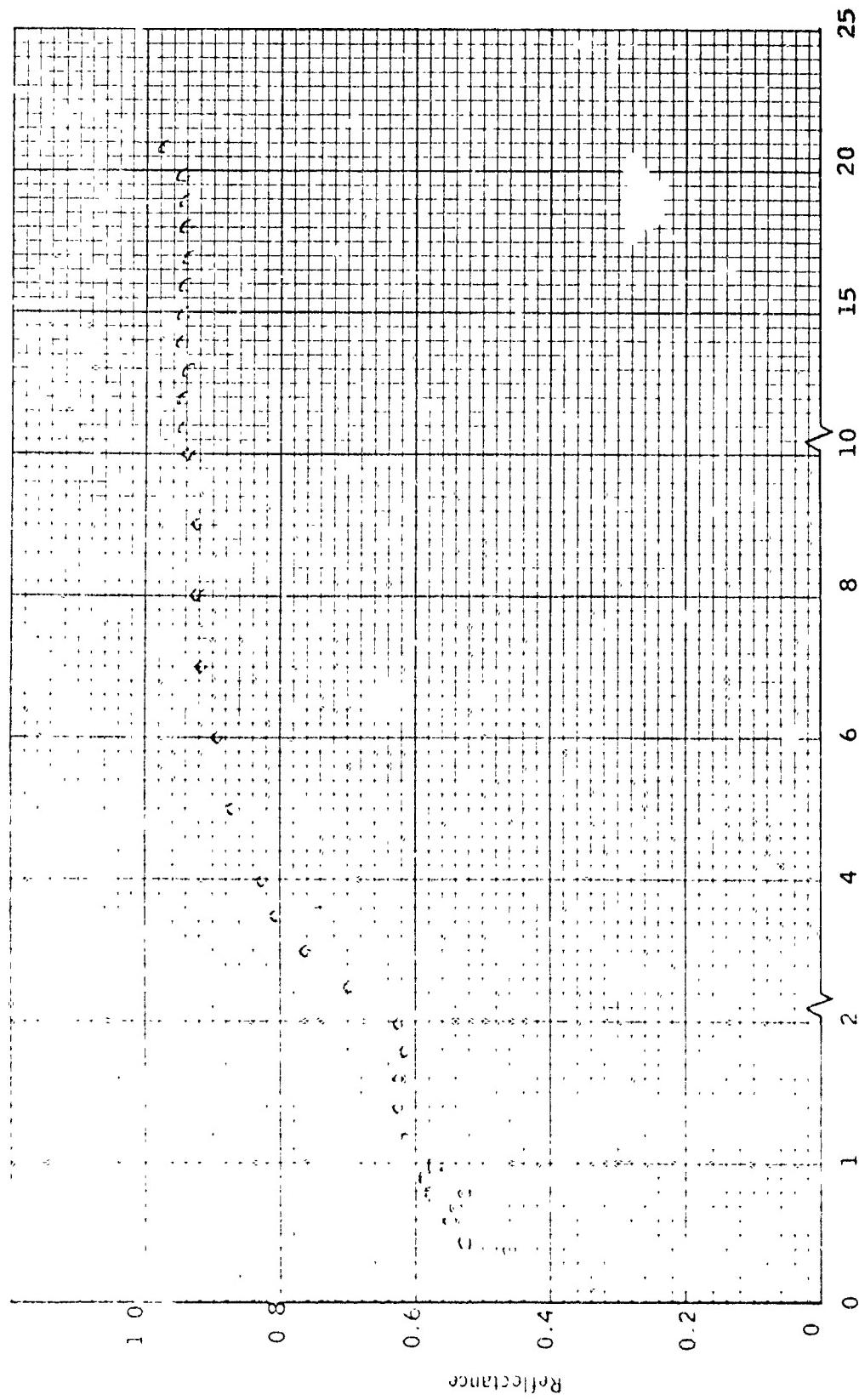


Fig. 115 Normal Spectral Reflectance of Specimen No. 31 Temperature RT

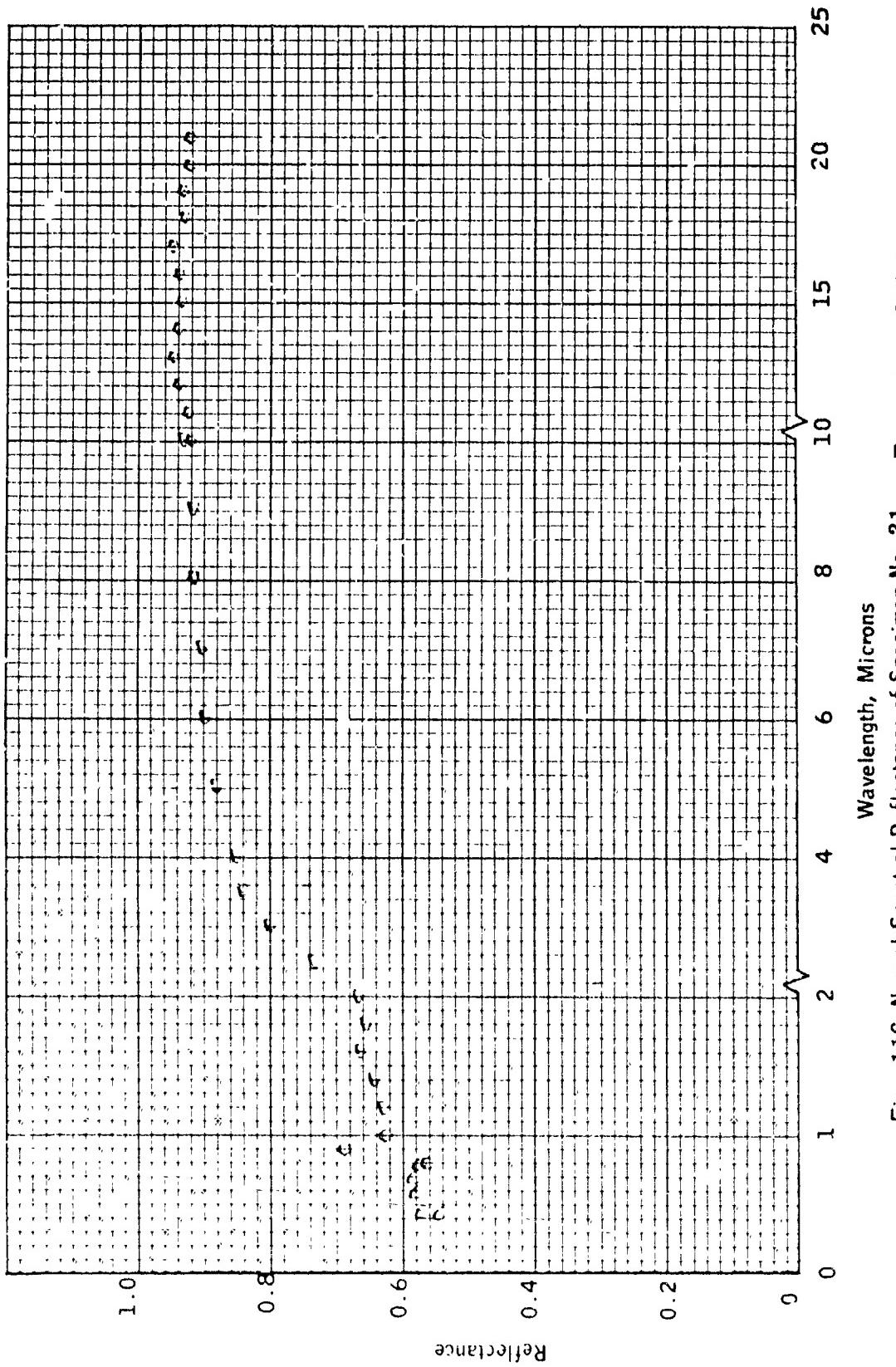


Fig. 116 Normal Spectral Reflectance of Specimen No. 31      Temperature 350°F

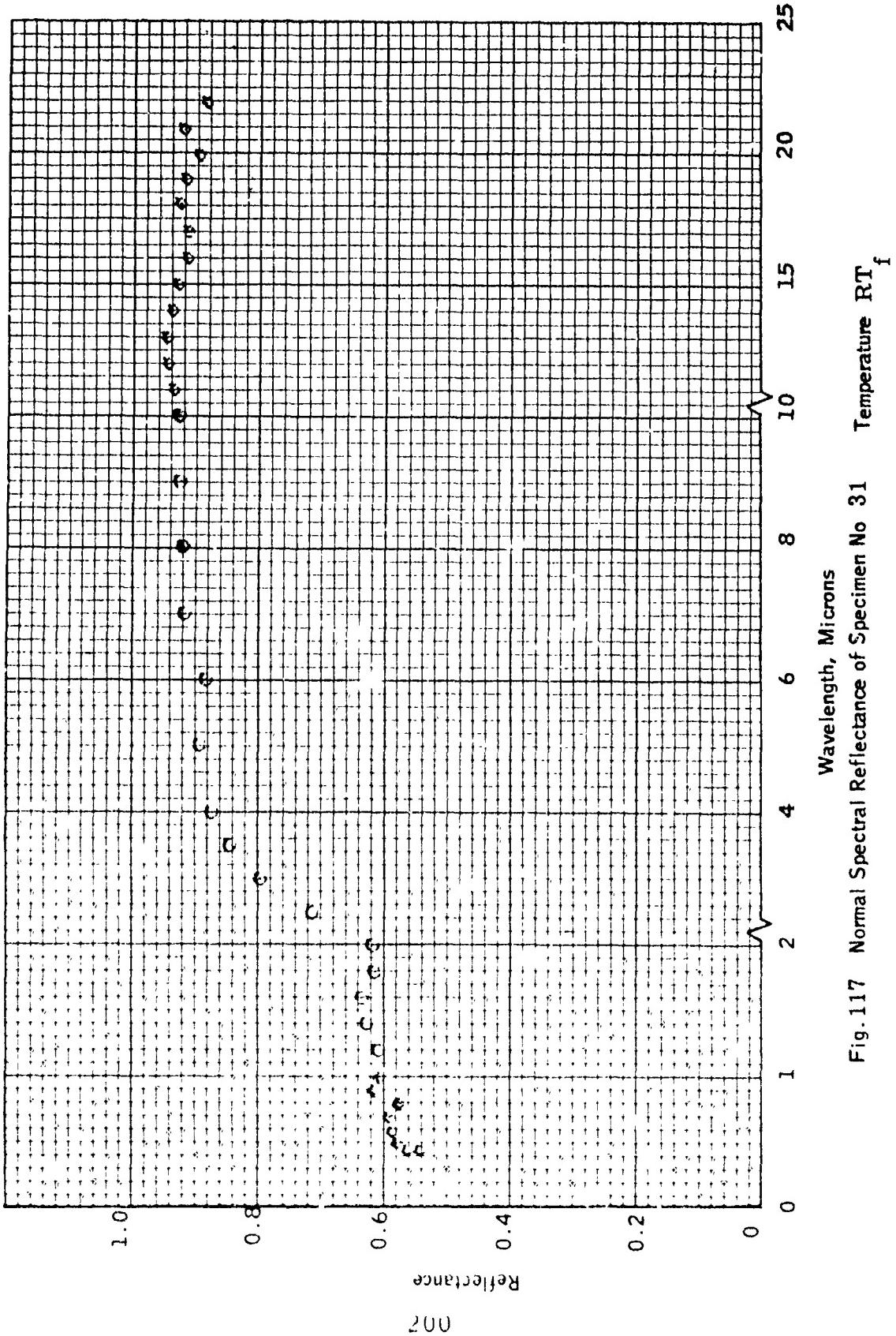


Fig. 117 Normal Spectral Reflectance of Specimen No. 31 Temperature  $RT_f$

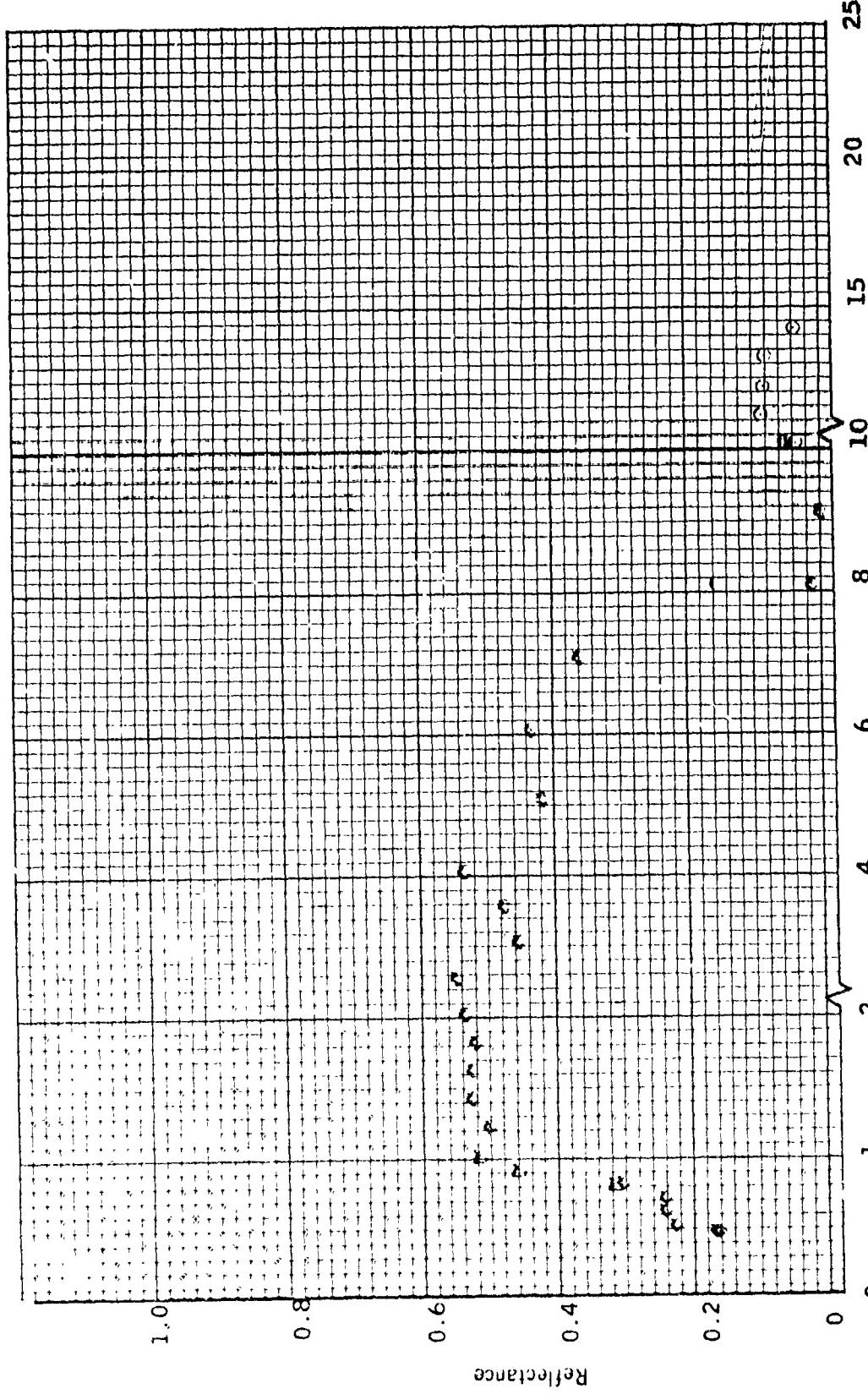


Fig. 118 Normal Spectral Reflectance of Specimen No. 35      Temperature RT HT-800 F

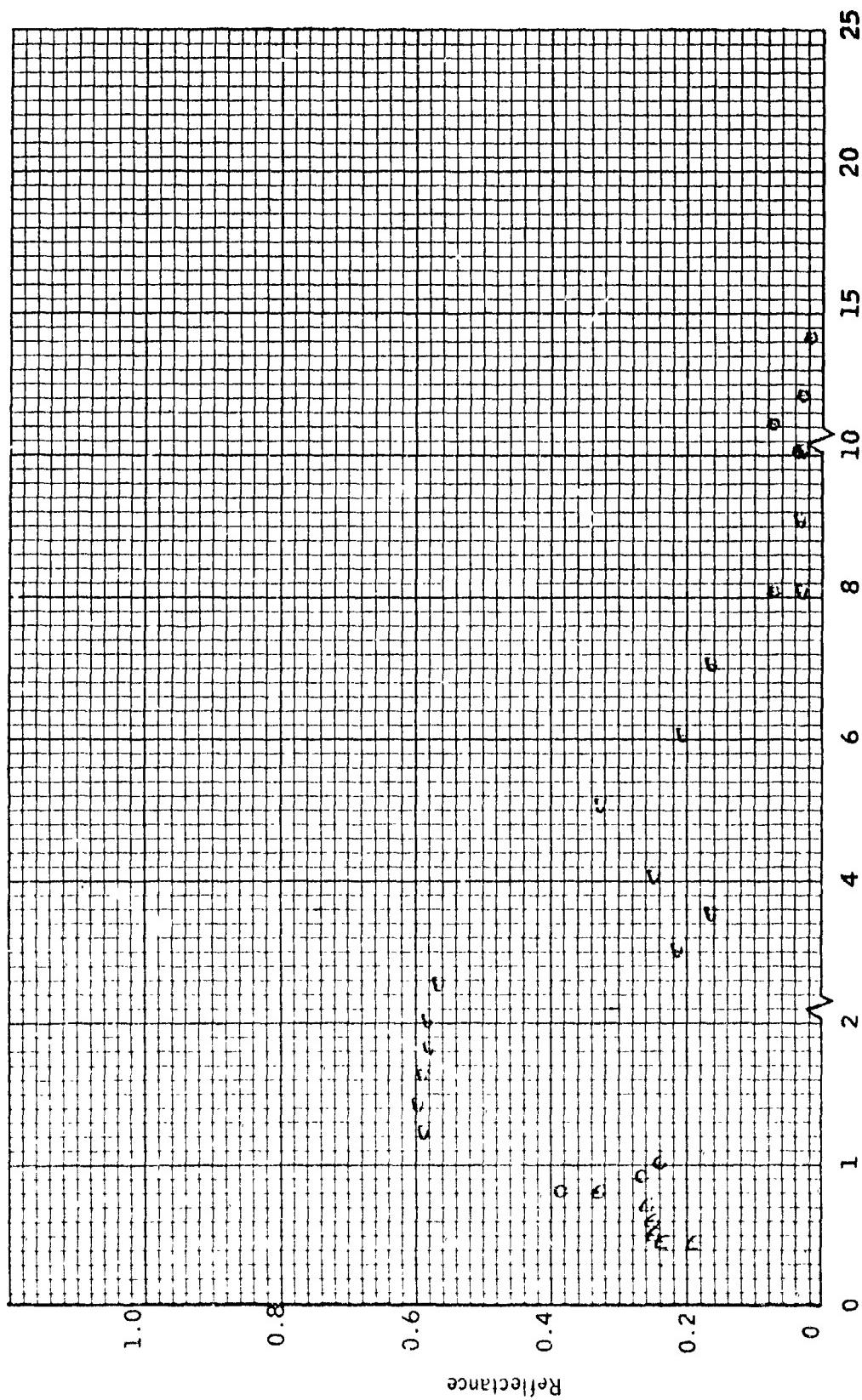


Fig. 119 Normal Spectral Reflectance of Specimen No. 35 Temperature RT

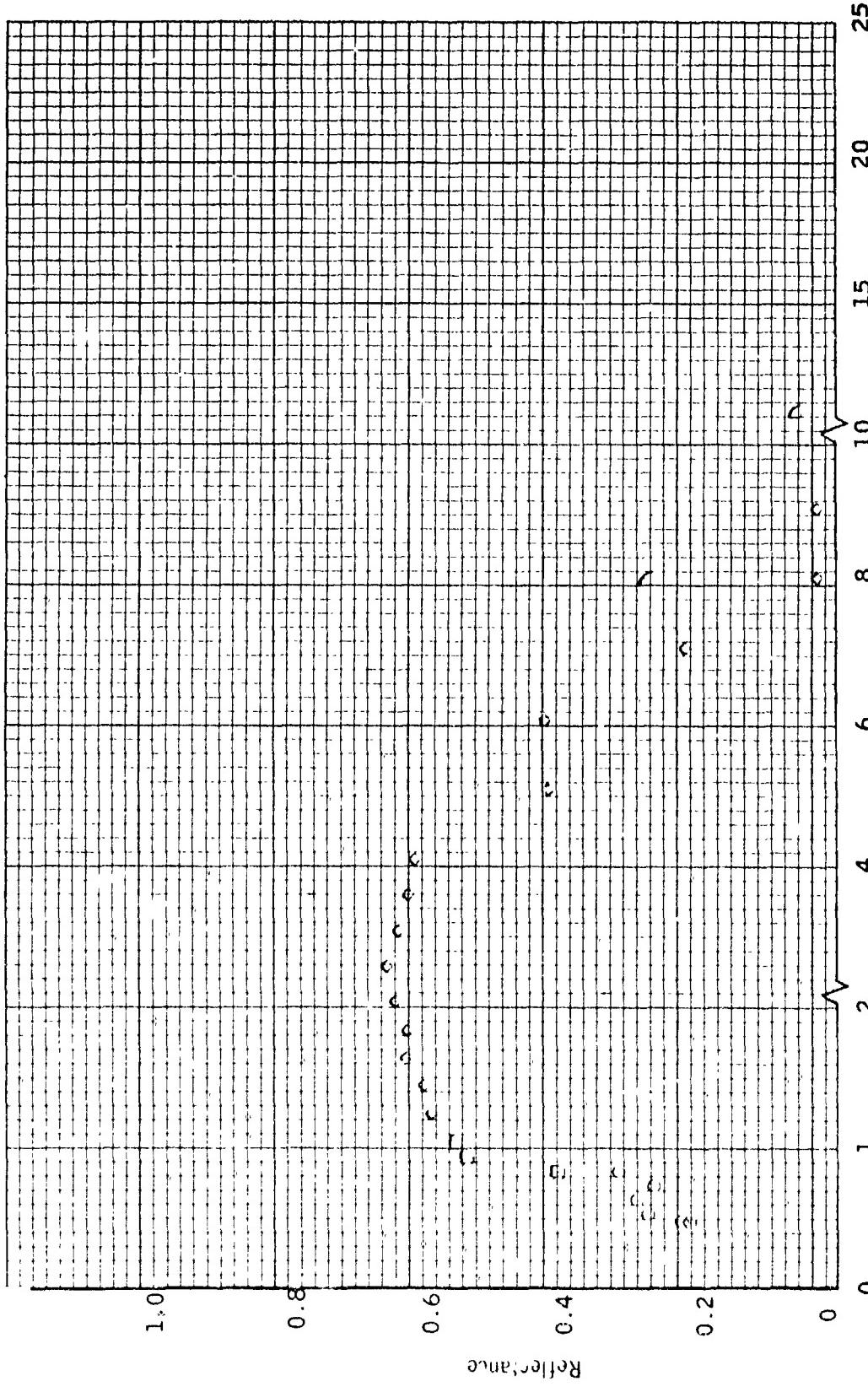


Fig. 120 Normal Spectral Reflectance of Specimen No. 35 Temperature 350°F

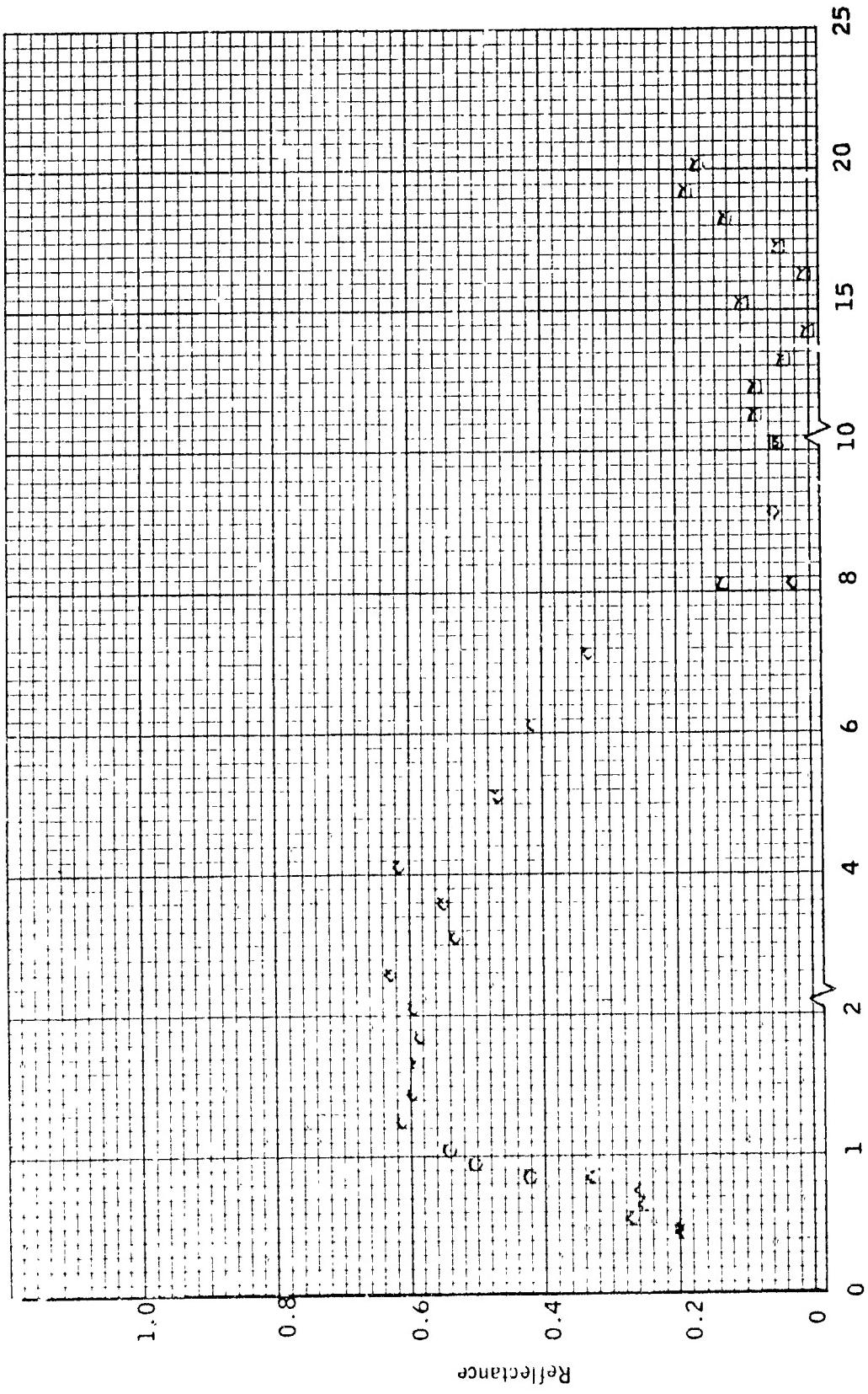


Fig. 121 Normal Spectral Reflectance of Specimen No. 35 Temperature  $RT_f$

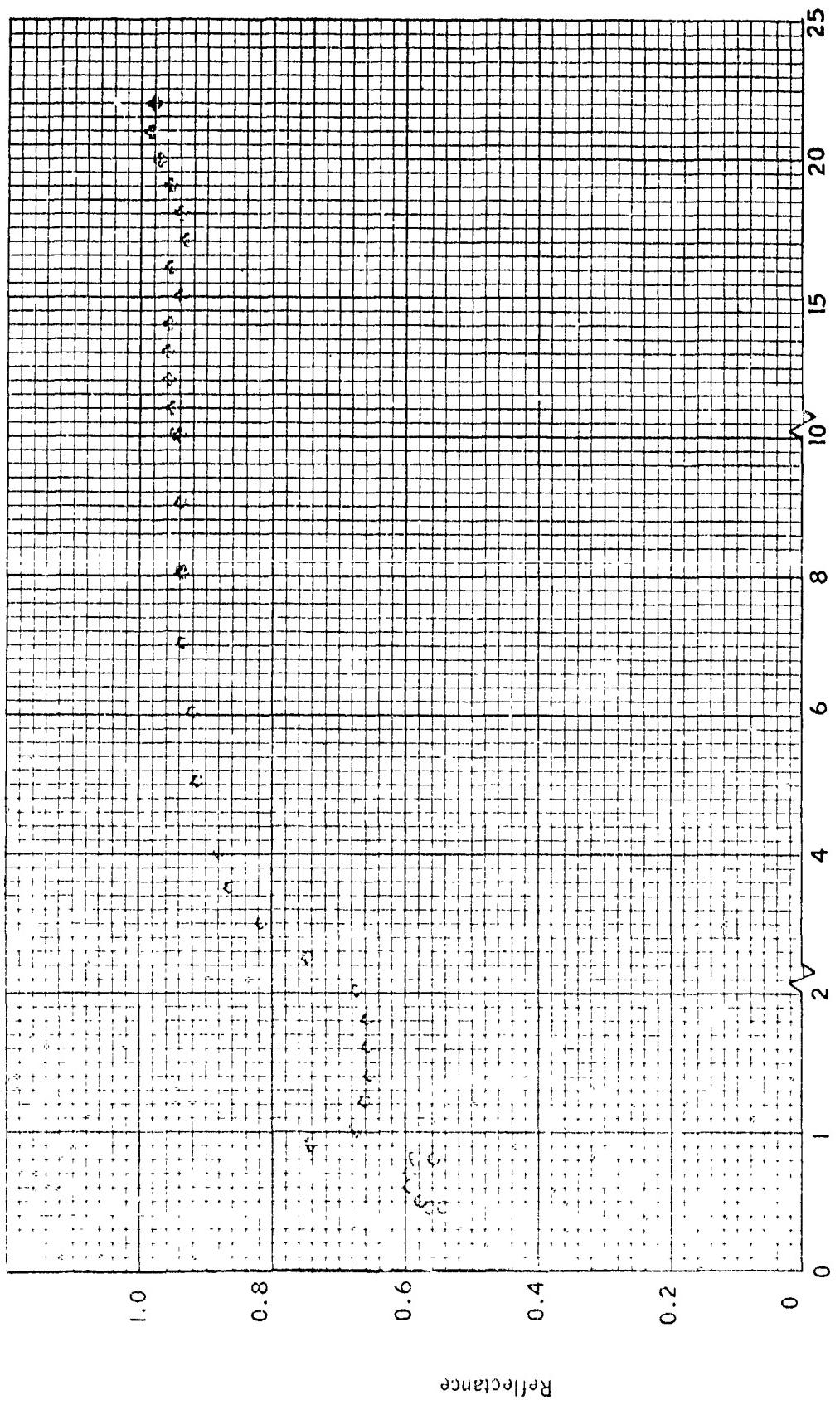


Fig. 122 Normal Spectral Reflectance of Specimen No 152 Temperature RT

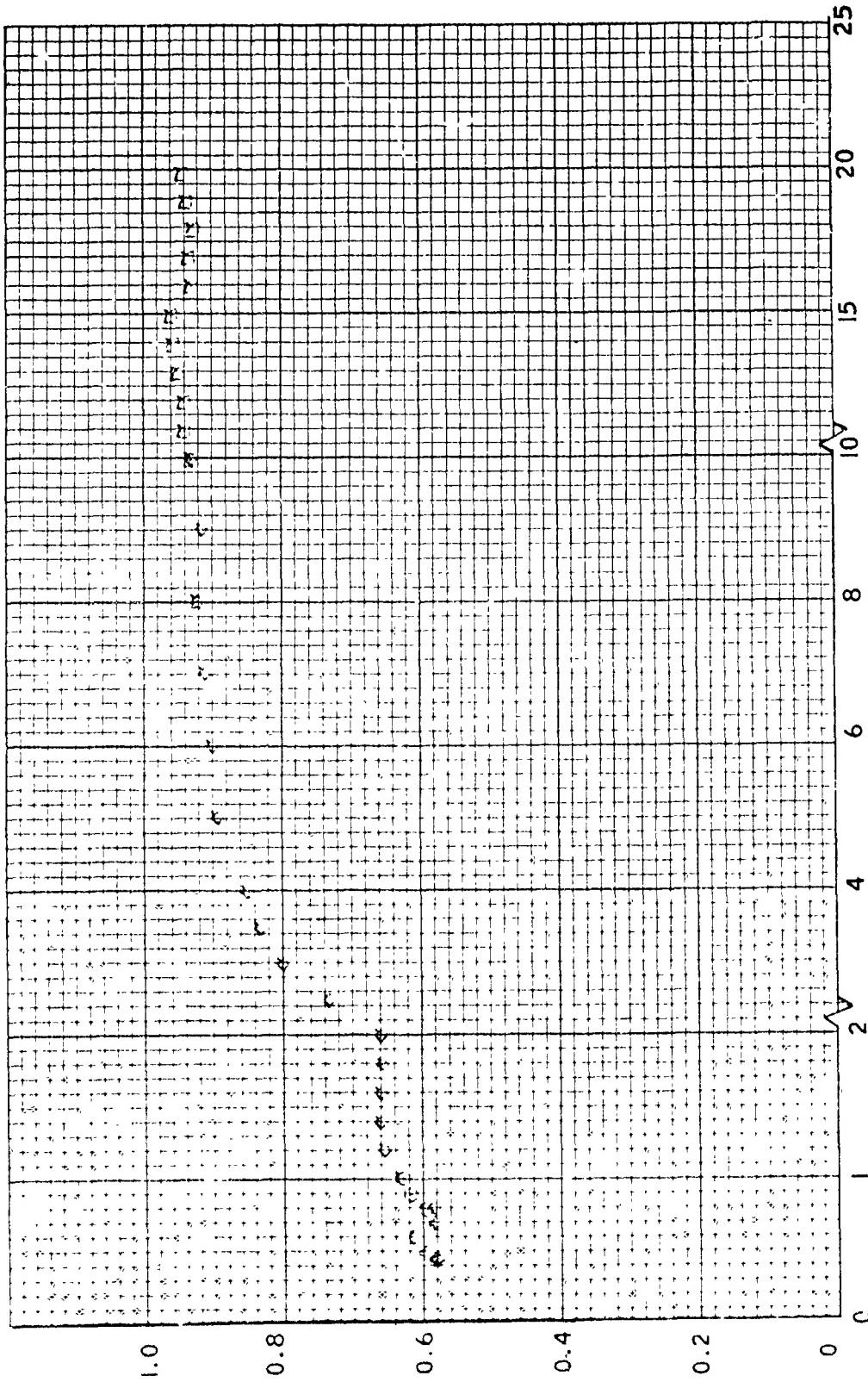


Fig. 123 Normal Spectral Reflectance of Specimen No 152 Temperature RT, MF

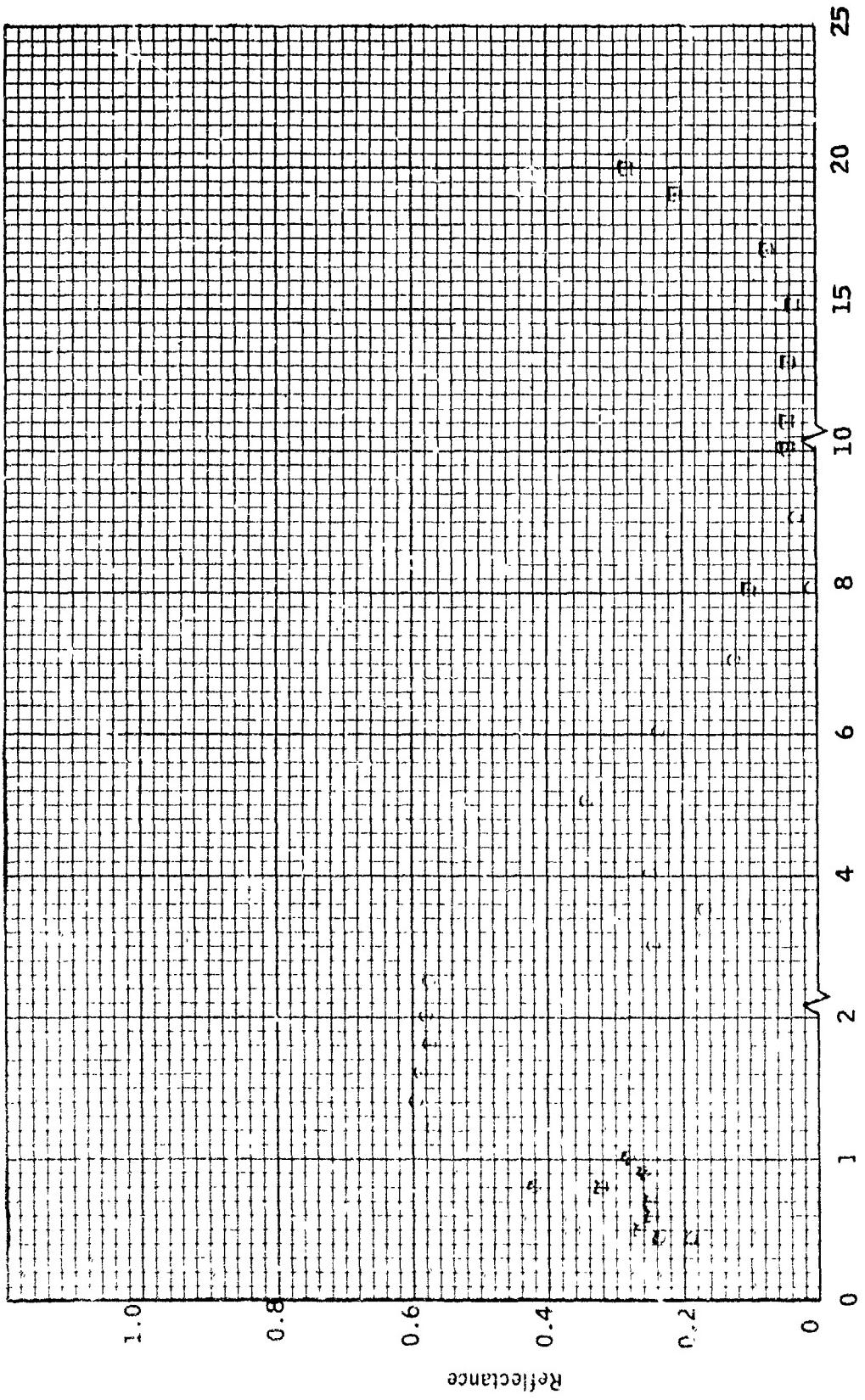


Fig. 124 Normal Spectral Reflectance of Specimen No. 158 Temperature RT

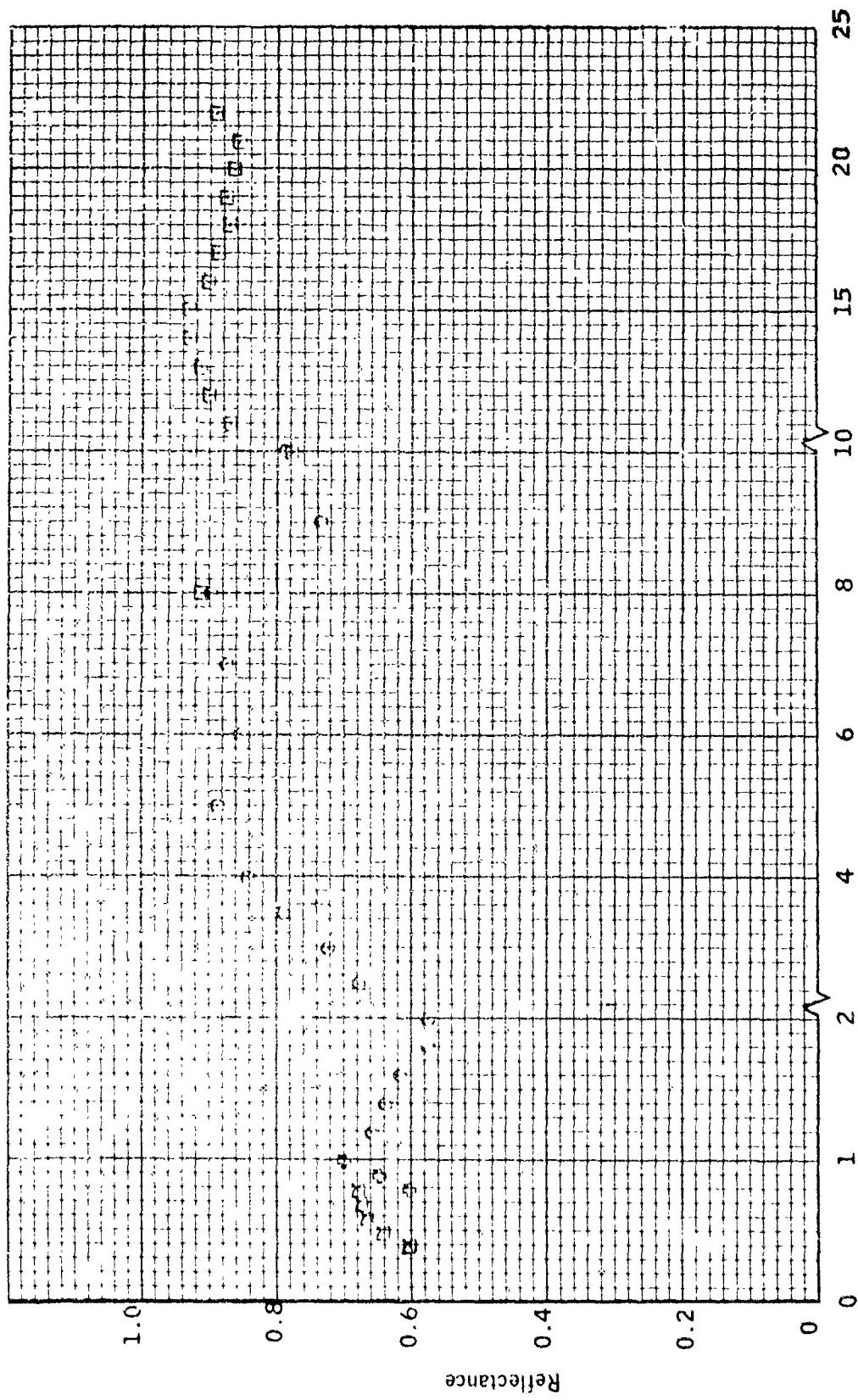


Fig 125 Normal Spectral Reflectance of Specimen No 212 Temperature RT

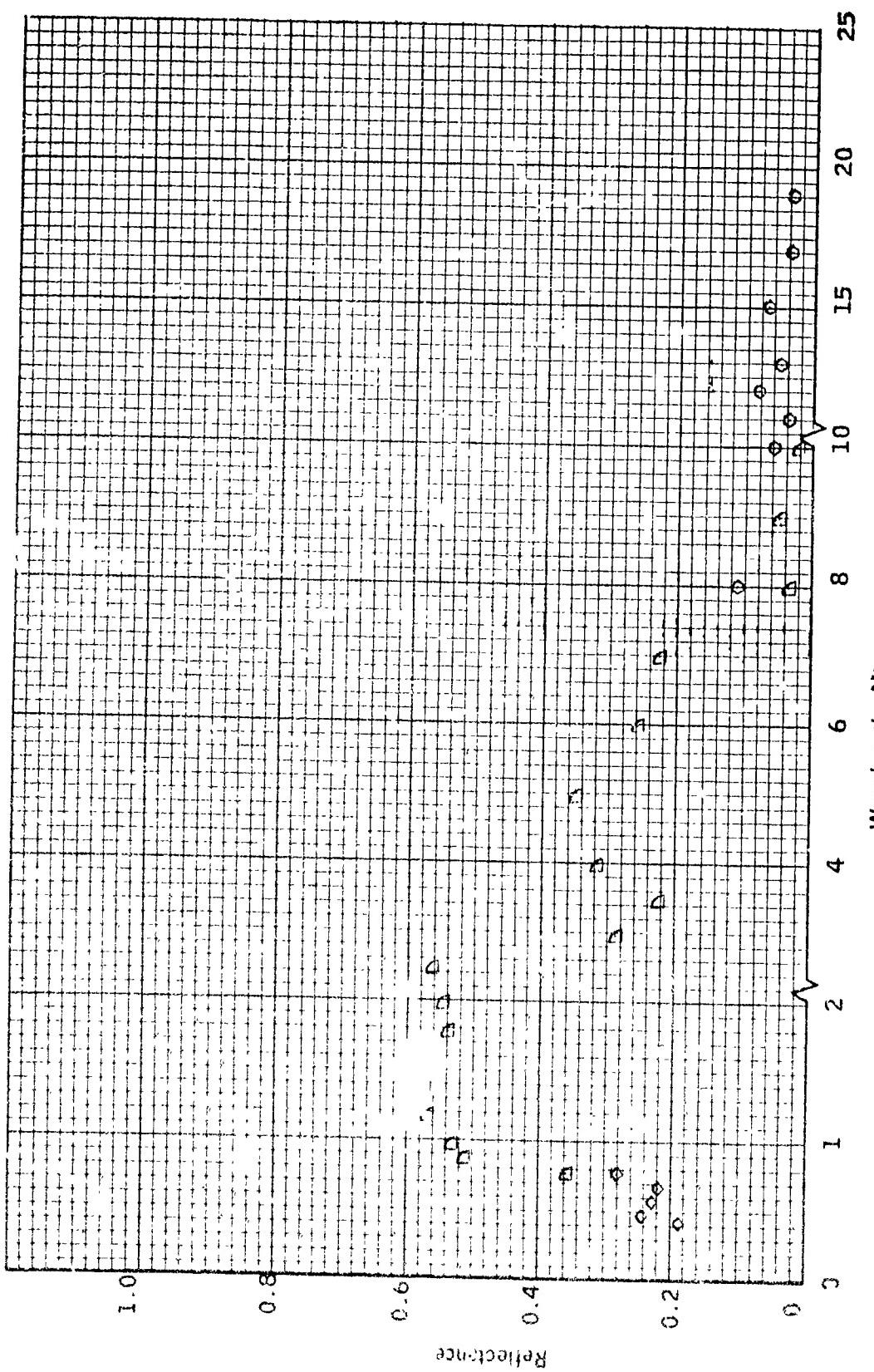


Fig. 126 Normal Spectral Reflectance of Specimen No. 213 Temperature RT

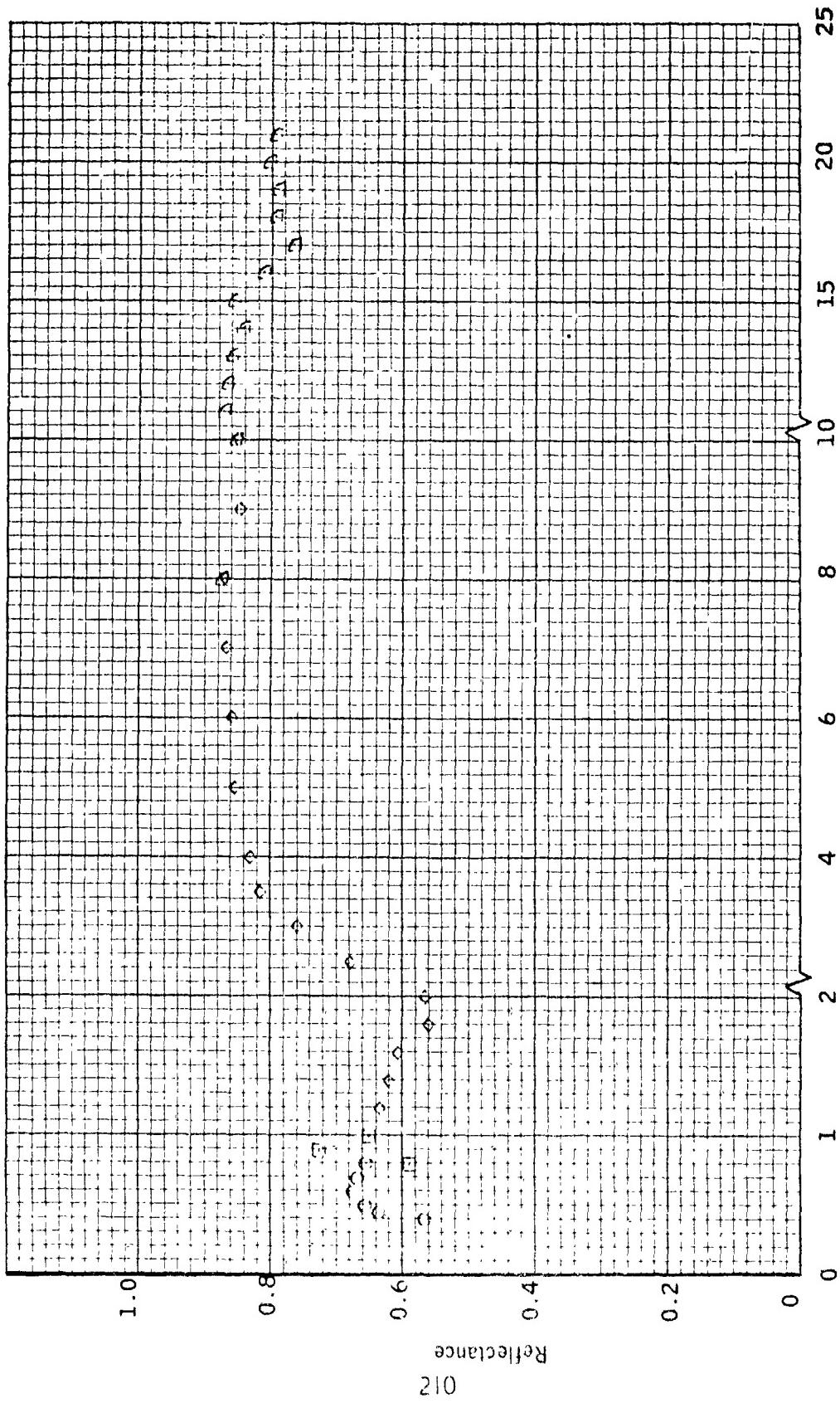


Fig. 127 Normal Spectral Reflectance of Specimen No. 214 Temperature RT

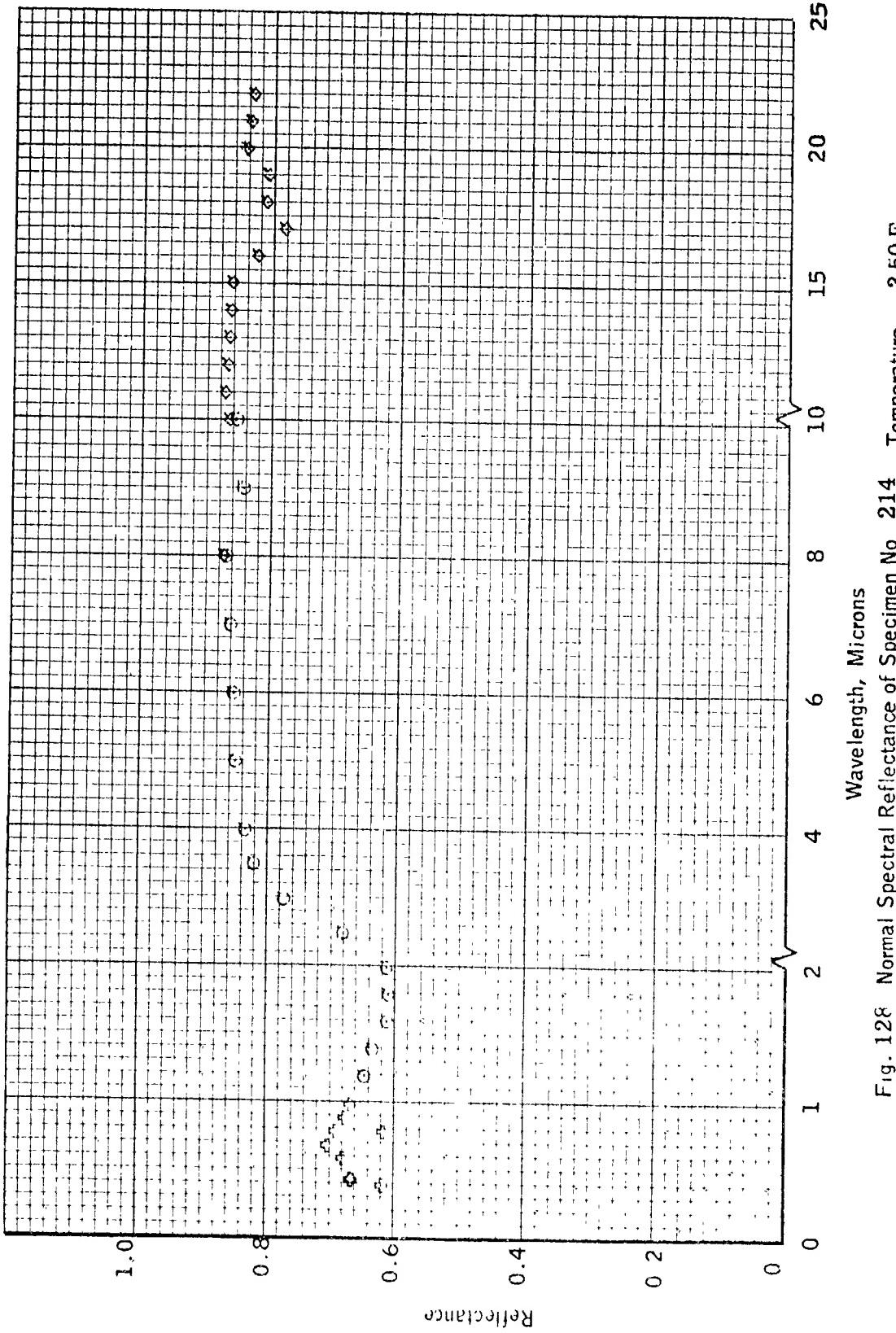


Fig. 128 Normal Spectral Reflectance of Specimen No. 214 Temperature 350 F

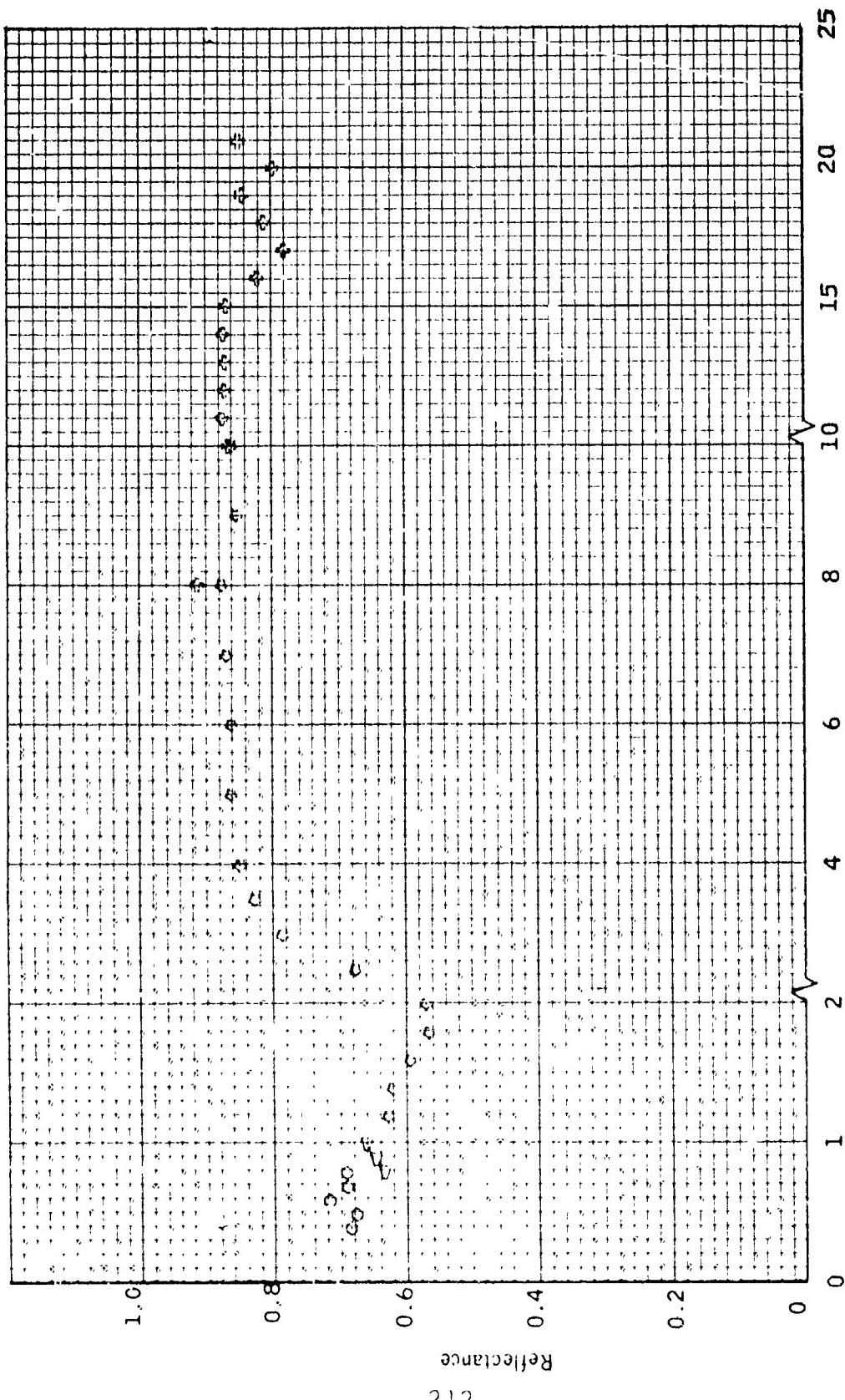


Fig. 129 Normal Spectral Reflectance of Specimen No. 214 Temperature  $RT_f$

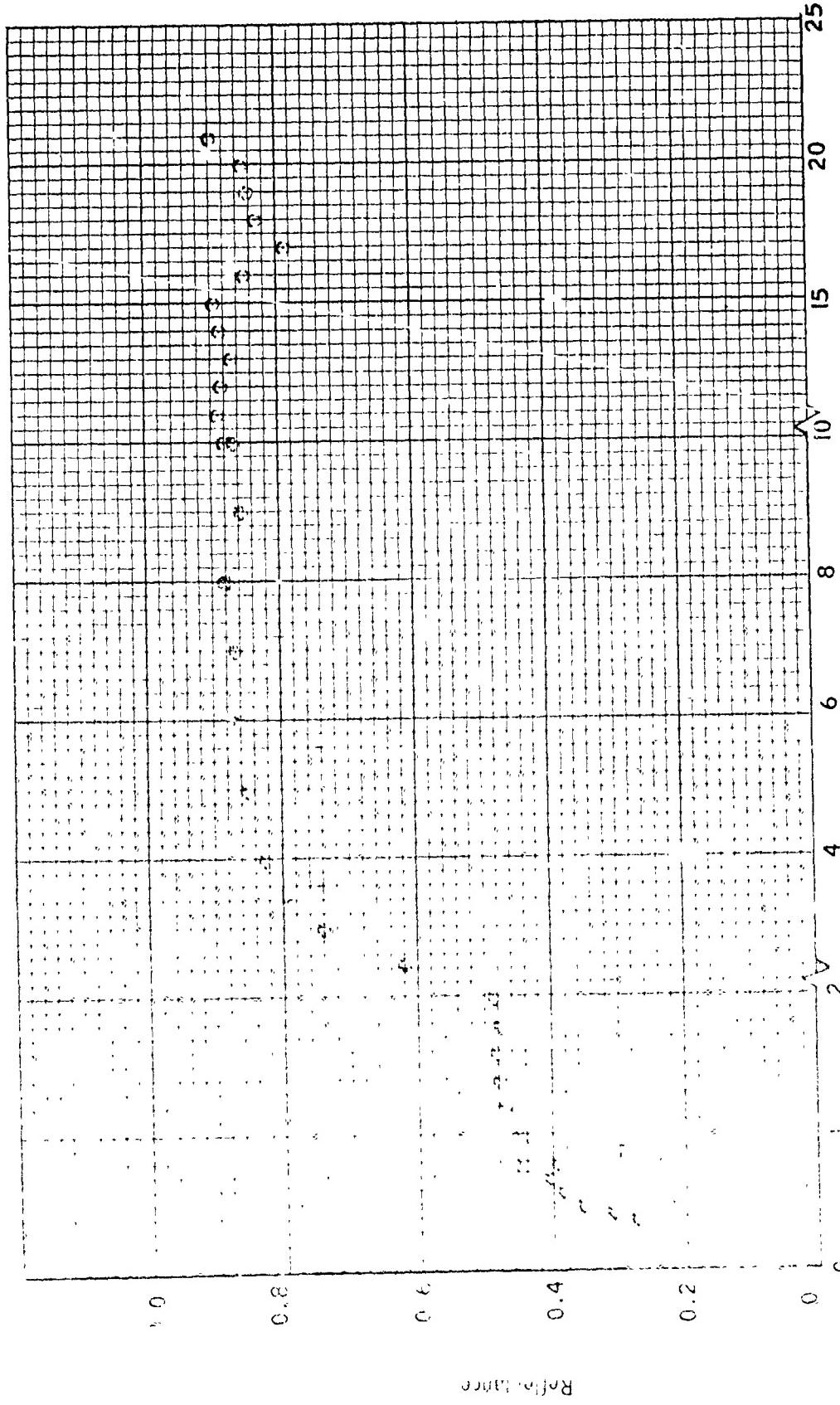


Fig. 130 Normal Spectral Reflectance of Specimen No. 214 Temperature RT, HT-800 F

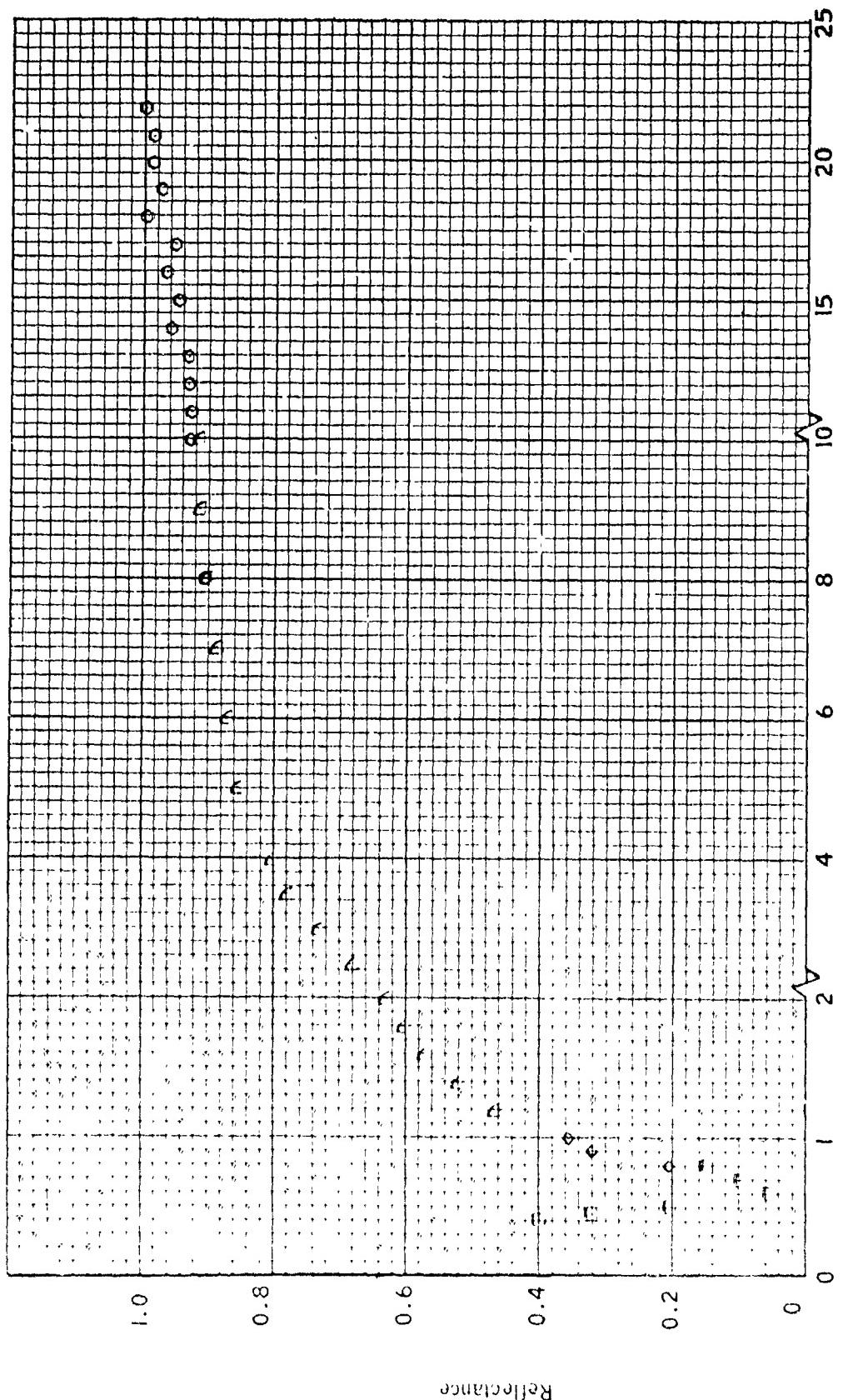


Fig. 131 Normal Spectral Reflectance of Specimen No. 38 Temperature RT

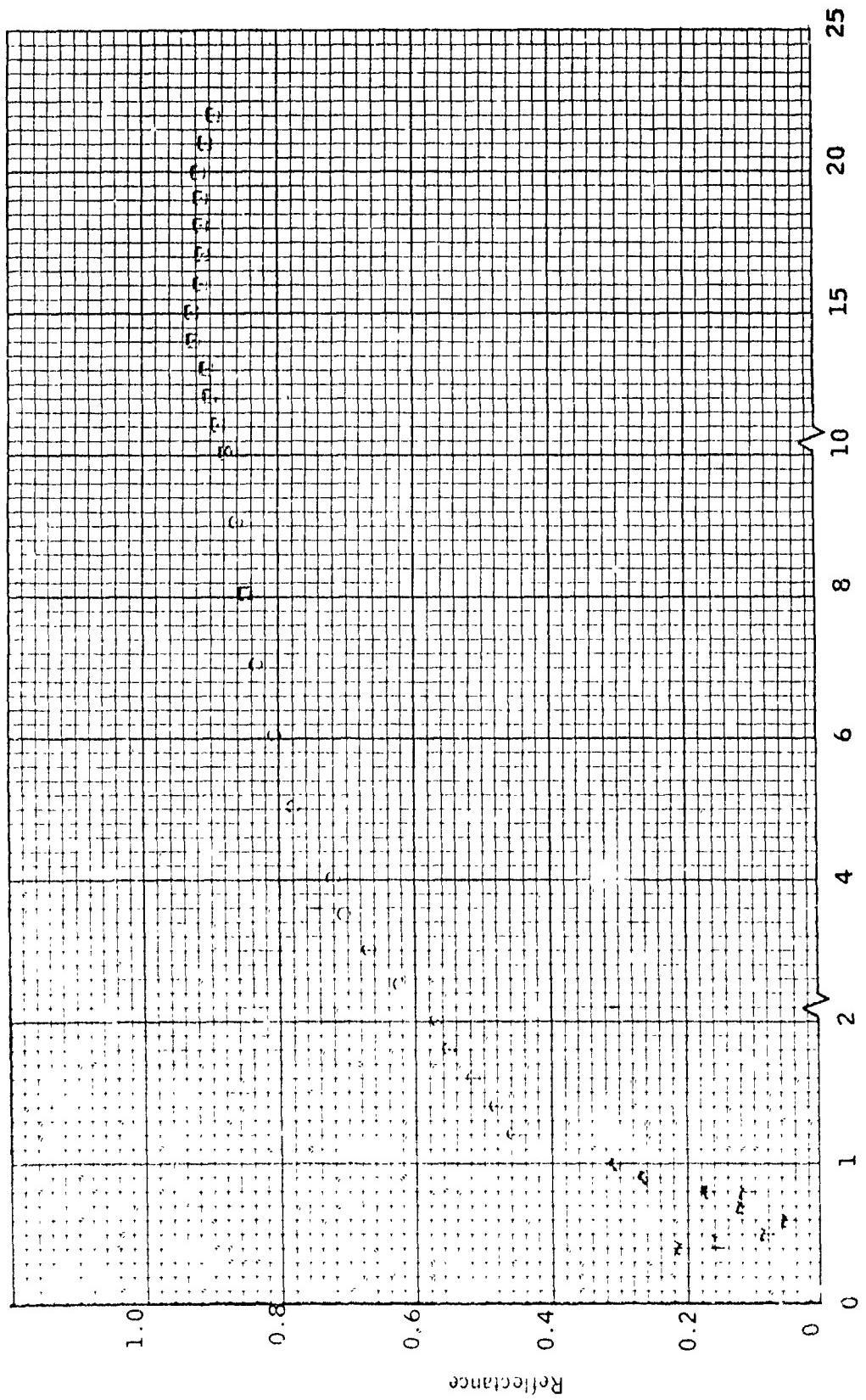


Fig. 132 Normal Spectral Reflectance of Specimen No 43 Temperature RT

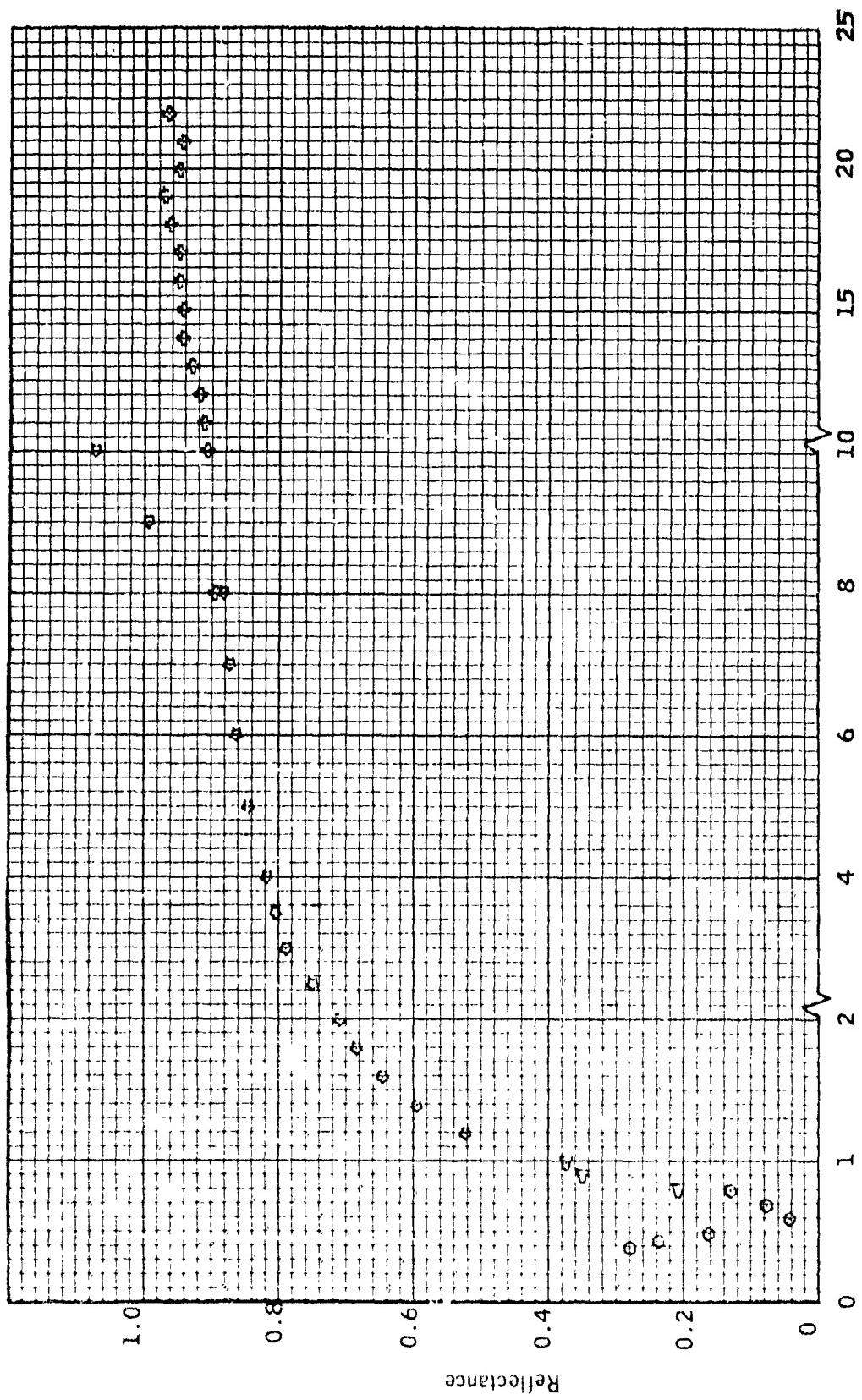


Fig. 133 Normal Spectral Reflectance of Specimen No. 203 Temperature RT

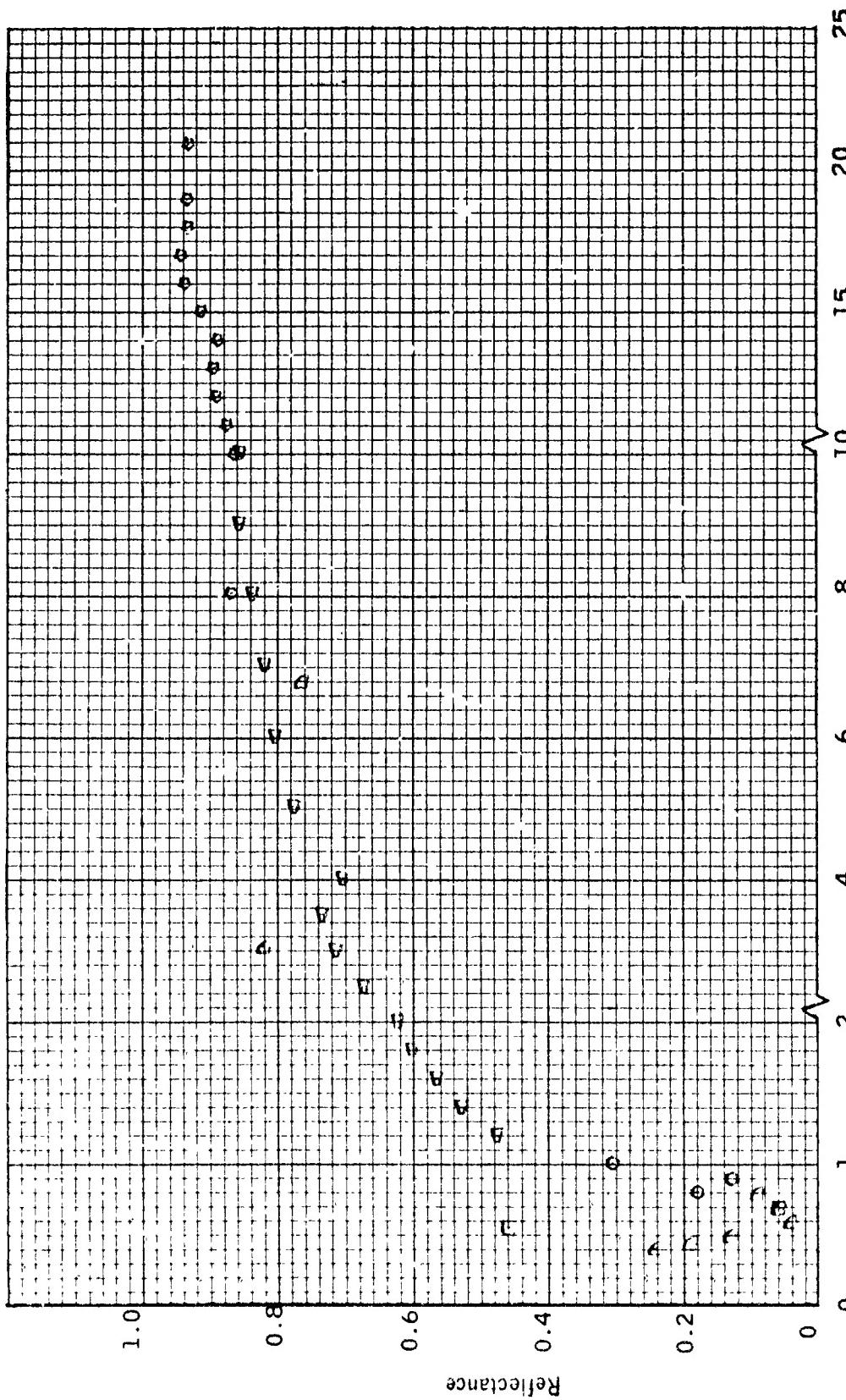


Fig. 134 Normal Spectral Reflectance of Specimen No. 207 Temperature RT  
Wavelength, Microns

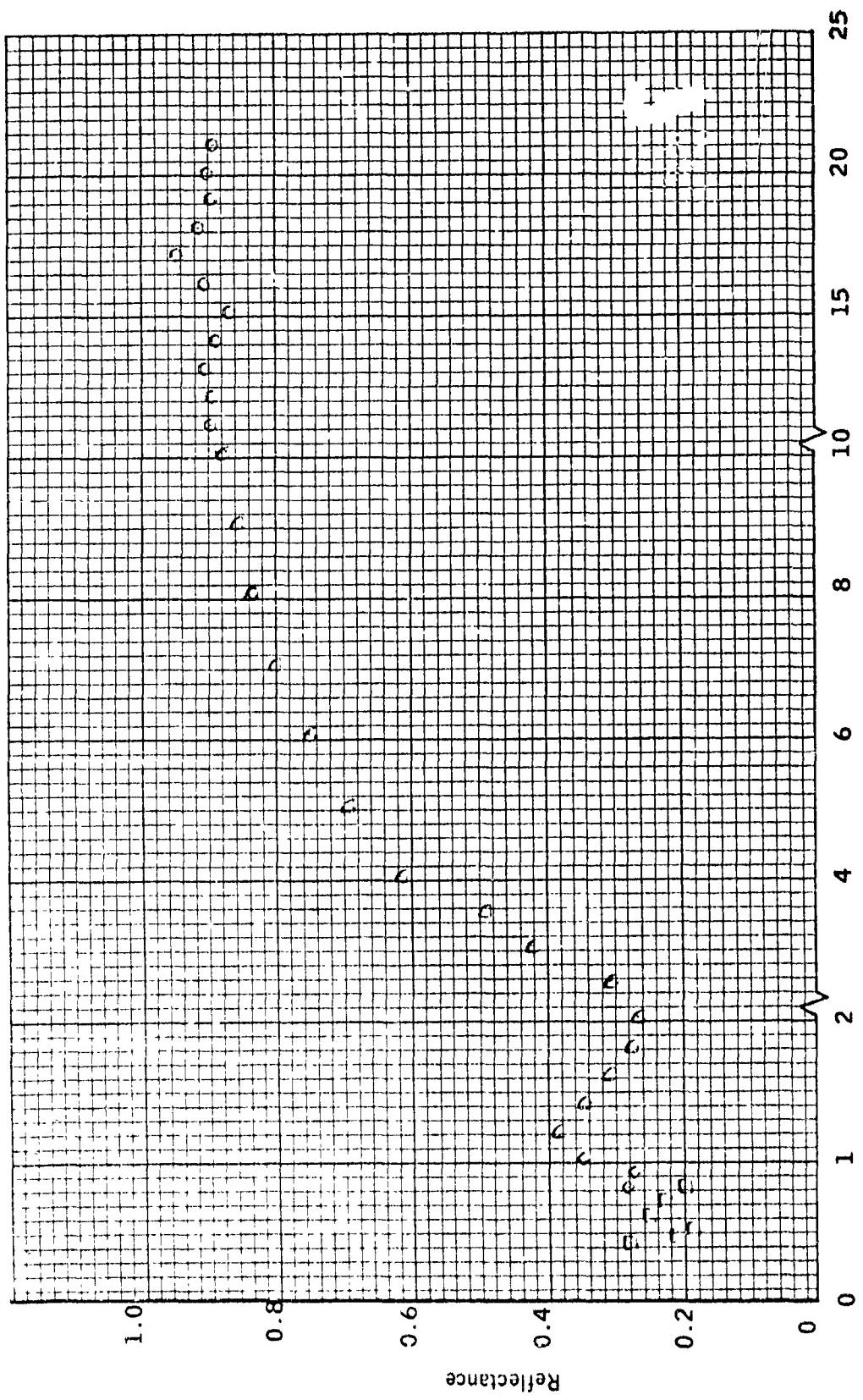


Fig.135 Normal Spectral Reflectance of Specimen No. 45 Temperature RT

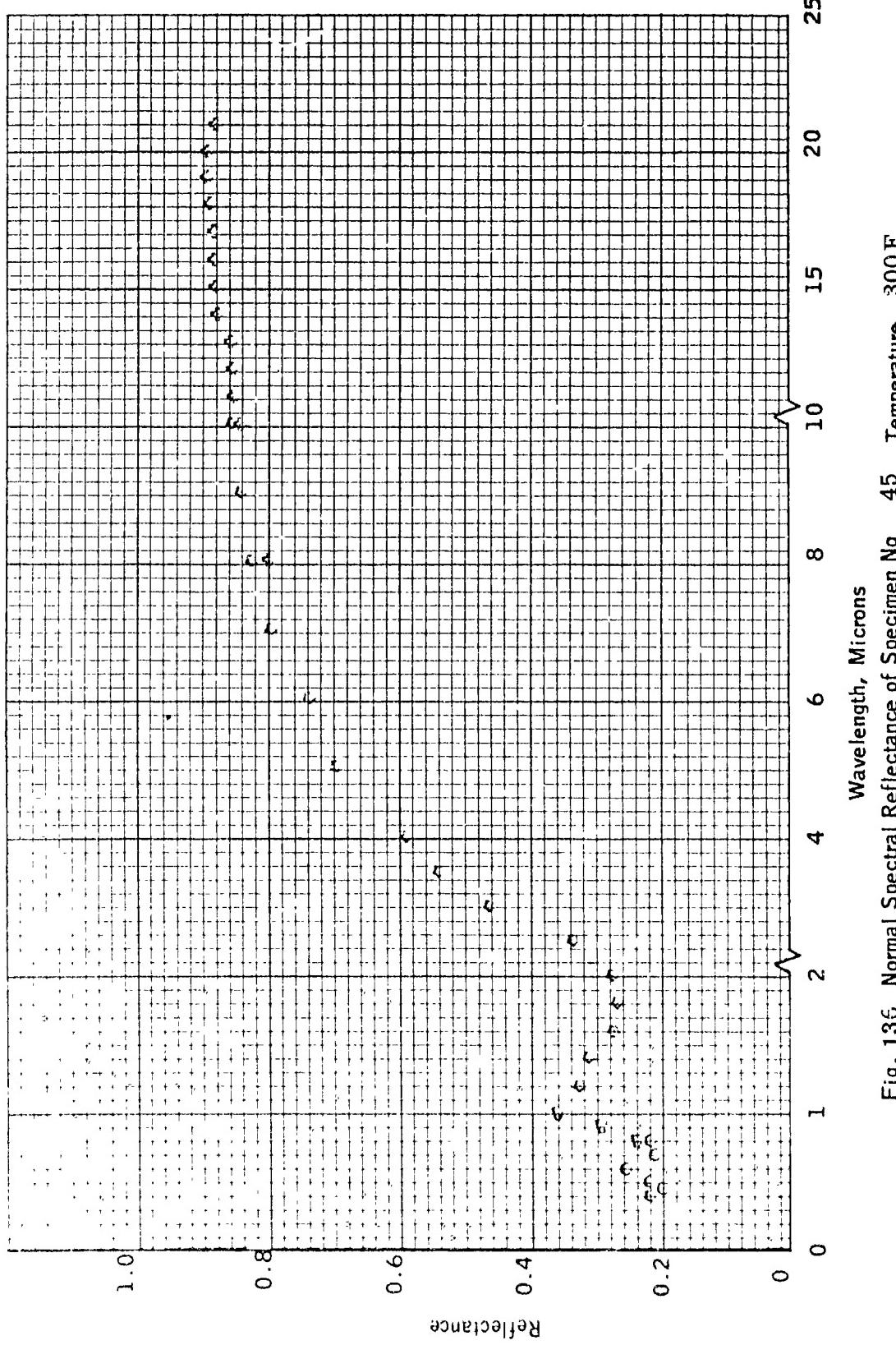


Fig. 136 Normal Spectral Reflectance of Specimen No. 45 Temperature 300°F

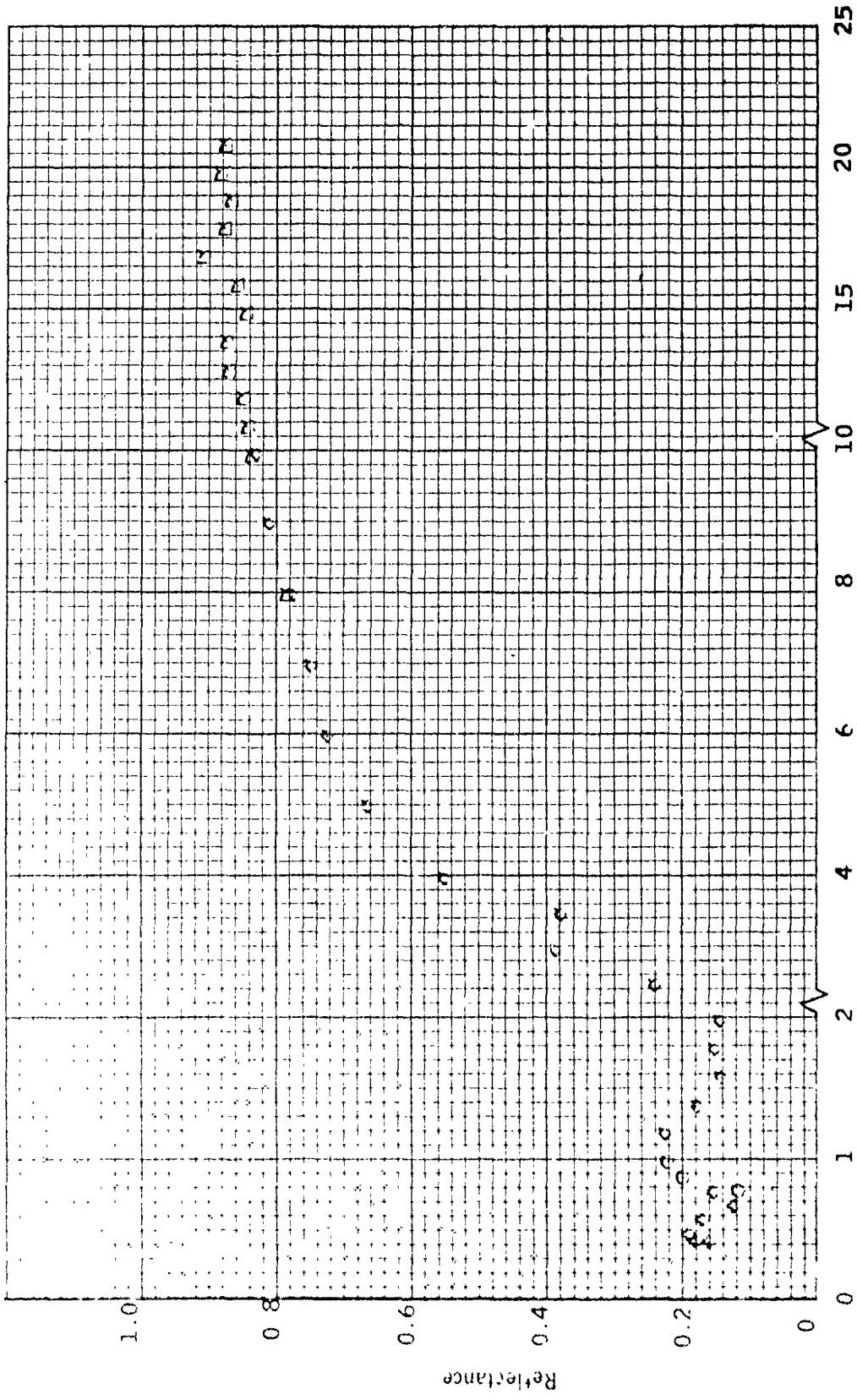


Fig. 137 Normal Spectral Reflectance of Specimen No 45 Temperature 600 F

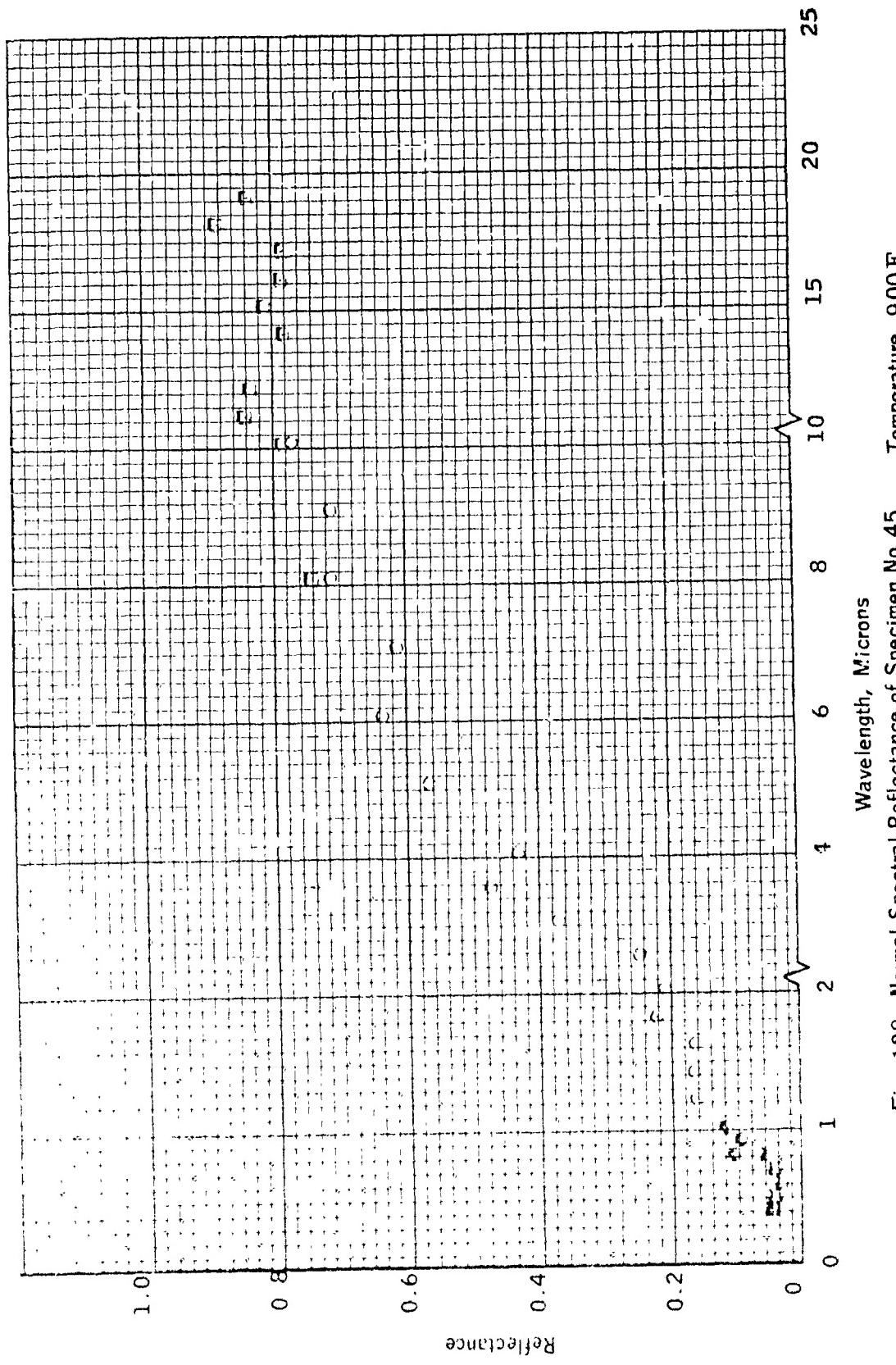


Fig. 138 Normal Spectral Reflectance of Specimen No 45  
Temperature 900°F

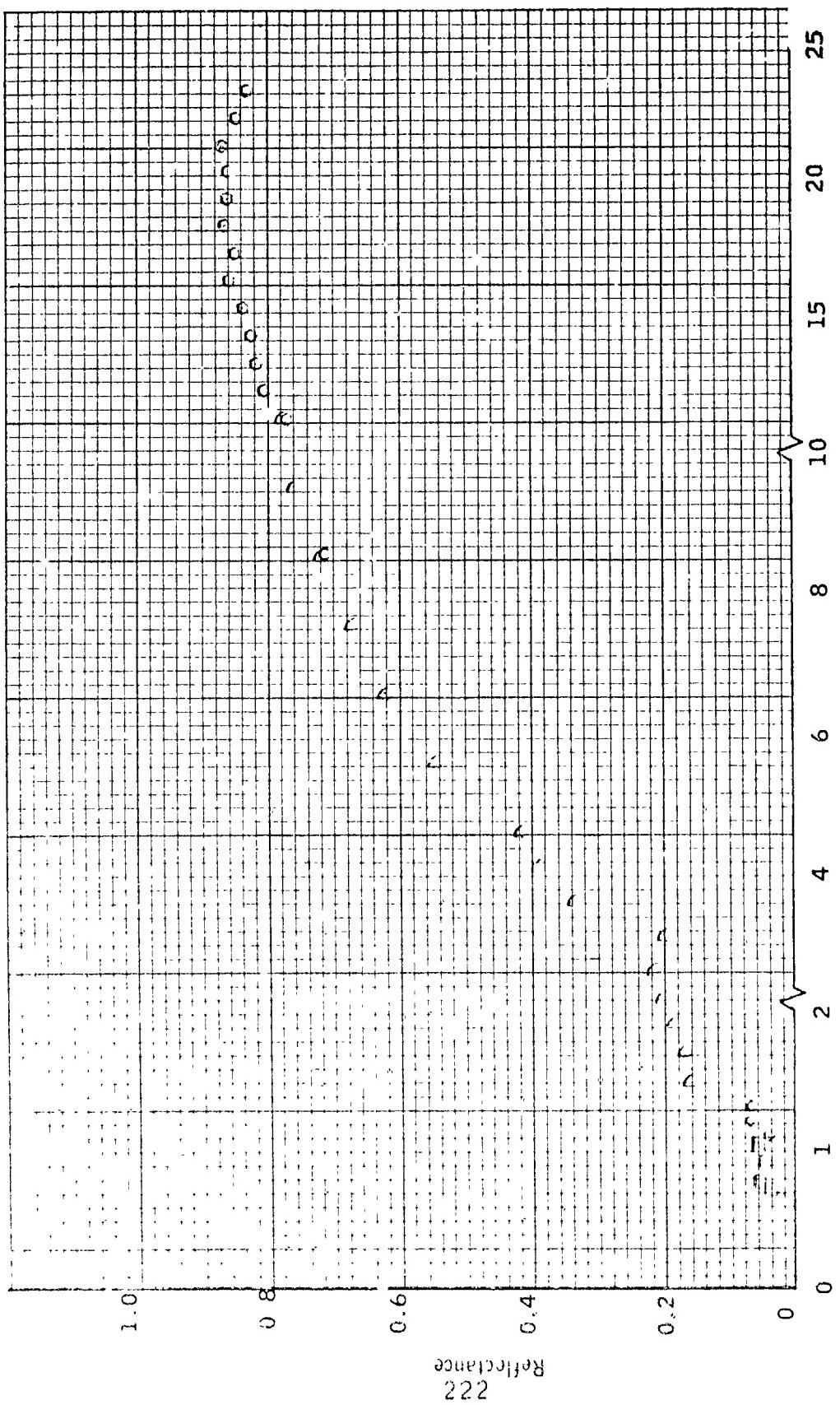


Fig. 139 Normal Spectral Reflectance of Specimen No. 45 Temperature  $RT_f$

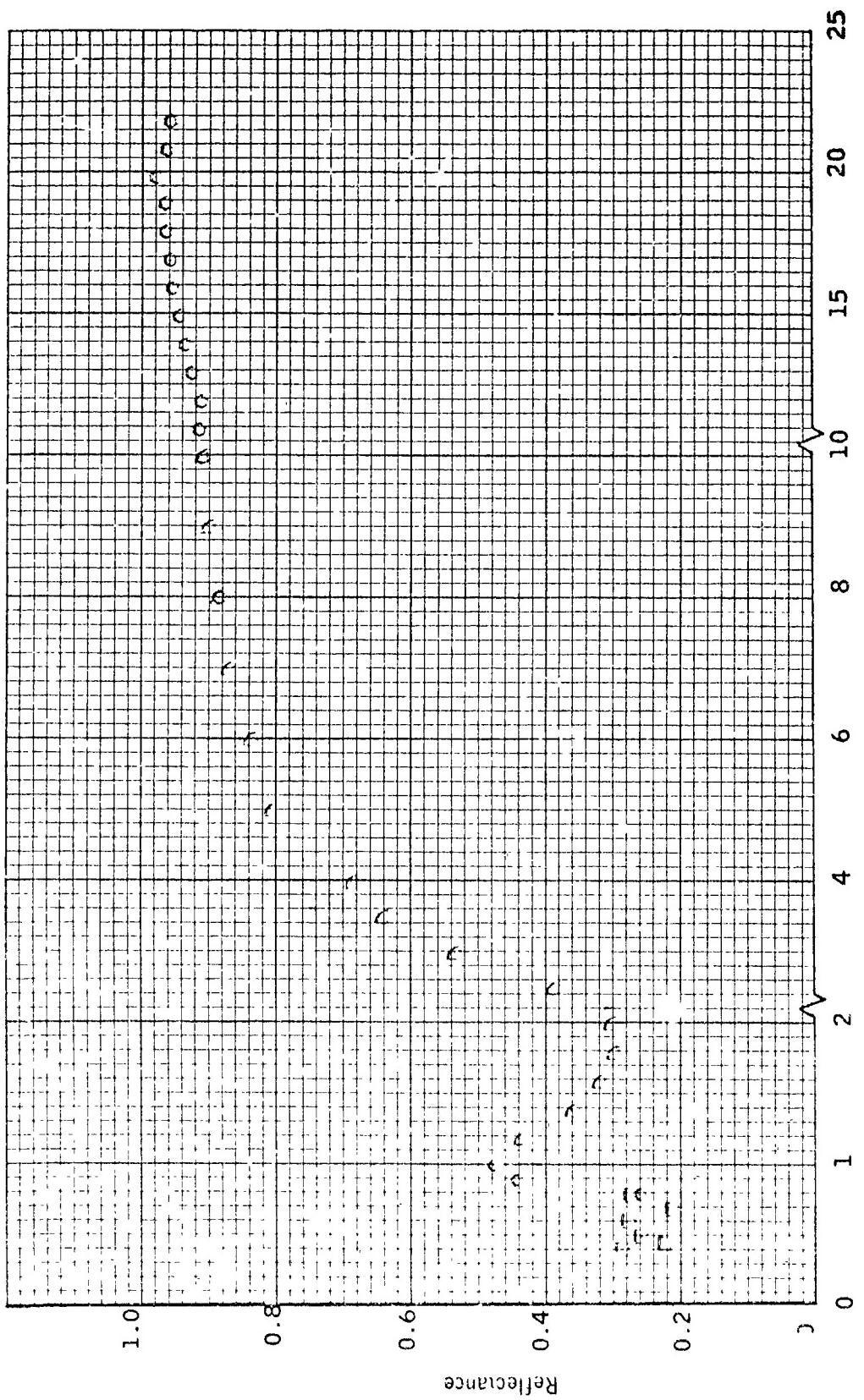


Fig. 140 Normal Spectral Reflectance of Specimen No. 48 Temperature RT



Fig. 141 Normal Spectral Reflectance of Specimen No 48 Temperature 500°F

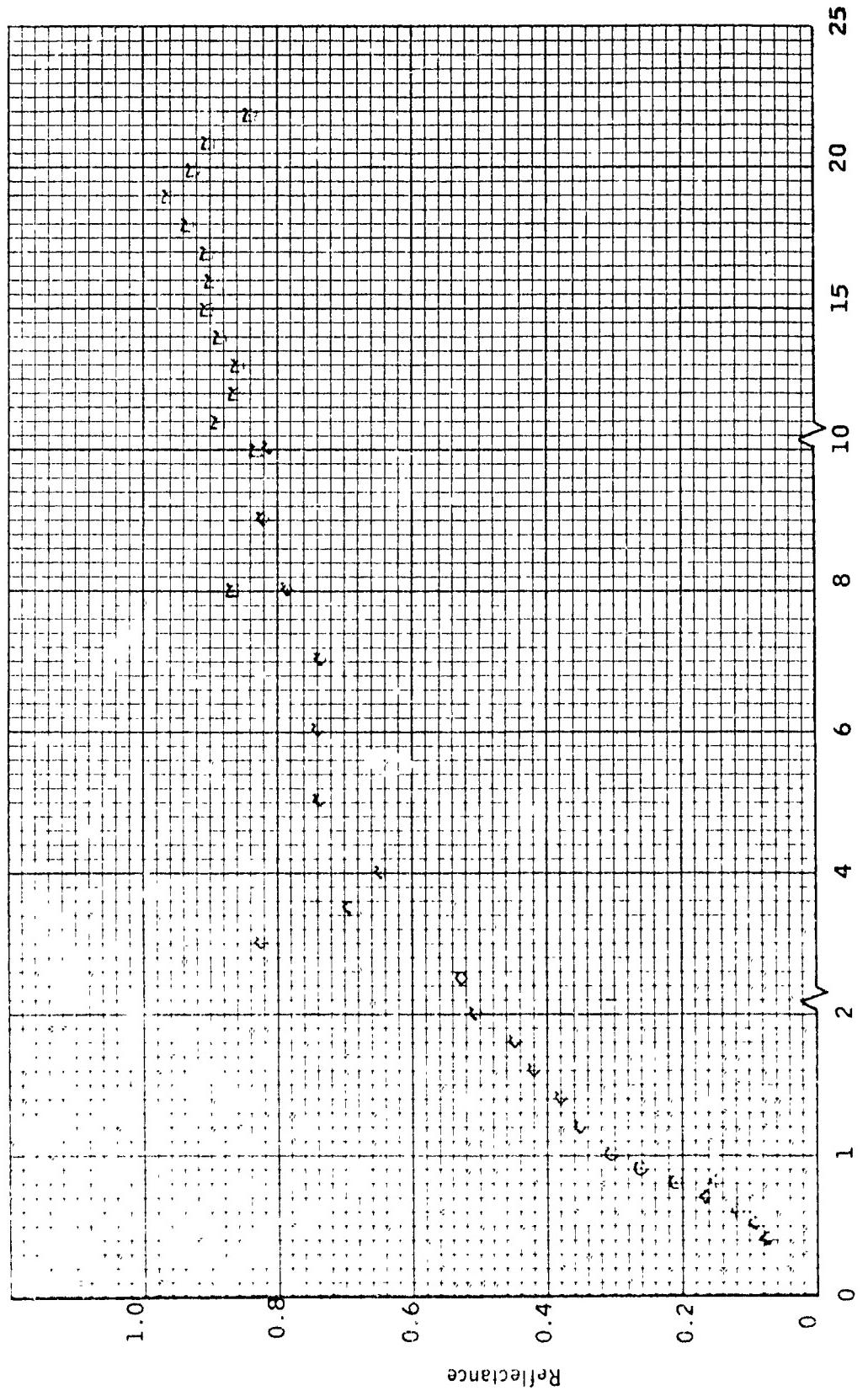


Fig. 142 Normal Spectral Reflectance of Specimen No. 48 Temperature 1000°F

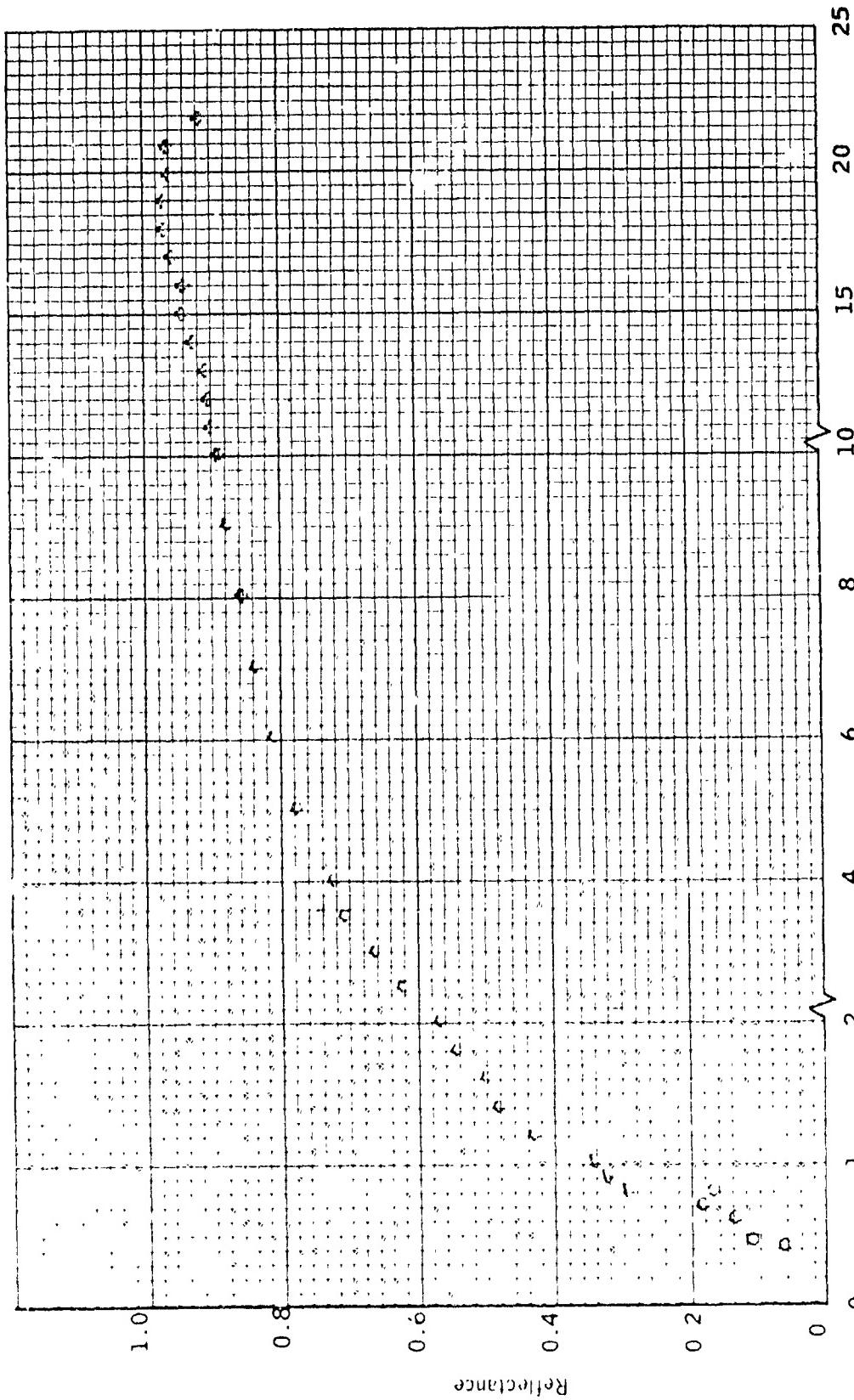


Fig. 143 Normal Spectral Reflectance of Specimen No. 48 Temperature  $RT_f$  (1000°F)

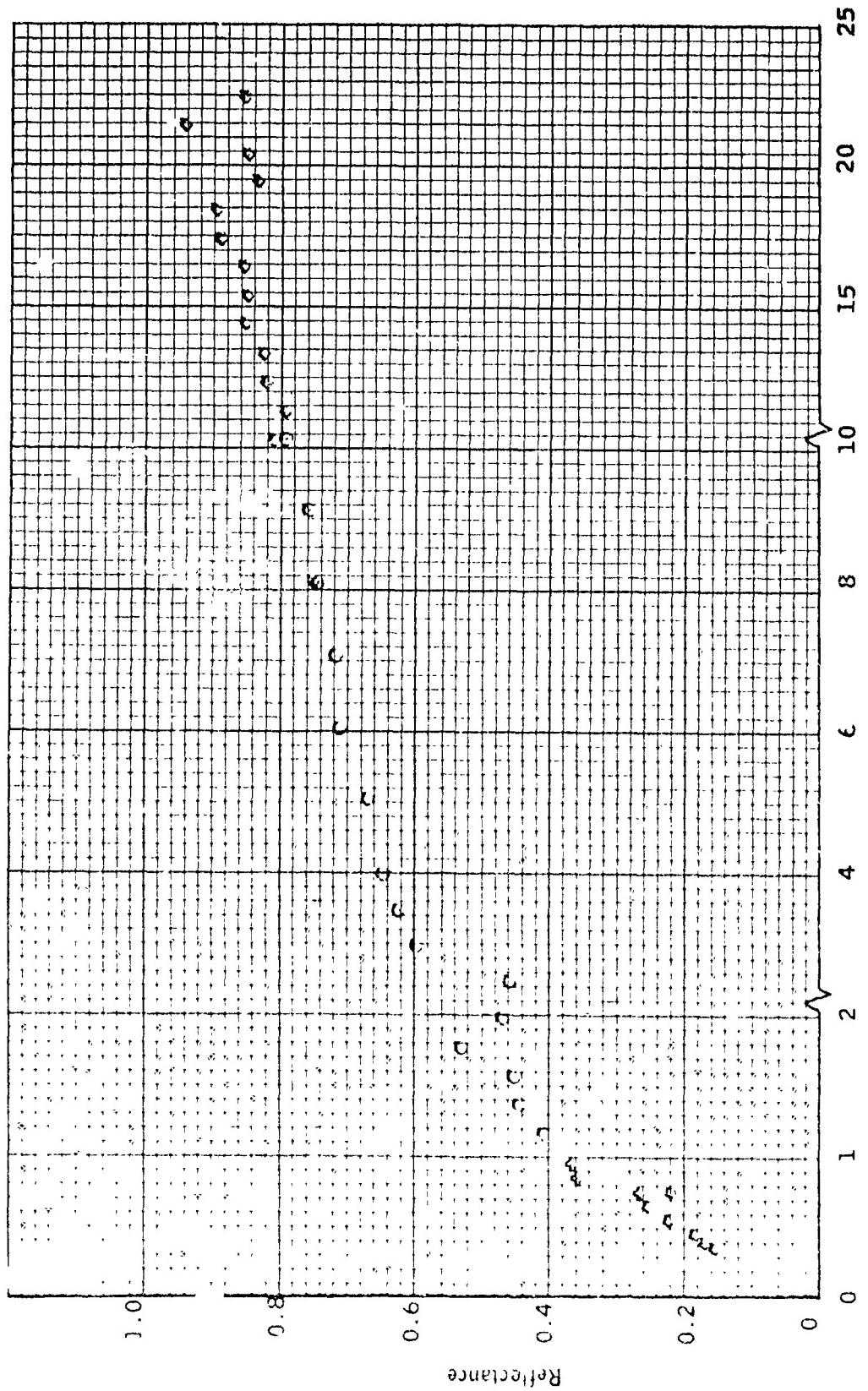


Fig. 144 Normal Spectral Reflectance of Specimen No 48 Temperature 1300°F

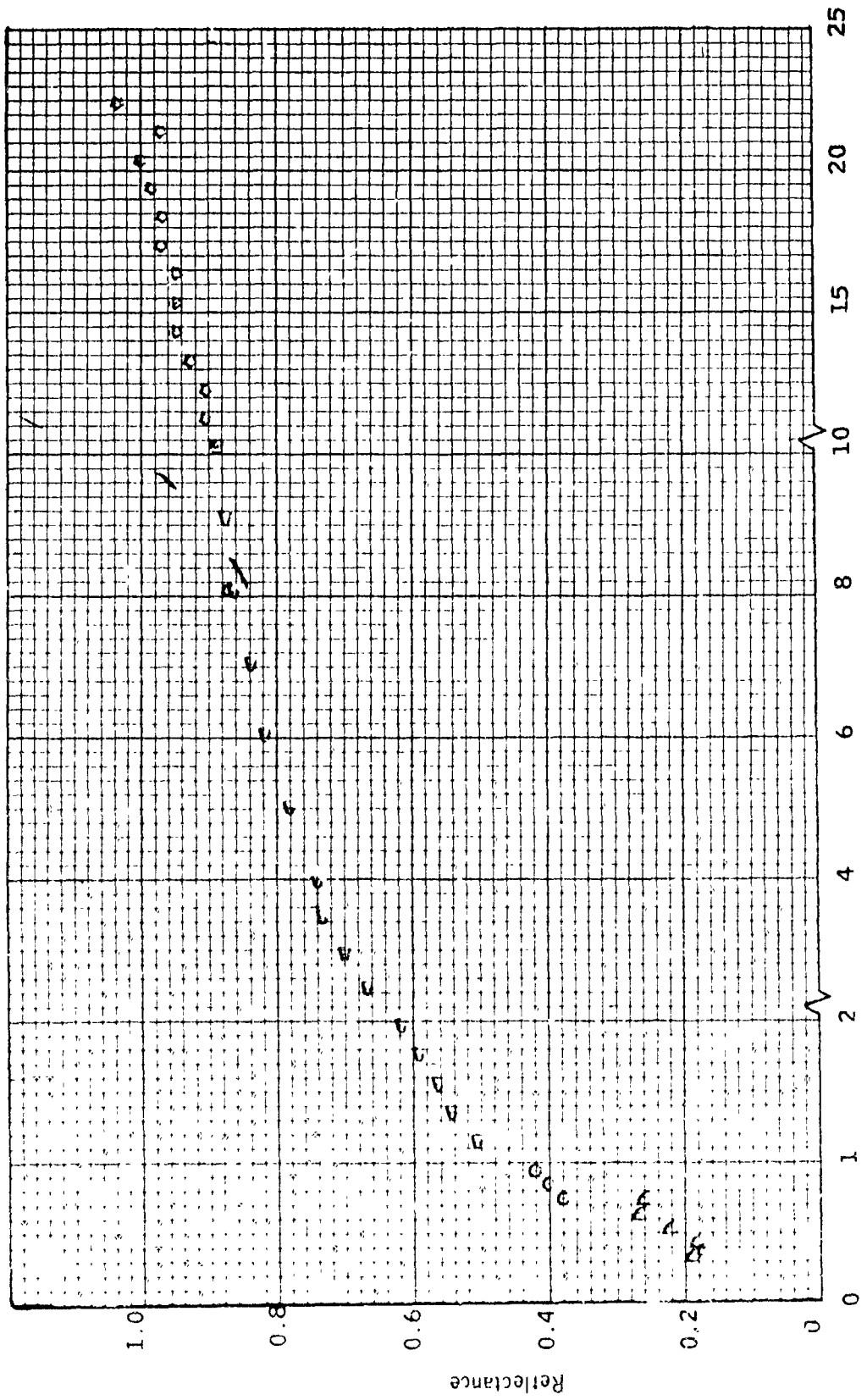


Fig. 145 Normal Spectral Reflectance of Specimen No. 48 Temperature  $RT_f$  (1300°F)

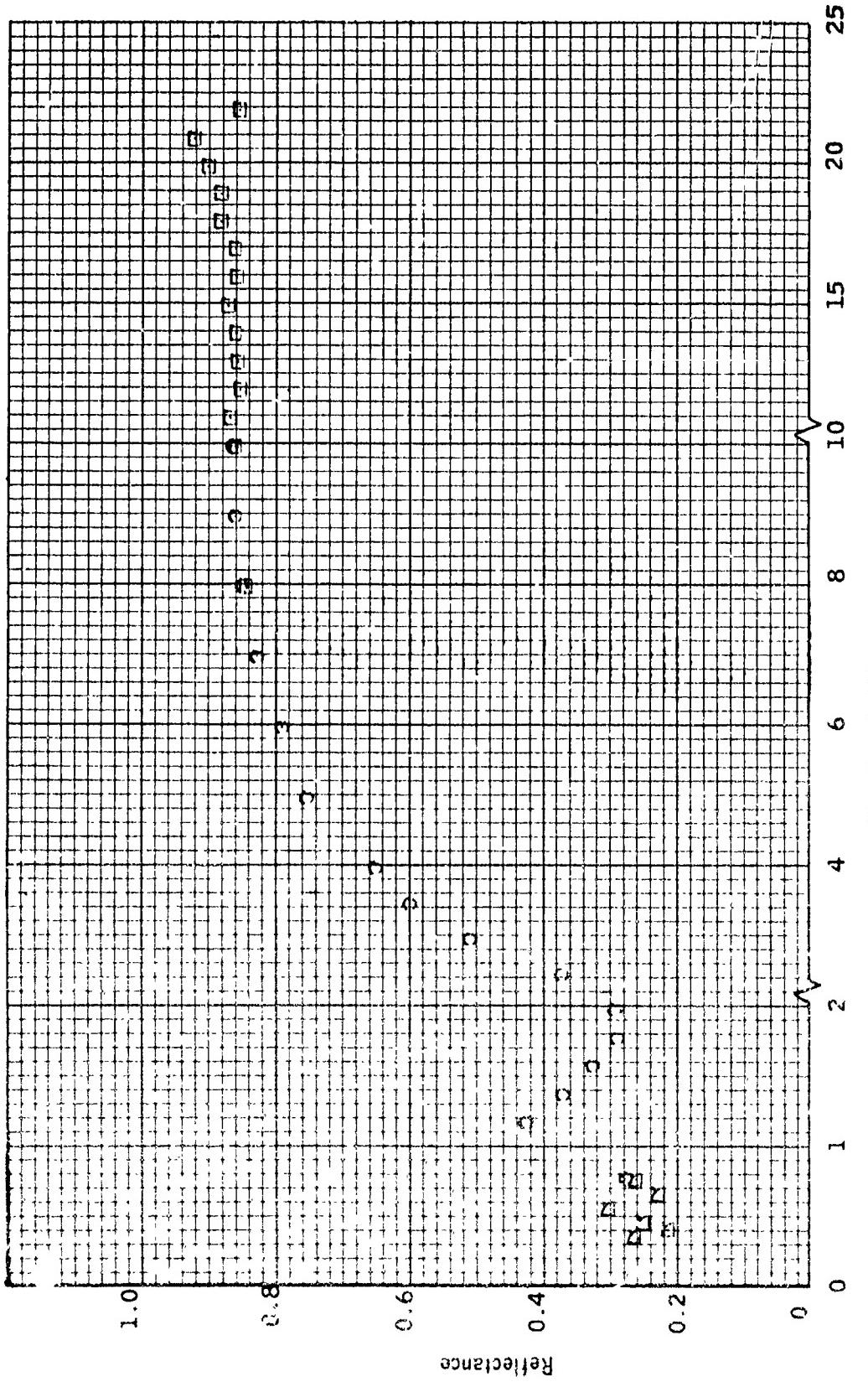


Fig. 146 Normal Spectral Reflectance of Specimen No. 178 Temperature RT

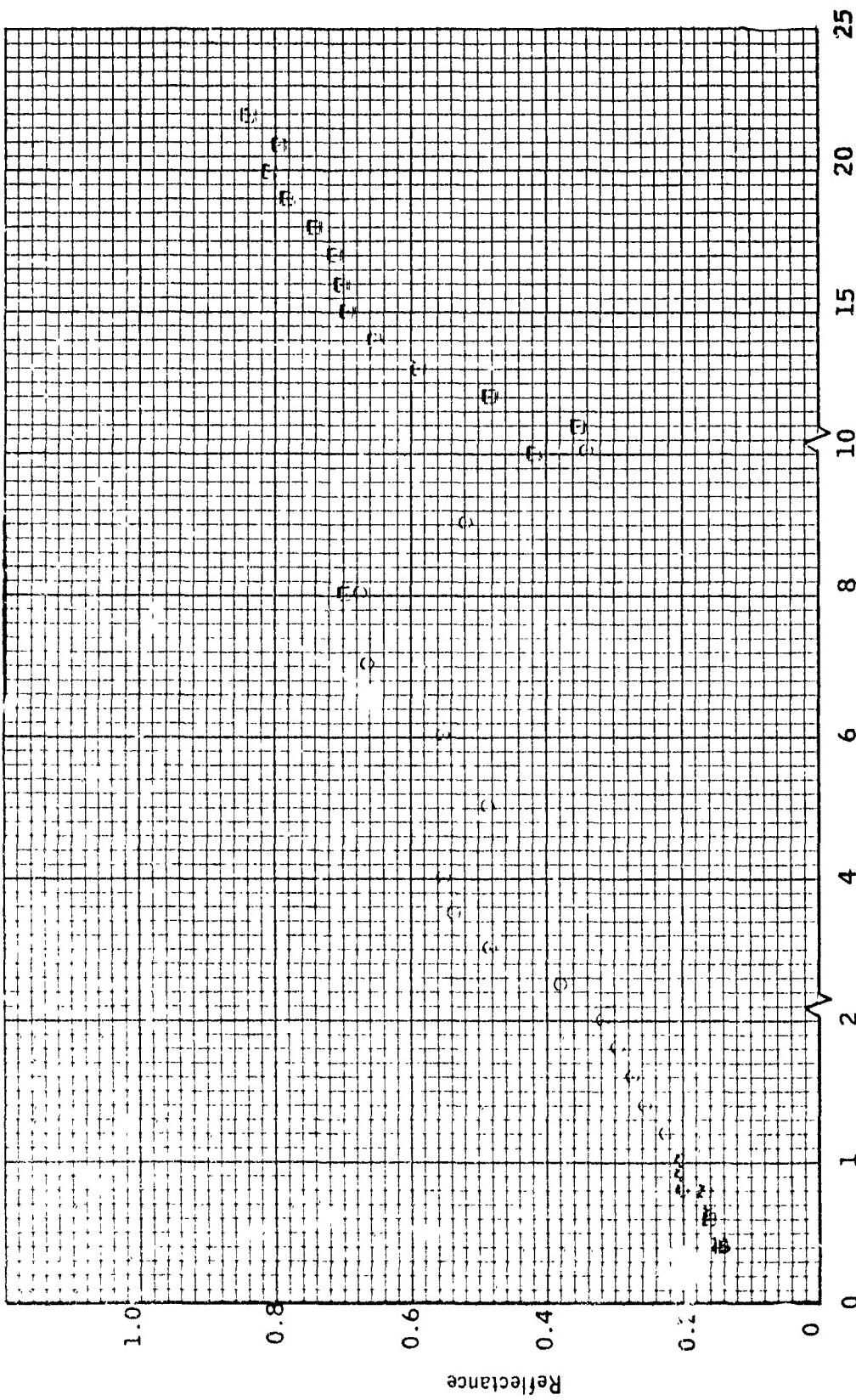


Fig. 147 Normal Spectral Reflectance of Specimen No. 37 Temperature RT

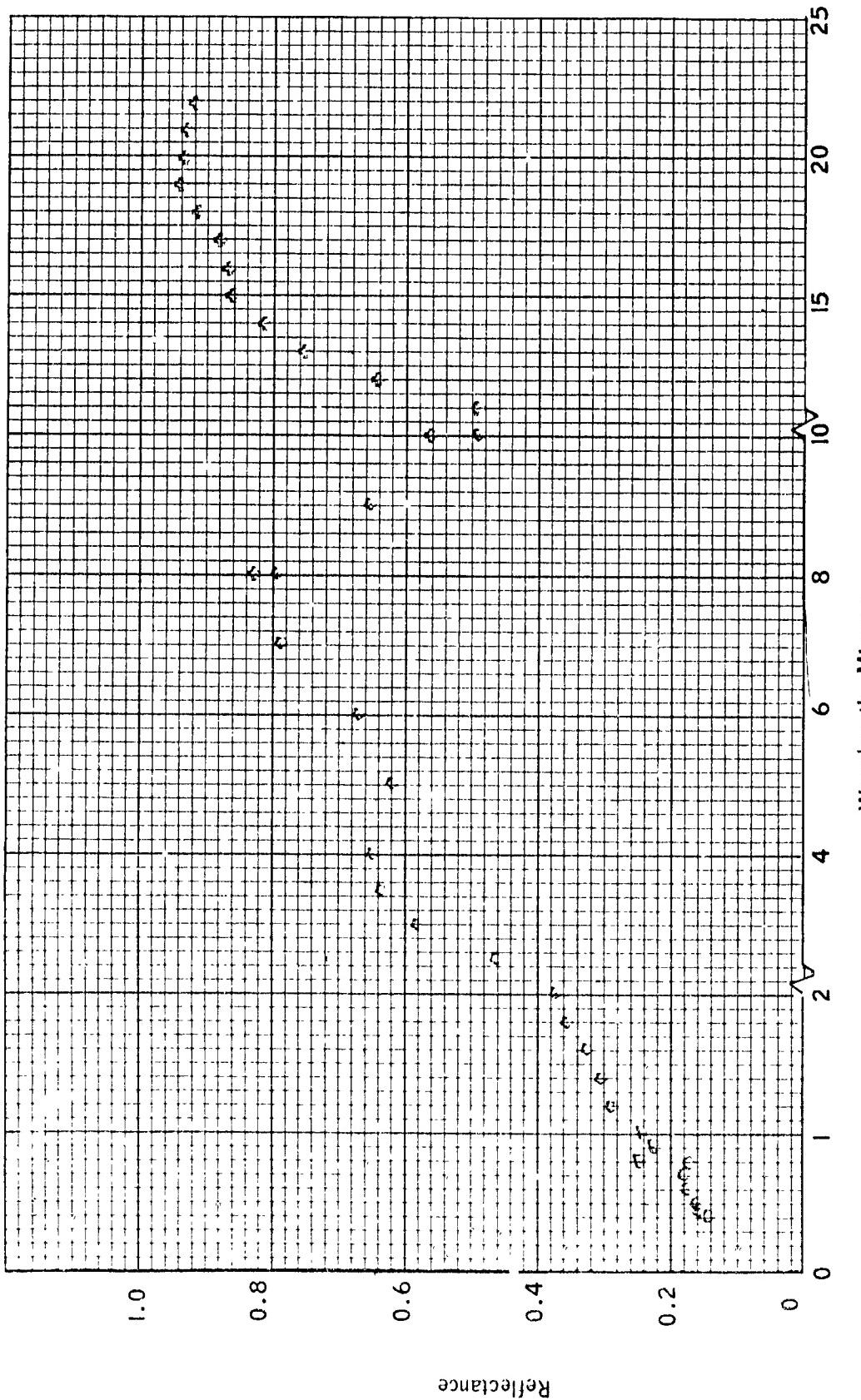


Fig. 148 Normal Spectral Reflectance of Specimen No. 41 Temperature RT

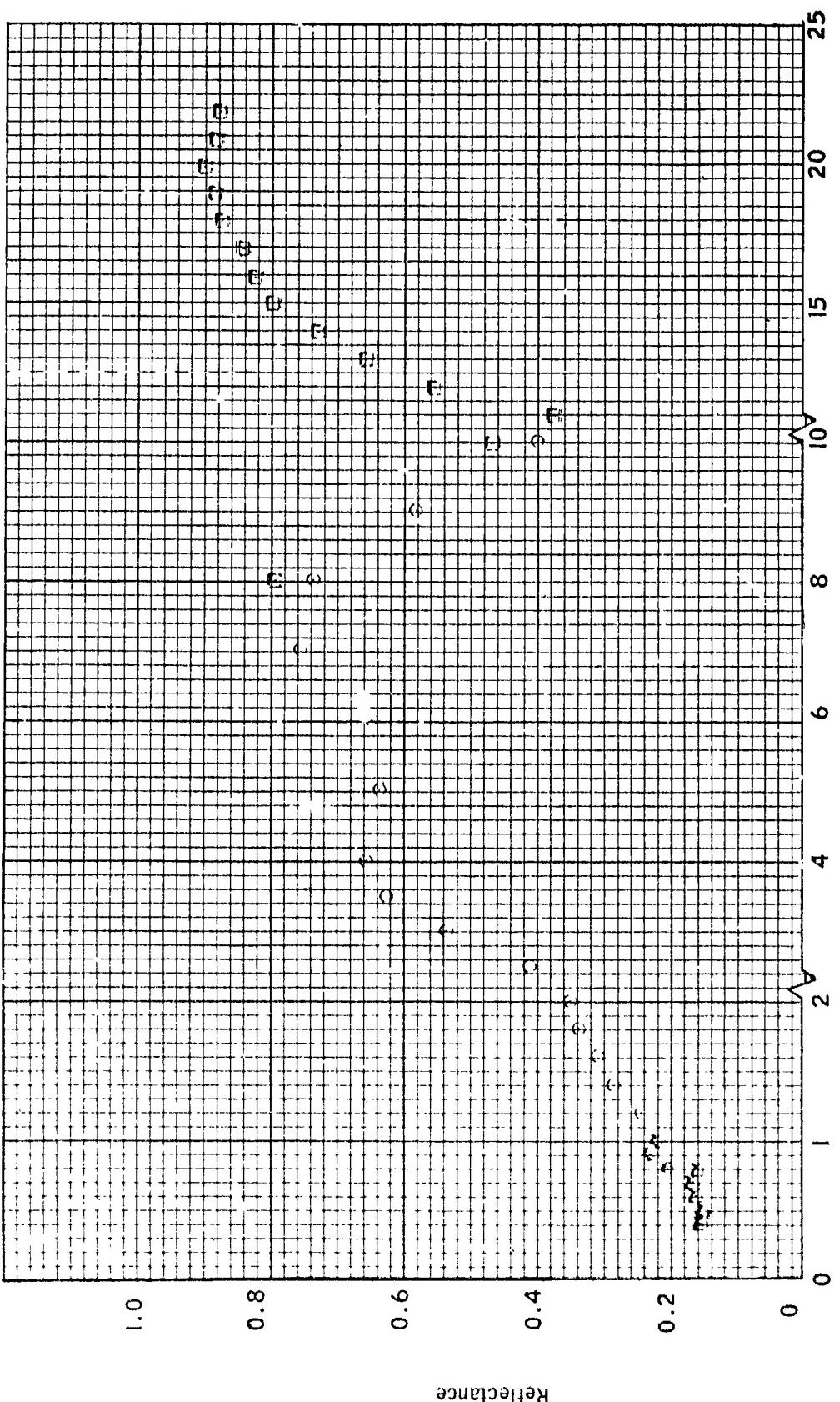


Fig. 149 Normal Spectral Reflectance of Specimen No. 47 Temperature RT

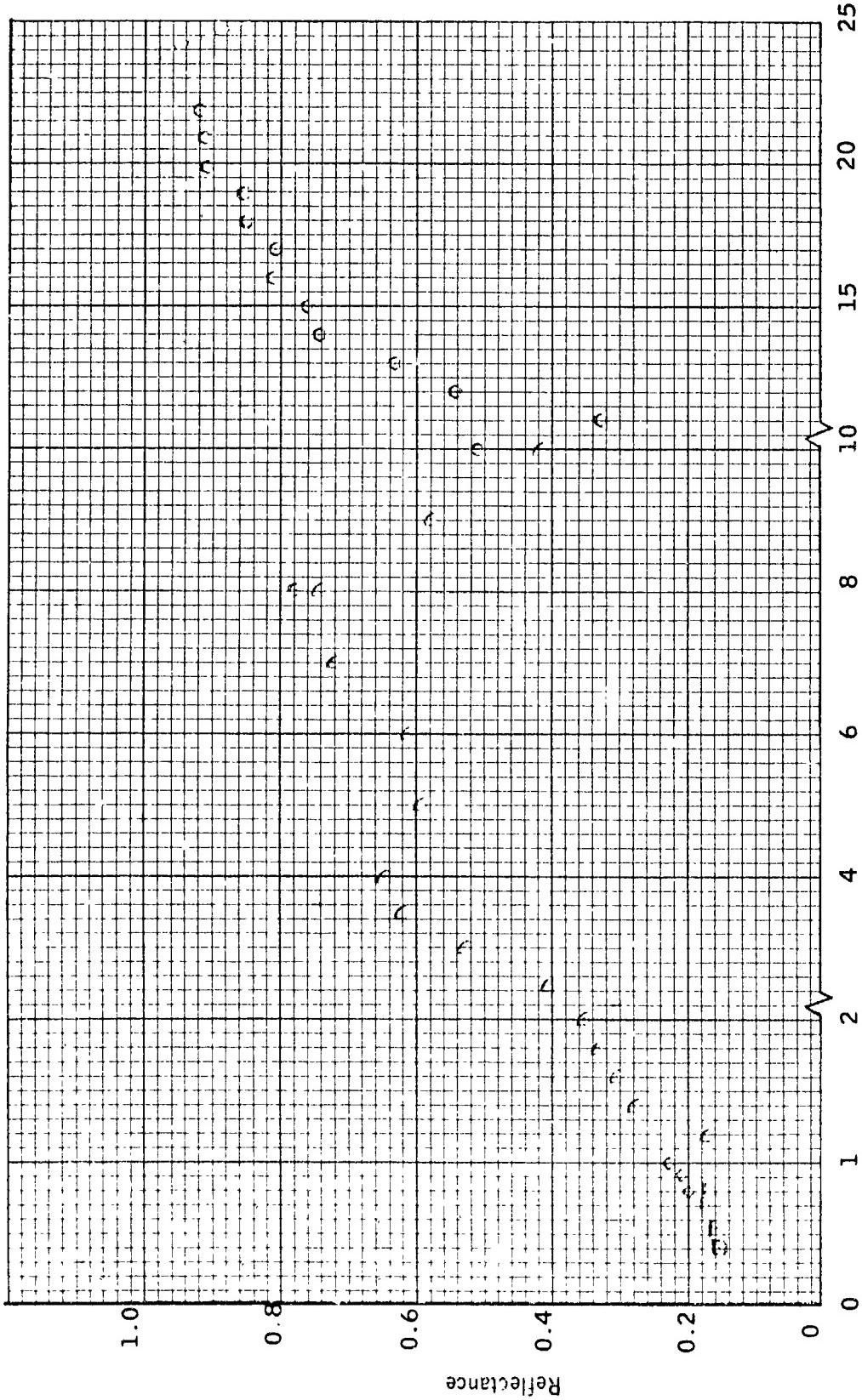


Fig. 150 Normal Spectral Reflectance of Specimen No. 47 Temperature 500°F

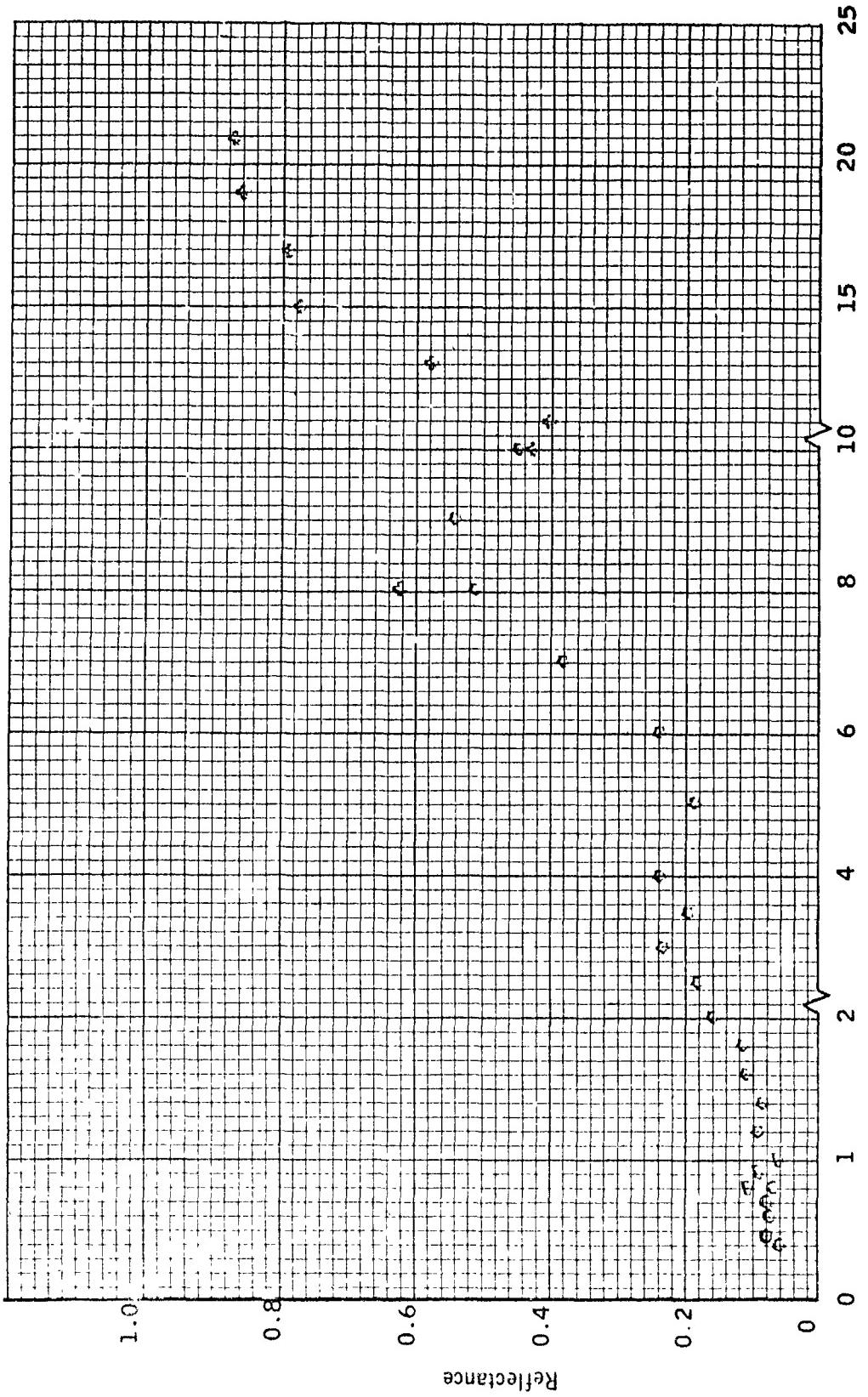


Fig. 151 Normal Spectral Reflectance of Specimen No. 47 Temperature 1000°F

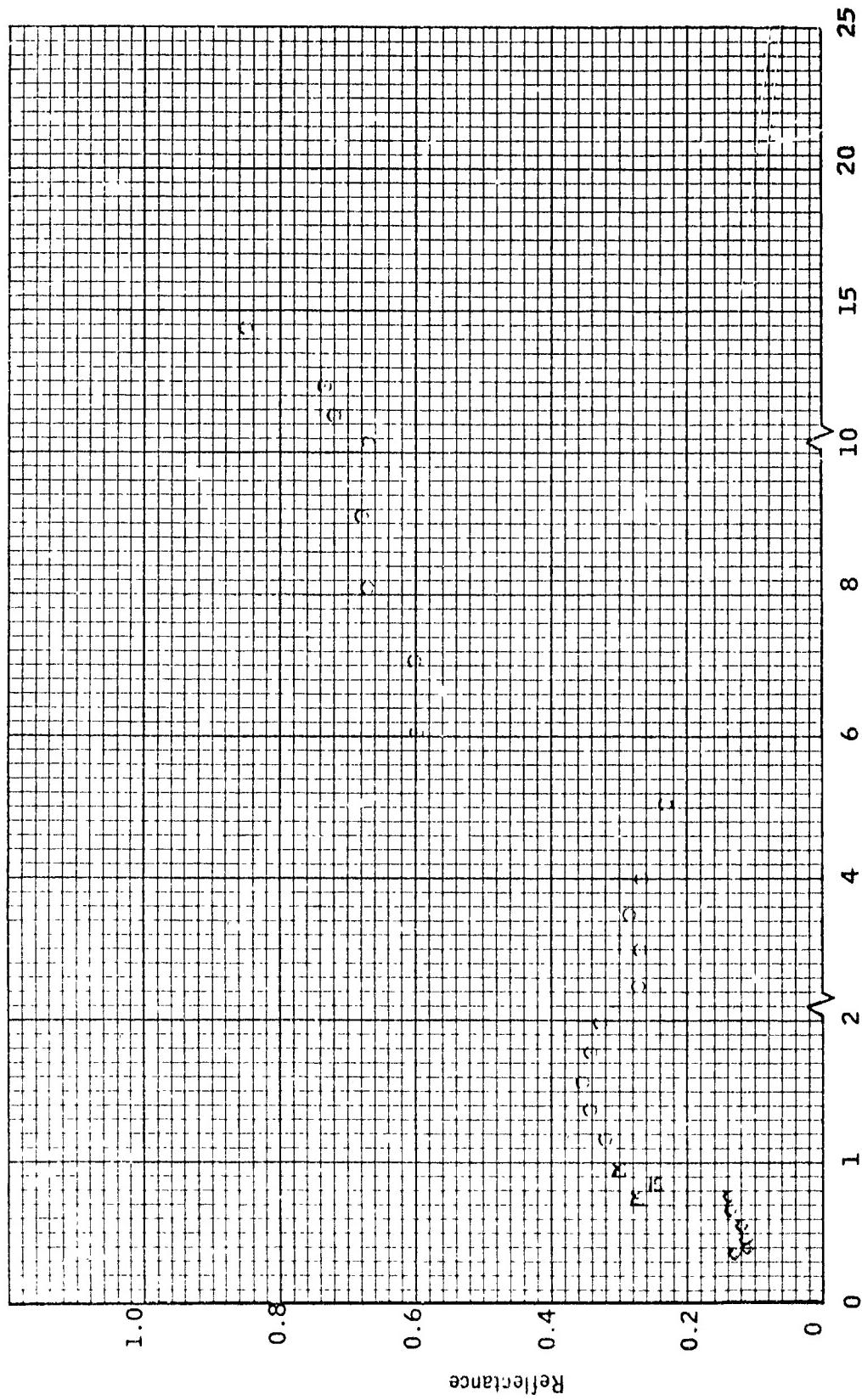


Fig. 152 Normal Spectral Reflectance of Specimen No. 47 Temperature 1300°F

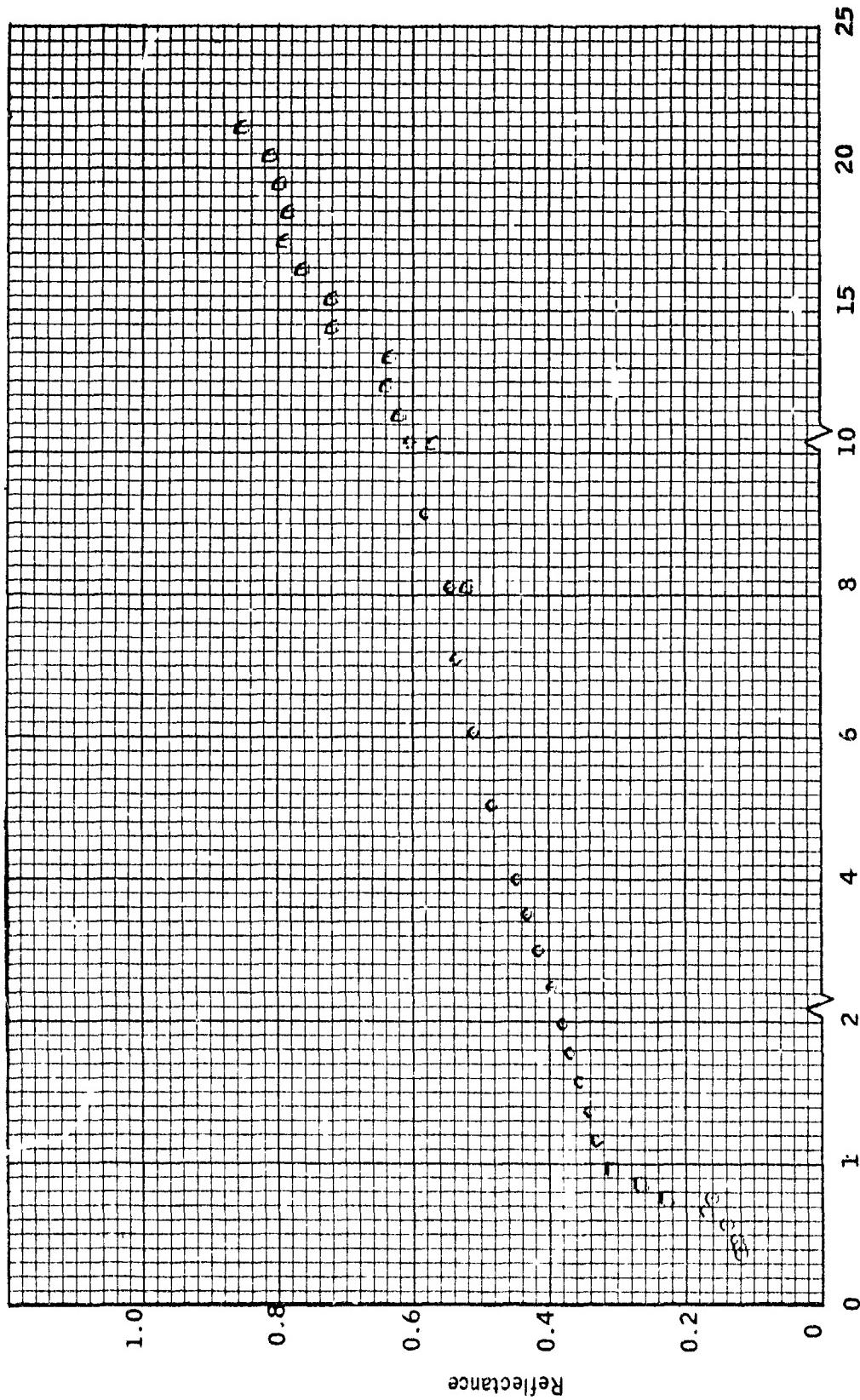


Fig. 153 Normal Spectral Reflectance of Specimen No. 47 Temperature  $RT_f$

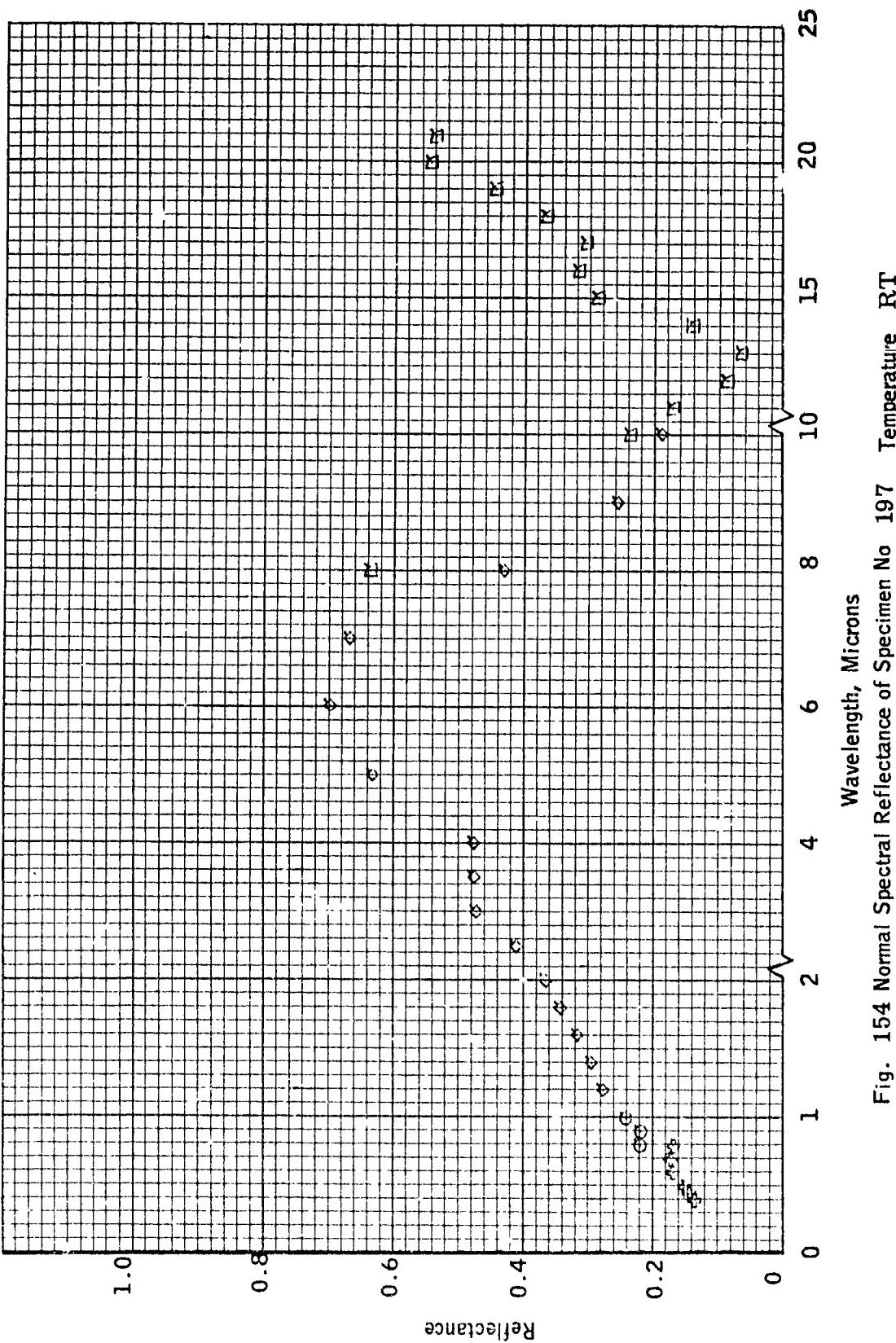


Fig. 154 Normal Spectral Reflectance of Specimen No. 197 Temperature RT

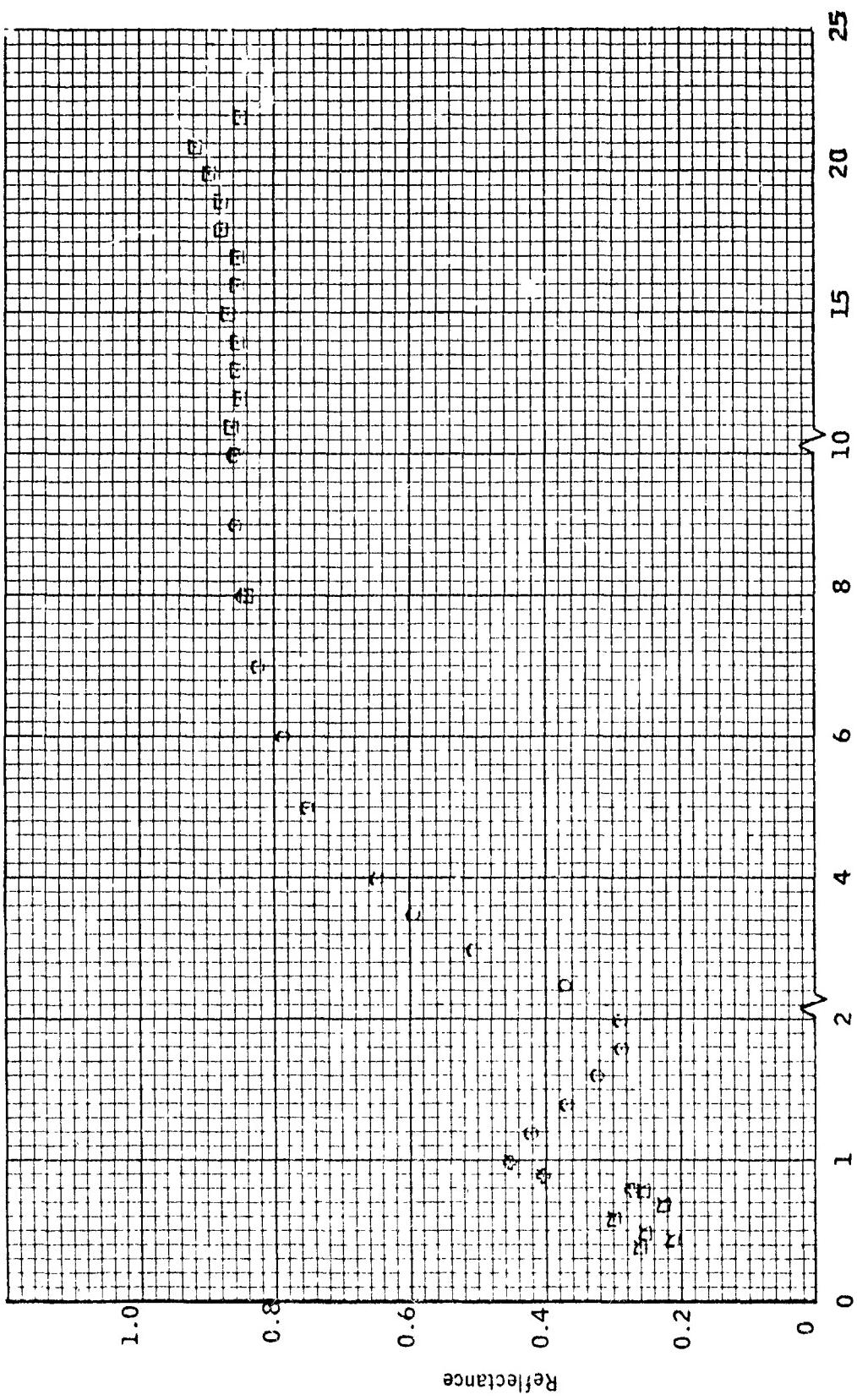


Fig. 155 Normal Spectra! Reflectance of Specimen No 199 Temperature RT

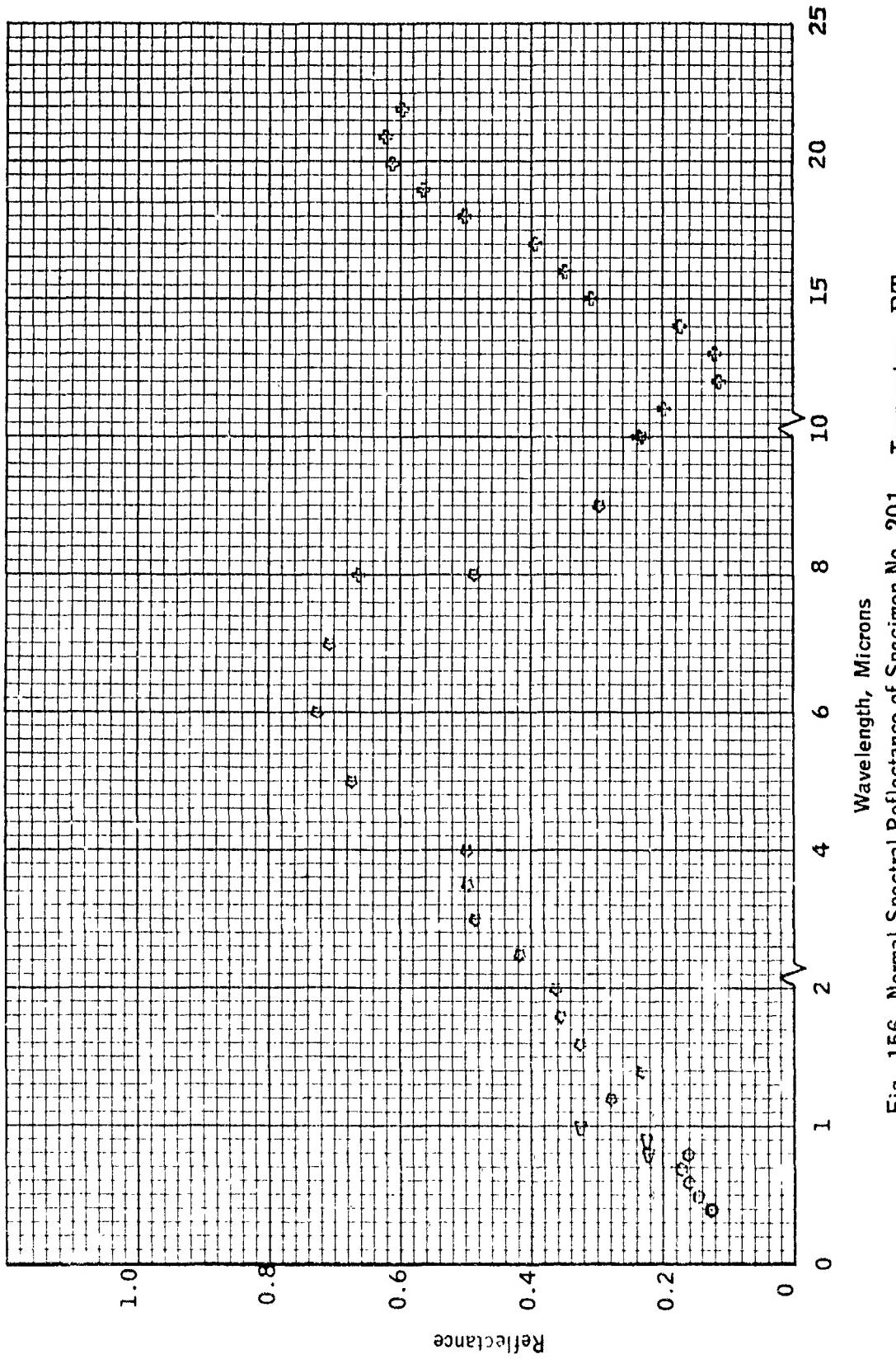


Fig. 156 Normal Spectral Reflectance of Specimen No. 201 Temperature RT

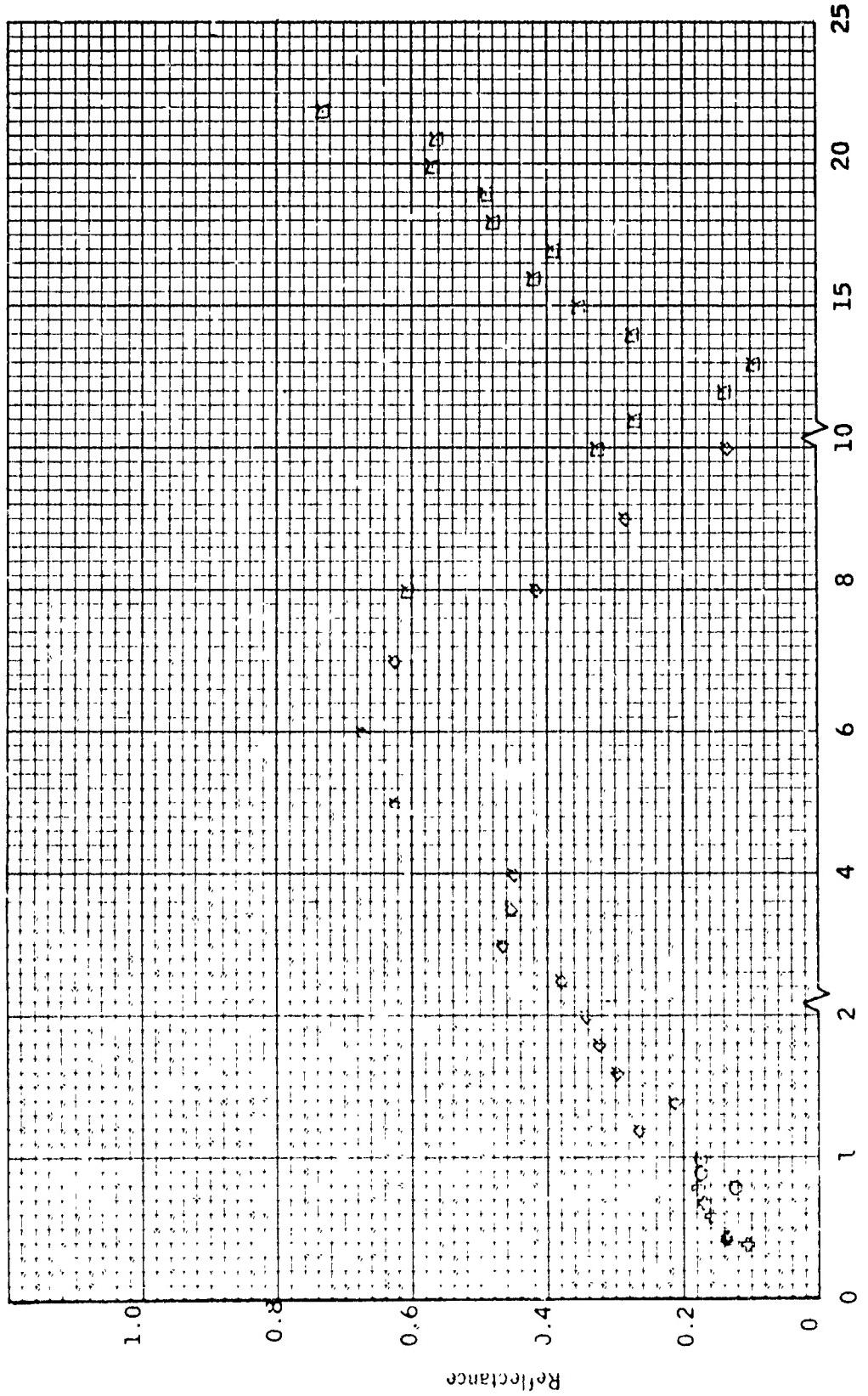


Fig. 157 Normal Spectral Reflectance of Specimen No. 201 Temperature 500°F

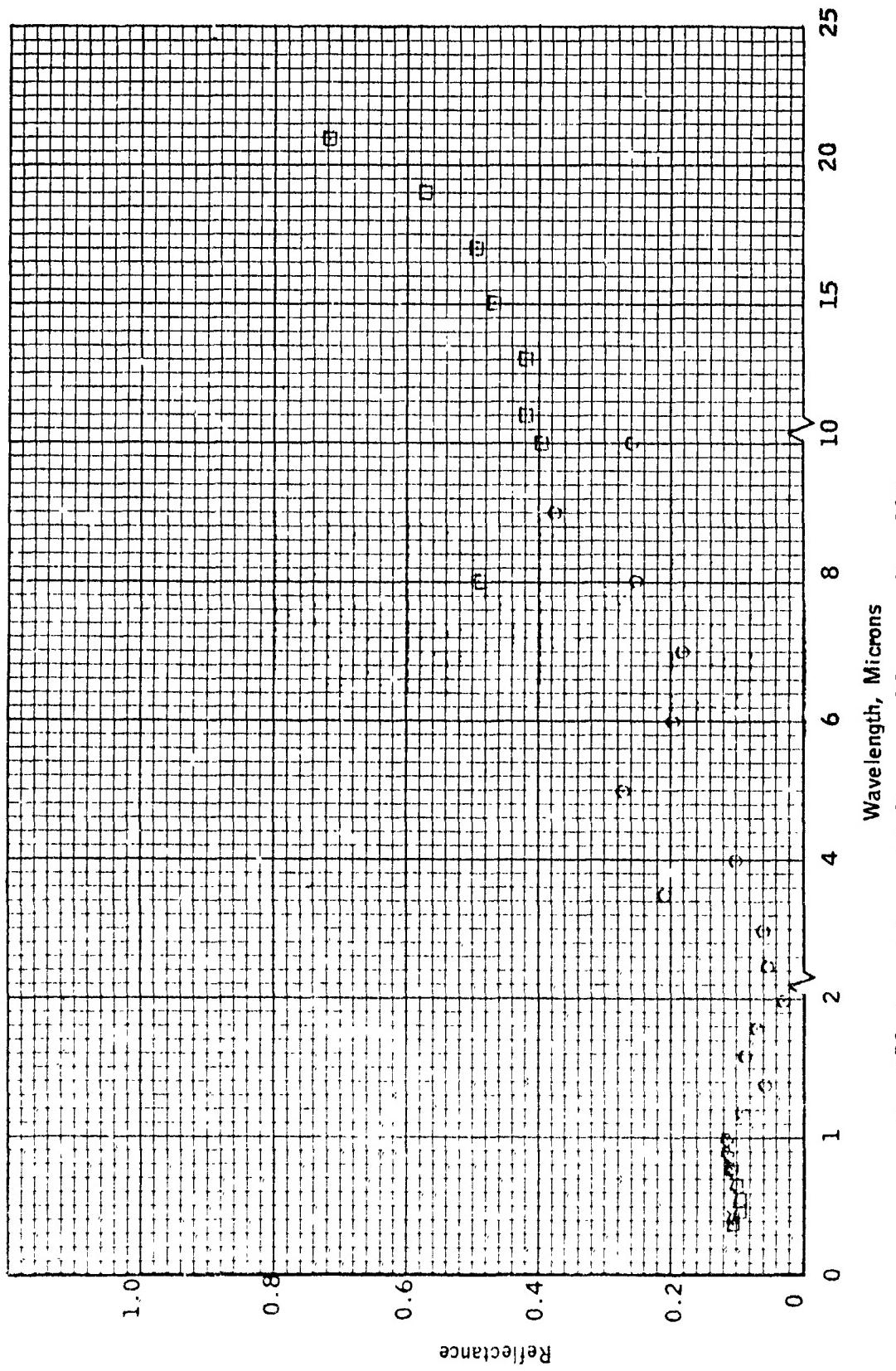


Fig. 158 Normal Spectral Reflectance of Specimen Nu 201 Temperature 1000°F

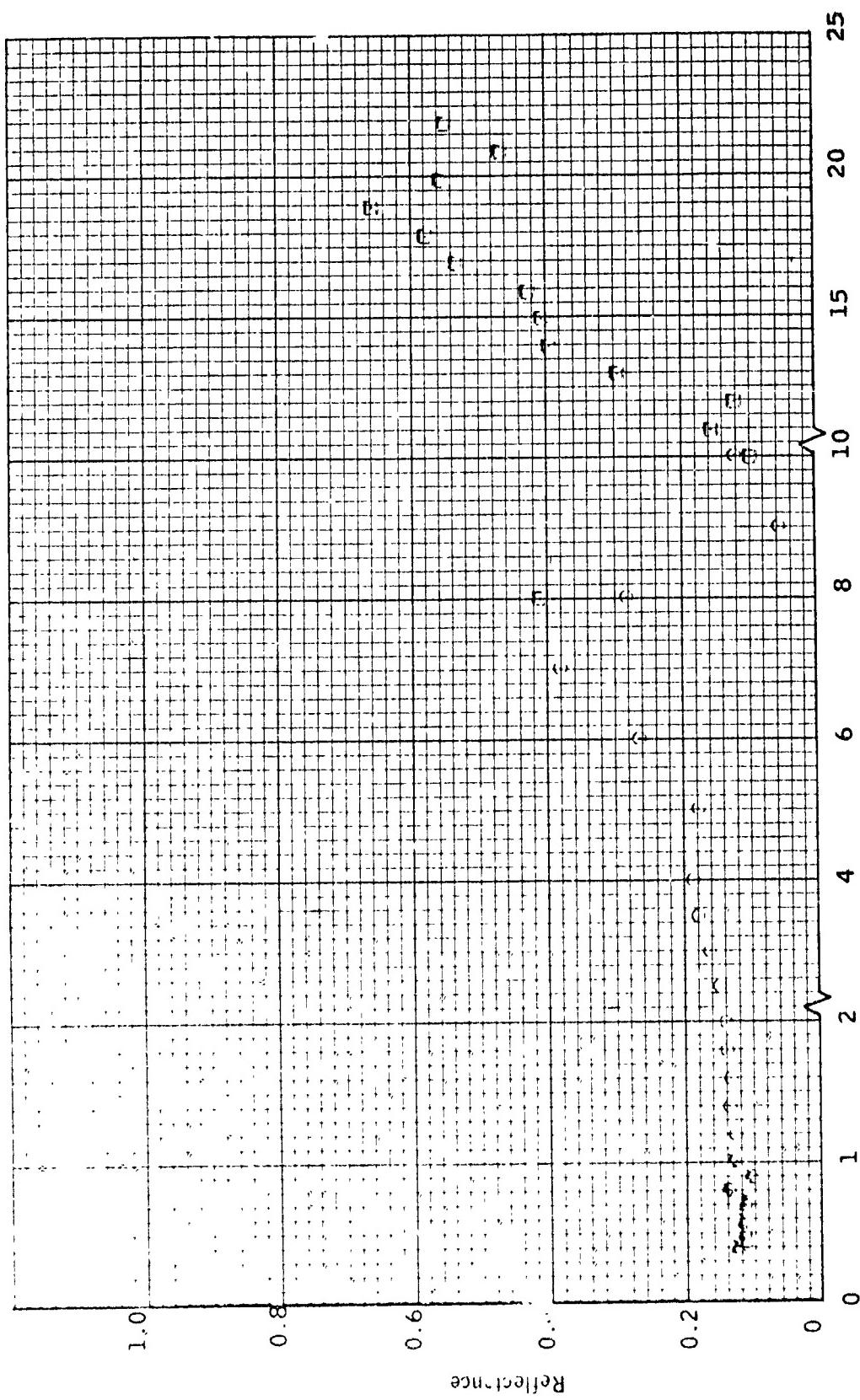


Fig. 159 Normal Spectral Reflectance of Specimen No. 201 Temperature RT, HT-1500 F

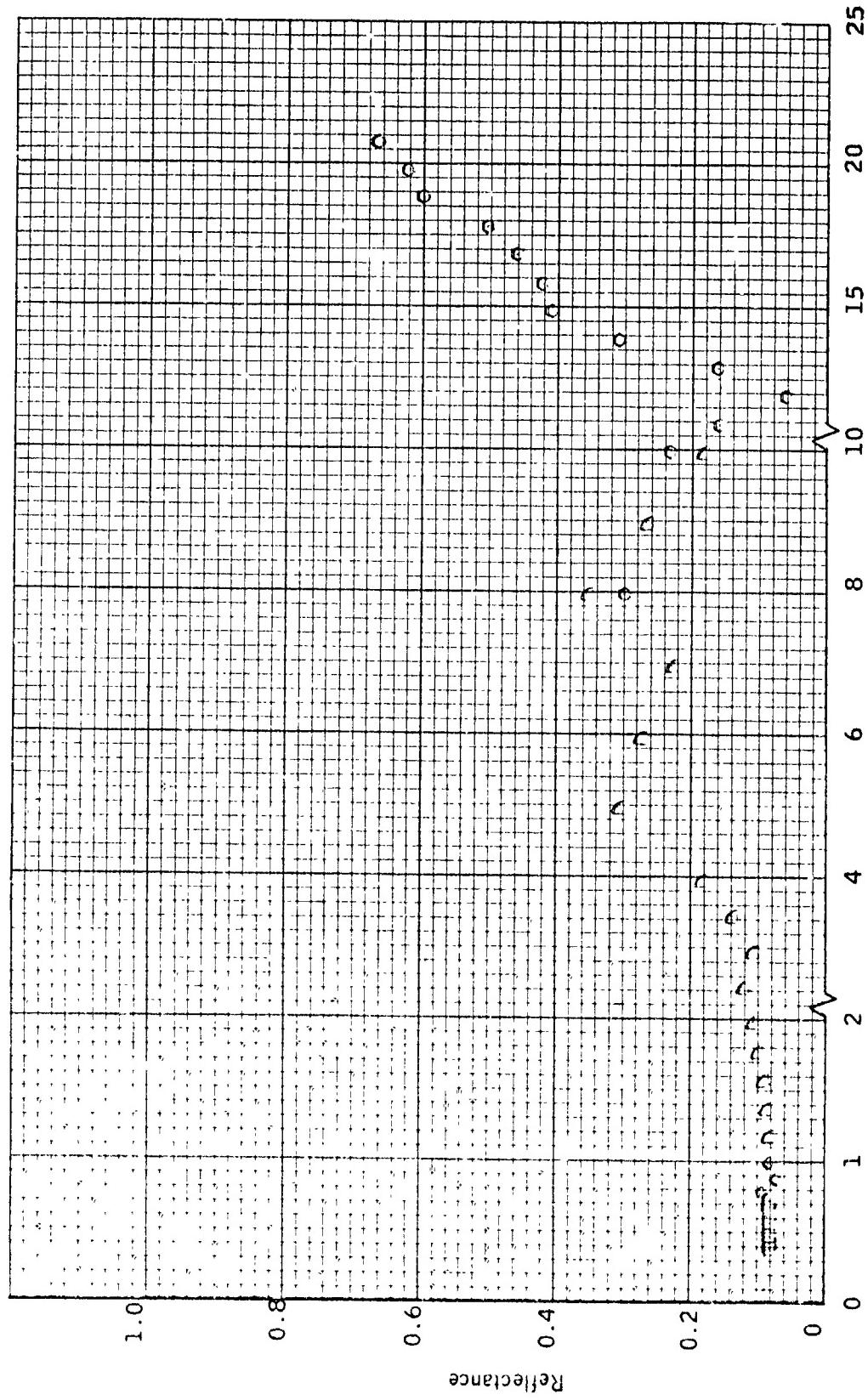


Fig. 160 Normal Spectral Reflectance of Specimen No. 201 Temperature  $RT_f$

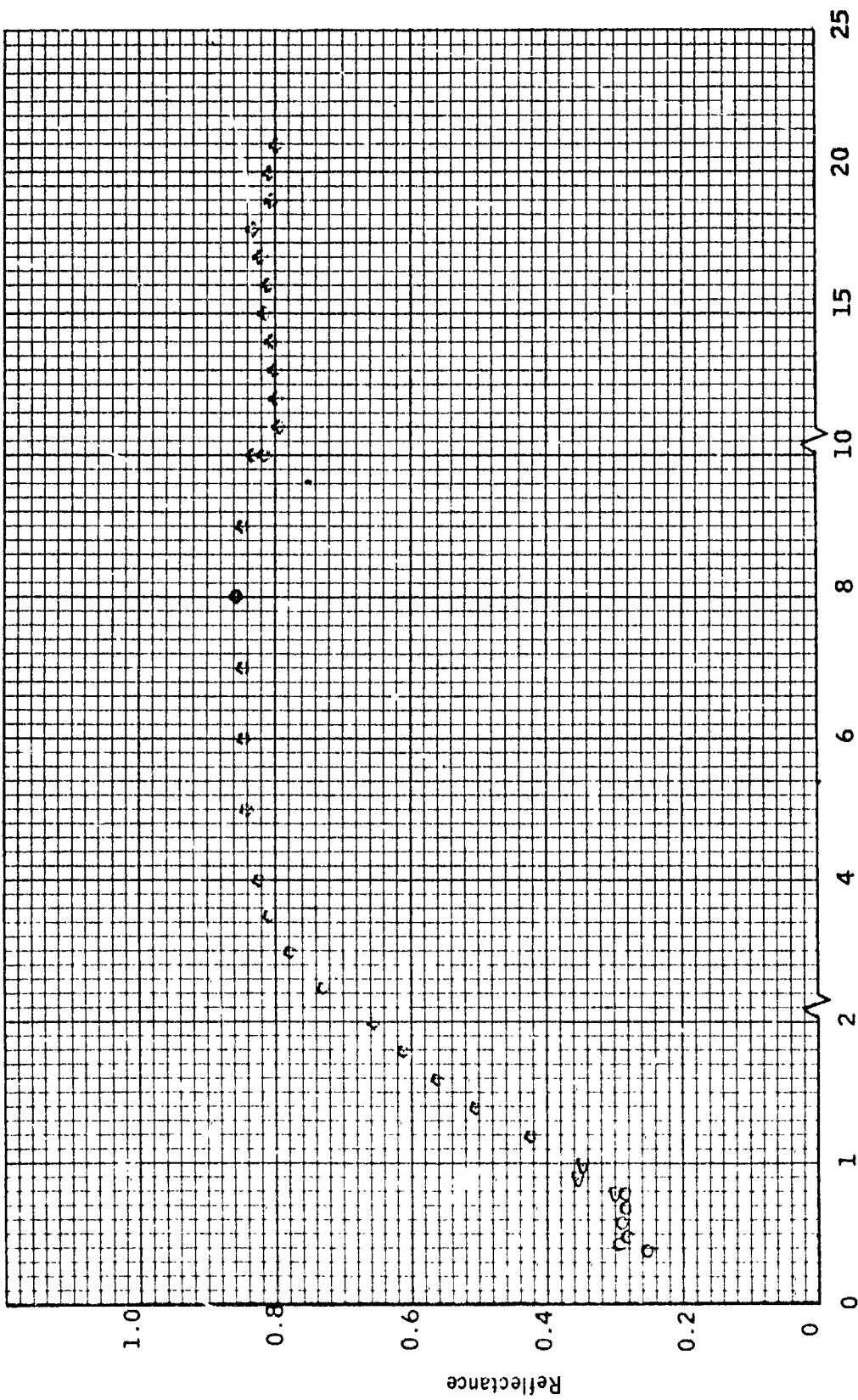


Fig. 161 Normal Spectral Reflectance of Specimen No. 162 Temperature RT

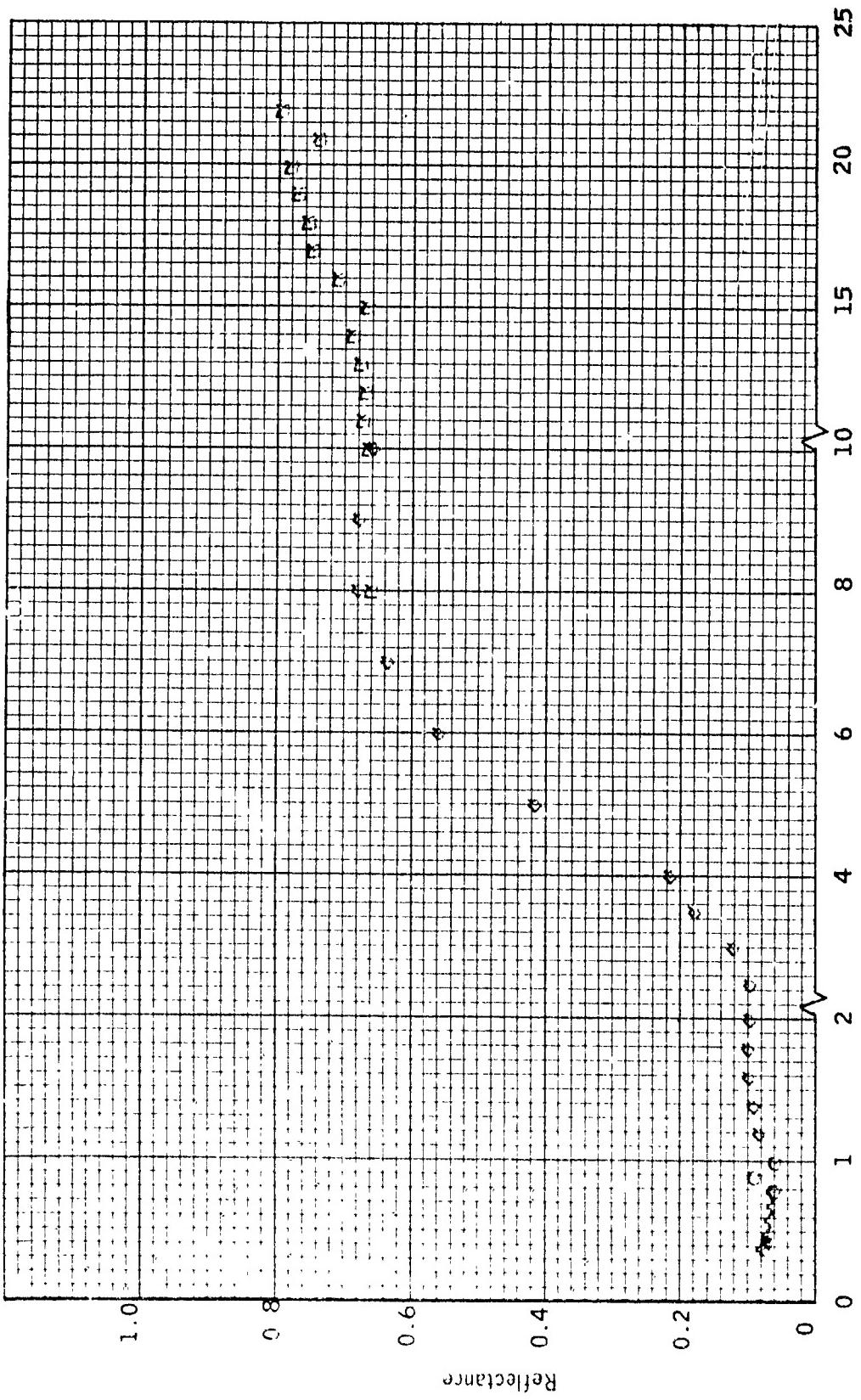


Fig. 162 Normal Spectral Reflectance of Specimen No. 164 Temperature RT

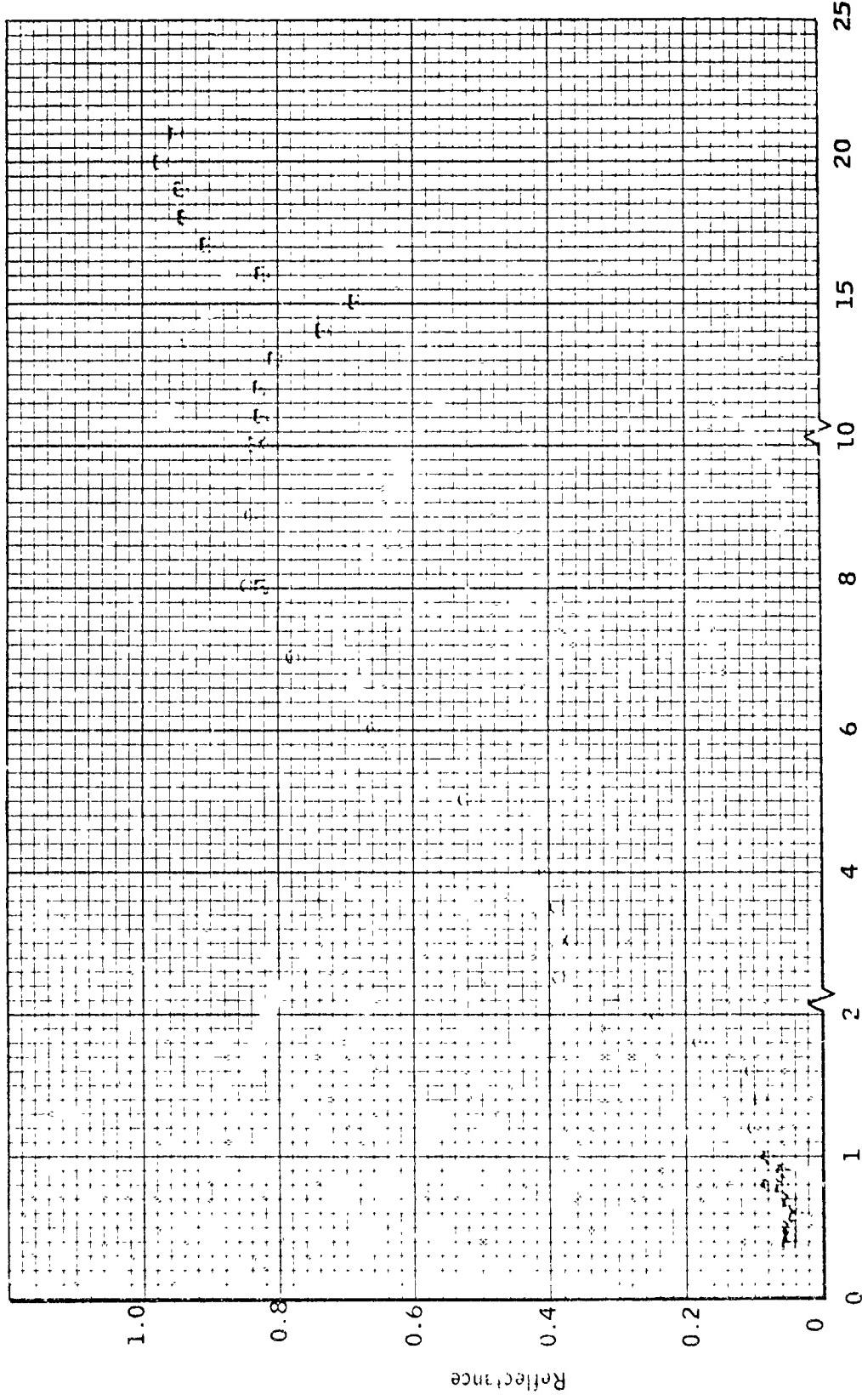


Fig. 163 Normal Spectral Reflectance of Specimen No. 166 Temperature RT

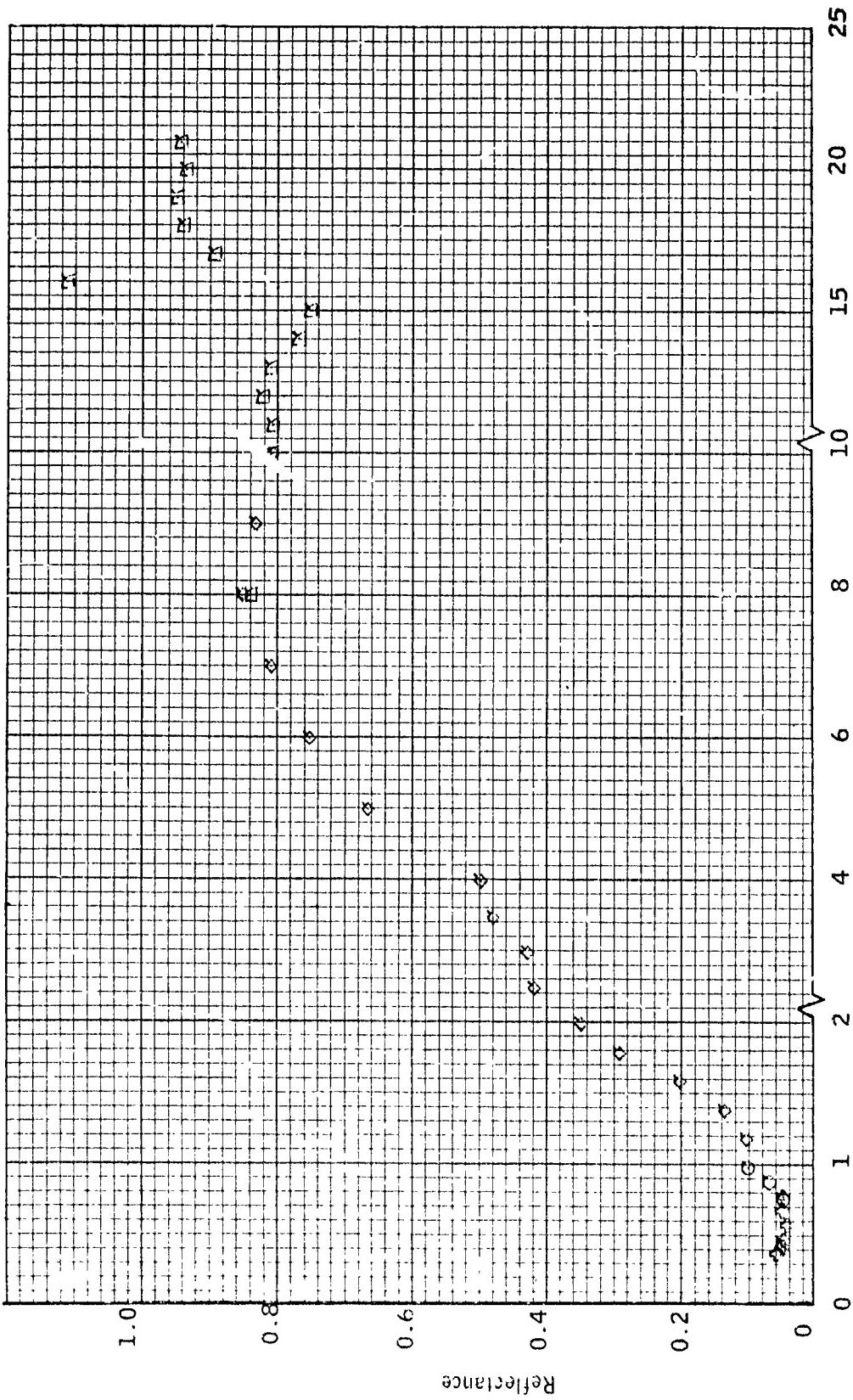


Fig. 164 Normal Spectral Reflectance of Specimen No. 166 Temperature 400°F

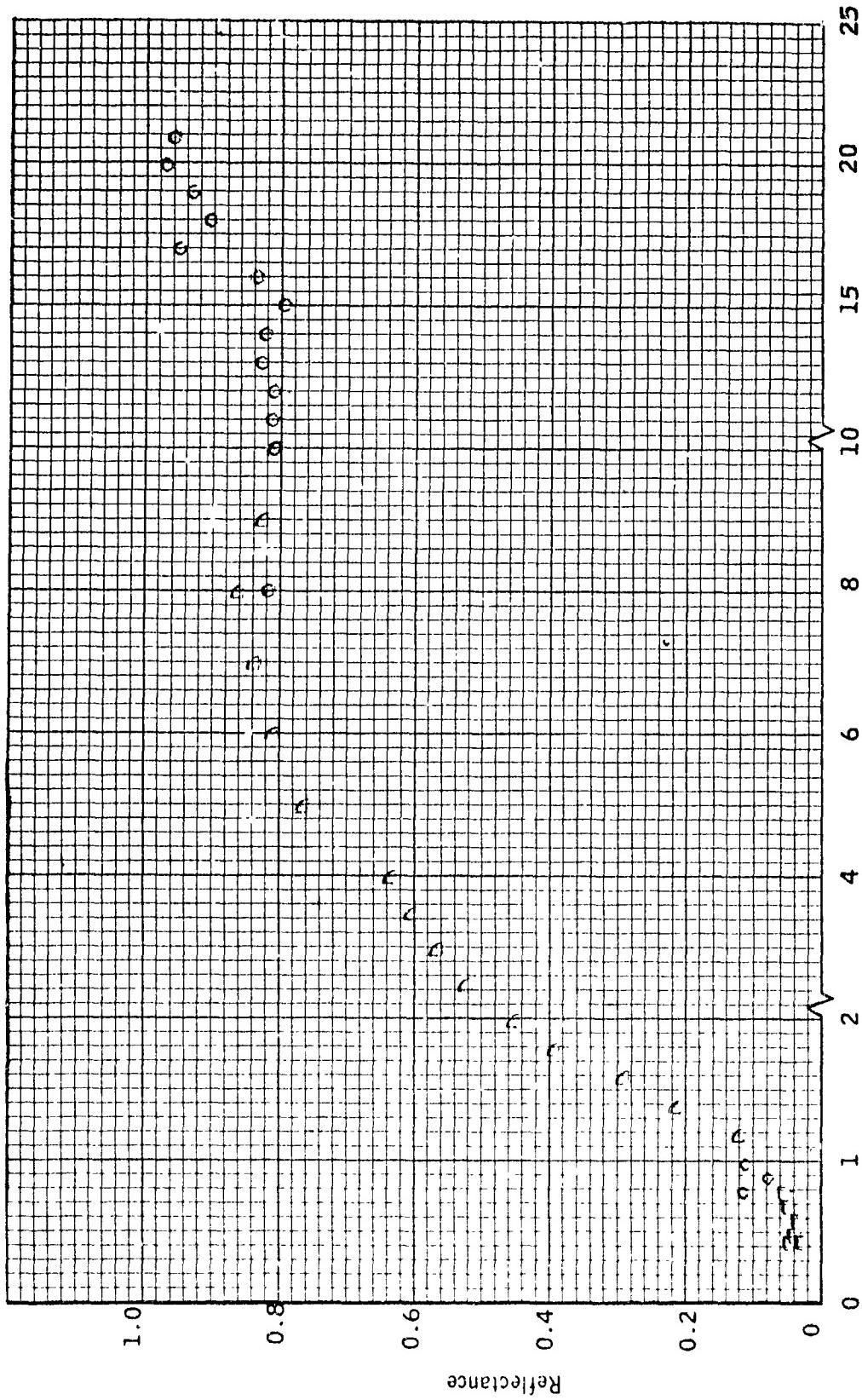


Fig. 165 Normal Spectral Reflectance of Specimen No. 166 Temperature 800°F



Fig. 166 Normal Spectral Reflectance of Specimen No. 166 Temperature 1200°F

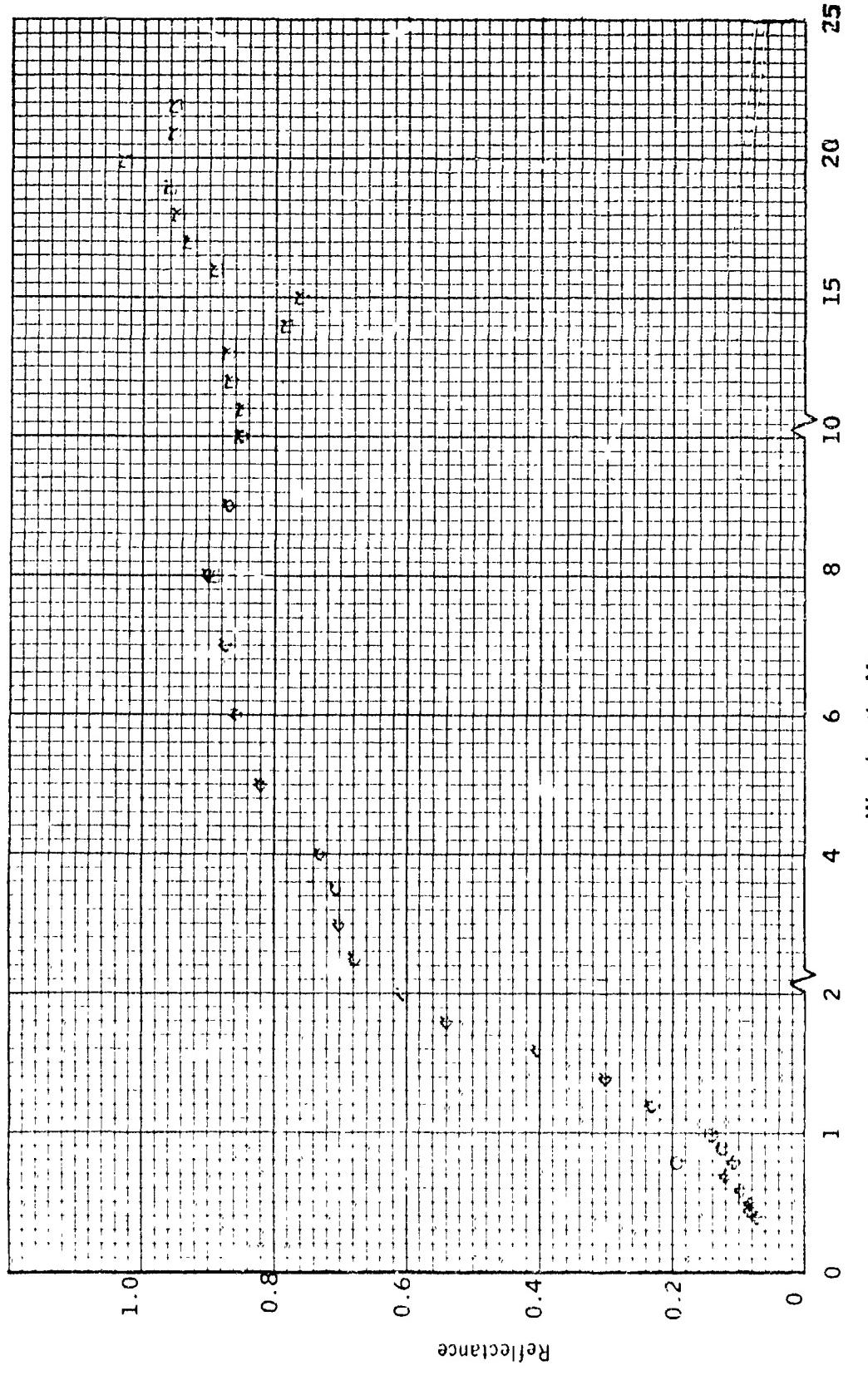


Fig. 167 Normal Spectral Reflectance of Specimen No. 166 Temperature  $RT_f$

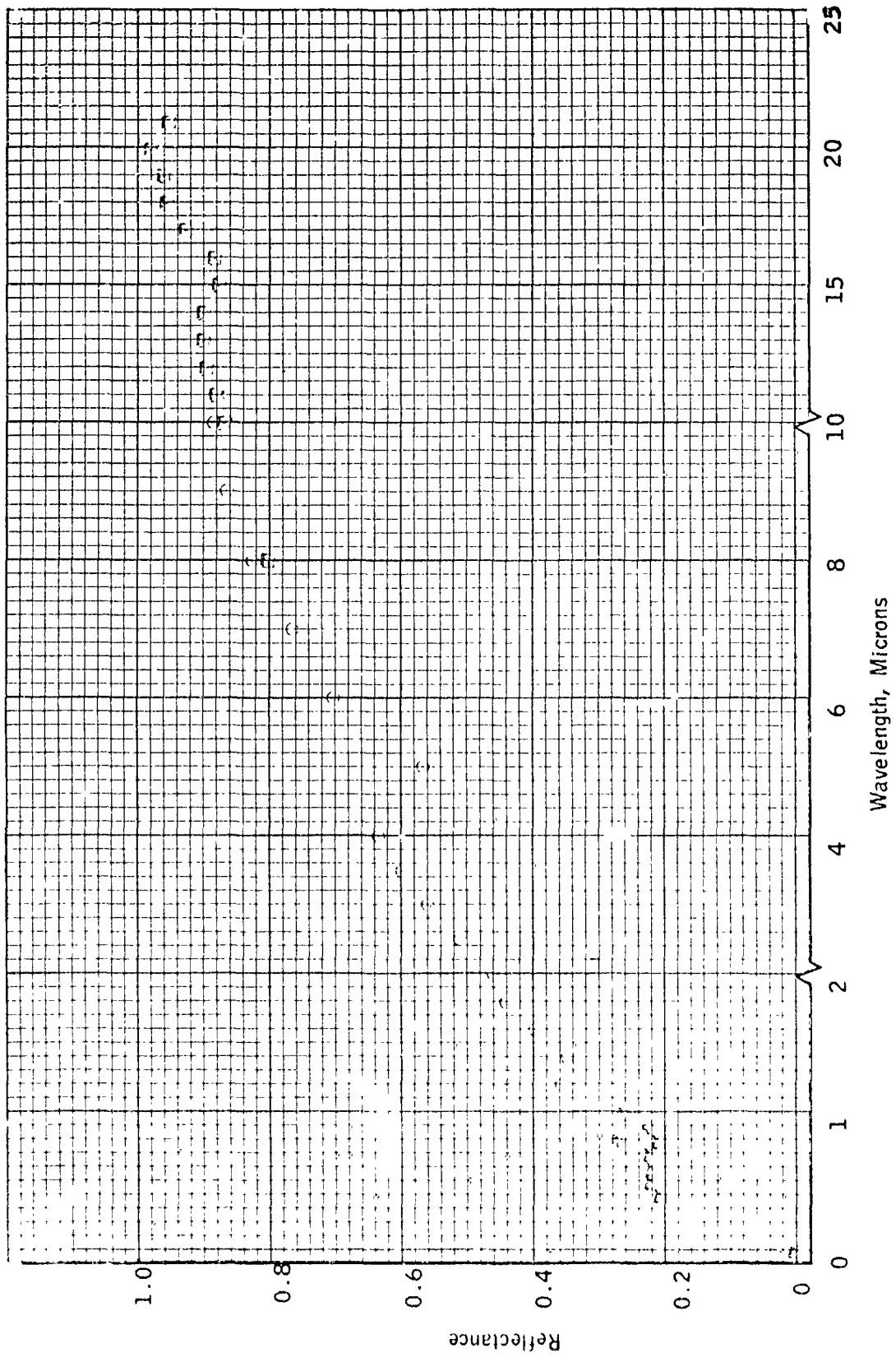


Fig. 168 Normal Spectral Reflectance of Specimen No 173 Temperature RT

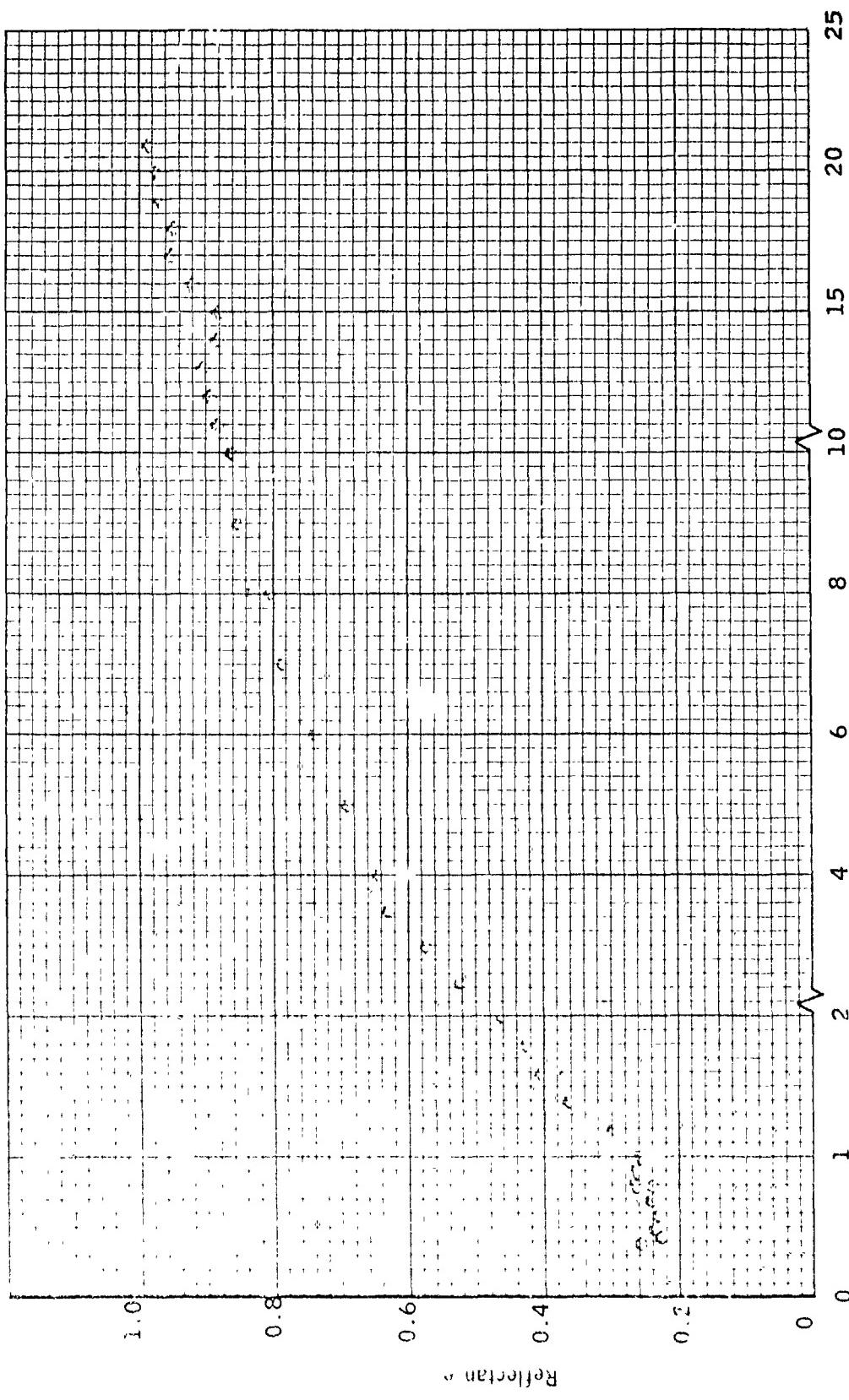


Fig. 169 Normal Spectral Reflectance of Specimen No 173 Temperature 400F

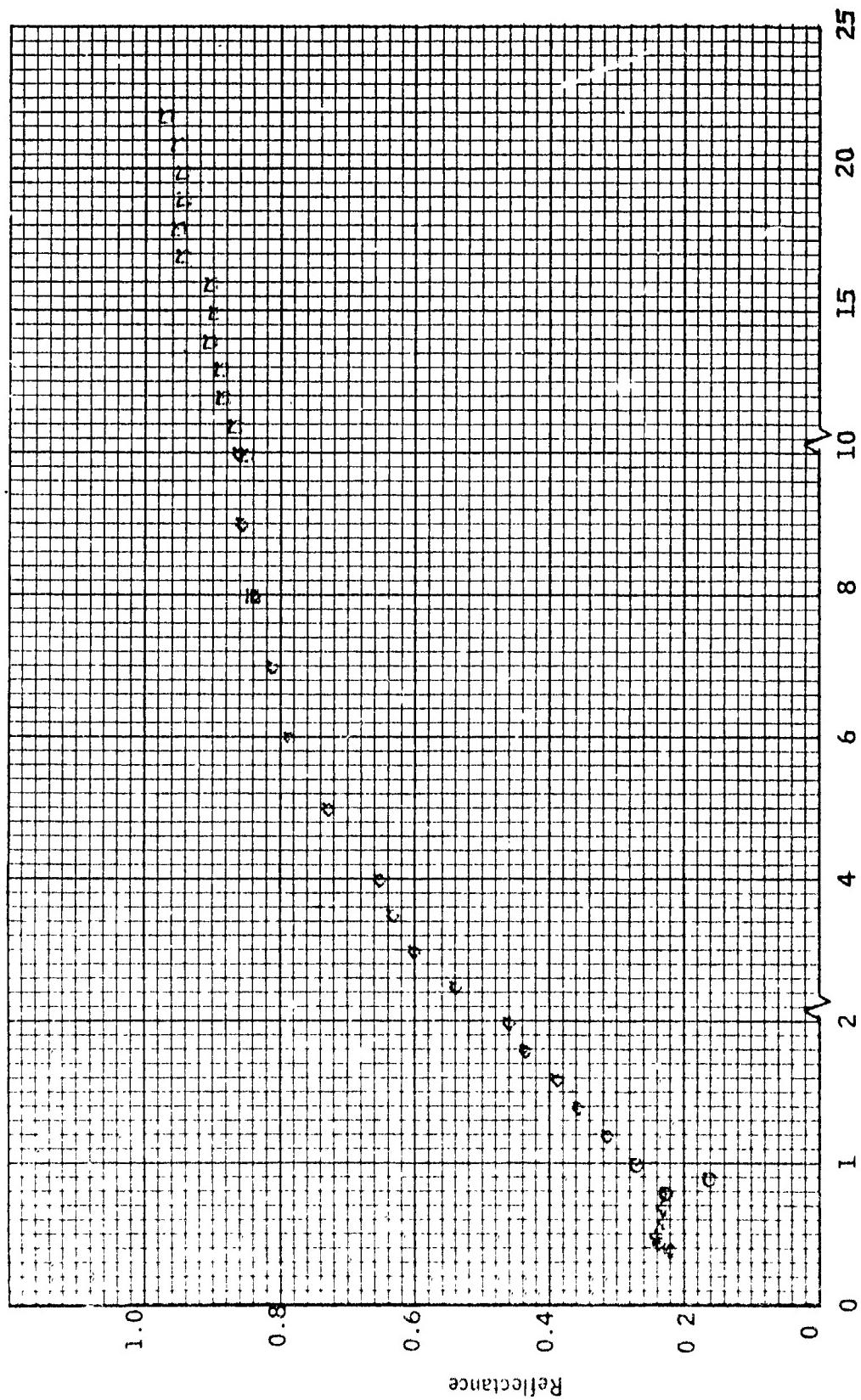


Fig. 170 Normal Spectral Reflectance of Specimen No. 173 Temperature 800 F

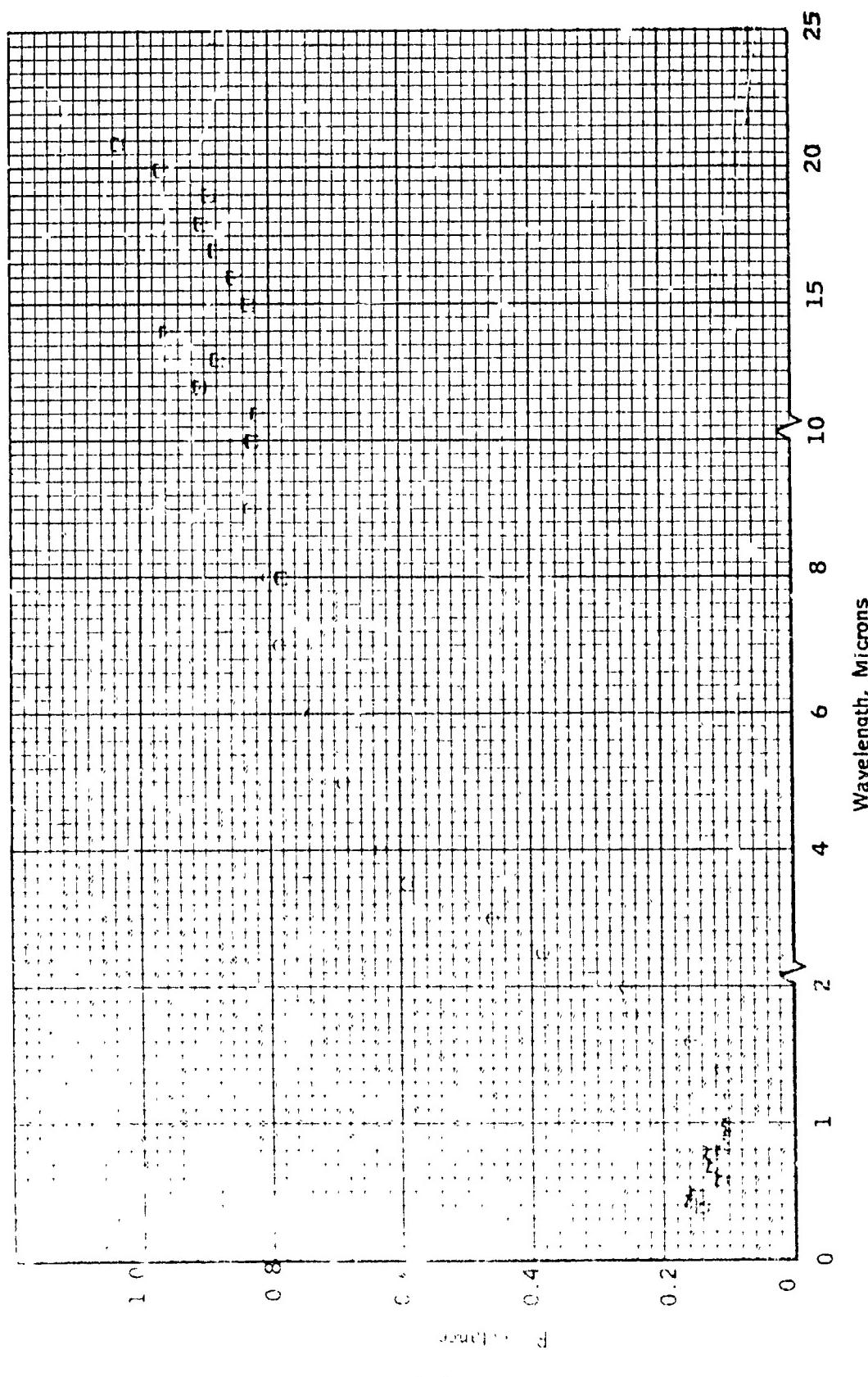


Fig. 171 Normal Spectral Reflectance of Specimen No. 173 Temperature 1200 F

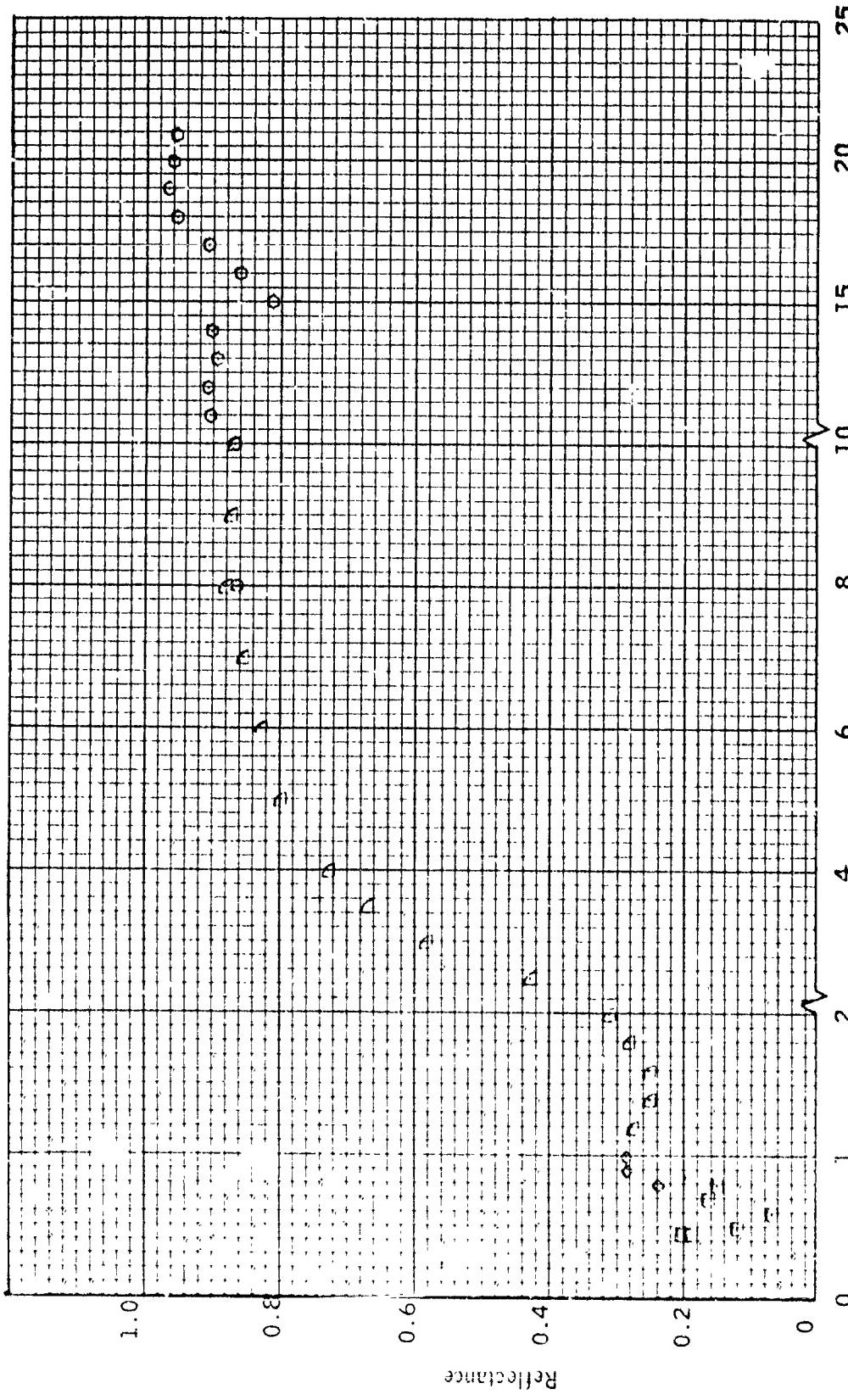


Fig. 172 Normal Spectral Reflectance of Specimen No 173 Temperature  $RT_f$

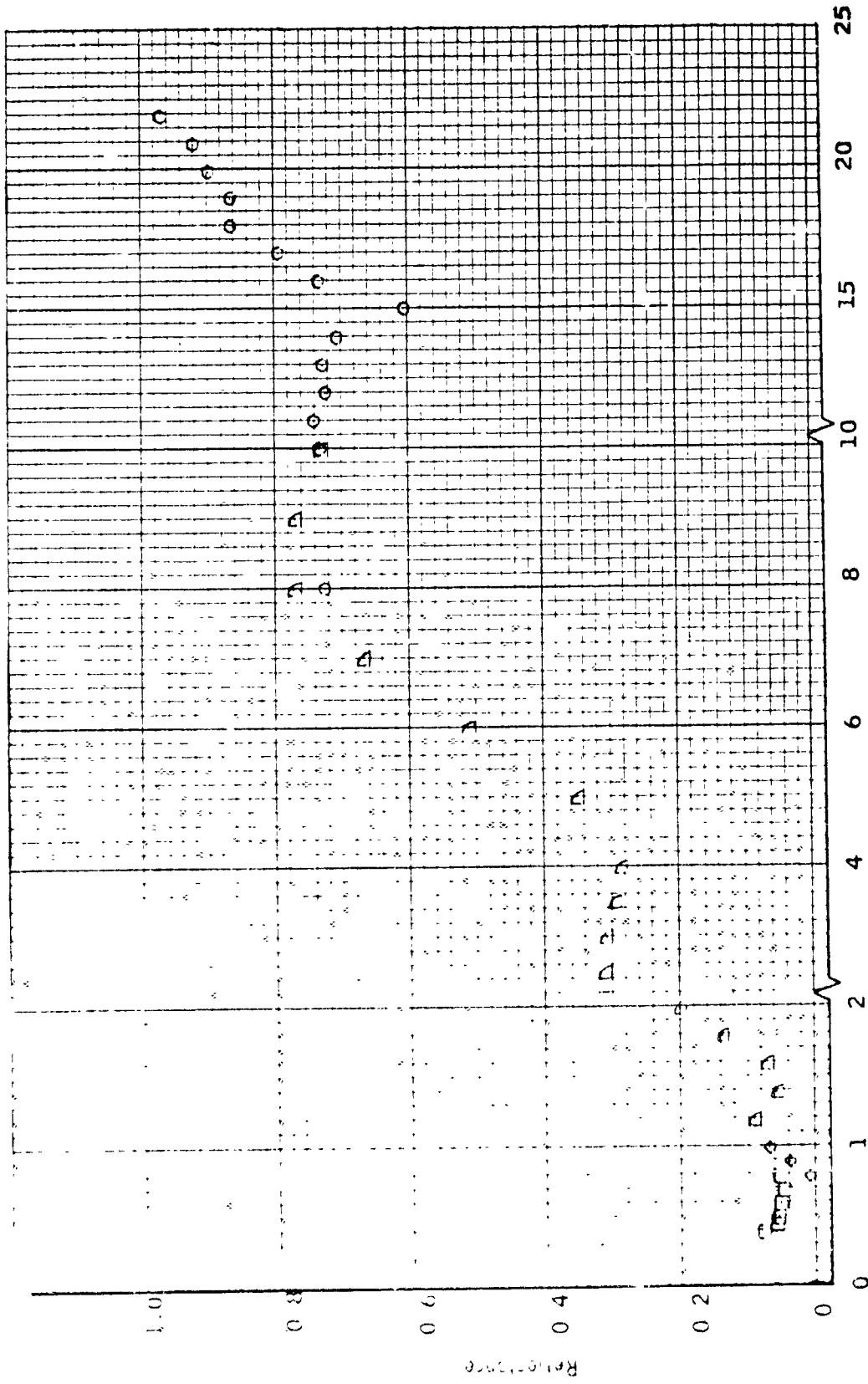


Fig. 173 Normal Spectral Reflectance of Specimen No 177 Temperature RT

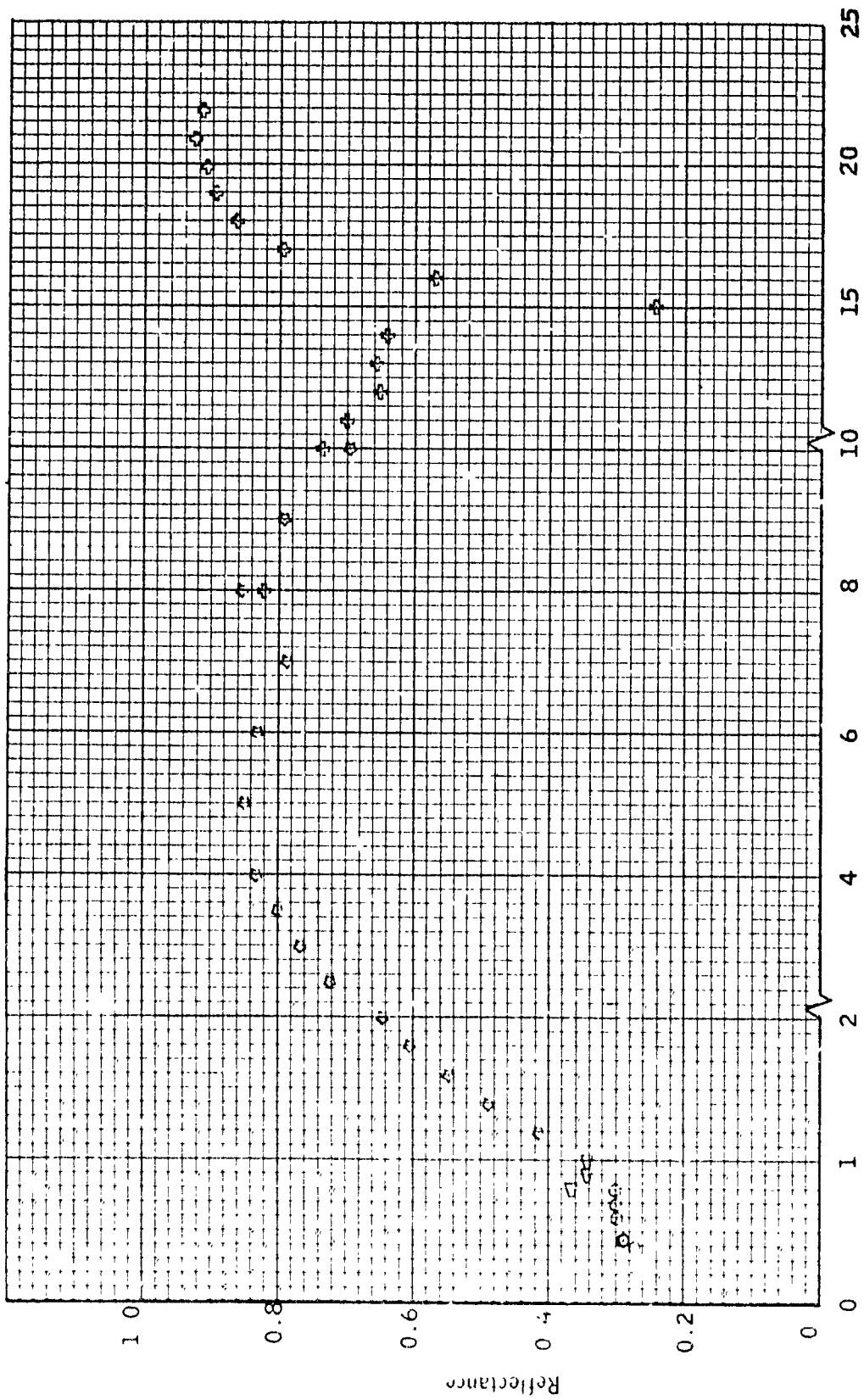


Fig. 174 Normal Spectral Reflectance of Specimen No 177 Temperature RT-HT-1500 F

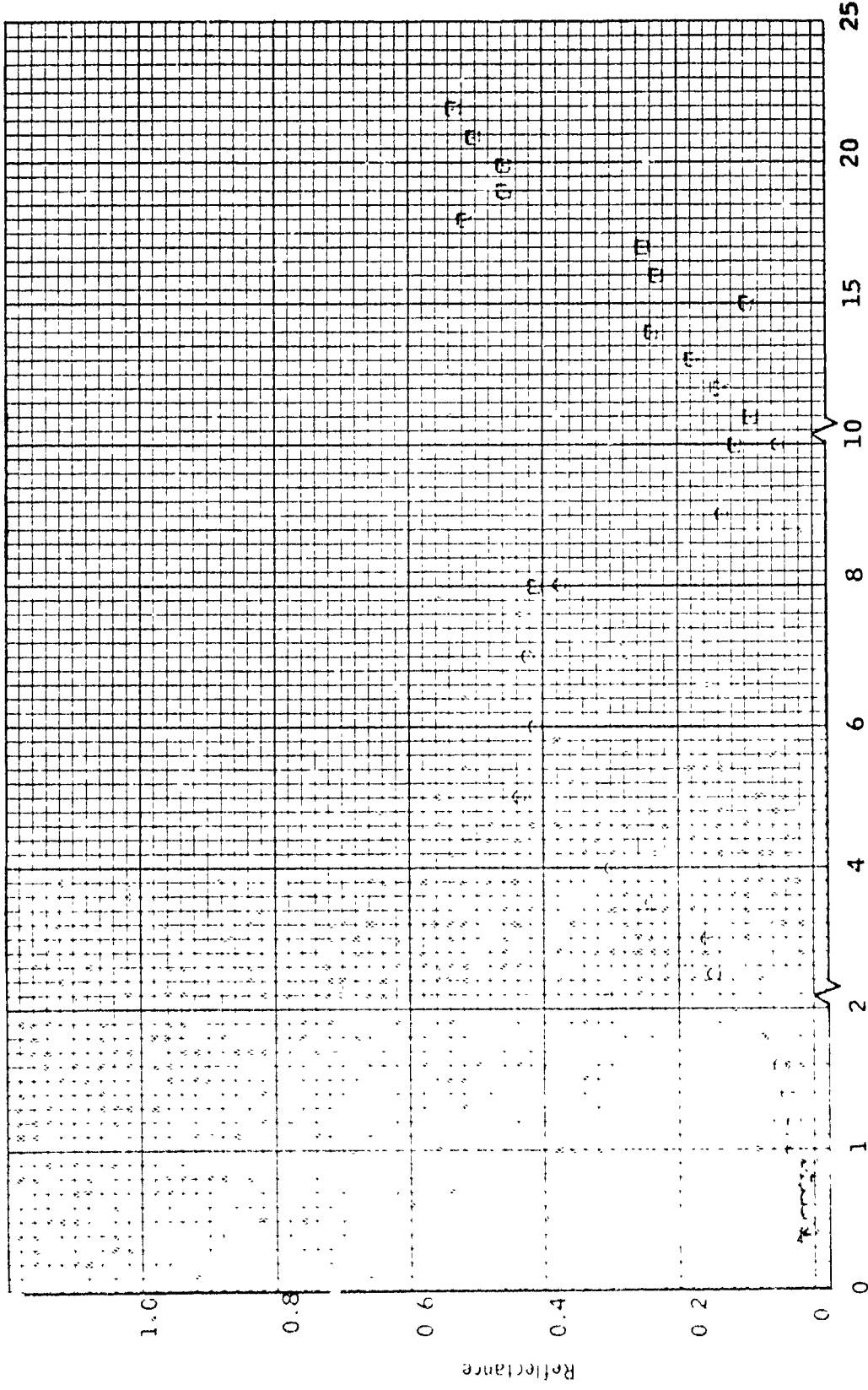


Fig. 175 Normal Spectral Reflectance of Specimen No 167 Temperature RT

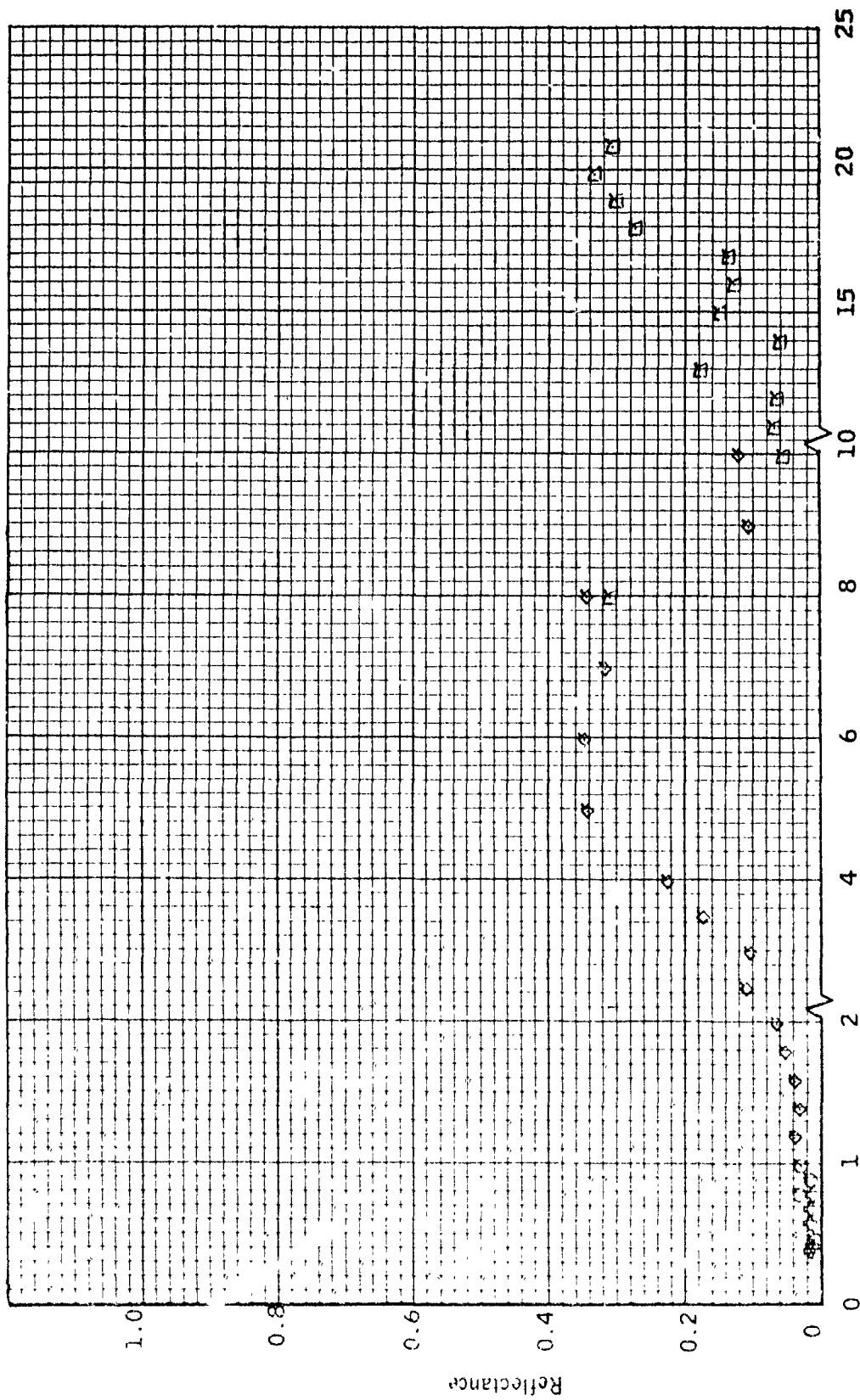


Fig. 176 Normal Spectral Reflectance of Specimen No 174 Temperature RT

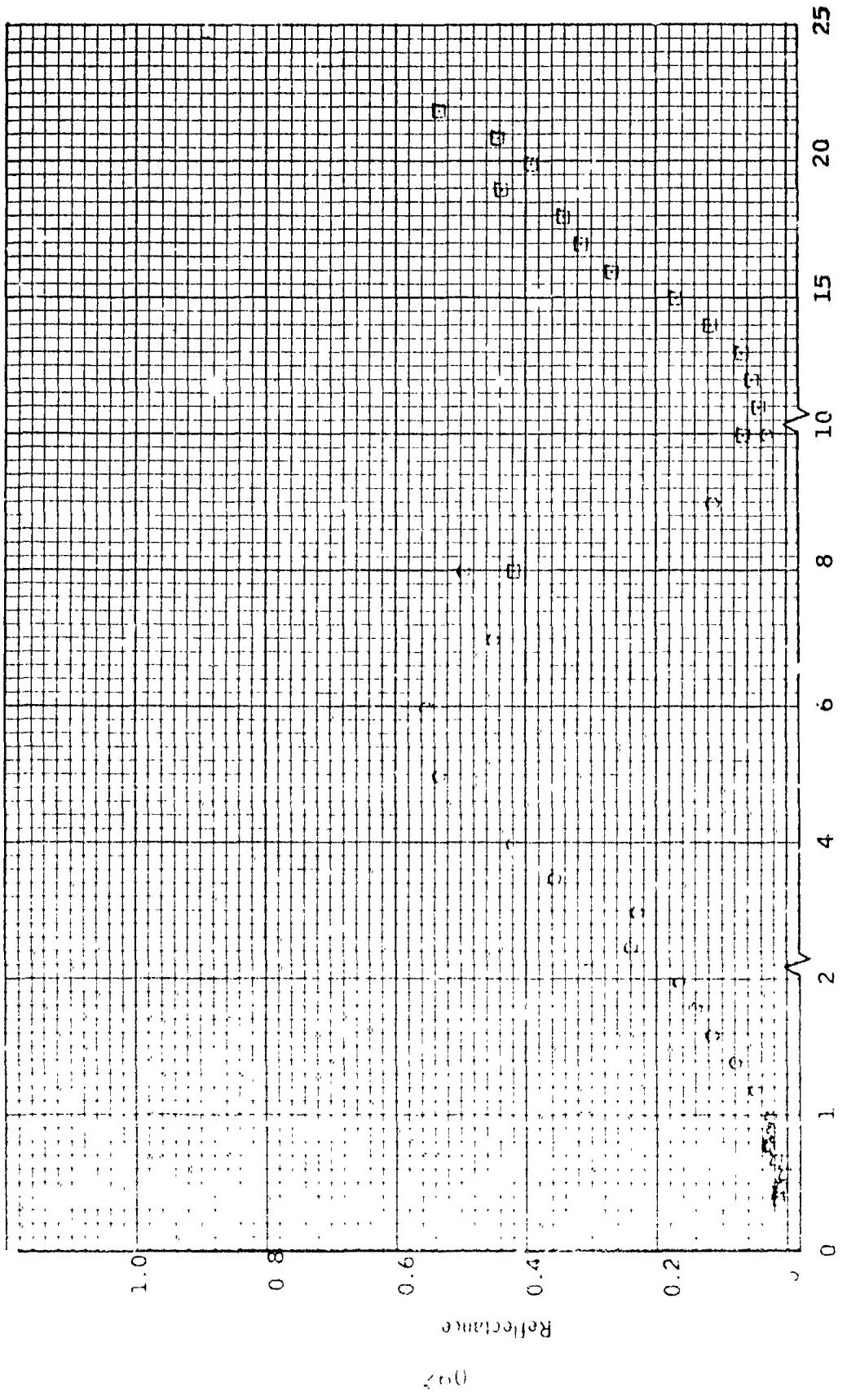


Fig. 177 Normal Spectral Reflectance of Specimen No 176 Temperature RT

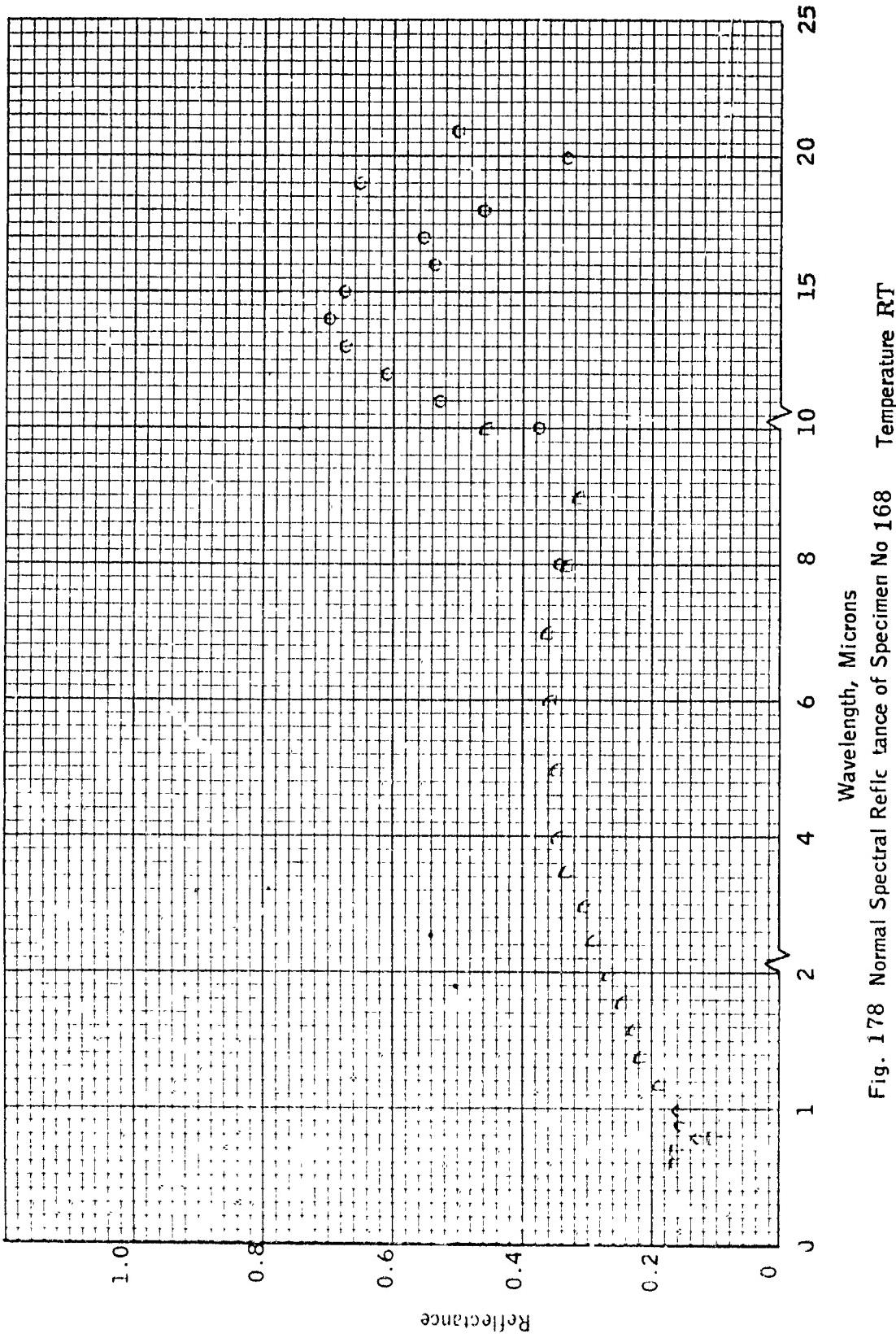


Fig. 178 Normal Spectral Reflectance of Specimen No 168 Temperature RT

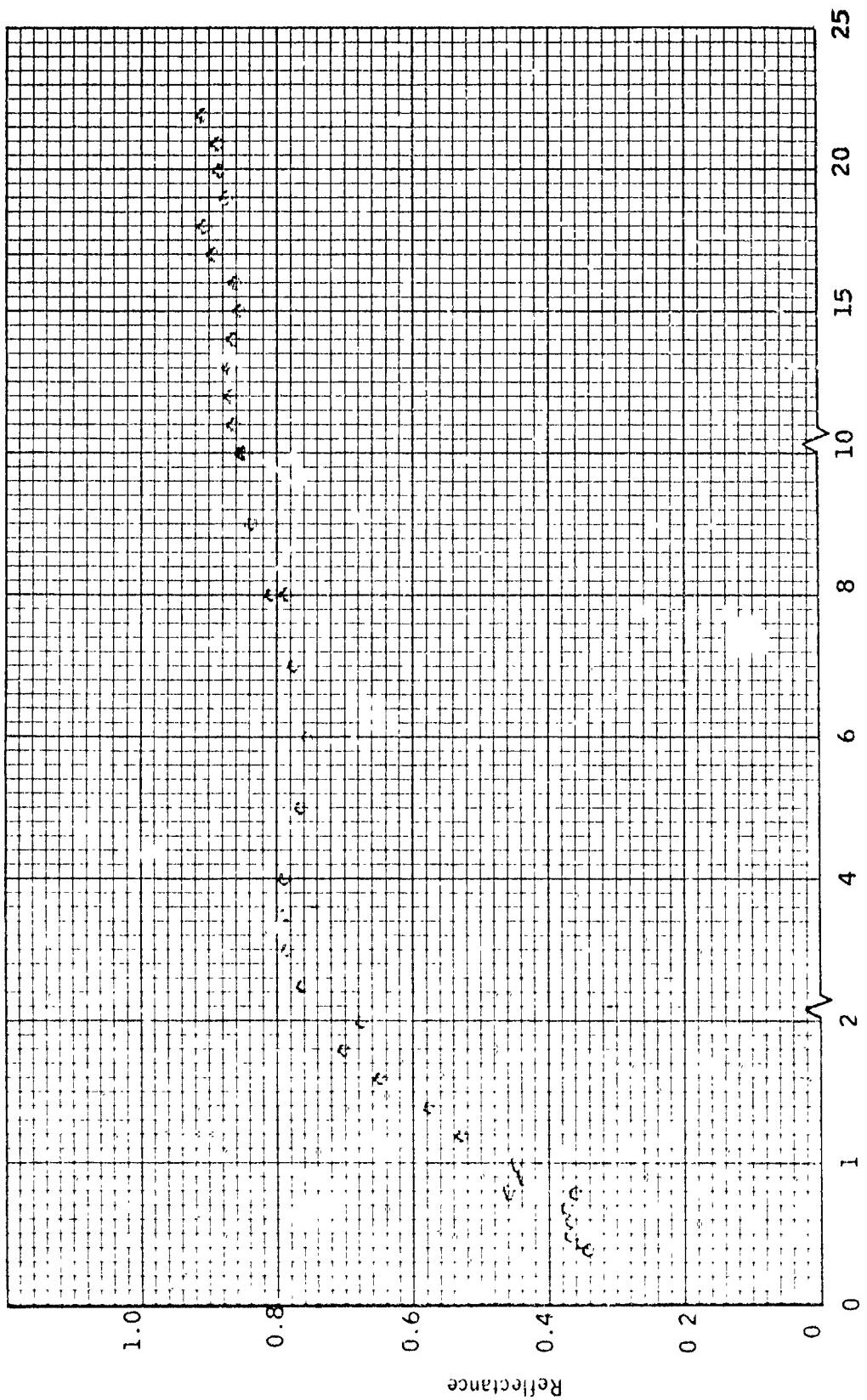


Fig. 179 Normal Spectral Reflectance of Specimen No 170 Temperature RT

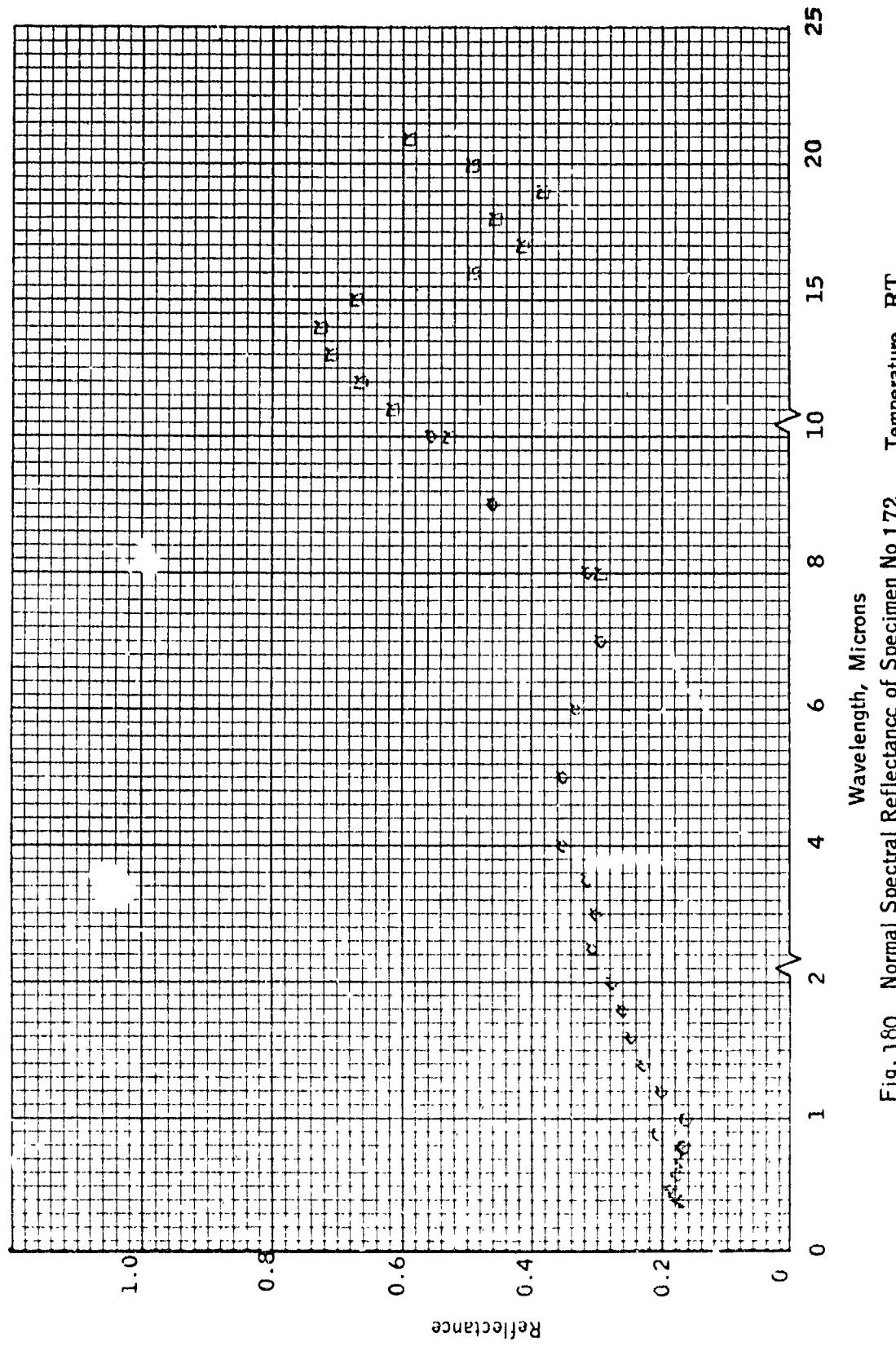


Fig. 180 Normal Spectral Reflectance of Specimen No 172 Temperature RT

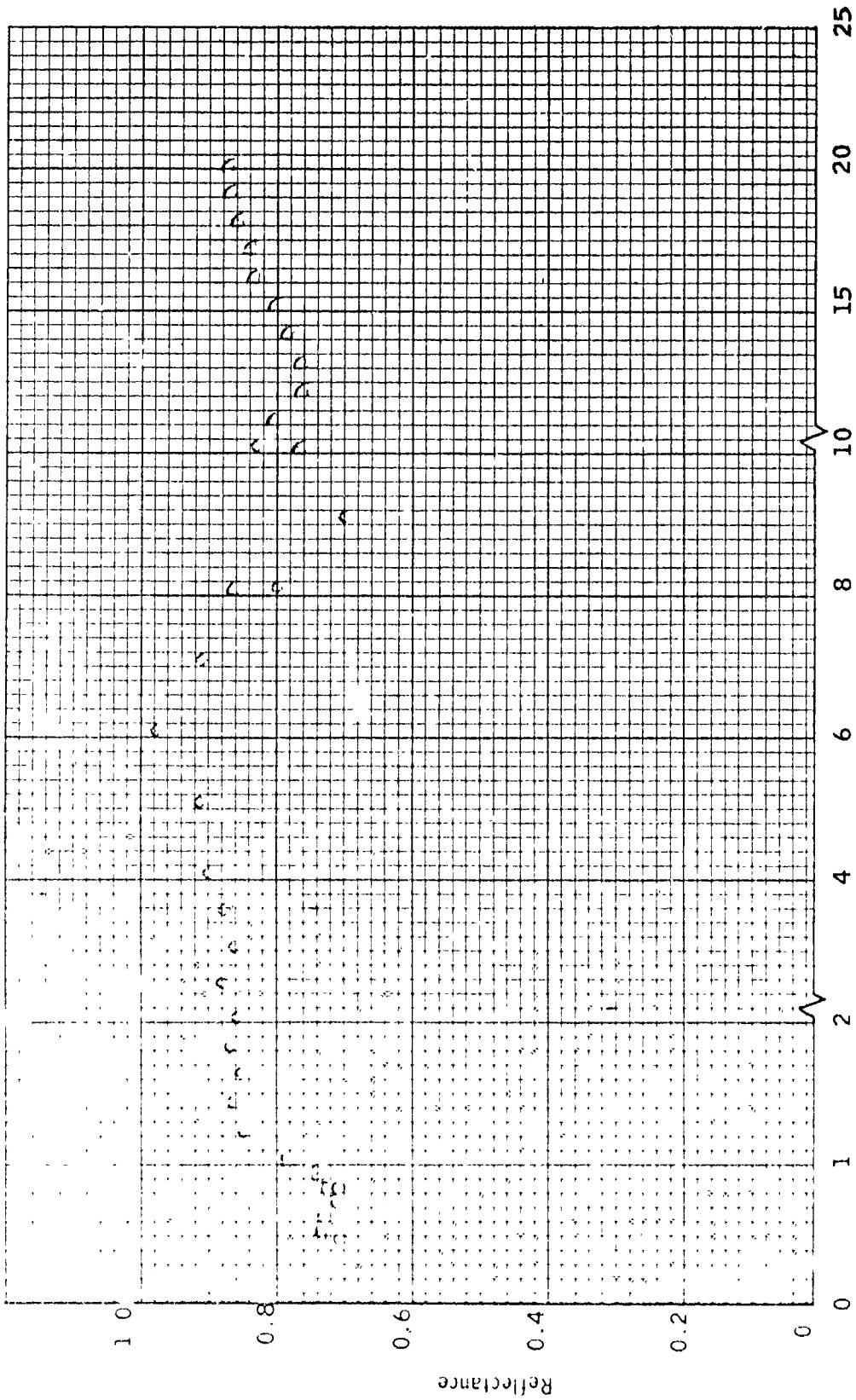


Fig. 181 Normal Spectral Reflectance of Specimen No. 18 Temperature RT

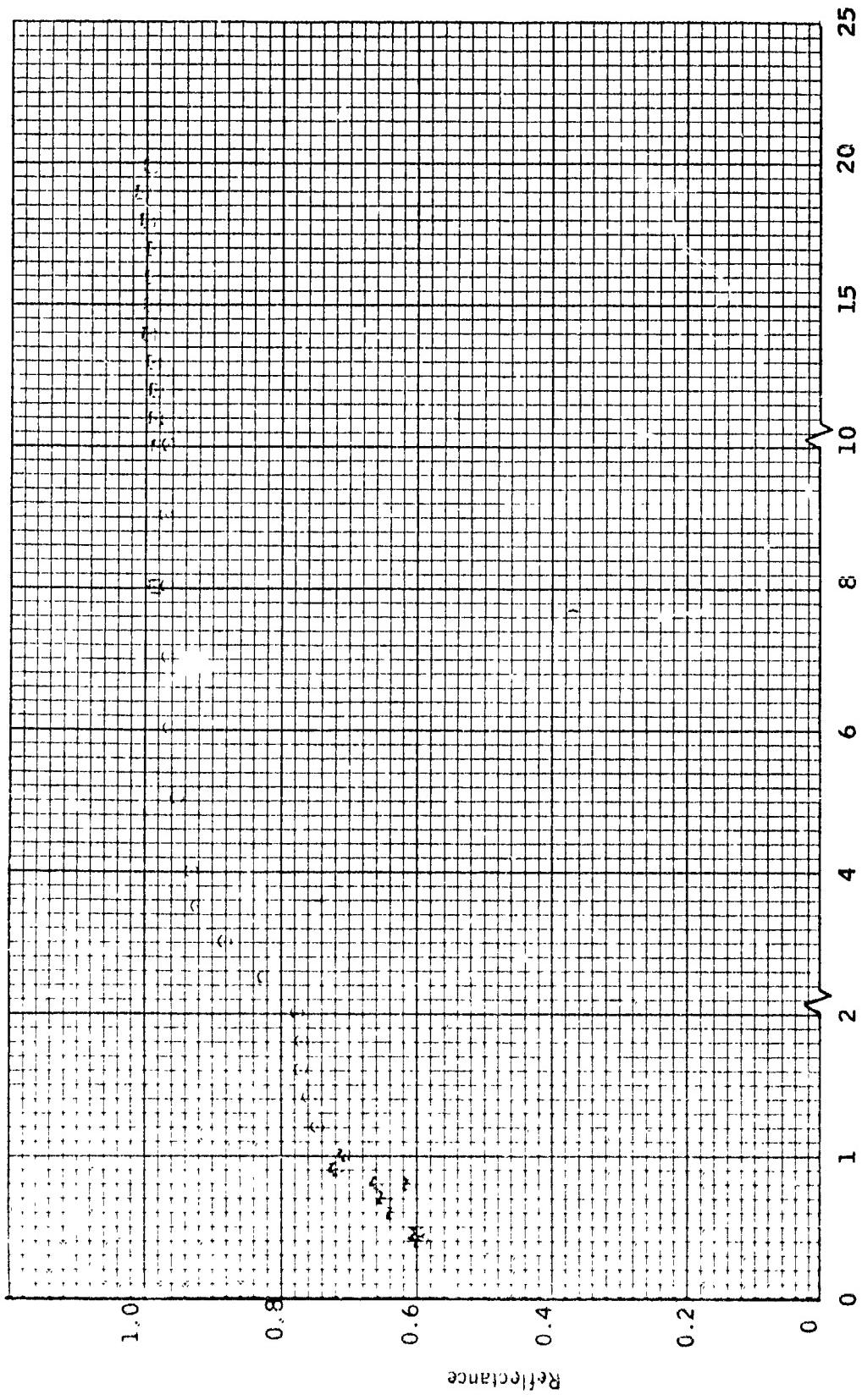


Fig. 182 Normal Spectral Reflectance of Specimen No 156 Temperature RT

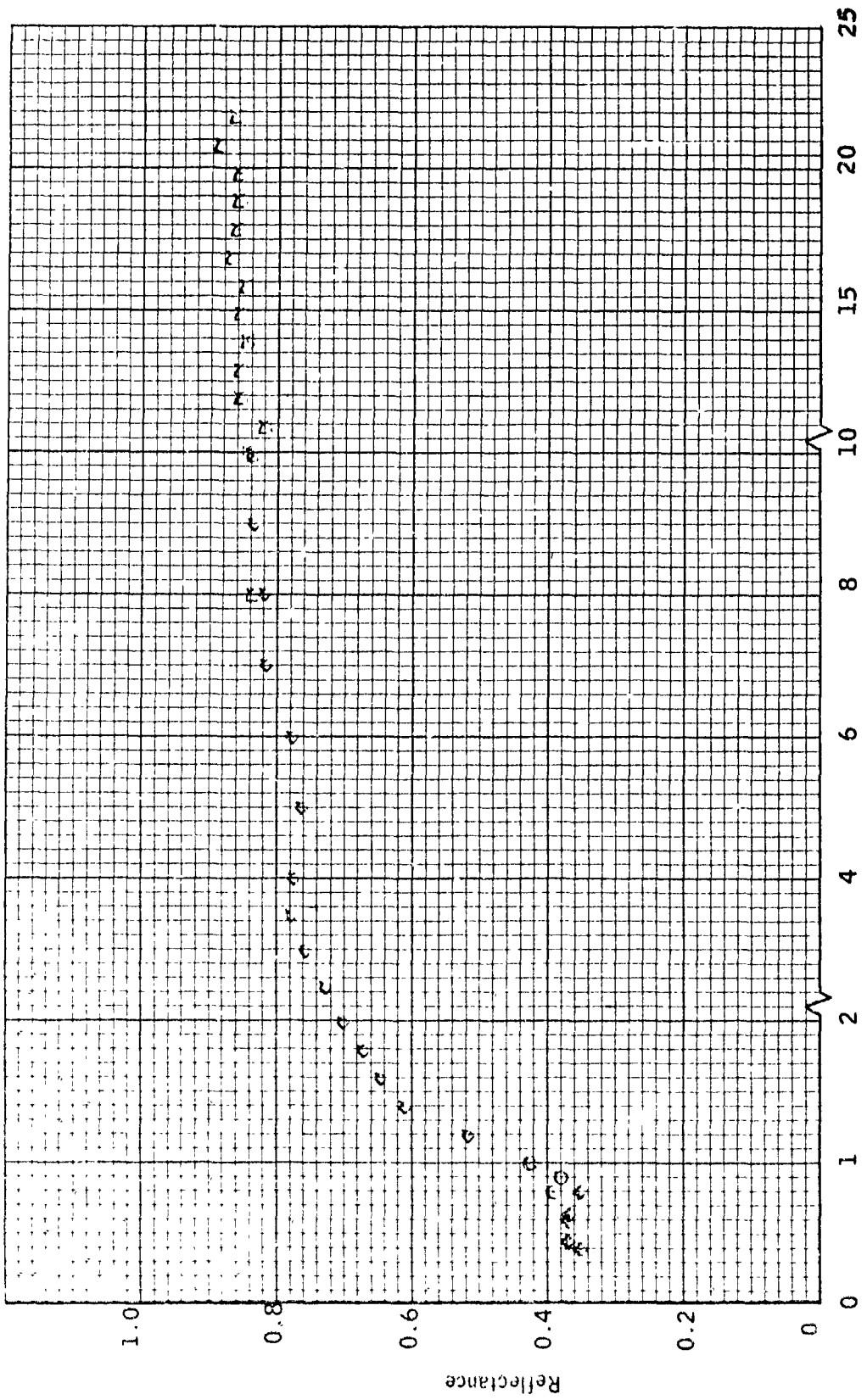


Fig. 183 Normal Spectral Reflectance of Specimen No 171 Temperature RT

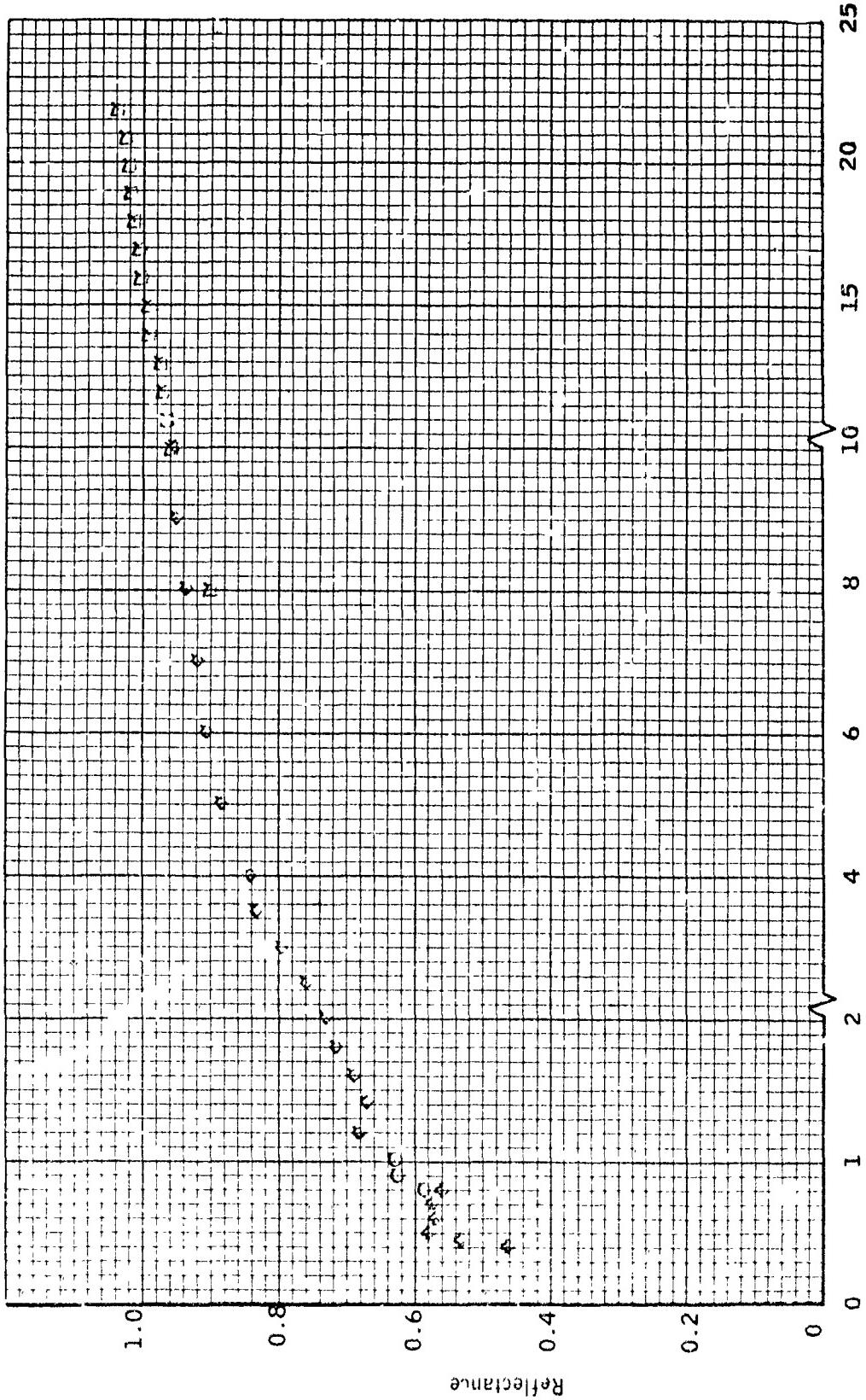


Fig. 184 Normal Spectral Reflectance of Specimen No. 4C Temperature RT

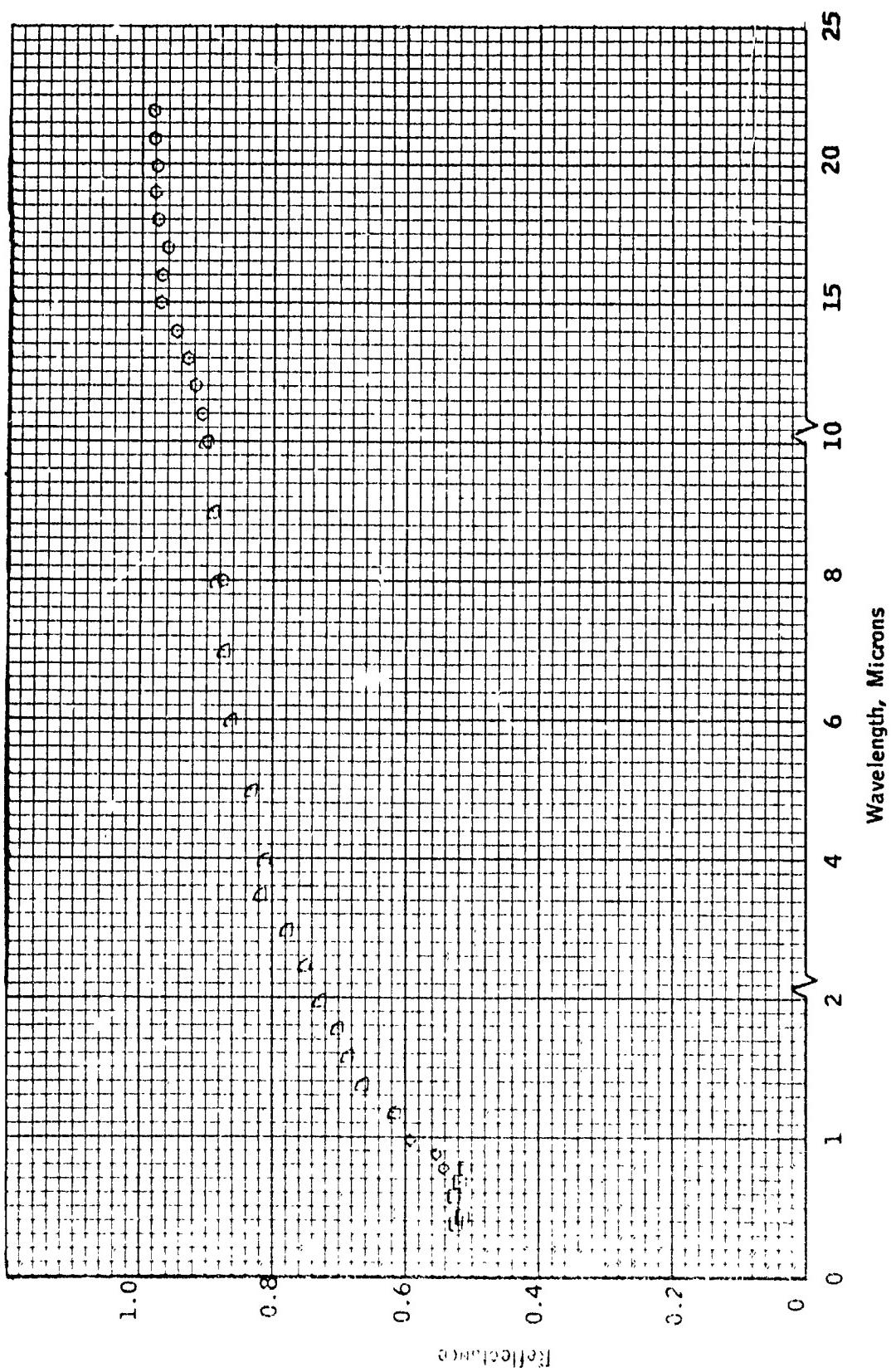


Fig. 186 Normal Spectral Reflectance of Specimen No 202 Temperature RT